



# **DESIGN GUIDELINES BMB&A**

**Approval Design Department** R&D Team

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# **R&D TEAM DESIGN GUIDELINES BMB&A**

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With lots of outstanding advantages compared with other materials, steel is increasingly used in civil and industrial buildings. Over the past hundreds of years, a vast majority of researches on steel structure conducted in order to improve safety for the structure as well as decrease cost of the building. To pursue the dream of becoming the largest steel company in Vietnam and over the world, design department made effort continuously to both finish all projects on schedule and study and apply newest achievements over the world to buildings. Standards and documents from the most developed country in steel field America used to guarantee the stability of the structure as well as strict requirements of architects and weight from customers.

From the first design guide published in 2011 in this latest design guide, there are more additional theories and guidelines for constructing SAP model and calculating connection, mezzanine floors, loads for building, etc.… according to the newest standards. In addition, many practical pictures, tables, serving for computing and calculating supplemented as well as mistakes are also updated and revised. The object of this editing is to make a standard of designing for all designers and support the other department to understand more about the processing of the design department.

In the compilation process, mistakes cannot be avoided so it would be so helpful to receive any feedback from readers to more complete in the next edition.

Thanks and Best regards,

Chief of Approval Design,

Mai Xuan Quang

# **TABLE OF CONTENT**



# **GENERAL INFORMATION**

#### **Introduction**

(1) The Design Department manual outlines the design process requirements of the BMB&A Standard in BMB&A J/S CO. (2) The Design manual establishes rules and standards to ensure that parameters and figures are presented uniformly throughout in BMB&A J/S CO.

**Scope of the design manual**

(3) The design manual is binding for all Design Team, Estimator Team and Office in the BMB&A J/S CO.

# **CHAPTER 1. MATERIALS 1.1. Materials**

## *Table 1 1 Material Specifications (unless indicated in drawings)*



## **1.2. Plates**



# **1.3. Hot Roll Members**

## Refer Appendix A for more sections.



# **1.4. Cold Form Sections**





(\*) Not available at Hung Yen Factory.

# **1.5. Sheeting**





## **1.6. Slab**



# **1.7. Bolts**



# **1.8. Bracing**



# **CHAPTER 2. CODES AND LOADS**

## **2.1. Codes and Manuals**

The Pre-Engineered Building described in these calculations was designed according to the latest U.S.A. Buildings and Design Codes that have been referred to in the design:

- 1. "Minimum Design Loads for Buildings and Other Structures", ASCE 7-10.
- 2. "International Buildings Code", IBC– 2012.
- 3. "Metal Building Systems Manual 2012" issued by MBMA.
- 4. "American Institute of Steel Construction" Allowable Stress Design, AISC 360-10.
- 5. Cold formed components have been designed in accordance with: "American Iron and Steel Institute" Cold Formed Steel Design Manual, AISI S100-2007
- 6. Welding has been applied in accordance with:
- "American Welding Society", AWS-2010.
- 7. Wind speed & live load for Vietnam Project has been applied in accordance with: "Load and Effects - Design Standard" TCVN 2737:1995.
- 8. Earthquake load has been considered in accordance with: "Uniform Building Code", UBC 1997 Edition.

The above codes are to be used for the design of buildings by BMB&A design engineers unless otherwise specified in the Contract Information Form.

## **2.2. Design Loads**

BMB&A Pre-Engineered Buildings are designed according to ASCE 7-10.

## **2.2.1. Dead load (DL)**

#### **Dead Load on Roof Sheeting**

This includes the self-weight of rigid frames and imposed dead load due to secondary elements like roof sheeting, purlins, insulation…

Following are some standard dead loads (in kN/m2) as below:



\* Calculate manually dead load when using 2 layers insulation and purlin spacing is smaller than 1.2 m



*Fiberglass Air Bubble PU (Poly Urethane)*







*PE (Poly Ethylene) Sandwich Rock Wool*

*Figure 2.1 Types of Insulation*

#### **Mezzanine dead load**

Concrete slab (in  $kN/m2$ ) Decking panel/ Checker plate/ Grating (in kN/m2) Finishing (in kN/m2) Wall load on beams, joists (in kN/m) Stair/ elevator (in kN) These values are determined depending on the specific weight & the sizes of material.

Following are some specific weight that BMB&A is used to calculate:



#### **2.2.2. Collateral load (AL)**

Collateral load here means superimposed dead load (in kN/m2) includes:

Ceiling (Gypsum board) HVAC duct Lighting fixtures

Sprinklers

Besides, some collateral systems (such as HVAC duct, sprinklers) have live load; we calculate them into live load (LL).

#### **2.2.3. Live load**

#### **Roof live load (LLr)**

The roof live load is used to design purlin, depends on the tributary area of rigid frames. It includes the weight of labor & accessary to install, repair the roof.

Refer to ASCE 7-10, Table 4-1 Minimum Uniformly Distributed Live Loads, L<sub>o</sub>, and Minimum Concentrated Live Loads. For built-up frames, minimum uniformly distributed live load on the roof is 1.0 kN/m<sup>2</sup>. *ASCE 7-10 section 4.8.2* allows the use of 0.57kN/m² as live load for roof and purlins. Roof live loads as per other building codes should be verified before proceeding with your design. Some customers/consultants may require pattern loading in live load applications.

#### **Live load on frame (LL)**

Refer to *TCVN 2737:1995 (Load and Effects - Design Standard), section 4.3.1*, minimum uniformly distributed live load on the frame is  $0.3 \text{ kN/m}^2$ .



#### **Mezzanine live load (FL)**

For floor loads of different occupancy, refer to CHAPTER 6. MEZZANINE FLOOR DESIGN.

#### **2.2.4. Wind load (WL)**

The wind load pressure is determined in accordance with *Chapter 26-Chapter30, ASCE 7-10*. Wind loads are governed by wind speed, roof slope, wave height and open wall conditions of the building. BMB&A's steel buildings are not designed for a wind speed less than 110 km/h.

Wind load on frame *W* depends on importance factor  $I_{\nu}$ .

$$
W = q_{\rm x} \times \left( GC_{\rm pf} - GC_{\rm pi} \right) \times B
$$

Where:  $W =$  wind design pressure on frame

 $-q_z$  = velocity pressure evaluated at height z above ground (N/m<sup>2</sup>);

 $q_z = 0.613 \times K_z \times K_{z} \times K_d \times V^2(N/m^2)$ ; *V* in m/s. *(27.3-1, ASCE 7-10)* 

- *GC*<sub>pf</sub> = product of the equivalent external pressure coefficient and gust-effect factor to be used in determination of wind loads for MWFRS of low-rise buildings

- *GC* <sub>*pi*</sub> = product of the internal pressure coefficient and gust-effect factor to be used in determination of wind loads for buildings

 $-B = bay$  spacing (m)



*Figure 2.4 Outline of process for determining Wind load*

#### **Risk categorization**

Building and other structures shall be classified, based on the risk to human life, health, and welfare associated with their damage or failure by nature of their occupancy or use, refer to *Table 1.5-1, ASCE 7-10* for purposes of applying flood, wind, snow, earthquake, and ice provisions. Each building or other structure shall be assigned to the highest applicable risk category or categories.

#### *Table 2 1 Risk Category of Buildings and Other Structures in Flood, Wind, Snow, Earthquake, and Ice Loads (Table 1.5-1, ASCE 7-10)*



<sup>a Buildings and other structures containing toxic, highly toxic, or explosive substances shall be eligible for</sup> classification to a lower Risk Category if it can be demonstrated to the satisfaction of the authority having jurisdiction by a hazard assessment as described in Section 1.5.2 ASCE 7-10 that a release of the substances is commensurate with the risk associated with that Risk Category.



#### *Table 2 2 Risk Category of Buildings for different codes*

Minimum design loads for structures shall incorporate the applicable importance factors given in *Table 1.5-2, ASCE 7-10.*

*Table 2 3 Importance Factors by Risk Category of Buildings and Other Structures for Snow, Ice, and Earthquake Loads* a



#### **Exposing categories <sup>1</sup>**

#### *Table 2 4 Exposure Categories*



For a site located in the transition zone between exposure categories, the category, resulting in the largest wind forces shall be used.





<sup>1</sup>ASCE 7-10, Section 26.7.3.

#### **Basic wind speed V 50y,3s**

Basic wind speed depends on every national standard. So we need to convert it into the value  $V_{50y,3s}$  in ASCE 7-10 with design 3-second gust wind speeds (m/s) at 10m above ground for Exposure C. In this manual, we will summarize by below table.



#### *Table 2 6 Wind return period for different codes*

#### **Example**

**Vietnam Location:** converting 20 years return period to 50 years return period by following a formula:

$$
\frac{V_T}{V_{50}} = 0.36 + 0.1 \times \ln(12T) \tag{C26.5-2, ASCE 7-10}
$$

$$
\frac{V_{20}}{V_{50}} = 0.36 + 0.1 \times \ln(12T) = 0.36 + 0.1 \times \ln(12 \times 20) = 0.908 \rightarrow V_{50} = \frac{V_{20}}{0.908}
$$

**Thailand Location:** If converting gust duration from 1h to 3s.

See *Figure C26.5-1 (ASCE 7-10)*. Maximum speed average over ts to hourly wind speed



*Figure 2.5 Converting gust duration diagram*

According to the above diagram, we see

$$
\frac{V_{3s}}{V_{1b}} = 1.52 \rightarrow V_{3s} = 1.52 V_{1b}
$$

## **Design wind speed VTy,3s**

Using the *Figure 26.5-1A, 1B, 1C ASCE 7-10* to determine the design wind speed, it depends on risk category of building

#### *Table 2 7 Design wind returns period*



Velocity pressure conversion factor:

$$
\frac{V_{Ty,3s}}{V_{50y,3s}} = 0.36 + 0.1 \times \ln(12T) \tag{C26.5-2, ASCE 7-10}
$$

$$
\rightarrow \text{Design wind speed: } V_{T_y,3s} = V_{50y,3s} \times \left[0.36 + 0.1 \times \ln(12T)\right]
$$

#### **Velocity pressure exposure coefficient Kz**

According to Table 27.3-1, ASCE 7-10, Velocity pressure exposure coefficient Kz depends on exposure and height of building above the ground.

Height above ground level, z		<b>Exposure</b>	
(m)	$\sf{B}$	$\mathsf C$	D
$0 - 4.6$	0.57	0.85	1.03
6.1	0.62	0.90	1.08
7.6	0.66	0.94	1.12
9.1	0.70	0.98	1.16
12.2	0.76	1.04	1.22
15.2	0.81	1.09	1.27
18.0	0.85	1.13	1.31
21.3	0.89	1.17	1.34
24.4	0.93	1.21	1.38
27.4	0.96	1.24	1.40
30.5	0.99	1.26	1.43
36.6	1.04	1.31	1.48
42.7	1.09	1.36	1.52
48.8	1.13	1.39	1.55
54.9	1.17	1.43	1.58
61.0	1.20	1.46	1.61
76.2	1.28	1.53	1.68
91.4	1.35	1.59	1.73
106.7	1.41	1.64	1.78
121.9	1.47	1.69	1.82
137.2	1.52	1.73	1.86
152.4	1.56	1.77	1.89

*Table 2 8 Velocity pressure exposure coefficients, Kh and Kz (Table 27.3-1, ASCE 7-10)*



#### **NOTE**

1. The velocity pressure exposure coefficient Kz may be determined from the following formula:

For  $15 \text{ ft} \leq z \leq z_g$ ,  $K_z = 2.01 \left( z/z_g \right)^{2/\alpha}$ For  $\zeta < 15 \text{ ft}$ ,  $K_z = 2.01 \left( 15 / z_g \right)^{2/\alpha}$ 

- 2.  $\alpha$  and  $\zeta$  are tabulated in Table 26.9.1 ASCE 7-10.
- 3. Linear interpolation for intermediate values of height z is acceptable.
- 4. Exposing categories are defined in Table 2 4.

### **Topographic factor Kzt**

The topographic factor for the site  $K_{\alpha t}$  is taken to be 1 in most cases. For more detail, see *Section 26.8.1 ASCE 7-10*.

#### **Wind directionality factor Kd**

Wind directionality factor  $K_d$  depends on type & shape of the building, see table below.





#### **Gust effect factor**

The gust-effect factor for a rigid building or low-rise building is permitted to be taken as 0.85.

For flexible or dynamically sensitive building or high-rise building, calculate G value according to section 26.9 ASCE 7-10.

#### **Enclosure classification**

For the purpose of determining internal pressure coefficients, all buildings shall be classified as enclosed, partially enclosed, or open



#### **Internal pressure coefficient (GCpi)**

Internal pressure coefficients, (GC<sub>pi</sub>), shall be determined from *Table 26.11-1 ASCE 7-10* based on building enclosure classifications.







#### **NOTE:**

1. Plus and minus signs signify pressures acting toward and away from the internal surfaces, respectively.

2. The values of  $(GC_{pi})$  shall be used with  $q_z$  or  $q_\theta$  as specified.

3. Two cases shall be considered to determine the critical load requirements for the appropriate condition:

(i) A positive value of  $(GC_{pi})$  applied to all internal surfaces

(ii) A negative value of  $(GC_{pi})$  applied to all internal surfaces

#### **2.2.5. Crane load (CR)**

Crane loads are determined using the crane data available from the crane manufacturer. Crane data include wheel load, curb weight, crane weight, wheelbase, ends hook approach (used when two cranes operate in one aisle) and minimum vertical and horizontal clearances.

Refer CHAPTER 5. CRANE SYSTEMS DESIGN for more detail.

#### **2.2.6. Earthquake**

#### *2.2.6.1 Seismic design concept*

An effective seismic design generally includes:

1. Selecting an overall structural concept, including the layout of a lateral-force-resisting system that is appropriate to the anticipated level of ground shaking. This includes providing a redundant and continuous load path to ensure that a building responds as a unit when subjected to ground motion2. Determining codeprescribed forces and deformations generated by the ground motion, and distributing the forces vertically to the lateral-force-resisting system. The structural system, configuration, and site characteristics are all considered when determining these forces.

3. Analysis of the building of the combined effects of gravity and seismic loads to verify that adequate vertical and lateral strength and stiffness are achieved to satisfy the structural performance and acceptable deformation levels prescribed in the governing building code.

4. Providing details to assure that the structure has the sufficient inelastic deformability to undergo fairly large deformations when subjected to a major earthquake. Appropriately detailed members possess the necessary characteristics to dissipate energy by inelastic deformations.

#### *2.2.6.2 Structural Response*

If the base of a structure is suddenly moved, as in a seismic event, the upper part of the structure will not respond instantaneously but will lag because of the inertial resistance and flexibility of the structure. The resulting stresses and distortions in the building are the same as if the base of the structure was to remain stationary while timevarying horizontal forces are applied to the upper part of the building. These forces, called inertia forces, are equal to the product of the mass of the structure times acceleration, i.e.,  $F = ma$  (the mass m is equal to weight divided by the acceleration of gravity, i.e.,  $m = w/g$ ). Because earthquake ground motion is three-dimensional (one vertical and two horizontal), the structure, in general, deforms in a three-dimensional manner. Generally, the inertia forces generated by the horizontal components of ground motion require greater consideration for seismic design since adequate resistance to vertical seismic loads is usually provided by the member capacities required for gravity load design. In the equivalent static procedure, the inertia forces are represented by equivalent static forces.

#### *2.2.6.3 Load path*

Buildings are generally composed of vertical and horizontal structural elements.

The vertical elements commonly used to transfer lateral forces on the ground are:

- 1) Shear walls;
- 2) Braced frames;
- 3) Moment-resisting frames.

The horizontal elements that distribute lateral forces to the vertical elements are:

- 1) Diaphragms, such as floor and roof slabs;
- 2) Horizontal bracing that transfers large shears from discontinuous walls or braces.

The seismic forces that are proportional to the mass of the building elements are considered to act as their centers of mass. All of the inertia forces originating from the masses on and off the structure must be transmitted to the lateral-force-resisting elements, and then to the base of the structure and into the ground.

A complete load path is a basic requirement for all buildings. There must be a complete lateral-forceresisting system that forms a continuous load path between the foundation, all diaphragm levels, and all portions of the building for proper seismic performance. The general load path is as follows. Seismic forces originating throughout the building, mostly in the heavier mass elements such as diaphragms, are delivered through connections to horizontal diaphragms; the diaphragms distribute these forces to vertical force-resisting elements such as shear walls and frames; the vertical elements transfer the forces into the foundation; and the foundation transfers the forces into the supporting soil.

If there is a discontinuity in the load path, the building is unable to resist seismic forces regardless of the strength of the elements. Interconnecting the elements needed to complete the load path is necessary to achieve good seismic performance. Examples of gaps in the load path would include a shear wall that does not extend to the foundation, a missing shear transfer connection between a diaphragm and vertical elements, a discontinuous chord at a diaphragm's notch, or a reentrant corner, or a missing collector.

A good way to remember this important design strategy is to ask yourself the question, "How does the inertia load get from here (meaning the point at which it is generated) to there (meaning the shear base of the structure, typically the foundations)?"

#### *2.2.6.4 The design base shear according to UBC 97*

The total design base shear in a given direction shall be determined following *section 1630.2 UBC 97:*

$$
V = \frac{C_{v} \times I}{R \times T} \times W
$$
 (kN)

Besides, V value must be satisfied below equation:

$$
0.11C_a \times I \times W \le V = \frac{C_v \times I}{R \times T} \times W \le \frac{2.5C_a \times I}{R} \times W
$$

For seismic zone 4, the total design base shear shall also not be less than the following:

$$
V \ge \frac{0.8Z \times N_v \times I}{R} \times W
$$

#### *2.2.6.5 Seismic coefficient (Cv, Ca)*

The seismic coefficient is determined by Table 2 11 and Table 2 12, it depends on soil profile type (Table 2 13) and seismic zone factor Z (Table 2 15).



#### Table 2 11 Seismic coefficients, C<sub>v</sub> (Table 16-R, UBC 97)

*Table 2 12 Seismic coefficients, Ca (Table 16-Q, UBC 97)*

Soil profile type	Seismic zone factor, Z				
	$Z = 0.075$	$Z = 0.15$	$Z = 0.2$	$Z = 0.3$	$Z = 0.4$
$S_A$	0.06	0.12	0.16	0.24	$0.32N_{v}$
$S_B$	0.08	0.15	0.20	0.30	$0.40N_{v}$
$S_C$	0.09	0.18	0.24	0.33	0.40N <sub>v</sub>
$S_{D}$	0.12	0.22	0.28	0.36	$0.44N_{v}$
$S_{E}$	0.19	0.30	0.34	0.36	0.36Nv
$S_F$	Refer to site-specific Geotechnical investigation and dynamic site response				
	analysis to determine $C_{v}$				

The seismic coefficients  $C_{\nu}$  and  $C_{a}$ , given in *Tables 16-R and 16-Q UBC 97*, are site-dependent ground motion coefficients that define the seismic response throughout the spectral range. They are measures of expected ground acceleration at a site.

For a given earthquake, a building on soft soil types such as  $S_C$  or  $S_D$  experiences a greater force than if the same building were located on a rock, type  $S_A$  or  $S_B$ . This is addressed in the UBC through the  $C_a$  and  $C_v$  coefficients, which are calibrated to soil type  $S_B$  with a value of unity. Instead of a single coefficient, two coefficients,  $C_a$  and C<sub>v</sub>, are used to distinguish the response characteristics of short-period and long-period buildings. Long period buildings are more affected by soft soils than short-period buildings.

In SAP software, a designer needs to choose Soil profile type & Seismic zone factor, the program will calculate  $C_{\rm a}$ ,  $C_{\rm v}$  automatically according to UBC 97.



## *2.2.6.6 Soil profile types*







#### **NOTE**

Use **Soil Types SD** if there is not any requirement from customers.

#### *Table 2 14 Soil profile types for different codes*



#### *2.2.6.7 Seismic zone & Seismic zone factor (Z)*

*Table 2 15 Seismic zone factor, Z (Table 16-I, UBC 97)*

Zone					
	0.075	$\cap$ 15 ◡…⊥◡	0.20 $\backsim$ . $\backsim$	0.30	0.40



**Note:** The zone shall be determined from the seismic zone map in Figure 16-2 UBC 97 The map accounts for the geographical variations in the expected levels of earthquake ground shaking and gives the estimated peak horizontal acceleration on a rock having a 10% chance of

being exceeded in a 50-year period (or 500-year return period).

The value of the seismic zone coefficient Z can be considered the peak ground acceleration of the percentage of gravity.

It means, the peak ground acceleration:  $a_g = Z \times g$ 

For example,  $Z = 0.4$  indicates a peak ground acceleration of 0.4g equal to 40% of gravity. For the buildings are not located in the United States, refer to *Appendix Chapter 16 UBC 97* to determine seismic zone.



*Figure 2.6 The United States Seismic zones map with 500-year return period*



### **Vietnam**

Use *Appendix H - TCVN 9386:2012* to determine ground acceleration for every area in Vietnam.

In this standard, the peak ground acceleration was surveyed considered on rock, 500-year return period. **Thailand**

According to Section 1653 UBC 97, we have a seismic zone for some areas in Thailand.

*Table 2 17 Seismic zone in Thailand*





Fig. 8 Thailand hazard maps for PGA corresponding to a probability of exceedance of 10% in 50 years *Figure 2.7 Thailand Seismic zones map with 500-year return period*

#### **Myanmar**

The seismic zone of Myanmar is divided into I, II, III, IV, V corresponding to seismic zone 2B, 3, 4 of UBC 97.



*Figure 2.8 Myanmar Seismic zones map with 500-year return period (Figure 3.4.1.5 MNBC 2016)*

### **Philippines**

The Philippines archipelago is divided into 2 seismic zones only. Zone 2 covers the provinces of Palawan (except Busuanga), Sulu and Tawi-Tawi while the rest of the country is under zone 4.

*Table 2 18 Seismic zone factor, Z in Philippines (Table 208-3, NSCP 2015)*





*Figure 2.9 Philippines Seismic zones map with 500-year return period (Figure 208-1 NSCP 2015)*

#### **Cambodia**

Use UBC 97 to design a building in Cambodia. Almost buildings no need to design earthquake.

#### **Indonesia**

According to Section 1653 UBC 97, we have a seismic zone for some areas in Indonesia.





*Figure 2.10 Indonesia Seismic zones map with 500-year return period*

## *2.2.6.8 Important factor (I)*

Occupancy category	Occupancy or functions of structure	Seismic importance factor, I	Seismic importance factor, Ip1	Wind importance factor, Iw
1. Essential facilities 2	Group I, Division 1 Occupancies having surgery and emergency treatment areas Fire and police stations Garages and shelters for emergency vehicles and emergency aircraft Structures and shelters in emergency- preparedness centers Aviation control towers Structures and equipment in government communication centers and other facilities required for emergency response Standby power-generating equipment for Category 1 facility Tanks or other structures containing housing or supporting water or other fire-suppression material or equipment required for the protection of Category 1, 2 or 3 structures	1.25	1.50	1.15
2. Hazardous facilities	Group H, Divisions 1, 2, 6 and 7 Occupancies and structures therein housing or supporting toxic or explosive chemicals or substances Non-building structures housing, supporting or containing quantities of toxic or explosive substances that, if contained within a building, would cause that building to be classified as a Group H, Division 1, 2 or 7 Occupancy	1.25	1.50	1.15
3. Special occupancy structures3	Group A, Divisions 1, 2 and 2.1 Occupancies Buildings housing Group E, Divisions 1 and 3 Occupancies with a capacity greater than 300 students Buildings housing Group B Occupancies used for college or adult education with a capacity greater than 500 students Group I, Divisions 1 and 2 Occupancies with 50 or more residents incapacitated patients, but not included in Category 1 Group I, Division 3 Occupancies All structures with an occupancy greater than 5,000 persons Structures and equipment in power-generating stations, and other public utility facilities not included in Category 1 or Category 2 above, and required for continued operation	1.00	1.00	1.00
4. Standard occupancy structures3	All structures housing occupancies or having functions not listed in Category 1, 2 or 3 and Group U Occupancy towers	1.00	1.00	1.00
5. Miscellaneous structures	Group U Occupancies except for towers	1.00	1.00	1.00

*Table 2 20 Occupancy category (Table 16-K, UBC 97)*
For steel building, we usually use important factor  $I = 1$ .



## *2.2.6.9 The total seismic dead load (W)*

According to *section 1630.1.1 UBC 97*, we can determine total seismic dead load W through mass source in SAP.



*W = Mass source = DL + AL + 0.25 (LL + FL + CR) (kN)*



#### **NOTE**

Load Pattern FL is fully applied floor load. Each Crane will contribute one Crane Load Pattern.

#### *2.2.6.10 Numerical coefficient representative of the inherent over strength and global ductility capacity of lateral force-resisting systems (R)*

The coefficient R shown in *Table 16-N UBC 97* is a measure of ductility and over strength of a structural system, based primarily on performance of similar systems in past earthquakes.

A higher value of R has the effect of reducing the design base shear. For example, for a steel special momentresisting frame, the factor has a value of 8.5, whereas for ordinary moment-resisting frame, the value is 4.5. This reflects the fact that a special moment-resisting frame performs better during an earthquake.

*Table 2 21 Structural systems (Table 16-N UBC 97)*

<b>Basic structural</b> system	Lateral force resisting system description	$\mathsf{R}$	$\Omega_{0}$	Height limit for seismic zones 3 and 4 (feet) x 304.8 for mm
1. Bearing wall	1. Light-framed walls with shear panels			
system	a. Wood structural panel walls for structures three stories or less	5.5	2.8	65
	b. All other light-framed walls 2. Shear walls	4.5	2.8	65
	a. Concrete	4.5	2.8	160
	b. Masonry	4.5	2.8	160
	3. Light steel-framed bearing walls with tension- only bracing	2.8	2.2	65
	4. Braced frames where bracing carries gravity	4.4	2.2	160
	load	2.8	2.2	$\sim$
	a. Steel b. Concrete c. Heavy timber	2.8	2.2	65
2. Building frame system	1. Steel eccentrically braced frame (EBF) 2. Light-framed walls with shear panels	7.0	2.8	240
	a. Wood structural panel walls for structures			
	three stories or less	6.5	2.8	65
	b. All other light-framed walls 3. Shear walls	5.0	2.8	65
	a. Concrete	5.5	2.8	240
	b. Masonry 4. Ordinary braced frames	5.5	2.8	160
	a. Steel	5.6	2.2	160
	b. Concrete	5.6	2.2	
	c. Heavy timber 5. Special concentrically braced frames	5.6	2.2	65
	a. Steel	6.4	2.2	240
3. Moment-	1. Special moment-resisting frame (SMRF)			
resisting frame	a. Steel	8.5	2.8	N.L.
system	b. Concrete	8.5	2.8	N.L.
	2. Masonry moment-resisting wall frame	6.5	2.8	160
	(MMRWF) 3. Concrete intermediate moment-resisting frame (IMRF)	5.5	2.8	
	4. Ordinary moment-resisting frame (OMRF)			
	a. Steel	4.5	2.8	160
	b. Concrete 5. Special truss moment frames of steel (STMF)	3.5 6.5	2.8 2.8	240
4. Dual systems	1. Shear walls a. Concrete with SMRF	8.5	2.8	N.L.



## *Table 2 22 R and factors for nonbuilding structures (Table 16-P UBC 97)*



For steel building, we usually use R=4.5.



### *2.2.6.11 Structure period (T)*

Structure period or elastic fundamental period of vibration, of the structure in the direction under consideration.

#### **Method A**

For all buildings, the value T may be determined from *section 1630.2.2 UBC 97*

$$
T = T_A = C_t \left( b_n \right)^{3/4} \text{ (s)}
$$

Where:

 $-C_t$  = the structural coefficient depends on the material of the frame, is always input in English unit in SAP software.

- $C_t = 0.035(0.0853)$  for steel moment-resisting frames.
- $C<sub>t</sub> = 0.030 (0.0731)$  for reinforced concrete moment-resisting frames and eccentrically braced frames.
- $C_t = 0.020(0.0488)$  for all other buildings.

 $-h_n$  = height above the base to level that is uppermost in the main portion of the structure. In SAP software, it's measured from the elevation of the specified bottom story/minimum elevation level to the (top of the) specified top story/maximum elevation level. (m)

The designer can modify h<sub>n</sub> value by choosing an option "User specified" in Lateral load elevation range & fill in Min Z/ Max Z.



#### **Method B**

The fundamental period T may be calculated using the structural properties and deformation characteristics of the resisting elements in a properly substantiated analysis. The analysis shall be in accordance with the requirements of *Section 1630.1.2 UBC 97*. The fundamental period T may be computed by using the following formula:

$$
T_B = 2\pi \sqrt{\frac{m}{k}} = 2\pi \sqrt{\left(\sum_{i=1}^n w_i \delta_i^2\right)} \div \left(g \sum_{i=1}^n f_i \delta_i\right)
$$
 (s)

Where:

 $-f_i =$  lateral force at level i (kN)

 $\delta_i$  = horizontal displacement at Level i relative to the base due to applied lateral force f (m)

- g = acceleration due to gravity,  $g = 9.81 \left( \frac{m}{s^2} \right)$ 

-  $w_i$  = that portion of the total seismic dead load assigned to level i. (kN)

SAP software use this method to calculate T if designer choose "Program calc" in Time period.



- If the seismic zone is zone 4 then:

• If  $T_B \leq 1.3 T_A \rightarrow T = T_B$ 

• If 
$$
T_B > 1.3T_A \rightarrow T = T_A
$$

- If the seismic zone is zone 1, 2A, 2B, 3 then:

- If  $T_R > 1.4 T_A \rightarrow T = T_A$
- If  $T_B > 1.4 T_A \rightarrow T = T_A$

#### *2.2.6.12 Near-source factor (Na, Nv)*

For seismic zone 4 (Z=0.4), we need near-source factor  $N_a$ ,  $N_v$  to determine seismic coefficient  $C_a$ ,  $C_v$  corresponding. They depend on seismic source type and the distance from the site to seismic source.

Seismic source type	Closest distance to known seismic source		
	$\leq$ 2 km	5 km	$\geq 10$ km
	15		
	13		

*Table 2 23 Near-source factor Na (Table 16-S UBC 97)*

#### *Table 2 24 Near-source factor NV (Table 16-T UBC 97)*



The purpose of  $N_a$  and  $N_v$  is to increase the soil-modified ground motion parameters,  $C_a$  and  $C_v$ , when there are active faults capable of generating large-magnitude earthquakes within 15 km of a seismic zone 4 site. So that

#### $N_a, N_{n} \ge 1$ .

In SAP software, a designer needs to choose "Seismic source type" & "Dist. to source", the program will be calculated  $N_a$ ,  $N_v$  automatically according to UBC 97.



#### *2.2.6.13 Seismic source types*



*Table 2 25 Seismic source type (Table 16-U UBC 97)*

The seismic source types labeled A, B, or C *(Table 16-U UBC 97)* is used to identify earthquake potential and activity of faults in the immediate vicinity of the structure.

They are defined in terms of the slip rate of the fault and the maximum magnitude of earthquake that may be generated at the fault. The highest seismic risk is posed by seismic source type A, which is defined by a maximum moment magnitude of 7.0 or greater and a slip rate of 5 mm/year or greater.

Moment magnitude (M) was introduced in 1979 by Hanks and Kanamori and has since become the most commonly used method of describing the size of a microseism. Moment magnitude measures the size of events in terms of how much energy is released.

The slip rate is how fast the two sides of a fault are slipping relative to one another, as determined from geodetic measurements, from offset man-made structures, or from offset geologic features whose age can be estimated. It is measured parallel to the predominant slip direction or estimated from the vertical or horizontal offset of geologic markers.

## **2.3. Load Combinations**

Based on design method (LRFD or ASD), ASCE 7-10 define load combination coefficients for each method. BMB&A uses allowable stress design (ASD) method, see *section 2.4.1 ASCE 7-10.* 

### **2.3.1. For Frame Structure**

- 1. Dead Load (DL)
- 2. Dead Load (DL) + Live Load (LL)
- 3. Dead Load (DL) + Live Load (Floor/Crane)

4. Dead Load (DL) +  $0.75$  Live Load (LL) +  $0.75$  Live Load (Floor/Crane)

5. Dead Load (DL) + [0.6 Wind Load (WL) or 0.7 Earthquake load (EL)]

6a. Dead Load (DL) + 0.75 Live Load (LL) + 0.75 Live Load (Floor/Crane) + 0.45 Wind Load (WL)

6b. Dead Load (DL) +  $0.75$  Live Load (LL) +  $0.525$  Earthquake load (EL)

7. 0.6 Dead Load (DL) + 0.6 Wind Load (WL)

8. 0.6 Dead Load (DL) + 0.7 Earthquake load (EL)

### **2.3.2. For Cold-Formed Section**

1. Dead Load (DL) + Roof Live Load (LLr)

2. Dead Load (DL) + 0.6 Wind Load (WL)

3. 0.6 Dead Load (DL) + 0.6 Wind Load (WL)

All of these are basic combination. For more loads (snow load, rain load, flood load..., etc.) see section 2.4 ASCE 7-10.

## **2.4. Serviceability Consideration**

*Table 2 26 Deflection Limitations*

<b>CONTRUCTION</b>		<b>LOAD &amp; DEFORMALOAD</b>	<b>REMARK</b>
1/ Rafter (Vertical deflection) <sup>2</sup>	LL	$DL + LL$	
a/ Supporting plaster ceiling	L/360	L/240	
b/ Supporting non-plaster ceiling	L/240	L/180	
c/ Not supporting ceiling	L/180	L/120	
2/ Column (Horizontal deflection) <sup>3</sup>	DL + WL 10 yr.	H/60	10 Yr. Wind
II/ Purlin1	$DL + LL$	$DL + WL 10 yr.$	10 Yr. Wind
1/ Purlin (Vertical deflection)	L/150	L/150	
2/ Girt (Horizontal deflection)	L/120	L/120	
III/ Floor members (Vertical deflection)	LL.	$DL + LL$	
Beam, Joist, Decking, Checker Plate	L/360	L/240	
IV/Top Running Cranes <sup>4</sup>		$DL + CR$	
1/ Runway Beam (Vertical deflection)		L/600	
2/ Runway Beam (Horizontal deflection)		L/400	
V/ Crane Bracket (Horizontal deflection) <sup>3</sup>		DL + CR or WL 10 yr.	Crane lateral
1/ Cab or Radio - Operator cranes		$H/240$ or $\leq 5.08$ cm	or 10 Yr.
2/ Pendant - Operator cranes		H/100	Wind

2 International Building Code - IBC 2012, Table 1604.3, page 335.

<sup>3</sup> Metal Building Systems Manual 2012, Table 3.3, page 331.

4 Metal Building Systems Manual 2012, Table 3.5, page 333.

## **CHAPTER 3. PRE-ENGINEERED BUILDING (PEB) SYSTEM**

Planning of the pre-engineered buildings (low rise metal buildings) and arranging different building components is a very important step for the designer before proceeding with the design of each component. The Following building configurations are significantly affecting the building Stability and Cost:

- 1. Main Frame configuration (orientation, type, roof slope, eave height)
- 2. Roof purlins spacing
- 3. Wall girts (connection & spacing)
- 4. End wall system
- 5. Expansion joints
- 6. Bay spacing
- 7. Bracing systems arrangement
- 8. Mezzanine floor beams/columns (orientation & spacing)
- 9. Crane systems



*Figure 3.1 Pre-engineered building system*



*Figure 3.2 Pre-engineered building system*

## **3.1. Main Frame Configuration**

Main frame is the basic supporting component in the PEB systems; main frames provide the vertical support for the whole building plus providing the lateral stability for the building in its direction while lateral stability in the other direction is usually achieved by a bracing system.

The width of the building is defined as the out-to-out dimensions of girts/eave struts and these extents define the sidewall steel lines. Eave height is the height measured from bottom of the column base plate to top of the eave strut. Rigid frame members are tapered using built-up sections following the shape of the bending moment diagram. Columns with fixed base are straight. Also the interior columns are always maintained straight.

#### **3.1.1. Main frame orientation**

Building should be oriented in such a way that the length is greater than the width. This will result in more number of lighter frames rather than less number of heavy frames. This also will reduce the wind bracing forces results in lighter bracing systems.

#### **3.1.2. Main frame types**

#### *3.1.2.1 Clear span*

Clear Span rigid frames are single gable frames and offer full-width clear space inside the building without interior columns. This type of frame is extensively used anywhere an unobstructed working area is desired in diverse applications such as auditoriums, gymnasiums, aircraft hangars, showrooms and recreation facilities.

The deepest part of the frame is the knee, the joint between the rafter and the column, which is generally designed as horizontal knee connection. An alternate design of knee joint is as vertical knee connection that is employed for flush side-wall construction. Clear Span rigid frames are appropriate and economical when:

i) Frame width is in the range 24m-30m.

ii) Headroom at the exterior walls is not critical.



*Figure 3.4 Column reaction of clear span frame*

#### *3.1.2.2 Multi – span*

When clear space inside the building is not the crucial requirement then Multi-Span rigid frames offer greater economy and theoretically unlimited building size. Buildings wider than around 90m experience a build-up of temperature stresses and require temperature load analysis and design. Multi-span rigid frames have straight interior columns, generally hot-rolled tube sections pin connected at the top with the rafter. When lateral sway is critical, the interior columns may be moment connected at the top with the rafter, and in such a situation built-up straight columns are more viable than hot-rolled tube columns.

Multi-Span rigid frame with an interior column located at ridge requires the rafter at ridge to have a horizontal bottom flange in order to accommodate horizontal cap plate. Multi-Span rigid frame is the most economical solution for wider buildings (width > 24m) and is used for buildings such as warehouses, distribution centers and factories. The most economical modular width in multi-span buildings is in the range 18m-24m.

The disadvantages of such a framing system include:

- The susceptibility to differential settlement of column supports.
- Locations of the interior columns are difficult to change in future.
- Longer un-braced interior columns especially for wider buildings.
- Horizontal sway may be critical and governing the design in case of internal columns pined with rafter.



*Figure 3.5 Multi - span frame*



*Figure 3.6 Column reaction of multi - span frame*

#### *3.1.2.3 Lean – to*

Lean-To is not a self-contained and stable framing system rather an add-on to the existing building with a single slope. This type of frame achieves stability when it is connected to an existing rigid framing. Usually column rafter connection at knee is pinned type, which results in lighter columns. In general, columns and rafters are straight except that rafters are tapered for larger widths (> 12m). For clear widths larger than 18m, tapered columns with moment resisting connections at the knee are more economical. Lean-To framing is typically used for building additions, equipment rooms and storage.

For larger widths "Multi-Span-Lean-To" framing can be adopted with exterior column tapered and moment connected at the knee.



*Figure 3.8 Column reaction of lean - to frame*

#### *3.1.2.4 Mono-slope*

Mono-slope or single-slope framing system is an alternative to gable type of frame that may be either Clear Span or multi-span. Mono-Slope configuration results in more expensive framing than the gable type. Mono-slope framing system is frequently adopted where:

- Rainwater needs to be drained away from the parking areas or from the adjacent buildings
- Larger headroom is required at one sidewall
- A new building is added directly adjacent to an existing building and it is required to avoid:
- The creation of a valley condition along the connection of both buildings.
- The imposition of additional loads on the columns and foundations of the existing building.



*Figure 3.9 Mono-slope frame*

For larger widths "mono-slope-multi-span" framing will be more economical when column free area inside the building is not an essential requirement.



*Figure 3.12 Mono-slope frame with 2 spans*



*Figure 3.13 Mono-slope frame with 3 spans*



*Figure 3.14 Mono-slope frame with 4 spans*

#### *3.1.2.5 Space saver*

Space Saver framing system offers straight columns, keeping the rafter bottom flange horizontal for ceiling applications with rigid knee connection. Selection of Space Saver is appropriate when:

- The frame width is between 6m to 18m and eave height does not exceed 6m.
- Straight columns are desired.
- Roof slope  $\leq 5\%$  are acceptable.
- Customer requires minimum air volume inside the building especially in cold storage warehouses



*Figure 3.16 Column reaction of space saver frame*

#### *3.1.2.6 Roof system*

A Roof System framing consists of beam (rafter) resting onto a planned or an existing substructure. The substructure is normally made of concrete or masonry. The rafter is designed in such a way to result in only vertical reaction (no horizontal reaction) by prescribing a roller support condition at one end. The roller supports are provided at one end by means of roller rods.







*Figure 3.18 Detail C – typical pinned arrangement Figure 3.19 Detail D – typical rolled arrangement* 

A Roof System is generally not economical for spans greater than 12m although it can span as large as 36m. This is due to fact that the Roof System stresses are concentrated at mid-span rather than at the knees.

### *3.1.2.7 Multi – gable*

Multi-Gable buildings are not recommended due to maintenance requirement of valley region, internal drainage and bracing requirement inside the building at columns located at valley. Especially in snow areas, Multi-Gable framing should be discouraged. However, for very wide buildings this type of framing offers a viable solution due to:

- Reduced height of ridge and thus the reduced height of interior columns
- Temperature effects can be controlled by dividing the frame into separate structural segments







*Figure 3.23 Typical detail at valley Figure 3.24 Recommended interior drainage arrangement*

Thus, Multi-Gable buildings are more economical than Multi-Span buildings for very wide buildings. Multi-Gable frames may be either Clear Spans or Multi-Spans. The columns at the valley location should be designed as rigidly connected to rafters on either side using a vertical type of connection.

### **3.1.3. Roof slope**

Optimum roof slope:



#### **3.1.4. Eave height**

Eave height is governed by:

- Clear height at eave (head clearance)
- Mezzanine clear heights below beam and above joist
- Crane beam/ Crane hook heights

Minimize eave height to the bare minimum requirement since the eave height affects the price of the building by adding to the price of sheeting, girts and columns. If columns are unbraced eave height affects the frame weight significantly. Also higher eave heights increase the wind loads on the building.

If eave height to width ratio becomes more than 0.8 then the frame may have a fixed based design in order to control the lateral deflection.



*Figure 3.25 Eave height in the building*

# **3.2. End Wall Systems**

### **3.2.1. Generals**

The standard end wall is designed as post & beam (all connections are pinned) the lateral stability is provided by the diaphragm action, in the absence of this shear diaphragm wind bracing are required.

End rigid frame are used in case of:

- Future extension is intended; in this case only wind posts are required.
- Crane running to the end wall
- Open for access condition prevails at the end wall
- X-bracing is not allowed at end wall in the case of by-framed end wall.

### **3.2.2. Post & Beam End Wall Rafter**

All end wall rafters are designed as simple beams over the end wall posts. They are comprised of built-up sections or hot rolled sections.

#### **Design concept**

**Under gravity load (Dead load + Live Load):** The top flange of the end wall rafters is under compression and it is braced against lateral torsional buckling at every purlin location.

Under uplift loads (Dead + Wind Load): The bottom flange of the end wall rafters is under compression and it is unbraced against lateral torsional buckling. The buckling length is the distance between end wall columns, this may significantly reduce the section bending capacity or flange braces are to be used.



### **3.2.3. End Wall Post**

All end wall posts are flush with the end wall structural line in end post & beam gable type and they are supporting end wall rafters. In all cases end wall posts are oriented so that end wall wind pressure is producing bending moments about the column major axis. End wall posts are comprised of built-up sections or hot rolled sections.

#### **Design concept**

End wall posts are the supporting elements for end wall girts/block walls for wind loads (pressure or suction) which produce bending moment about posts major axes, end wall posts are designed as simple beam supported at foundation level (base plate) and are connected with Spanner at roof purlins level.

For Post & Beam end walls additional vertical loads from end wall rafters are transmitted to posts producing axial loads (compression or tension).

End wall post buckling length about major axis is the column length.

Flanges unsupported length is depending on the end wall type and position.

## **3.3. Expansion Joints**

The purpose to add the expansion joint is separating the building into many areas in order that every area has allowable temperature deformation. When separating the foundation into 2 parts, expansion joint become settlement joint.

*Section L7 AISC 360-10* recommends that the length of building needs to be added expansion joint or research NRC 1974 for more.



Fig. 1: Maximum allowable building length without expansion joints for various design temperature changes.

Unless more specifc site information is available, most engineers assume a range of 50° to 70° F (10° to 21° C) for continuously heated and air-conditioned buildings. Using that assumption, most steel, rectangular, framed confguration buildings with symmetrical stiffness can tolerate 460 ft (140 m) between expansion joints. Or the designer can refer to section 11.1.2 TCVN 5575:2012 to know the length & width of building with expansion joint to be added.





*Figure 3.26 Purlin/girt connection at expansion joint*



*Figure 3.27 Eave strut connection at expansion joint*



*Figure 3.28 Roof & wall panels at expansion joint*



*Figure 3.29 Panel connection at expansion joint*



*Figure 3.30 Gutter at expansion joint*



*Figure 3.31 Expansion joint at ridge*

## **3.4. Bay Spacing**



*Figure 3.32 Bay spacing*

The most economical bay spacing is around 8m for the standard loads as following:



For greater loads than standard loads the economical bay spacing tends to decrease.

For buildings with heavy cranes (crane capacity  $> 10$  Tons) the economical bay spacing ranges between 6m and 7m.

Smaller end bays than interior bays will taper off the effect of higher deflection and bending moment in end bays as compared to interior bays and help reduce the weights of purlins/girts in the end bays. This will avoid the need of nested purlins/girts in the end bays and result in uniform size of purlin/girt sizes.

Some buildings require bay spacing more than 10m in order to have a greater clear space at the interior of the building in Multi-Span buildings. Such a situation can be handled by providing jack beams that support the intermediate frames without interior columns. Thus the exterior columns will have bay spacing of say 6m while the interior columns are spaced at 12m. Intermediate frames allow the purlin to span for 6m.



*Figure 3.33 Jack beam plan with bay Spacing 12m*



*Figure 3.34 Jack beam at side wall*



*Figure 3.35 Elevation of jack beam at side wall*



*Figure 3.36 Section A - Jack beam at side wall*



*Figure 3.37 Detail 1 - Jack beam at side wall*



*Figure 3.38 Interior jack beam*



*Figure 3.39 Elevation - Interior jack beam*



Figure 3.40 Jack beam at middle span of rafter



*Figure 3.41 Jack beam at intermediate span of rafter*

## **3.5. Bracing Systems Arrangement**

Bracing is a structural system used to provide stability in a structure in a direction where applied forces on that structure would otherwise make it unstable. Whether it is a force due to wind, crane or seismic applications, the bracing system will always eventually transmit that load down to the column base and then to the foundations. The rules of arranging different types of bracing systems are as follows:

### **3.5.1. Bracing for wind and seismic loads in the longitudinal direction**

1. In long buildings, braced bays shall be provided in intervals not to exceed 5 bays.

2. Sidewall bracing shall be generally placed in the same bays of roof bracing. This may not be possible at times due to openings in the sidewalls. In such cases, sidewall bracing shall be placed in bays adjacent to those containing the roof bracing with a consideration that load transfers to the adjacent bays.

3. Roof rod bracing shall not cross the ridgeline.

4. Cables/rods braces shall not exceed 15m in length. If a cross bracing contains rods longer than 15m, then the bracing should be broken to two sets of bracings with a strut member between them so that the rod/cable lengths shall not exceed 15 m.

5. Sidewall bracing shall be comprised of any one of the following types:

- **Cables**
- Rods or angles.
- Portal frame with/without rods or angles.

6. There shall be only one type of bracing in the same sidewall. Do not mix different types/materials in the same sidewall.

7. It is preferable to use only one type of wall bracing in the whole building otherwise the lateral loads (especially seismic loads) will not be divided equally between bracing lines. For cases when this will result in excessive weight for bracing system advanced calculation is to be done to determine the force that will be carried by each type depending on its stiffness and location.

8. Do not use rod/cable Ø25mm for roof bracing because its weight make a large vertical deflection.



*Figure 3.42 Cable or rod bracing at roof & wall of a braced bay*





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*Figure 3.45 Portal frame Figure 3.46 Cable or rod bracing with strut tube*



*Figure 3.47 Angle bracing with strut tube*

#### **3.5.2. Wind and seismic bracing in P&B end wall**

1. End wall bracing is not required for a fully sheeted P&B end wall with flush girt construction. If P&B end walls have by-framed girts then this end wall needs bracing.

2. If required, bracing in P&B end walls shall comprise cables or rods, unless otherwise specified by the customer. In such a case the end wall members shall be either built-up or hot-rolled members.

3. If an end wall requires bracing and the customer requests that no bracing to be placed in the plane of the end wall, then it is recommended that the load in the plane of the end wall is transferred back to the first rigid frame through additional roof bracing in the end bay.

#### **3.5.3. Crane Bracing**

1. In crane buildings, bracing has to be designed for longitudinal crane loads for top running or underhung cranes. The bracing shall be placed in intervals not to exceed 5 bays.

2. Longitudinal bracing for top running cranes shall be comprised of any one of the following types.

- Angles or pipes;
- Portal frame with rods (or angles);
- Portal frame without rods (or angles).

3. Longitudinal bracing for top running cranes shall be of only one type in the same longitudinal plane of a building.

4. Longitudinal bracing for underhung cranes shall consist of either rods or angles.

5. Lateral bracing for underhung cranes (attached to crane brackets), if any shall consist of either rods or angles.

6. Whenever a brace rod is used for crane bracing, the minimum diameter of that rod shall be 20mm.

7. A brace rod shall not exceed 15m in length. If angles are used the critical slenderness ratio of a bracing angle shall not exceed 300.

## **4.1. Main frame design procedure and constraints CHAPTER 4. MAIN FRAME AND CONNECTION**

**4.1.1. Design procedure** *4.1.1.1 Stress Unity Checks*

Combined Stress Unity Check

$$
\frac{f_a}{F_a} + \frac{f_{bx}}{F_{bx}} + \frac{f_{by}}{F_{by}} \le 1.0
$$

Where fa, fbx and fby are actual axial, major axis bending and minor axis bending stresses respectively. Fa, Fbx and Fby are corresponding allowable stresses. If section fails in combined unity check then check the allowable stresses:

(1) If **Mn/Omega Capacity** or **Pnc/Omega Capacity** are much lower than **Mn/Omega No LTB** or **Pnt/Omega Capacity** then it implies that the member is not properly braced then try one of the following:

	Pr	Pnc/Omega	Pnt/Omega
	Force	Capacity	Capacity
Axial	1252.377	2184.504	3250.850
	Mr	Mn/Omega	Mn/Omega
	Moment	Capacity	<b>No LTB</b>
Major Moment	3.762	327.434	327.434
Minor Moment	0.000	187.005	

*Figure 4 1 Stress check in SAP*

• For rafters and exterior columns (with sheeted side walls) adding flange braces with roof purlins or wall girts will adjust the allowable stresses for the unbraced flange.

• For exterior columns (without sheeted side walls) then providing EB (strut tubes) adequately connected to bracing system at an appropriate height would reduce the unbraced length and adjust the allowable stress.

• For interior I-section columns they can also be braced by means of EB if allowed and adequately connected to bracing system.

• For interior I-section columns that brace points cannot be added in the design then stress ratios can be improved by increasing flanges width or by minor adjustment in the flange thickness.

• For columns connected with mezzanine beams/joists columns are considered braced at mezzanine level.

• For columns supporting top running crane beam the columns are considered laterally braced at the level of carne beam top flange.

(2) If allowable stresses are sufficiently high and still the section is failing in unity check, then unity check ratio can be improved by increasing the following in the given order:

- Increasing the web depth
- Increasing the flange width
- Increasing the flange thickness
- (3) If Shear stress unit ratio  $f_V/F_V > 1.0$  increase web thickness.

#### *4.1.1.2 Controlling Deflections*

Refer Table 2 26 Deflection Limitations for Deflection Limitations.

If lateral deflection exceeds the prescribed limit (normally  $H/60$ ) then check the  $H/W$ idth ratio. If  $H/B > 0.75$ then fixing the base would result in more economical frame. If H/B<0.75 then increase the web depth at knee of both column and rafter (difference between knee depth of column and rafter < 200mm)

In multi-span frames before going for the option of fixing the exterior column at base, check whether fixing the tops of interior columns control the lateral sway. If not then fix the exterior column bases.

If Vertical deflection ∆v exceeds the prescribed limit (normally Span/180), increase the web depth at knee of both column and rafter. A slight increase in the rafter depth at ridge will also help control the vertical deflection.

#### **4.1.2. Design constraints** *4.1.2.1 Standard*

$$
\frac{b_w}{t_w} < 180 \, ; \, \frac{b_f}{t_f} < 31
$$
\n
$$
\frac{b}{b_f} < 5 \, ; \, \frac{t_f}{t_w} < 2.5
$$
\n
$$
kI
$$

Compression element:  $\frac{kL}{s} < 200$ *r*  $\lt$ 

Tension element:  $\frac{kL}{s} < 300$ *r*  $\lt$ 



### *4.1.2.2 Fabrication limitation for built-up section*





#### **NOTE**

Width of continuous flange should be constant along the one welded piece.

• Variation of thickness at any butt weld splice of continuous flange/web within the one welded piece should be limited to maximum 6mm.

• Width flange/web having constant width, use steel plate gouge like below table.



#### *4.1.2.3 Shipping limitation*

Maximum fabricated out-to-out length of the piece is **14m** for transportation by truck (in Vietnam), and **11.7m** for transportation by dry cargo container (foreign) except **13.5m** for Cambodia.

#### *4.1.2.4 Other guidelines*

- (1) At knee connection, maximum difference between column depth and rafter depth is 200mm.
- (2) In a tapered section, the minimum difference in web depth at start and end should be 100mm.
- (3) Minimum base plate thickness  $= 14$  mm.
- (4) Minimum base plate width  $= 164$  mm.
- (5) Minimum splice plate thickness = 12 mm.
- (6) Minimum splice plate width  $= 164$  mm.
- (7) Minimum anchor bolt diameter = M20 (except end-wall post M16).
- (8) Minimum splice bolt  $= M16$ .

#### *4.1.2.5 Optimization*

To produce the most economical frame profiles, let apply the following rules:

(1) Minimize number of splices in the columns and rafters by providing maximum possible lengths regardless of the material savings that can be produced otherwise. Section lengths should be multiple of 3m i.e., 3m, 6m, 9m and 12m in order to reduce the scrap.

- (2) In case of different bay spacing avoid using more than 3 frames.
- (3) Different frame should be adopted if saving of 5% on all frames with a minimum of 1.0ton is ascertained.

(4) When different frames have to be used due to different bay spacing, maintain the same web cuts for all such frames.

(5) Minimize the number of different flange widths in a frame. Maximum different widths of flanges in all the frames should preferably be less than three.

(6) As much as possible maintain uniformity in the base plate detail and anchor bolt sizes for all the frames.

(7) Try to locate the splices at the locations where the bending moment is least and/or where the depth is least in a frame.

(8) Try to follow the shape of bending moment diagram for the controlling load combination in the configuration of the frame by maintaining the stress unity check ratios closer to 1.

## **4.1.3. Instruction to build Sap2000 Model**

This section provides step-by-step instructions for building a basic SAP2000 model.



#### *4.1.3.1 Defining*

In this Step, the basic grid that will serve as a template for developing the model will be defined. Then a material will be defined and sections will be selected.

a. Setting Default Units (KN, m, C) through: **File > New Model Form.**



*Figure 4 2 New model form*
## b. Setting up geometry in 2 ways:

(1) Creating Generally Grid System: **New Model form > Grid Only**

(2) Define Grid System Data: **Define>Coordinate Systems/Grids>GLOBAL>Modify/Show System**



*Figure 4 3 Quick Grid Lines form*

*Figure 4 4 Define Grid System Date form*

# c. Define Materials through: **Define > Materials**



*Figure 4 5 Material Property Data*

## d. Define Frame Sections

#### Define Frame Section through: **Define/Section Properties/Frame Sections**



*Figure 4 6 I/Wide Flange Section form Figure 4 7 Nonprismatic Section Definition form*

#### Define Tendon Section through: **Define/Section Properties/Tendon Sections**



*Figure 4 8 Tendon Section Data form (Diameter 16mm)*

e. Define Load patterns, load cases and load combinations

- Define Load Patterns through: **Define > Load Patterns**
- Define Load Cases through: **Define > Load Cases**
- Define Load Combinations through: **Define > Load Combinations** or using **\*.s2k** text file

In "**1A - LOAD APPLICATION AND PURLIN - Under 18m**" file, go to sheet "2" and enter the number for each load pattern, click **COMBINATION** then click **EXPORT TO .TXT** which will display a window to specify position for saving file. Click the Save button to save file.



*Figure 4 9 Design loading combination sheet*

In SAP2000, you have to define at least one load combination before export .s2k text file. Click the **Define > Load Combinations** command, in the **Define Load Combinations** form which has just appeared, click the **Add New Combo** button and create an combination as shown in figure below.



*Figure 4 10 Create load combination COMB1*

After create any combination, click the **File > Export > Sap2000 .s2k Text File** command to display the **Choose Tables for Export to Text File** form shown in figure below.



*Figure 4 11 Choose Tables for Export to Text File form*

Make sure that all tables, load patterns, and Open File After Export option are selected before click the **OK** button. Copy all data in combination.txt text file and paste to .s2k text file at highlight text shown in figure below.



*Figure 4 11 Choose Tables for Export to Text File form*

Save and close file. Return to SAP and import this file. Click the **File > Import > SAP2000 .s2k Text File** command to display **Import Tabular Database** shown in figure below. Click the OK button, specify created .s2k text file in appeared window and click the Done button to import file.



*Figure 4 13 Import Tabular Database form*



*Figure 4 14 Click the Done button to finish importing .s2k text file*

## *4.1.3.2 Modeling*

## a. Draw Frame Objects

In this step, frame objects with the associated sections previously defined are drawn using the grids and snap-to options, and generated using **Edit>Draw Frame/Cable.**

The **Properties of Object** pop-up form for frames will appear as shown in figure below.

Properties of Object	×
Line Object Type	<b>Straight Frame</b>
Section	1C-250212
<b>Moment Releases</b>	Continuous
XY Plane Offset Normal	
<b>Drawing Control Type</b>	None <space bar=""></space>

*Figure 4 15 Properties of Object form*

Use default values as seen in figure above to draw because we will assign information to frames after dividing them with appropriate length.

## b. Draw Bracing System

To draw EB and Tendon/Cable system, using Draw Frame/Cable button.



*Figure 4 16 Bracing System*

## c. Edit Modeling

## **Divide frame objects**

SAP2000 allows members to be sub-divided into multiple objects after they are drawn to accommodate changes in geometry (this differs from the internal meshing done during analysis where the number of objects remains the same).

Select rafters and vertical lines, then click the **Edit > Edit Lines > Divide Frames** command to show Divide Selected Frames form. In this form, make sure that Break at intersection with selected Joints, Frames, Area Edges and Solid Edges is selected before click the OK button.



*Figure 4 17 Divide Selected Frames form*



*Figure 4 18 Replicate form*

#### **Assign member end releases and restraint**

- Assign Frame Releases through: **Assign > Frame > Releases/Partial Fixity**
- Assign Joint Restraints through: **Assign > Joint > Restraints**
- Assign Frame Sections through: **Assign > Frame > Frame Sections**
- **•** Assign Unbraced Length Ratio (ULR):

ULR is specified as a fraction of the frame object length. Multiplying these factor times the frame object length gives the unbraced length for the object. There are three types of ULR include ULR in major axis, minor axis, and lateral torsional buckling.

**(1) ULR (major)** is ULR for buckling about the frame object major axis (occur when object is compressed). This item is taken to be the maximum distance between two major bracing divided by actual length of object and it is usually specified as Program Determined.

**(2) ULR (minor)** is ULR for buckling about the frame object minor axis (occur when object is compressed). This item is taken to be the maximum distance between two bracing points (bracing member frame into web or flange of the braced member) divided by actual length of object.

**(3) ULR (LTB)** is ULR for lateral-torsional buckling for the frame object (occur when object is flexed). This item is taken to be the maximum distance between two bracing points (bracing member frame into flange of the braced member) divided by actual length of object.



## **NOTE**

Assume purlin spacing and girt spacing is **1.5m.**

**With rafters**, bracing members are purlins and flange braces. To conservative, just locations that have both purlin and flange brace are considered as bracing point.

Therefore, unbraced length is spacing between flange braces.

**With outer columns**, bracing members are girts and flange braces.

Therefore, unbraced length is the greater value of spacing between flange brace or distance from ground to the first flange brace.

**With inner columns**, bracing members are longitudinal steel members and portal frame. Unbraced length is the largest of unbraced lengths.

Steel Frame Design Overwrites for AISC 360-10 Item Description Unbraced length factor for lateral-Item Value ▲ torsional buckling for the frame  $\overline{1}$ Current Design Section Program Determined object. This item is specified as a  $\overline{2}$ Framing Type Program Determined fraction of the frame object length. Multiplying this factor times the  $\overline{3}$  $0<sub>meqa0</sub>$ Program Determined frame object length gives the unbraced Consider Deflection?  $\overline{4}$ No. length for the object. Specifying 0 5 Deflection Check Type Program Determined means the value is program determined. Program Determined 6 DL Limit, L /  $\overline{7}$ Super DL+LL Limit, L / Program Determined 8 Live Load Limit, L / Program Determined 9 | Total Limit, L7 Program Determined 10 | Total--Camber Limit, L/ Program Determined Program Determined 11 | DL Limit, abs Program Determined 12 | Super DL+LL Limit, abs 13 | Live Load Limit, abs Program Determined 14 | Total Limit, abs Program Determined 15 | Total-Camber Limit, abs Program Determined 16 Specified Camber Program Determined 17 Net Area to Total Area Ratio Program Determined 18 | Live Load Reduction Factor Program Determined 19 | Unbraced Length Ratio (Major) Program Determined 20 Unbraced Length Ratio (Minor)  $0.25$ 21 Unbraced Length Ratio (LTB)  $0.25$ 22 Effective Length Factor (K1 Major) Program Determined Explanation of Color Coding for Values: 23 Effective Length Factor (K1 Minor) Program Determined Blue: All selected items are program 24 Effective Length Factor (K2 Major) Program Determined  $\overline{\phantom{a}}$ determined **Black:** Some selected items are user Set To Prog Determined (Default) Values: Reset To Previous Valuesdefined Red: Value that has changed during All Items Selected Items All Items Selected Items the current session.  $0K$ Cancel

*Figure 4 19 Steel frame design overwrites form*

# d. Assign loads through Assign/Frame Loads.

In this step, the dead, live and wind loads will be applied to the model.

Calculate frame loads which will be assigned to frame using "**1A - LOAD APPLICATION AND PURLIN - Under 18m**" file.



## **NOTE**

It will be easier to assign frame load when frames are not be divided into different length members.

## *4.1.3.3 Analyzing*

In this part SAP2000 will assemble and solve the global matrix. The following steps are needed:

Setting Analysis Option through: **Analysis > Set Option:** check the available DOFs. If you are analyzing a plane truss, check UX and UY, leave the UZ, RX, RY and RZ blank.

From the **Analysis** menu, select **Run**.Click the **Run Now** button to start running.



*Figure 4 20 Set Load Cases to Run form*

When the analysis is finished, the program automatically displays a deformed shape view of the model, and the model is locked. Check the general stability of model by clicking the Start Animation button at the lower right hand of monitor. If any member is deformed unusually, you should unlock model and revise it.

## *4.1.3.4 Display the result*

- Displaying the deformed shape through: **Display > Show Deformed Shape**
- Return normal model through: F4 or the **Display > Show Undeformed Shape**
- Show Model Definition or Analysis Result through: Ctrl+T or the **Display > Show Table**

## *4.1.3.5 Design Steel Frame*

Click the **Design menu > Steel Frame Design > View/Revise Preferences** command. The **Steel Frame Design Preferences** form shown in figure below displays.



*Figure 4 21 Steel Frame Design Preferences form* 

Make sure that the Design Code is set to **AISC 360-10**, the Framing Type is set to **OMF**(ordinary-moment-frame) and the Design Provision is set to **ASD**.

Review the information contained in the other items and then click OK to accept the selections.

Click the **Design menu > Steel Frame Design > Select Design Combos** command to access the Design Load Combinations Selection form. Uncheck **Automatically Generate Code-Based Design Load Combinations**, then click OK to accept the changes.



*Figure 4 22 Design Load Combinations Selection form*

Click the **Design menu > Steel Frame Design > Start Design/ Check of Structure** command to start the steel frame design process. When the design is complete, stress ratio are displayed on the model as a color pallet.



*Figure 4 23 Model after checking*

Click the **Design menu > Steel Frame Design > Display Design Info** command to show **Display Steel Design Results (AISC 360-10)** form. In the **Design Output** drop-down list, select P-M Ratio Colors & Values, then click OK to accept.

To see detail information of checking strength, right click on the **1K** member to show the **Steel Stress Check Information** form.





Steel Stress Check Data AISC 360-10					$\mathbf{x}$
File					
		AISC 360-10 STEEL SECTION CHECK (Summary for Combo and Station)			Units KN, m, C $\vert \cdot \vert$
Units $E$ KN, $m$ , $C$					
Frame : 148 Length: 6.067	X Mid: 3.000 Y Mid: 8.000	Combo: COMB1 Shape: 1K	Frame Type: OMF	Design Tupe: Brace	
Loc : 1.517	Z Mid: 8.450	Class: Non-Compact		Princpl Rot: 0.000 degrees	
<b>Provision: ASD</b>	Analysis: Direct Analysis				
$D/C$ Limit=1.000	2nd Order: General 2nd Order		Reduction: Tau-b Fixed		
$A1phaPr/Pv=0.024$	AlphaPr/Pe=0.057 Tau b=1.000		$EA$ factor=0.800	$EI$ $Factor=0.800$	
$0$ megaB=1.670	OmegaC=1.670	OmegaTY=1.670	$0$ megaTF= $2.000$		
$0$ megaU=1.670	0 $neq$ a $U - R I = 1.500$	0 $neg$ a $UT = 1.670$			
$A = 0.005$	$133 = 2.402E - 04$	$r33=0.218$	$S33 = 8.135E - 04$	$A \cup 3 = 0.002$	
$J = 0.000$	$I22=5.152E-86$	$r22=0.032$	$S22=6.283E-05$	$Av2 = 0.003$	
$E=199947978.8$ RLLF=1.000	$FQ = 345000.000$ Fu=448000.000	$R_{U} = 1.072$	$z33=0.001$ $z22=9.753E-05$	$Cw = 0.000$	
STRESS CHECK FORCES & MOMENTS (Combo COMB1)					
Location	Pr	Mr33 Mr22	Ur2	Ur3 Tr	
1.517	$-26.092$	$-160.279$ $3.700E - 05$	$-40.525$	9.069E-04 $-1.846E - 05$	
PMM DEMAND/CAPACITY RATIO	$(H1-1b)$				
D/C Ratio:	$0.851 = 0.037 + 0.814 + 0.000$				
		= (1/2)(Pr/Pc) + (Mr33/Mc33) + (Mr22/Mc22)			
AXIAL FORCE & BIAXIAL MOMENT DESIGN		$(H1-1b)$			
Factor Major Bending	.L 4.167	K2 K1 1.000 1.000	<b>B1</b> 1.000	<b>B2</b> Cm 1.000 1.000	
<b>Minor Bending</b>	0.250	1.000 1.000	1.000	1.000 0.838	
	Lith	Kltb C <sub>b</sub>			
LTB.	0.250	1.000 1.343			
	Pr Force	Pnc/Omega Pnt/Omega Capacity Capacity			
Axial	$-26.092$	1035.000 353.527			
	Mr	Mn/Omega Mn/Omega			
	Moment	Capacity No LTB			
Major Moment	$-160.279$	196.790 196.798			
Minor Moment	$3.700E - 05$	13.717			
<b>SHEAR CHECK</b>					
	Ur Force	Un/Omega Stress Capacity Ratio	Status <b>Check</b>		
Major Shear	40.525	0.316 128.123	0K		
Minor Shear	$9.069E - 04$	284.594 $3.187E - 06$	<b>OK</b>		
BRACE MAXIMUM AXIAL LOADS					
	P	P			
Axial	Comp $-26.092$	<b>Tens</b> N/C			

*Figure 4 25 Steel Stress Check Data form*

Click the Details button to show more detail information.



## **NOTE**

Unbraced length ratio for major bending is 4.167 which is taken to be distance between two bracing point in major axis (25m) divided by actual length of member (6m). You should carefully review this ratio to ensure that the design process is consistent with your expectations.

To check whether all steel frames passed the stress check, click the **Design menu > Steel Frame Design > Verify All Members Passed** command to display as figure below.



*Figure 4 26 Checking capacity of all steel objects*

## *4.1.3.6 Check Deflection Limitt*

Refer Section 2.4 Serviceability Consideration for more details.

To optimize weight of building, unlock model, modify section and irritate above steps until total weight is smallest.

## **4.1.4. Export Joint Reaction**

## *4.1.4.1 Transform from wind load with return period wind speed of 700 years to wind load with those of 50 years*

In **Load Case Data – Linear Static** form, change **Scale Factor** for all Windload Case from **1** to **0.625** as Figure 4 27.

$$
\frac{V_T}{V_{50}} = 0.36 + 0.1 \ln(12T)
$$
\n
$$
\Rightarrow V_{50} = \frac{V_{700}}{0.36 + 0.1 \ln(12 \times 700)} = 0.791V_{700} \Rightarrow WL_{50} = 0.791^2 \times WL_{700}^2 = 0.626 \times WL_{700}^2
$$



*Figure 4 27 Load Case Data – Linear Static form*

# *4.1.4.2 Changing label name of joints*

Select the **Edit > Change Labels** command to access **Interactive Name Change** form. After enter all the information as figure below, select the **Edit > Auto Relabel > All In List** command to change label name.



*Figure 4 28 Change Label Form*





*Figure 4 29 Joints in X-Y Plan @Z=0 Figure 4 30 Label name after changing*

# *4.1.4.3 Displaying the Joint Forces/Joint Reactions*

# **Export Joint Reactions to Excel**

When steel columns locate on concrete columns, specify all steel columns

and show table **Element Joint Forces – Frames**.

When steel columns locate on foundations, specify all base joints

and show table **Joint Reactions**.

# **Create JOINT REACTIONS Results**

Open file "**7 - JOINT REACTIONS**" and move to sheet "**SteelColumn**", and then enter all the information about the project and Joint Labels Plan.

In Excel file that has just appeared, filter all base joints. Then insert one column on the left of column "E". Copy all data from column "B" to column "K".

In sheet "**InputS**" in file "**7 – JOINT REACTIONS**":

- Delete all data in area from columns A to J and from rows 4 to the end of sheet.
- Select cell "A4" and paste data into this. Select cell "X6" and change content in this cell to "Wind load Y".
- Move to sheet "SteelColumn" and click the HIDE button.
- Set Print Area before print PDF file.

Recheck all information; make sure that Vertical Reactions Vz for Dead load and Live load are always positive.

# **4.1.5. Design Example 4a: Low-Rise Building Sap Model**



*Figure 4 31 Low-rise building model*





# *4.1.5.1 Defining*



*Figure 4 33 Setting up geometry and default units*

## Define Section Properties as table below:

#### *Table 4 2 Frame Section Properties*



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# *4.1.5.2 Modeling* **Create general model**



*Figure 4 35 End wall frame*



## **Define load patterns and load combinations**

*Figure 4 36 Define Load Patterns form*

# **Assign load**



*Wind load 3 (WL3) Figure 4 38 Load applied on end-wall frame*

#### **Divide frame objects**

Draw vertical lines that are located at  $x=6$ m,  $x=17.5$ m,  $x=23.5$ m,  $x=26.5$ m,  $x=32.5$ m and  $x=44$ m as shown in figure below.



*Figure 4 39 Vertical lines to divide rafter*

Select rafters and vertical lines, then click the **Edit > Edit Lines > Divide Frames**  command to show Divide Selected Frames form. In this form, make sure that **Break at intersection with selected Joints, Frames, Area Edges and Solid Edges** is selected before click the OK button.



*Figure 4 40 Divide Selected Frames form*



*Figure 4 41 Main frame after dividing* 

*Figure 4 42 End wall frame*



*Figure 4 43 Main frame sections* 



*Figure 4 44 End wall frame sections*

#### **Calculate ULR for each member in frame**

- Assume purlin spacing and girt spacing are 1.5m.

- **With rafters**: unbraced length for 1K, 3K and 4K are 1.5m, for 2K-384148 are 3m and for KDH, IPE-200 are their length (no bracing)

- **With outer columns**: unbraced length for 1C and CDH are 5m (base on architectural drawing).

- **With inner columns** (2C264184): there are not any ST or portal frame, so unbraced length for this column is taken to be its length or ULR is 1.



#### *Table 4 3 Unbraced length ratio*

Select 1K section members and click the **Design > Steel Frame Design > View/Revise Overwrites** command to show **the Steel Frame Design Overwrites** for AISC 360-10 form.

At the Value column, type 0.25 for both **Unbraced Length Ratio (Minor) and Unbraced Length Ratio (LTB).**



*Figure 4 45 Steel frame design overwrites form*

Repeat these steps to assign ULR for the other member. **Replicate Frame**

Select frame at X-Z Plan @ Y=8 and then select the Edit > Replicate.

To generate 8 additional frames with 8m bay spacing, enter like below:



*Figure 4 46 Replicate form*

## **Draw Bracing System**



*Figure 4 47 Column brace system*

# *4.1.5.3 Analyzing*



*Figure 4 48 Model after running analysis*

## *4.1.5.4 Design Steel Frame*



*Figure 4 49 Model after checking*



*Figure 4 50 Main frame at plane Y=8*

To check whether all steel frames passed the stress check, click the **Design menu > Steel Frame Design > Verify All Members Passed** command to display as figure below.

SAP2000	
	All steel frames passed the stress/capacity check.

*Figure 4 51 Checking capacity of all steel objects*

## *4.1.5.5 Check Deflection Limitation*

Create combos for deflection checking:



#### **Vertical Deflection**

Click the **Display > Show Deformed Shape** command which will show a Deformed Shape form. Select **COMB1 (DL + LL)** from **Case/Combo Name**  drop-down list and click OK to accept.



*Figure 4 52 Deformed Shape form*

Checking if vertical deflection (U3) of mid points of rafter are excess deflection limit. Put the cursor at mid points of rafter to see if deflection of that point is excess the limit or not.





*Figure 4 54 Deflection of point 113 – 0.0755m*

#### **Horizontal Deflection**

Checking if horizontal deflection of combination CV of top of columns are excess deflection limit. Put the cursor at points located at the top of column to see if the horizontal deflection (U1 or U2) of that point is excess the limit or not.

Compute horizontal deflection limit:

$$
\Delta = \frac{b}{60} = \frac{8}{60} = 0.133 \, \text{m}
$$



*Figure 4 55 Deflection of point 28 – 0.0225m*



*Figure 4 56 Deflection of point 34 – 0.0221m*

## *4.1.5.6 Export joint reaction*

Transform from wind load with return period wind speed of 700 years to wind load with those of 50 years



*Figure 4 57 Load Case Data – Linear Static form*



*Figure 4 58 Labels after change*



*Figure 4 59 Column reactions file*

▼ Element Joint Forces - Frames Units: As Noted											
	Frame Text	Joint Text	<b>OutputCase</b> Text	<b>CaseType</b> Text	F1 KN	F <sub>2</sub> KN	F <sub>3</sub> KN	M1 KN-m	M2 $KN-m$		
▶	24	P15	<b>DL</b>	LinStatic	6.049	0	18.24	0	0		
	24	65	<b>DL</b>	LinStatic	$-6.049$	$\overline{0}$	$-14.976$	0	47.1819		
	24	P15	LL	LinStatic	13.418	$\overline{0}$	32.929	$\overline{0}$	$\overline{0}$		
	24	65	LL	LinStatic	$-13.418$	0	$-32.929$	0	104.6626		
	24	P15	WL1	LinStatic	$-16.6$	0	$-35.755$	$\overline{0}$	ū		
	24	65	WL1	LinStatic	11.281	0	35,755	$\overline{0}$	$-108.7342$		
	24	P15	WL <sub>2</sub>	LinStatic	$-15.227$	0	$-23.277$	$\overline{0}$	$\overline{0}$		
	24	65	WL <sub>2</sub>	LinStatic	2.269	$\overline{0}$	23.277	$\overline{0}$	$-68.237$		
	24	P15	WL3	LinStatic	0	0	0	0	0		
	24	65	WL3	LinStatic	n	$\overline{0}$	$\Omega$	$\overline{0}$	$\overline{0}$		
	28	P17	<b>DL</b>	LinStatic	$-6.049$	0	18.24	0	$\overline{0}$		
	28	78	<b>DL</b>	LinStatic	6.049	0	$-14.976$	$\overline{0}$	$-47.1819$		
	28	P17	LL	LinStatic	$-13.418$	0	32.929	$\overline{0}$	$\overline{0}$		
	28	78	LL	LinStatic	13.418	0	$-32.929$	$\overline{0}$	-104.6626		
	28	P17	WL1	LinStatic	2.835	0	$-22.773$	0	0		
	28	78	WL1	LinStatic	$-13.506$	0	22.773	$\overline{0}$	63.7311		
	28	P17	WL <sub>2</sub>	LinStatic	0.091	0	$-9.624$	$\overline{0}$	$\overline{0}$		
	28	78	WL <sub>2</sub>	LinStatic	$-3.124$	0	9.624	$\overline{0}$	12.5392		
	28	P17	WL3	LinStatic	0	0	$\overline{0}$	$\overline{0}$	$\vert 0 \vert$		
	28	78	WL3	LinStatic	۵I	$\overline{0}$	$\Omega$	$\Omega$	0  ٠		

*Figure 4 60 Element Joints Forces – Frame table*

	А	в	C	D	E	F	G	н	I	п	К	L
L			<b>TABLE: Element Joint Forces - Frames</b>									
2	Fram $\sqrt{*}$		Joint T OutputCa v CaseTy		$\overline{\phantom{a}}$	$\overline{\mathbf{v}}$ F <sub>1</sub>	$\overline{\phantom{a}}$ F <sub>2</sub>	$\overline{\mathbf v}$ F <sub>3</sub>	$\overline{\phantom{a}}$ M1	v M <sub>2</sub>	M <sub>3</sub>	▼ FrameEle
4	39	<b>P1</b>	<b>DL</b>	LinStatic		0.046	4.454E-08	4.103	0	$\Omega$	0.000004456 39-1	
5	39	<b>P1</b>	LL.	LinStatic		0.094	$-1.381E-07$	3.52	$\mathbf{0}$	$\Omega$	0.00001021 39-1	
3	39	<b>P1</b>	WL1	LinStatic		$-2.29$	3.974E-07	$-5.041$	0	0	$-0.00001329$ 39-1	
	$.0$ 39	<b>P1</b>	WL <sub>2</sub>	LinStatic		$-3.892$	1.578E-07	$-4.642$	$\overline{0}$	o	$-0.000006482$ 39-1	
2	39	<b>P1</b>	WL3	LinStatic		$-0.00693$	$-2.62$	$-9.232$	0	0	$-0.000945$ 39-1	
4	45	P <sub>2</sub>	DL	LinStatic		8.674E-18	6.306E-18	5.427	0	0	0.000003164 45-1	
.6	45	P <sub>2</sub>	LL	LinStatic		1.995E-17	3.163E-18	5.867	0	0	0.000007239 45-1	
	.8 45	P <sub>2</sub>	WL1	LinStatic		$-3.816E-17$	$-1.155E-17$	$-9.038$	0	0	$-0.000009329$ 45-1	
	$10 \, 45$	P <sub>2</sub>	WL <sub>2</sub>	LinStatic		1.18E-16	$-6.739E-18$	$-6.419$	$\mathbf 0$	0	$-0.000004362$ 45-1	
	245	P <sub>2</sub>	WL3	LinStatic		3.459E-16	$-4.216$	1.606	$\mathbf 0$	$\mathbf 0$	$-0.0006892$ 45-1	
	'4 47	P <sub>3</sub>	<b>DL</b>	LinStatic		$-4.337E-19$	$-5.117E-17$	5.843	0	o	0.000001241 47-1	
	6 47	P <sub>3</sub>	LL	LinStatic		$-8.674E-19$	$-2.341E-16$	6.353	$\mathbf 0$	$\Omega$	0.000003099 47-1	
	8 47	P <sub>3</sub>	WL1	LinStatic		0	2.878E-16	$-8.825$	$\overline{0}$	0	$-0.000004082$ 47-1	
	$10 \t 47$	P <sub>3</sub>	WL <sub>2</sub>	LinStatic		$-3.469E-18$	1.209E-16	$-6.087$	0	0	$-0.000001614$ 47-1	
	247	P <sub>3</sub>	WL3	LinStatic		$-1.49E-17$	$-4.586$	0.004948	0	0	$-0.000467$ 47-1	
	14 49	P <sub>4</sub>	<b>DL</b>	LinStatic		2.212E-17	$-6.696E-17$	5.531	0	0	$-1.967E-07$ 49-1	
	16 49	<b>P4</b>	LL	LinStatic		$-4.77E-18$	$-5.815E-16$	5.444	0	0	5.73E-08 49-1	
	8 49	P <sub>4</sub>	WL1	LinStatic		8.674E-18	8.165E-16	$-7.848$	$\mathbf 0$	$\Omega$	$-2.771E-07$ 49-1	
	$0$ 49	P <sub>4</sub>	WL <sub>2</sub>	LinStatic		$-3.123E-17$	$2.8E-16$	$-6.093$	0	$\bf{0}$	0.00000022 49-1	
	2 49	<b>P4</b>	WL3	LinStatic		$-5.658E-17$	$-4.956$	$-0.807$	0	$\Omega$	$-0.0002868$ 49-1	
	14 51	P <sub>5</sub>	<b>DL</b>	LinStatic		8.413E-17	$-4.799E-17$	6.055	0	$\Omega$	$-9.518E - 0751 - 1$	
	6 51	P <sub>5</sub>	LL.	LinStatic		$-1.301E-18$	4.268E-17	6.149	$\mathbf 0$	$\Omega$	$-0.00000165$ 51-1	
	$8 \, 51$	P <sub>5</sub>	WL1	LinStatic		1.084E-19	$-1.043E-16$	$-8.805$	0	$\bf{0}$	0.000001725 51-1	
00 51		P <sub>5</sub>	WL <sub>2</sub>	LinStatic		$-6.939E-18$	$-4.808E-18$	$-5.959$	0	0	0.000001136 51-1	
	2251	P <sub>5</sub>	WL3	LinStatic		$-4.479E-18$	$-5.326$	0.304	0	0	$-0.000127751-1$	
	54 155	P <sub>6</sub>	<b>I</b> DL	LinStatic		$-4.561E-16$	$-4.359E-07$	4.337	0	0	5.767E-19 155-1	
	56 155	P <sub>6</sub>	LL.	LinStatic		$-5.556E-15$	$-5.115E-07$	4.313	0	$\bf{0}$	$-1.627E-18$ 155-1	
	58 155	P <sub>6</sub>	WL1	LinStatic		0.077	3.508E-07	$-3.54$	$\mathbf 0$	$\Omega$	$-1.536E - 07$ 155-1	
	50 155	P <sub>6</sub>	WL <sub>2</sub>	LinStatic		0.078	2.328E-07	$-2.75$	$\overline{0}$	$\Omega$	$-1.081E-07$ 155-1	
	52 155	P <sub>6</sub>	WL3	LinStatic		$-9.178E - 17$	$-3.329$	$-9.442$	$\overline{0}$	$\mathbf{0}$	$-4.686E - 18$ 155-1	
	24 55	<b>P7</b>	<b>DL</b>	LinStatic		4.337E-19	$-1.793E-17$	6.055	0	0	9.518E-07 55-1	
	26 55	P7	LL	LinStatic		4.337E-19	1.63E-17	6.149	0	0	0.00000165 55-1	
	28 55	<b>P7</b>	WL1	LinStatic		$-3.469E-18$	$-2.704E-17$	$-5.605$	0	0	$-0.000001635$ 55-1	
	30 55	P7	WL <sub>2</sub>	LinStatic		$-3.469E-18$	2.121E-17	$-3.205$	0	0	$-2.239E-0755-1$	
32 55		<b>P7</b>	WL3	LinStatic		$-1.009E-18$	$-5.326$	0.304	0	0	0.0001277 55-1	
	94 57	P8	<b>DL</b>	LinStatic		$-1.262E-16$	$-7.876E-17$	5.531	0	0	1.967E-07 57-1	
	36 57	P <sub>8</sub>	LL	LinStatic		1.735E-18	$-6.953E-16$	5.444	0	$\Omega$	$-5.73E-08$ 57-1	
	98 57	P <sub>8</sub>	WL1	LinStatic		$-6.939E-18$	8.729E-16	$-4.005$	0	$\Omega$	$-4.853E - 0757 - 1$	
	<b>SALES</b>	mn	14.0.2	المتعدد ومسامره		4.2005.47	CADAC 47	4.704	$\mathbf{r}$	$\sim$	S SASE AZ EZ 4	

*Figure 4 61 Data of element joint forces after filter and sort*



## *Figure 4 62 Data after paste*

	А	B C	D	E	F	G	н			Κ	$\vert \vert$ $\sim$
	27 1. Sian Convention										
	Ζ Z										
		Joint									
	28 2. Reactions										
29	<b>Joint</b>	<b>OutputCase</b>	<b>Horizontal</b> <b>Reaction Hx</b>	<b>Horizontal</b> <b>Reaction Hy</b>	<b>Vertical</b> <b>Reaction</b> Vz	Moment Mx	<b>Moment</b> My	<b>Moment Mz</b>	<b>HIDE</b>		
30	<b>Text</b>	Text	KN	KN	KN	$KN-m$	$KN - m$	$KN-m$			
31	Point-P1	Dead Load	0.0	0.0	4.1	0.0	0.0	0.0			
32	Point-P1	Live Load	0.1	0.0	3.5	0.0	0.0	0.0			
33	Point-P1	<b>Left Windload Case1</b>	$-2.3$	0.0	$-5.0$	0.0	0.0	0.0			
34	Point-P1	Left Windload Case2	$-3.9$	0.0	$-4.6$	0.0	0.0	0.0			
35	Point-P1	Wind load Y	0.0	$-2.6$	$-9.2$	0.0	0.0	0.0			
36	Point-P2	<b>Dead Load</b>	0.0	0.0	5.4	0.0	0.0	0.0			
37	Point-P2	Live Load	0.0	0.0	5.9	0.0	0.0	0.0			
38	Point-P2	<b>Left Windload Case1</b>	0.0	0.0	$-9.0$	0.0	0.0	0.0			
39	Point-P2	<b>Left Windload Case2</b>	0.0	0.0	$-6.4$	0.0	0.0	0.0			
40	Point-P2	Wind load Y	0.0	$-4.2$	1.6	0.0	0.0	0.0			
41	Point-P3	Dead Load	0.0	0.0	5.8	0.0	0.0	0.0			
42	Point-P3	<b>Live Load</b> <b>Left Windload Case1</b>	0.0	0.0 0.0	6.4	0.0	0.0	0.0 0.0			
43	Point-P3		0.0		$-8.8$	0.0	0.0	0.0			
44	Point-P3	Left Windload Case2	0.0	0.0	$-6.1$	0.0	0.0				
45	Point-P3	Wind load Y	0.0	$-4.6$ 0.0	0.0	0.0	0.0	0.0 0.0			
46	Point-P4	Dead Load	0.0		$-5.5$	0.0	0.0				
47	Point-P4	<b>Live Load</b>	0.0	0.0 0.0	5.4	0.0	0.0	0.0 0.0			
48	Point-P4	<b>Left Windload Case1</b>	0.01		$-7.8$	0.0	0.0				
49	Point-P4	Left Windload Case2	0.01	0.01	$-6.1$	0.0	0.0	0.0 0.0			
50	Point-P4 Point-P5	Wind load Y Dead Load	0.0 0.0	$-5.0$ 0.0	$-0.8$ 6.1	0.0 0.0	0.0 0.0	0.0			
51 52	Point-P5	Live Load	0.0	0.0	6.1	0.0	0.0	0.0			
53	Point-P5	<b>Left Windload Case1</b>	0.0	0.0	$-8.8$	0.0	0.0	0.0			
54	Point-P5	Left Windload Case2	0.0	0.0	$-6.0$	0.0	0.0	0.0			
55	Point-P5	Wind load Y	0.0	$-5.3$	0.3	0.0	0.0	0.0			
56	Point-P6	Dead Load	0.0	0.0	4.3	0.0	0.0	0.0			
57	Point-P6	Live Load	0.0	0.0	4.3	0.0	0.0	0.0			
58	Point-P6	Left Windload Case1	0.1	0.0	$-3.5$	0.0	0.0	0.0			
59	Point-P6	Left Windload Case2	0.1	0.0	$-2.8$	0.0	0.0	0.0			
60	Point-P6	Wind load Y	0.0	$-3.3$	$-9.4$	0.0	0.0	0.0			
61	Point-P7	<b>Dead Load</b>	0.0	0.0	6.1	0.0	0.0	0.0			
62	Point-P7	Live Load	0.0	0.0	6.1	0.0	0.0	0.0			
	63 Point-P7	<b>Left Windload Case1</b>	0.0	0.0	$-5.6$	0.0	0.0	0.0			$\overline{\mathbf{v}}$
		<b>DESIGN SHEET OF BMB</b>	<b>InputS</b>	<b>SteelColumn</b>	$\left( \widehat{+}\right)$	÷. $\overline{4}$					$\mathbb F$

*Figure 4 63 Joint reactions after hide rows*

# **4.1.6. Design Example 4b: High-Rise Building Sap Model**



*Figure 4 64 High-rise building model*



*Figure 4 65 Setting up geometry and default units*









## Define Section Properties as table below:

#### *Table 4 4 Frame Section Properties*



# *4.1.6.2 Modeling*

# **Create general model**




*Main frame – GL iv16 to GL iv21 except GL iv18 Main frame – GL iv22*











*Figure 4 69 Jack beam and crane bracing*



*Figure 4 70 Crane runway beam & bracing – level +19.5m*



*Figure 4 71 Crane runway beam & bracing – level +19.5m*



*Figure 4 72 Mezzanine floor – level +6.5m, +13m*

## **Define load patterns and load combinations**

Click the **Define > Load Patterns** command to access the **Define Load Patterns** form. In **Load Patterns** area, create load patterns including DL, LL, FL1, FL2, FL3, WL1, WL2, WL3, WL4, WL5, WL6, WL7, WL8, EL1, EL2, CR1, CR2 as shown in figure below.



*Figure 4 73 Define Load Patterns form*



*Figure 4 74 Define Load Patterns form*

Select EL1 in Load Patterns area and then click the **Modify Lateral Load Pattern** button in Click to area to display the **1997 UBC Seismic Load Pattern** form. Enter all information as figure below.



*Figure 4 75 EL1 definition*

Select EL2 in **Load Patterns** area and then click the **Modify Lateral Load**  Pattern button in Click to area to display the **1997 UBC Seismic Load Pattern** form. Enter all information as figure below.

Note: Based on **UBC 97** standard, we have all the parameters for seismic zone 2B as above.



*Figure 4 76 EL2 definition*

Define mass source: click the **Define > Mass** Source command to show the Mass Source form.



*Figure 4 77 Mass Source form*

Click the **Modify/Show Mass source**  button to access the **Mass Source Data** form. Enter all information like figure below:



*Figure 4 78 Mass Source Data form*



*Figure 4 79 Design loading combination sheet*

Define load combinations follow ASCE 7-10 standard using .s2k text file.

In SAP2000, you have to define at least one load combination before export .s2k text file. Click the **Define > Load Combinations** command, in the **Define Load Combinations** form which has just appeared, click the **Add New Combo** button and create an combination as shown in figure below.



*Figure 4 80 Create load combination COMB1*

• After create any combination, click the **File > Export > Sap2000 .s2k Text File** command to display the **Choose Tables for Export to Text File** form shown in figure below.



*Figure 4 81 Choose Tables for Export to Text File form*

• Make sure that all tables, load patterns, and Open File After Export option are selected before click the **OK** button.

• Copy all data in combination.txt text file and paste to .s2k text file at highlight text shown in figure below.

```
TABLE: "LOAD PATTERN DEFINITIONS"
  LoadPat=DL DesignType=DEAD SelfWtMult=1
   \texttt{LoadPat=LL} \hspace{0.3cm} \texttt{DesignType=LIVE} \hspace{0.3cm} \texttt{SelfWtMult=0}\texttt{LoadPat=WL1} \qquad \texttt{DesignType=WIND} \qquad \texttt{SelfWtMult=0}AutoLoad=None
   LoadPat=WL2
                 DesignType=WIND
                                    SelfWtMult=0
                                                    AutoLoad=None
   LoadPat=WL3 DesignType=WIND SelfWtMult=0 AutoLoad=None
TABLE: "AUTO WAVE 3 - WAVE CHARACTERISTICS - GENERAL"
   WaveChar=Default WaveType="From Theory" KinFactor=1
SWaterDepth=45 WaveHeight=18
                                 WavePeriod=12
WaveTheory=Linear
       "COMBINATION DEFINITIONS"
TABLE:
  ComboName=COMB1    ComboType="Linear Add"
                                               AutoDesign=No
CaseType="Linear Static"
                           CaseName=DL ScaleFactor=1
SteelDesign=None ConcDesign=None
                                      AlumDesign=None
ColdDesign=None
TABLE: "FUNCTION - RESPONSE SPECTRUM - USER"
   Name=UNIFRS Period=0 Accel=1 FuncDamp=0.05
   Name=UNIFRS Period=1
                            Accel=1TABLE: "FUNCTION - TIME HISTORY - USER"
  Name=RAMPTH Time=0 Value=0
   Name=RAMPTH Time=1 Value=1
   Name = RAMPTH Time = 4Value=1
   Name=UNTFTH
                 Time=0Value=1
   Name=UNIFTH Time=1 Value=1
TABLE: "FUNCTION - POWER SPECTRAL DENSITY - USER"
  Name=UNIFPSD Frequency=0 Value=1
   Name=UNIFPSD Frequency=1 Value=1
TABLE: "FUNCTION - STEADY STATE - USER"
  Name=UNIFSS Frequency=0 Value=1
   Name=UNIFSS Frequency=1 Value=1
TABLE: "GROUPS 1 - DEFINITIONS"<br>GroupName=ALL Selection=Yes SectionCut=Yes Steel=Yes
Concrete=Yes Aluminum=Yes ColdFormed=Yes Stage=Yes<br>Bridge=Yes AutoSeismic=No AutoWind=No SelDesSteel=No
SelDesAlum=No SelDesCold=No MassWeight=Yes Color=Red
TABLE: "JOINT PATTERN DEFINITIONS"
  Pattern=Default
TABLE: "MASS SOURCE"
  MassSource=MSSSRC1
                       Elements=Yes Masses=Yes Loads=No
IsDefault=Yes
```
*Figure 4 82 Substitute combination defined by SAP with combination exported from excel.*

Save and close file. Return to SAP and import this file. Click the **File > Import > SAP2000 .s2k Text File** command to display **Import Tabular Database** shown in figure below. Click the OK button, specify created .s2k text file in appeared window and click the Done button to import file.



*Figure 4 83 Import Tabular Database form*



*Figure 4 84 Click the Done button to finish importing .s2k text file*

## **Assign loads**

In this step, the dead, live and wind loads will be applied to the model.

Calculate wind loads which will be assigned to frame using "**1A - LOAD APPLICATION AND PURLIN – All Height**" file.

• In sheet "**Input**", enter all the information as below:



*Figure 4 85 Input sheet*

• Go to sheet "**Left Monoslope Roof**" and then click the Calculation button to compute wind load blowing from left to right.



*Figure 4 86 Left Monoslope Roof sheet*

• Go to sheet "**Right Monoslope Roof**" and then click the Calculation button to compute wind load blowing from right to left.



*Figure 4 87 Right Monoslope Roof sheet*

Calculate dead load and floor load using "**2 - COMPOSITE BEAM & DECKING**" file.

## In sheet "**1**", enter all the information as figure below:



*Figure 4 88 Calculate load for mezzanine floor*

Use 2.96 kN/m2 for dead load and 2 kN/m2 for live load to apply on mezzanine floor. Calculate crane load using "**3 - CRANE BEAM DESIGN**" file. Enter all the information as figure below:



*Figure 4 89 Crane load for bay spacing 8m*



*Figure 4 90 Crane load for bay spacing 24m and 31m*

# **Assign Load**

After calculate load applied on frame, we assign them to frame as shown in these figure below.





*Wind load 1 (WL1)*



*Wind load 4 (WL4)*





*Wind load 6 (WL6)*



*Wind load 7 (WL7)*

*Wind load 8 (WL8)*



• End wall frame









*Wind load 1 (WL1)*

*Wind load 2 (WL2)*



*Wind load 3 (WL3)*

*Wind load 4 (WL4)*



*Wind load 7 (WL7*

*Wind load 8 (WL8)*

*Figure 4 92 Load applied End-wall Frame*



*Deal load (DL) – level +6.5m*



*Deal load (DL) – level +13m*



*Floor load 1 (FL1)*





*Floor load 3 (FL3)*

*Figure 4 93 Mezzanine floor*

• Crane bracket



*Figure 4 94 Crane load 1 (CR1)*



### **Divide frame objects and assign member section, end releases and restraint**

By the same manner in modeling low-rise building, we have model as shown in figure below:



*Main frame – GL iv4 to GL iv14*

*Main frame – GL iv15 & GL iv18*



*Main frame – GL iv16 to GL iv21 except GL iv18*

*Main frame – GL iv22*



*Elevation – GL ivB*



*Elevation – GL ivA*



*Jack beam and crane bracing*

뗆	ST264164 CR (500~800~500)	뗆	ST264164 띪 CR (500~800~500)	ST264164 CR (500~800~500)	띪	ST264164 CR (500~800~500)	띪	ST264164 CR (500~800~500)	띪	ST264164 CR (500~800~500)	꾒	$\overline{\phantom{a}}$ ←
뙶	CR (500~800~500) ST264164	뙶	CR (500~800~500) 띪 ST264164	CR (500~800~500) ST264164	띪	CR (500~800~500) ST264164	띪	CR (500~800~500) ST264164	副	CR (500~800~500) ST264164	뙶	$\overline{\phantom{0}}$

*Crane runway beam & bracing – level +19.5m*



*Crane runway beam & bracing – level +19.5m*



*Mezzanine floor – level +6.5m, +13m*

## **Calculate ULR for each member in frame**

- Assume purlin spacing and girt spacing are 1.5m.

- **With rafters**: unbraced length for 1K, 1KA, 5K, 3K are 1.5m, for 2K and 4k766184 are 3m and for KDH is their length (no bracing).

- **With outer columns**: unbraced length for C1, C2, C3 and CDH are 2.5m (based on architectural drawing).

- With the other members, set ULR as Program Determined.

### *Table 4 5 Unbraced length ratio*



Select 1K section members and click the **Design > Steel Frame Design > View/Revise Overwrites** command to show the **Steel Frame Design Overwrites** for AISC 360-10 form. At the Value column, type **0.25** for both **Unbraced Length Ratio (Minor)** and **Unbraced Length Ratio (LTB)**.



*Figure 4 96 Steel frame design overwrites form*

Repeat these steps to assign ULR for the other member.

# **Draw Bracing System**





*End wall GL iv1 End wall GL iv22*

 $\overline{5}$ 



Elevation GL ivB



*Elevation GL ivA*

# *4.1.6.3 Analyzing*



*Figure 4 98 Model after running analysis*

# *4.1.6.4 Design Steel Frame*



*Figure 4 99 Model after checking*

To see detail information of checking strength, right click on the **1K** member to show the **Steel Stress Check Information** form.

624 Frame ID					l1Κ Analysis Section					
Design Code	AISC 360-10				Design Section					
<b>COMBO</b>								STATION /----MOMENT INTERACTION CHECK-----//-MAJ-SHR---MIN-SHR-/		
ID	LOC.	RATIO	$=$		AXL + B-MAJ + B-MIN			RATIO	RATIO	
<b>COMB272</b>	3.00	$0.125(C) = 0.005 + 0.054 + 0.067$						0.060	0.001	
<b>COMB272</b>	6.00	$0.122(C) = 0.006 + 0.003 + 0.113$						0.025	0.001	
<b>COMB273</b>	0.00	$0.102(C) = 0.004 + 0.086 + 0.012$						0.100	0.001	
<b>COMB273</b>	3.00	$0.093(C) = 0.004 + 0.025 + 0.064$						0.050	0.001	
<b>COMB273</b>	6.00	$0.141(C) = 0.005 + 0.025 + 0.111$						0.018	0.001	
COMB274	0.00	$0.604(T) = 0.009 + 0.594 + 0.000$						0.648	0.000	
Modify/Show Overwrites- <b>Overwrites</b>		Display Details for Selected Item-		Details					Display Complete Details: Tabular Data	
⊙ Strenath	<b>C</b> Deflection			$\Box$ OK $\Box$	Cancel				Stylesheet: Default Table Format File	

*Figure 4 100 Steel Stress Check Information form*

Click the **Details** button to show more detail information.

Steel Stress Check Data AISC 360-10							
File							
							Units KN, m, C $\vert \mathbf{v} \vert$
AISC 360-10 STEEL SECTION CHECK Units				(Summary for Combo and Station)			
: KN, m, C							
Frame : 624	X Mid: 2.996		Combo: COMB274		<b>Design Type: Brace</b>		
Length: 6.000	Y Mid: 16.000	Shape: 1K		Frame Type: OMF			
Loc : 0.000	Z Mid: 23.150		<b>Class: Slender</b>		Princpl Rot: 0.000 degrees		
<b>Provision: ASD</b> $D/C$ Limit=1.000	Analysis: Direct Analysis 2nd Order: General 2nd Order			<b>Reduction: No Modification</b>			
$\alpha$ lphaPr/Pu=0.017	AlphaPr/Pe=0.009	Tau b=1.000		EA factor=1.000	EI factor=1.000		
$0$ megaB=1.670	$0$ mega $0 = 1.670$		OmegaTY=1.670	$0$ megaTF $= 2.000$			
$0$ megaU=1.670	$0$ meqa $0 - RI = 1.500$		OmegaUT=1.670				
$A = 0.011$	$133 = 0.001$	$r33=0.363$		$S33 = 0.003$	$A \cup 3 = 0.005$		
$J = 0.000$	$I22=2.798E-05$	$r22=0.051$		$S22=2.256E-04$	$Au2=0.005$		
E=199947978.8	Fu=345000.000	$Ru=1.072$		$233 = 0.004$	$Cw = 5.525E - 06$		
RLLF=1.000	Fu=448000.000			z22=3.462E-04			
STRESS CHECK FORCES & MOMENTS (Combo COMB274)							
Location	Pr	Mr33	Mr22	Ur <sub>2</sub>	Ur3	Tr	
0.000	39.372	368.641	0.053	88.682	$-0.002$	$1.802E - 04$	
D/C Ratio:	$0.604 = 0.009 + 0.594 + 0.000$	$(H1.2, H1-1b)$					
PMM DEMAND/CAPACITY RATIO				= $(1/2)(Pr/PC)$ + $(Mr33/Mc33)$ + $(Mr22/Mc22)$			
AXIAL FORCE & BIAXIAL MOMENT DESIGN Factor	-U	K1	$(H1.2, H1-1b)$ K2	<b>B1</b>	<b>B2</b>	$c_{m}$	
<b>Major Bending</b>	3.337	1.000	1.000	1.000	1.000	1.000	
<b>Minor Bending</b>	0.250	1.000	1.000	1.000	1.000	1.000	
LTB	Lith	Kltb	$c$ 1.282				
	0.250	1.000					
	Pr	Pnc/Omega	Pnt/Omega				
	Force	Capacity	Capacity				
Axial	39.372	1220.199	2215.437				
	Mr	Mn/Omega	Mn/Omega				
	Moment	Capacity	No LTB				
Maior Moment	368.641	620.908	620.908				
Minor Moment	0.053	53.512					
	Ur	Un/Omega	<b>Stress</b>	<b>Status</b>			
	Force	Capacity	Ratio	Check			
Major Shear	88.682	136.775	8.648	0K			
<b>Minor Shear</b>	0.002	676.283	$2.652E - 06$	0K			
	P	P					
<b>SHEAR CHECK</b> BRACE MAXIMUM AXIAL LOADS Axial	Comp 50.678	<b>Tens</b> N/C					

*Figure 4 101 Steel Stress Check Data form*



# **NOTE**

Unbrace length ratio for major bending is 3.337 which is taken to be distance between two bracing point in major axis (20m) divided by actual length of member (6m). You should carefully review this ratio to ensure that the design process is consistent with your expectations.

To check whether all steel frames passed the stress check, click the **Design menu > Steel Frame Design > Verify All Members Passed** command to display as figure below.



*Figure 4 102 Checking capacity of all steel objects*

# *4.1.6.5 Check Deflection Limitation*

Create combos for deflection checking:



### **Checking if vertical deflection (U3) of mid points of rafter are excess deflection limit.**

Compute deflection limit:  $\Delta = \frac{L}{100} = \frac{20}{100} = 0.167$ 120 120  $\frac{L}{20} = \frac{20}{120} = 0.167$  m

Put the cursor at mid points of rafter to see if deflection of that point is excess the limit or not.



*Deflection of point 563 – 0.0109m Deflection of point 538 – 0.0212m*

Click the Display > Show Deformed Shape command which will show the **Deformed Shape** form. Select **CV**  from **Case/Combo Name** drop-down list and click OK to accept.

**Checking if horizontal deflection of crane bracket are excess deflection limit**.

Compute horizontal deflection limit:  $\Delta = \frac{b}{100} = \frac{19.5}{100} = 0.195$ 100 100  $\frac{b}{\cos \theta} = \frac{19.5}{100} = 0.195$ 

Put the cursor at points located at the crane bracket to see if the horizontal deflection (U1 or U2) of that point is excess the limit or not.



*Deflection of point 249 – 0.1094m*

*Deflection of point 183 – 0.1237m*

## **Checking if vertical deflection of girder are excess deflection limit.**

Compute vertical deflection limit:  $\Delta = \frac{L}{\Delta t_0} = \frac{10}{24.0} = 0.042$ 240 240  $L = \frac{10}{24.0} = 0.042$  *m* 

Put the cursor at points located at the middle of girder to see if the vertical deflection (U1 or U2) of that point is excess the limit or not.



*Deflection of point 249 – 0.1094m*



*Deflection of point 196 – 0.0083m*

# *4.1.6.6 Export joint reaction*

Transform from wind load with return period wind speed of 700 years to wind load with those of 50 years



*Figure 4 103 Load Case Data – Linear Static form*



Changing label name of joints

*Figure 4 104 Labels after change*



*Figure 4 105 Column reactions file*



## **NOTE**

Change Collateral load to Earthquake load in Design load area. Do not insert any more rows or column in this sheet.

Run the model and specify all joints in active plan. Select the **Display > Show Tables** command to show Choose Tables for Display form. Make sure that the Table: **Joint Reactions** item and **DL, LL, FL1, FL2, FL3, CR1, CR2, WL1, WL2, WL3, WL4, WL5, WL6, WL7, WL8, EL1, EL2** in the Select Load Cases form are selected before click the OK button.

Units: As Noted				Joint Reactions							
Joint Text	<b>OutputCase</b> Text	<b>CaseType</b> Text	F <sub>1</sub> KN	F <sub>2</sub> KN	F <sub>3</sub> KN	M1 $KN-m$	M <sub>2</sub> KN-m	M3 KN-m			
P <sub>1</sub>	<b>DL</b>	LinStatic	13.56	2.351	145.032	0	0.615	0.0004886			
P <sub>1</sub>	LL	LinStatic	$-0.016$	0.308	5.022	$\overline{0}$	$-0.0543$	0.000203			
P <sub>1</sub>	WL1	LinStatic	$-34.426$	21.371	$-115.853$	$\vert 0 \vert$	$-26.1868$	$-0.0077$			
P <sub>1</sub>	WL <sub>2</sub>	LinStatic	$-36.311$	11.487	$-128.174$	0	$-28.7126$	$-0.0025$			
P <sub>1</sub>	WL3	LinStatic	$-48.534$	22.579	$-156.256$	$\overline{0}$	$-36.5107$	$-0.0067$			
P <sub>1</sub>	WL4	LinStatic	$-38.564$	12.268	$-128.847$	$\vert 0 \vert$	$-30.3766$	$-0.0016$			
P <sub>1</sub>	WL5	LinStatic	11.909	$-25.65$	$-77.189$	$\overline{0}$	13.1126	0.0035			
P <sub>1</sub>	WL6	LinStatic	43.611	21.896	217.576	$\overline{0}$	33.9604	$-0.0141$			
P <sub>1</sub>	WL7	LinStatic	36.851	12.104	193.237	0	27.2765	$-0.0086$			
P <sub>1</sub>	WL8	LinStatic	12.394	0.399	2.517	$\overline{0}$	13.1879	0.0017			
P <sub>1</sub>	EL1	LinStatic	$-21.793$	$-0.258$	$-99.633$	$\vert 0 \vert$	$-15.5845$	0.0015			
P <sub>1</sub>	EL <sub>2</sub>	LinStatic	$-0.553$	$-23.909$	$-89.724$	$\overline{0}$	$-0.0973$	$-0.1776$			
P <sub>1</sub>	CR1	LinStatic	7.419	$-1.959$	130.355	$\overline{0}$	4.925	$-0.0291$			
P <sub>1</sub>	CR <sub>2</sub>	LinStatic	$-6.089$	$-0.002696$	$-7.305$	0	$-4.1668$	0.0003708			
P <sub>1</sub>	FL <sub>1</sub>	LinStatic	7.622	0.864	63.127	$\vert 0 \vert$	0.3299	0.0002197			
P <sub>1</sub>	FL <sub>2</sub>	LinStatic	5.923	0.837	48.585	$\overline{0}$	$-0.6089$	0.0004404			
P <sub>1</sub>	FL <sub>3</sub>	LinStatic	1.698	0.027	14.541	0	0.9387	$-0.0002208$			
P <sub>2</sub>	DL	LinStatic	$-13.02$	$-0.002652$	240.528	0.1646	Ō	$-0.000006594$			
P <sub>2</sub>	LL	LinStatic	$-0.217$	0.001389	9.284	0.0007146	0	$-0.000006527$			
P <sub>2</sub>	WL1	LinStatic	$-25.872$	71.896	104.692	$-340.8842$	0	0.0008783			
P <sub>2</sub>	WL <sub>2</sub>	LinStatic	$-25.105$	38.505	111.136	$-182.5022$	01	0.0005207			

*Figure 4 106 Element Joints Forces – Frame table*



*Figure 4 107 Data of element joint forces after filter and sort*
	A	B	Ċ	D	Ė	F	Ġ	н	I	<b>I</b>	K	L	$AE$ $\rightarrow$	
1	TABLE:	<b>Joint Reactions</b>												
2	Joint		DutputCas(CaseType StepType		F <sub>1</sub>	F <sub>2</sub>	F <sub>3</sub>	M1	M <sub>2</sub>	M <sub>3</sub>				
3	Text	<b>Text</b>	<b>Text</b>	<b>Text</b>	KN	KN	KN	$KN-m$	KN-m	KN-m				
4	<b>P1</b>	<b>DL</b>	LinStatic		13.56	2.351	145.032	$\mathbf 0$	0.615	0.0004886				
5	P <sub>1</sub>	Щ	LinStatic		$-0.016$	0.308	5.022	$\mathbf{0}$	$-0.0543$	0.000203				
6	<b>P1</b>	WL1	LinStatic		$-34.426$	21.371	$-115.853$	$\bf{0}$	$-26.1868$	$-0.0077$				
7	<b>P1</b>	WL <sub>2</sub>	LinStatic		$-36.311$	11.487	$-128.174$	0	$-28.7126$	$-0.0025$				
8	<b>P1</b>	WL3	LinStatic		$-48.534$	22.579	$-156.256$	0	$-36.5107$	$-0.0067$				
9	<b>P1</b>	WL4	LinStatic		$-38.564$	12.268	$-128.847$	$\bf{0}$	$-30.3766$	$-0.0016$				
10	P1	WL5	LinStatic		11.909	$-25.65$	$-77.189$	0	13.1126	0.0035				
11	P1	WL6	LinStatic		43.611	21.896	217.576	0	33.9604	$-0.0141$				
$12$ P1		WL7	LinStatic		36.851	12.104	193.237	0	27.2765	$-0.0086$				
13 P1		WL8	LinStatic		12.394	0.399	2.517	0	13.1879	0.0017				
14 P1		EL <sub>1</sub>	LinStatic		$-21.793$	$-0.258$	$-99.633$	$\bf{0}$	$-15.5845$	0.0015				
15 P1		EL <sub>2</sub>	LinStatic		$-0.553$	$-23.909$	$-89.724$	0	$-0.0973$	$-0.1776$				
16 P1		CR <sub>1</sub>	LinStatic		7.419	$-1.959$	130.355	0	4.925	$-0.0291$				
17	$ p_1 $	CR <sub>2</sub>	LinStatic		$-6.089$	$-0.002696$	$-7.305$	$\bf{0}$	$-4.1668$	0.0003708				
18	P1	FL <sub>1</sub>	LinStatic		7.622	0.864	63.127	0	0.3299	0.0002197				
19	P1	FL <sub>2</sub>	LinStatic		5.923	0.837	48.585	$\bf{0}$	$-0.6089$	0.0004404				
20 P1		FL3	LinStatic		1.698	0.027	14.541	$\bf{0}$	0.9387	$-0.0002208$				
$21$ P <sub>2</sub>		DL	LinStatic		$-13.02$	$-0.002652$	240.528	0.1646	0	$-6.594E-06$				
22 P <sub>2</sub>		LL.	LinStatic		$-0.217$	0.001389	9.284	0.0007146	0	$-6.527E-06$				
23 P <sub>2</sub>		WL1	LinStatic		$-25.872$	71.896	104.692	$-340.8842$	0	0.0008783				
24 P <sub>2</sub>		WL <sub>2</sub>	LinStatic		$-25.105$	38.505	111.136	$-182.5022$	0	0.0005207				
25 P <sub>2</sub>		WL3	LinStatic		$-38.691$	71.907	188.878	$-340.9069$	0	0.0008838				
26 P <sub>2</sub>		WL4	LinStatic		$-27.666$	38.511	151.135	$-182.4868$	0	0.0005309				
$27$ P <sub>2</sub>		WL5	LinStatic		2.47	$-65.006$	$-41.431$	326.4516	0	$-0.0006454$				
28 P <sub>2</sub>		WL6	LinStatic		31.745	71.869	$-203.171$	$-341.1807$	0	0.0004952				
29 P <sub>2</sub>		WL7	LinStatic		29.694	38.479	$-184.574$	$-182.7928$	0	0.0001412				
30 P2		WL8	LinStatic		2.31	0.571	$-39.448$	$-3.3364$	0	0.0001707				
31 P <sub>2</sub>		EL <sub>1</sub>	LinStatic		$-18.885$	0.0062	95.678	0.0953	0	0.0001161				
32 P <sub>2</sub>		EL <sub>2</sub>	LinStatic		0.251	$-45.597$	$-5.425$	259.1744	0	$-0.0005969$				
33 P <sub>2</sub>		CR <sub>1</sub>	LinStatic		4.892	$-0.054$	$-24.413$	0.4486	0	0.00000749				
34 P <sub>2</sub>		CR <sub>2</sub>	LinStatic		$-6.019$	0.003564	43.775	0.045	0	0.00001829				
35 P <sub>2</sub>		FL <sub>1</sub>	LinStatic		$-7.292$	$-0.001562$	101.217	0.0874	0	9.21E-07				
36 P <sub>2</sub>		FL <sub>2</sub>	LinStatic		$-8.073$	0.009305	61.738	0.017	0	0.00004952				
37	P <sub>2</sub>	FL3	LinStatic		0.782	$-0.011$	39.479	0.0705	0	$-0.0000486$				
38	P <sub>3</sub>	DL	LinStatic		12.685	$-0.007224$	236.806	0.206	0	$-0.00001481$				
39 P3		LL	LinStatic		$-0.051$	$-0.003801$	5.423	0.0475	0	$-0.00001405$				
40 P3		WL1	LinStatic		$-26.321$	72.072	$-155.079$	$-345.5814$	0	$-0.0013$				
41 P3		WL <sub>2</sub>	LinStatic		$-25.027$	38.573	$-144.142$	$-184.9689$	0	$-0.0006286$				$\overline{\mathbf{v}}$
	4 $\mathbf{b}$		<b>DESIGN SHEET OF BMB</b>		<b>InputS</b>	SteelColumn	$^{\rm (+)}$	$\blacktriangleleft$					D.	

*Figure 4 108 Data after paste*

Select area from cell "W11" to cell "Z11" and click the Home > Insert command to insert more cells above active area. Revise load case name as shown in figure below.



Move to sheet "**SteelColumn**" and click the HIDE button.



### *Figure 4 109 Joint reactions after hide rows*

Select area from cell "A1" to cell "I795" and select the Page Layout > Print Area > Set Print Area command. Recheck all information; make sure that Vertical Reactions Vz for Dead load and Live load are always positive.

# **4.2. Pinned Base Plate Design**

## **4.2.1. Input**  *4.2.1.1 Loading*

Run the model and specify all joints that connecting as pin to foundation and have the same column section. Select the **Display > Show Tables** command to show **Choose Tables for Display** form. Make sure that the **Table: Joint Reactions** item and **all combinations** in the **Select Load Cases** form are selected before click the OK button.



*Figure 4 110 Choose Tables for Display form*

	Joint Reactions Units: As Noted								
	Joint Text	<b>OutputCase</b> Text	<b>CaseType</b> Text	F <sub>1</sub> KN	F <sub>2</sub> ΚN	F <sub>3</sub> KN	M1 KN-m	M <sub>2</sub> KN-m	M <sub>3</sub> KN-m
▶	P <sub>6</sub>	COMB1	Combination	0.094	0.001481	664.18	$\overline{0}$	0	
	P <sub>6</sub>	COMB <sub>2</sub>	Combination	$-4.435$	0.053	669.033	$\overline{0}$	0	
	P <sub>6</sub>	COMB3	Combination	$-4.083$	0.029	668.896	$\overline{0}$	0	
	P <sub>6</sub>	COMB4	Combination	$-4.399$	0.053	666.395	$\overline{0}$	0	
	P <sub>6</sub>	COMB5	Combination	$-3.979$	0.03	666.402	$\overline{0}$	0	
	P <sub>6</sub>	COMB6	Combination	$-0.107$	0.009685	667.222	$\overline{0}$	0	
	P <sub>6</sub>	COMB7	Combination	4.411	0.053	669.472	$\overline{0}$	0	
	P <sub>6</sub>	COMB8	Combination	4.177	0.029	669.31	$\overline{0}$	0	
	P <sub>6</sub>	COMB9	Combination	$-0.128$	$-0.005679$	666.773	$\overline{0}$	0	
	P <sub>6</sub>	COMB10	Combination	$-3.278$	0.04	666.989	$\overline{0}$	0	
	P <sub>6</sub>	COMB11	Combination	$-3.014$	0.022	666.887	$\overline{0}$	0	
	P <sub>6</sub>	COMB12	Combination	$-3.251$	0.04	665.011	$\overline{0}$	0	
	P <sub>6</sub>	COMB13	Combination	$-2.935$	0.023	665.016	$\overline{0}$	0	
	P <sub>6</sub>	COMB14	Combination	$-0.031$	0.007679	665.631	$\overline{0}$	0	
	P <sub>6</sub>	COMB15	Combination	3.357	0.04	667.318	$\overline{0}$	0	
	P <sub>6</sub>	COMB16	Combination	3.182	0.022	667.197	$\overline{0}$	0	
	P <sub>6</sub>	COMB17	Combination	$-0.047$	$-0.003844$	665.294	$\overline{0}$	0	
	P <sub>6</sub>	COMB18	Combination	0.7	0.002414	666.219	$\overline{0}$	0	
	P <sub>6</sub>	COMB19	Combination	$-0.634$	0.002213	666.093	$\overline{0}$	0	
	P <sub>6</sub>	COMB20	Combination	$-2.823$	0.041	668.518	$\overline{0}$	0	
	P <sub>6</sub>	COMB21	Combination	$-3.824$	0.041	668.423	$\overline{0}$	$\overline{0}$	

*Figure 4 111 Joint Reactions table*

In excel file just exported, filter largest value in "F3" column and enter this value and correspondence shear to Pmax and Vp in "Base Plate Calculation" file.

In Joint Reactions tables, select **File > Export Current Table > To Excel.** 



## **NOTE**

All value in Joint Reactions file are counter sign with value in "Base Plate Calculation" file, so we have to change the sign when enter axial force into calculation file. Shear forces will be taken as absolute values in all cases.

Do as above manner, filter the smallest value in "F3" column and enter this value and correspondence shear to Tmax and VT, filter largest absolute value in "F1" and "F2" column and take them to  $V_{\text{max}}$  and  $T_V$ .



**Maximum compression** Correspondence shear **Maximum tension** Correspondence shear **Maximum Shear** Correspondence tension



#### **NOTE**

Compressive force will have negative sign "-", and tensile force will have positive sign "+" or no sign. If there is not tensile force at base, value for it will be taken as 1.

## *4.2.1.2 Material, geometry parameter*

In order to illustrate a pinned base plate connection with full of dangerous cases, we assume information as figure below.



## **4.2.2. Checking base plate** *4.2.2.1 Base plate for concentric axial compressive load*



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# *4.2.2.2 Base plate for uplift load*





## *4.2.2.3 Design welding*

## **Weld capacity due to anchor tension**



## **Elastic method weld shear capacity**



## **Elastic method weld axial capacity**





## **4.2.3. Anchor bolt checking**

## *4.2.3.1 Checking for combination maximum tension* **Steel strength of a single anchor in tension**



## **Pullout of anchor in tension**



## **Steel strength of a single anchor in tension**



## **Interaction of tensile and shear forces of anchor**



## *4.2.3.2 Checking for combination maximum shear*

## **Steel strength of a single anchor in tension**



## **Pullout of anchor in tension**



## **Steel strength of anchors in shear**



## **Interaction of tensile and shear forces of anchor**



# **4.3. Fixed Base Plate Design**

## **4.3.1. Loading**

Run the model and specify all joints that connecting as fixed to foundation and have the same column section. Select the **Display > Show Tables** command to show **Choose Tables for Display** form. Make sure that the Table: Joint Reactions item and **all combinations** in the **Select Load Cases** form are selected before click the OK button.

Fdit		Load Patterns (Model Def.)				
<b>ii</b> -□ System Data <b>E-□ Property Definitions</b> <b>ii-□ Load Pattern Definitions</b> <b>ii-□ Other Definitions</b> <b>#</b> □ Load Case Definitions <b>ii</b> -□ Connectivity Data <b>E-□ Joint Assignments</b> <b>±</b> Frame Assignments <b>E-</b> Tendon Assignments <b>±</b> Options/Preferences Data <b>ii</b> -□ Miscellaneous Data <b>E-⊠ Joint Output</b>	⊕-□ MODEL DEFINITION (0 of 63 tables selected) 白図 ANALYSIS RESULTS (1 of 9 tables selected)					
<b>ii</b> -□ Displacements 白 <b>冈</b> Reactions M Table: Joint Reactions <b>ii-Π</b> Joint Masses <b>ii-□ Element Output</b> <b>E-□ Structure Output</b>	Select Load Cases Select COMB1 ▴ COMB10 <b>COMB100</b> 0K COMB101 <b>COMB102</b> <b>COMB103</b> Cancel <b>COMB104</b> <b>COMB105</b> <b>COMB106</b> <b>COMB107</b> COMB108 $\overline{\phantom{a}}$ Clear All	Show Unformatted Named Sets Save Named Set Show Named Set Delete Named Set				
Table Formats File	Current Table Formats File: Program Default	0K Cancel				

*Figure 4 113 Choose Tables for Display form*

In Joint Reactions tables, select **File > Export Current Table > To Excel**.



#### *Figure 4 114 Joint Reactions table*

In excel file just exported, filter largest value in "F3" column and enter this value, correspondence shear and moment M2 to  $P_{\rm max}, V_{\rm p}$  and  $M_{\rm p}$  in "Base Plate Calculation" file.

## **NOTE**



All values in Joint Reactions file are countered sign with value in "Base Plate Calculation" file, so we have to change the sign when enter axial force into calculation file. Shear forces and moment will be taken as absolute values in all cases.

Do as above manner, filter the smallest value in "F3" column and enter this value and correspondence shear and moment M2 to  $T_{\text{max}}$ ,  $V_{\text{T}}$  and  $M_{\text{T}}$ 

Filter all positive values of F3, and then filter the largest absolute value of M2. Enter this value, correspondence compressive force and shear to  $M1_{\text{max}}$ ,  $P_{\text{M1}}$  and  $V_{\text{M1}}$ .

Filter all negative values of F3, and then filter the largest absolute value of M2. Enter this value, correspondence tensile force and shear to  $M2_{\text{max}}$ ,  $P_{\text{M2}}$  and  $V_{\text{M2}}$ .

Filter all negative values of F3, and then filter the largest absolute value of F1. Enter this value, correspondence tensile force and shear to  $V_{\text{max}}$ ,  $T_{\text{V}}$  and  $M_{\text{V}}$ .



*Figure 4 115 Internal force for calculation*



## **NOTE**

Compressive force will have negative sign "-", and tensile force will have positive sign "+" or no sign. If there is not tensile force at base, value for it will be taken as 1.

## **4.3.2. Material, geometry parameter**

In order to illustrate a fixed base plate connection with full of dangerous cases, we assume information as figure below.



## **4.3.3. Checking base plate**





# *4.3.3.1 Checking base plate with combo Mp + Pmax*



# *4.3.3.2 Checking base plate with combo Tmax + M<sup>T</sup>*

It's always the case of base plate with large moment when considering combo momnent with tension.

$$
f_p = f_{p, \text{max}} = 9755.36 \frac{kN}{m^2}
$$



Figure 3.4.1. Base plate with large moment.

Determine bearing length Y:

- Equilibrium momentum with rotate center at the edge of compresion concrete zone:

$$
\frac{P_1}{b_1 - Y} \Big[ (b_1 - Y)^2 + (b_2 - Y)^2 + (b_3 - Y)^2 + (b_4 - Y)^2 + (b_5 - Y)^2 \Big] + q_{\text{max}} \frac{Y^2}{2} = M + P \Big( \frac{N}{2} - Y \Big)
$$
  
\n
$$
\Leftrightarrow R \Big( \sum_{i=1}^5 b_i^2 - 2Y \sum_{i=1}^5 b_i + 5Y^2 \Big) = M + P \Big( \frac{N}{2} - Y \Big) - q_{\text{max}} \frac{Y^2}{2}
$$
  
\n
$$
\Leftrightarrow R = \frac{M + P \Big( \frac{N}{2} - Y \Big) - q_{\text{max}} \frac{Y^2}{2}}{\Big( \sum_{i=1}^5 b_i^2 - 2Y \sum_{i=1}^5 b_i + 5Y^2 \Big)} = \frac{50 + 200 \Big( \frac{0.86}{2} - Y \Big) - 2068.14 \frac{Y^2}{2}}{12951 \times 10^{-4} - 2Y \times 2.15 + 5Y^2} = \frac{136 - 200Y - 1034.07Y^2}{12951 \times 10^{-4} - 4.3Y + 5Y^2}
$$

With:

the:

\n
$$
\sum_{\substack{i=1 \ i \neq j}}^{5} b_i^2 = 800^2 + 650^2 + 430^2 + 210^2 + 60^2 = 1295100 \, \text{mm}^2
$$
\n
$$
\sum_{i=1}^{5} b_i = 800 + 650 + 430 + 210 + 60 = 2150 \, \text{mm}
$$
\n
$$
R = \frac{P_1}{b_1 - Y}
$$

$$
\begin{aligned}\n&\text{Equilibrium force equation:} \quad \frac{P_1}{b_1 - Y} \left[ (b_1 - Y) + (b_2 - Y) + (b_3 - Y) + (b_4 - Y) + (b_5 - Y) \right] = P + q_{\text{max}} Y \\
&\Leftrightarrow R \left( \sum_{i=1}^5 b_i - 5Y \right) = P + q_{\text{max}} Y \\
&\Leftrightarrow (q_{\text{max}} + 5R) Y = R \sum_{i=1}^5 b_i - P \\
&\Leftrightarrow Y = \frac{R \sum_{i=1}^5 b_i - P}{(q_{\text{max}} + 5R)} = \frac{2.15R - 200}{2068.14 + 5R}\n\end{aligned}
$$

- Assume Y=0.001, set this value into Equation R: 2  $\frac{136 - 200 \times 0.001 - 1034.07 \times 0.001^2}{1205 \times 10^{-4} + 4.3 \times 0.001 + 5 \times 0.001^2} = 105.2$  $1295 \times 10^{-4} + 4.3 \times 0.001 + 5 \times 0.001$  $R = \frac{136 - 200 \times 0.001 - 1034.07 \times}{1005 + 0.0001 + 0.0001 + 0.0001}$  $=\frac{136-200 \times 0.001-1034.07 \times 0.001^2}{1295 \times 10^{-4}+4.3 \times 0.001+5 \times 0.001^2}$ 

Set R into Equation Y: 
$$
Y = \frac{2.15 \times 105.2 - 200}{2068.14 + 5 \times 105.2} = 0.01
$$

$$
\Delta Y = \frac{0.01 - 0.001}{0.001} = 9 > 0.05 \implies \text{Recalculate until this change not excess } 0.05
$$

Set Y=0.01 into Equation R: 
$$
R = \frac{136 - 200 \times 0.01 - 1034.07 \times 0.01^2}{1295 \times 10^{-4} - 4.3 \times 0.01 + 5 \times 0.01^2} = 106.89
$$

$$
Y = \frac{2.15 \times 106.89 - 200}{2068.14 + 5 \times 106.89} = 0.01146
$$

 $\frac{0.01146 - 0.01}{0.01} = 0.146 > 0.05$ 0.01  $\Delta Y = \frac{0.01146 - 0.01}{8.0011} = 0.146 > 0.05$  => Recalculate until this change not excess 0.05

$$
\text{Set Y=0.01146 into Equation R: } R = \frac{136 - 200 \times 0.01146 - 1034.07 \times 0.01146^2}{1295 \times 10^{-4} - 4.3 \times 0.01146 + 5 \times 0.01146^2} = 107.16
$$

$$
Y = \frac{2.15 \times 107.16 - 200}{2068.14 + 5 \times 107.16} = 0.01167
$$

$$
\Delta Y = \frac{0.01167 - 0.01146}{0.01146} = 0.018 < 0.05 \implies \text{OK}
$$

Determine maximum anchor tension:  $R = \frac{1}{1 - x^2} \Rightarrow P_1 = R(h_1)$ 1  $R = \frac{P_1}{I} \Rightarrow P_1 = R(h_1 - Y) = 107.16(0.8 - 0.01167) = 84.5$  $=\frac{I_1}{b_1 - Y}$   $\Rightarrow$   $P_1 = R(b_1 - Y) = 107.16(0.8 - 0.01167) = 84.5$  kN

Tension in one outermost bolt:  $T_{\text{max}} = \frac{1}{\sqrt{2}}$ 1  $\frac{84.5}{11.6}$  = 42.25 /2 4/2  $T_{\text{max}} = \frac{P_1}{n_1/2} = \frac{84.5}{4/2} = 42.25 \text{ kN}$ 

Determine minimum plate thickness with m  $(Y \le m)$ :

$$
\frac{1}{\Omega}F_y \frac{Bt_p^2}{4} = f_p BY(m - \frac{Y}{2})
$$

$$
\rightarrow t_p = \sqrt{\frac{4\Omega f_p Y(m - \frac{Y}{2})}{F_y}} = \sqrt{\frac{4x1.67x9755.36x11.67x(145 - \frac{11.67}{2})}{34.5x10^4}} = 17.51 \text{ mm}
$$

Determine minimum plate thickness follow tension in bolt: 2 1 4 *p y Bt*  $\frac{1}{\Omega}F_y \frac{E_y}{4} = P_1 x$ 

$$
\rightarrow t_p = \sqrt{\frac{4\Omega P_1 x}{BF_y}} = \sqrt{\frac{4x1.67x84.5x76}{212x34.5}} = 24.22 \text{ mm}
$$

1

Calculate similarly for combos  $M1_{\text{max}} \& P_{M1}$ ,  $M2_{\text{max}} \& T_{M2}$ ,  $V_{\text{max}} \& M_V$  we find that required minimum  $\qquad \qquad \Longrightarrow$ base plate thickness is 44.62mm which is still smaller than currently thickness of 50mm.

Maximun tension in anchor bolts governed by combination  $M2_{\text{max}}$  &  $T_{M2}$  $\qquad \qquad \blacksquare$ 

Use these force to checking bolts:  $\qquad \qquad \blacksquare$ 

Total axial force  $P = 50$  kN

Total shear force  $V_g = 50$  kN

Total moment  $M = 200$  kNm

Maximum anchor tension Tmax = 82.41 kN

Shear for one bolt  $V = 5$  kN

## *4.3.3.3 Design welding*

#### **Weld capacity due to anchor tension.**

Load angle factor when force perpendicular with welding:

$$
\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5
$$

Width of effective area resisting tension force of one bolt:

 $b_{\text{eff}} = 2a = 2 \times 70 = 140$  mm

Allowable bearing capacity of column flange weld:

$$
\frac{R_{w}}{\Omega} = \frac{1}{\Omega} \chi 0.6 F_{\text{EXX}} \frac{\sqrt{2}}{2} b_{f\text{-flange}} = \frac{1}{2} x 1.5 x 0.6 x 43 x \frac{\sqrt{2}}{2} x 0.7 = 9.58 \frac{kN}{cm}
$$

Force apply on column flange weld:

$$
\frac{T_{\text{max}}}{b_{\text{eff}}} = \frac{82.41}{14} = 5.89 \frac{kN}{cm}
$$
\nDemand/capacity

\ndefined:

\n
$$
\frac{5.89}{9.58} = 0.615 < 1
$$
\nOK

#### **Elastic method weld shear capacity**

Load angle factor when force parallel with welding:

$$
\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(0) = 1.0
$$

Total web weld length:

$$
L_{\text{shear}} = 2d = 2x600 = 1200 \text{ mm}
$$

Allowable bearing capacity of column web weld:

$$
\frac{R_{\nu}}{\Omega} = \frac{1}{\Omega} \chi 0.6 F_{\text{EXX}} \frac{\sqrt{2}}{2} b_{\text{f\_web}} = \frac{1}{2} \times 1.0 \times 0.6 \times 43 \times \frac{\sqrt{2}}{2} \times 0.6 = 5.47 \frac{kN}{cm}
$$

Force apply on column web weld:

$$
\frac{V_{\text{max}}}{L_{\text{shear}}} = \frac{200}{120} = 1.67 \frac{kN}{cm}
$$
\nDemand/capacity

\nDetting: 
$$
\frac{1.67}{5.47} = 0.305 < 1
$$

\nOK

#### **Elastic method weld axial capacity**

Load angle factor when force perpendicular with welding:

$$
\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5
$$
  
Total flange well length

 $L = 4b_f = 4 \times 212 = 848$  mm

160

Inertia moment of weld:

$$
I = 4b_f (d/2)^2 + 2(d-2t_f)^3 / 12 = 4 \times 212 \times (600/2)^2 + 2 \times (600-2 \times 12)^3 / 12 = 108170.5 \text{ cm}^3
$$

Allowable bearing capacity of column web weld:

$$
\frac{R_{w}}{\Omega} = \frac{1}{\Omega} \chi 0.6 F_{\text{EXX}} \frac{\sqrt{2}}{2} b_{f\text{-flange}} = \frac{1}{2} x 1.5 x 0.6 x 43 x \frac{\sqrt{2}}{2} x 0.7 = 9.58 \frac{kN}{cm}
$$

Force apply on column flange weld:

$$
\max\left(\frac{T_{\text{max}}}{L} + \frac{M_T(d/2)}{I}, \frac{T_{M2}}{L} + \frac{M2_{\text{max}}(d/2)}{I}\right) = \max\left(\frac{200}{84.8} + \frac{5000(60/2)}{108170.5}, \frac{50}{84.8} + \frac{20000(600/2)}{109170.5}\right) = 6.14
$$
  
Demand/capacity welding:  $\frac{6.14}{9.58} = 0.64 < 1$   
OK

#### **4.3.4. Anchor bolt checking**

#### **Checking for combination maximum tension**

Total axial force  $P = 50$  kN Total shear force  $V_g = 50 \text{ kN}$ Total moment  $M = 200$  kNm Maximum anchor tension  $T_{max} = 82.41 \text{ kN}$ Shear for one bolt  $V = 5$  kN

### *4.3.4.1 Steel strength of a single anchor in tension*

Allowable tensile capacity of each bolt:

$$
\frac{R_n}{\Omega} = \frac{1}{\Omega} F_n \frac{\pi d_b^2}{4} = \frac{1}{2} 37.5x \frac{\pi 2.7^2}{4} = 107.35 \text{ kN}
$$
  
Demand/Capacity:  $\frac{82.41}{107.35} = 0.77 < 1$   
OK

#### *4.3.4.2 Pullout of anchor in tension*

The pullout strength in tension of a single headed bolt:

$$
\frac{N_{\rho m}}{\Omega} = \psi_{c,P} \frac{A_{bg} 8 f_c'}{\Omega} = 1.4 \frac{8.83 \times 8 \times 2}{2.14} = 92.43 \text{ kN}
$$
  
With:  $A_{bg} = \frac{\sqrt{3}}{2} F^2 - \frac{\pi d_b^2}{4} = \frac{\sqrt{3}}{2} 4.1^2 - \frac{\pi \times 2.7^2}{4} = 8.83 \text{ cm}^2 \text{ (F is nominal dimension of bolt's head, take } F \approx 1.5 d_b \text{)}$ 

Assume there is no cracking in concrete so modification factor for pullout shall be taken as 1.4

$$
\Rightarrow \text{ Demand/Capacity} = \frac{82.41}{92.43} = 0.89 < 1
$$

$$
\longrightarrow \quad OK
$$

## *4.3.4.3 Steel strength of anchors in shear*

Allowable shear capacity of each bolt:

$$
\frac{V_{sa}}{\Omega} = \frac{0.8F_{m}\pi d_{b}^{2}/4}{\Omega} = \frac{0.8 \times 22.5 \times \pi \times 2.7^{2}/4}{2} = 51.53 \text{ kN}
$$
  
Demand/Capacity  $= \frac{5}{51.53} = 0.097 < 1$   
OK

## *4.3.4.4 Interaction of tensile and shear forces of anchor:*

 $\frac{0.89 + 0.097}{1.2} = 0.823 < 1$ 1.2  $\frac{+0.097}{-0.823}$  = 0.823 <  $\Rightarrow$  OK

#### **Checking for combination maximum shear**

Total axial force  $P = 50$  kN Total shear force  $V_g = 200 \text{ kN}$ Total moment  $M = 50$  kNm Maximum anchor tension  $T_{max} = 22.74 \text{ kN}$ Shear for one bolt  $V = 20$  kN

#### *4.3.4.5 Steel strength of a single anchor in tension*

Allowable tensile capacity of each bolt:

$$
\frac{R_n}{\Omega} = \frac{1}{\Omega} F_n \frac{\pi d_b^2}{4} = \frac{1}{2} 37.5 \times \frac{\pi 2.7^2}{4} = 107.35 \text{ kN}
$$
  
Demand/Capacity:  $\frac{22.74}{107.35} = 0.21 < 1$ 

#### *4.3.4.6 Pullout of anchor in tension*

The pullout strength in tension of a single headed bolt:

$$
\frac{N_{p\mu}}{\Omega} = \psi_{c,p} \frac{A_{bg} 8f_c'}{\Omega} = 1.4 \frac{8.83 \times 8 \times 2}{2.14} = 92.43 \text{ kN}
$$
  
With:  $A_{bg} = \frac{\sqrt{3}}{2}F^2 - \frac{\pi d_b^2}{4} = \frac{\sqrt{3}}{2}4.1^2 - \frac{\pi \times 2.7^2}{4} = 8.83 \text{ cm}^2$  (F is nominal dimension of bolt's head, take  $F \approx 1.5d_b$ )

Assume there is no cracking in concrete so modification factor for pullout shall be taken as 1.4

$$
\text{Demand}/\text{Capacity} = \frac{22.74}{92.43} = 0.25 < 1
$$
\nOK

## *4.3.4.7 Steel strength of anchors in shear*

Allowable shear capacity of each bolt:

$$
\frac{V_{sa}}{\Omega} = \frac{0.8F_m \pi d_b^2 / 4}{\Omega} = \frac{0.8 \times 22.5 \times \pi \times 2.7^2 / 4}{2} = 51.53 \text{ kN}
$$
  
Demand/Capacity  $= \frac{20}{51.53} = 0.39 < 1$   
OK

# *4.3.4.8 Interaction of tensile and shear forces of anchor:*

$$
\frac{0.25 + 0.39}{1.2} = 0.53 < 1 : \text{OK}
$$

# **4.4. Horizontal Knee Connection Design**

## **4.4.1. Input** *4.4.1.1 Loading*

Run the model and specify all rafter having section 1K. Select the **Display** > Show Tables command to show Choose Tables for Display form. Make sure that the Table: Element Forces - Frames item and all combinations in the Select Load Cases form are selected before click the OK button.



*Figure 4 116 Choose Tables for Display form*

In Element Forces tables, select **File > Export Current Table > To Excel**.

Units: As Noted				<b>Element Forces - Frames</b>					
Frame Text	<b>Station</b> $\mathbf{m}$	<b>OutputCase</b> Text	CaseType Text	P KN	V <sub>2</sub> KN	$V_3$ KN	KN-m	M2 $KN-m$	
41	$\overline{0}$	COMB1	Combination	$-9.237$	$-36.652$	$-0.001523$	$-0.0001683$	0.0011	
41	3	COMB1	Combination	$-8.64$	$-24.724$	$-0.001523$	$-0.0001683$	0.0057	
41	6 <sup>1</sup>	COMB1	Combination	$-8.057$	$-13.061$	$-0.001523$	$-0.0001683$	0.0102	
41	0	COMB <sub>2</sub>	Combination	10.687	53.495	0.003232	$-0.00005324$	$-0.0128$	
41	3	COMB <sub>2</sub>	Combination	10.924	40.785	0.003232	$-0.00005324$	$-0.0225$	
41	6	COMB <sub>2</sub>	Combination	11.148	27.809	0.003232	$-0.00005324$	$-0.0322$	
41	0	COMB3	Combination	$-4.345$	39.083	$-0.001714$	$-0.00002205$	$-0.0045$	
41	3	COMB3	Combination	$-4.108$	31.261	$-0.001714$	$-0.00002205$	0.0006913	
41	6	COMB3	Combination	$-3.885$	23.173	$-0.001714$	$-0.00002205$	0.0058	
41	0	COMB4	Combination	4.005	16.883	0.001195	$-0.00003876$	$-0.0118$	
41	3	COMB4	Combination	4.242	17.099	0.001195	$-0.00003876$	$-0.0154$	
41	6	COMB4	Combination	4.465	17.049	0.001195	$-0.00003876$	$-0.019$	
41	0	COMB5	Combination	$-10.615$	4.607	$-0.003614$	$-0.000006423$	$-0.0035$	
41	3	COMB5	Combination	$-10.378$	8.978	$-0.003614$	$-0.000006423$	0.0073	
41	6	COMB5	Combination	$-10.155$	13.083	$-0.003614$	$-0.000006423$	0.0182	
41	0	COMB6	Combination	32.367	23.847	0.021	$-0.0004379$	0.016	
41	3	COMB6	Combination	32.604	14.789	0.021	$-0.0004379$	$-0.0471$	
41	6	COMB6	Combination	32.828	5.465	0.021	$-0.0004379$	$-0.1102$	
41	0	COMB7	Combination	5.329	6.621	$-0.002999$	$-0.00001421$	$-0.0219$	
41	3	COMB7	Combination	5.565	0.84	$-0.002999$	$-0.00001421$	$-0.0129$	

*Figure 4 117 Joint Reactions table*

In excel file just exported, in Station column, select station that use K-type connection, usually "0". Filter the smallest value in "M3" column and enter this value, correspondence shear V2 and axial force P to "Bolt Connections" file.



#### **NOTE**

Sign of force in sap will be the same sign of force in calculation file. a. Filter the largest value in "M3" column and enter this value, correspondence shear V2 and axial force P to "Bolt Connections" file.

b. Filter the largest value in "V2" column and enter this value, correspondence moment M3 and axial force P to "Bolt Connections" file.

#### *4.4.1.2 Material and geometry*

In order to illustrate a K-type connection with full of dangerous cases, we assume information as figure below.



## **4.4.2. Determind internal force for K connection**

Because interal force are in the end of rafter so we have to transfer this load into internal force at top of column

#### **Internal force for Combo of negative moment**



#### **Internal force for Combo of positive moment**



#### **Internal force for Combo of shear max**



Positive Moment 44.78 54.73 200.00 -234.02 278.80

Shear 203.98 29.85 -50.00 166.09 37.89

166

## **4.4.3. Checking moment end plate (external flange)**

### **Flexural yielding**

Calculate total length of yielding line of external flange from Table 4-4 DG16:

$$
Y = \frac{b_{p}}{2} \left( \frac{b_{1}}{p_{f,i}} + \frac{b_{2}}{s} + \frac{b_{0}}{p_{f,o}} - \frac{1}{2} \right) + \frac{2}{g} \left[ b_{1} \left( p_{f,i} + 0.75 p_{b} \right) + b_{2} \left( s + 0.25 p_{b} \right) \right] + \frac{g}{2}
$$
  
=  $\frac{248}{2} \left( \frac{730}{50} + \frac{630}{78.74} + \frac{850}{50} - \frac{1}{2} \right) + \frac{2}{100} \left[ 730 \left( 50 + 0.75 x 100 \right) + 630 \left( 78.74 + 0.25 x 100 \right) \right] + \frac{100}{2}$   
= 8030.65 mm

With:  $s = \frac{1}{2} \sqrt{b_p g} = \frac{1}{2} \sqrt{248 \times 100} = 78.74$  mm





#### **No prying bolt moment strength**



Demand/Capacity ratio

\n
$$
D/C = \frac{371.34}{513.65} = 0.72
$$
\nOrclusion

\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.95 < \frac{M_{up}}{\Omega} = 513.65
$$
\nUniform

\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.95 < \frac{M_{up}}{\Omega} = 513.65
$$
\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.95 < \frac{M_{up}}{\Omega} = 513.65
$$
\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.95 < \frac{M_{up}}{\Omega} = 513.65
$$
\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.95 < \frac{M_{up}}{\Omega} = 513.65
$$
\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.95 < \frac{M_{up}}{\Omega} = 513.65
$$
\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.95 < \frac{M_{up}}{\Omega} = 513.65
$$
\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.95 < \frac{M_{up}}{\Omega} = 513.65
$$
\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.95 < \frac{M_{up}}{\Omega} = 513.65
$$
\nUniform

\n
$$
0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 424.39 = 381.
$$

 $q_{\rm u}$ 

O

 $(b)$  intermediate

qu

 $(c)$  thin



Maximum prying force of inner bolts:

 $d_2 = 62$  $\mathbf{u}$ ಕೆ

 $\ddot{\phi}$  $\ddot{\Phi}$ 

 $\ddot{\phi}$ ó ਠੱ

$$
Q_{\text{max},i} = \frac{w' t_p^2}{4a_i} \sqrt{F_{p}^2 - 3 \left(\frac{F_i'}{w' t_p}\right)^2} = \frac{101 \times 16^2}{4 \times 45.72} \sqrt{3450^2 - 3 \left(\frac{93.94}{101 \times 16}\right)^2} = 47.10 \text{ kN}
$$

With:  $w' = b_p / 2 - d_b = 248 / 2 - 22 = 102$  mm

 $(a)$  thick

$$
a_i = 25.4 \left( 3.682 \left( \frac{t_p}{d_b} \right)^3 - 0.085 \right) = 25.4 \left( 3.682 \left( \frac{16}{20} \right)^3 - 0.085 \right) = 45.72 \text{ mm}
$$
  

$$
F_i' = \frac{t_p^2 F_y \left( 0.85 \frac{b_p}{2} + 0.80 w^i \right) + \frac{\pi d_b^3 F_w}{8}}{4 p_{f,i}} = \frac{16^2 \times 0.345 \left( 0.85 \frac{248}{2} + 0.8 \times 102 \right) + \frac{\pi \times 20^3 \times 0.75}{8}}{4 \times 50} = 94.30 \text{ km}
$$

Maximum prying force of outer bolts:

$$
Q_{\text{max,o}} = \frac{w' t_{p}^2}{4a_{q}} \sqrt{F_{py}^2 - 3\left(\frac{F_{q}'}{w' t_{p}}\right)^2} = \frac{101 \times 16^2}{4 \times 45.72} \sqrt{3450^2 - 3\left(\frac{93.94}{101 \times 16}\right)^2} = 47.10 \text{ kN}
$$

With:  $w' = b_p / 2 - d_b = 248 / 2 - 22 = 102$  mm

$$
a_{\rho} = \min\left(L_{\text{ev}}; 25.4\left(3.682\left(\frac{t_{p}}{d_{b}}\right)^{3} - 0.085\right)\right) = \min\left(50; 25.4\left(3.682\left(\frac{16}{20}\right)^{3} - 0.085\right)\right) = 45.72 \text{ mm}
$$

$$
F_{\text{v}}' = \frac{t_{p}^{2}F_{\text{v}}\left(0.85\frac{b_{p}}{2} + 0.80w^{4}\right) + \frac{\pi d_{b}^{3}F_{\text{m}}}{8}}{4p_{\text{f,o}}} = \frac{16^{2} \times 0.345\left(0.85\frac{248}{2} + 0.8 \times 102\right) + \frac{\pi \times 20^{3} \times 0.75}{8}}{4 \times 50} = 94.30 \text{ kN}
$$



#### **Bolts shear**



#### **Bolts bearing under shear load**

Distance from edge of outermost hole to edge of plate:

$$
l_{k1} = L_w - d_k/2 = 50 - 22 / 2 = 39
$$
mm

Distance from edge of hole to hole:

$$
l_c = p_b - d_b = 100 - 22 = 78 \,\mathrm{mm}
$$

Bolts bearing capacity at outermost bolts:

$$
r_{n2} = 2(0.6F_n l_c t) = 1.2l_c t F_n = 1.2x7.8x1.6x44.8 = 670.92 \text{ kN}
$$
  
Bolts bearing capacity between bolts:

 $r_{n2} = 2(0.6 F_u l_c t) = 1.2 l_c t F_u = 1.2 \times 7.8 \times 1.6 \times 44.8 = 670.92 \text{ kN}$ Maximum bearing capacity:

 $r_{n(\text{max})} = 2(0.6 F_u 2 dt) = 2.4 dt F_u = 2.4 \times 2 \times 1.6 \times 44.8 = 344.06 \text{ kN}$ 





## **Shear yielding of plate**

Shear yielding strength of plate:

$$
V_a = \frac{Puffop}{2} = \frac{\max(476.08, 234.02, 166.09)}{2} = 238.04 \text{ kN}
$$
  
Demand shear strength:

 $\frac{\max(476.08, 234.02, 166.09)}{2} = 238.04$  $\frac{a}{2}$   $\frac{1}{2}$   $\frac{1}{2}$   $\frac{1}{2}$   $\frac{2}{2}$  $V_a = \frac{Puftop}{2} = \frac{\max(476.08, 234.02, 166.09)}{2} = 238.04 \text{ kN}$ Demand/Capacity ratio:

$$
D/C = \frac{238.04}{491.84} = 0.48 : OK
$$

#### **Shear rupture of plate**

Shear rupture strength of plate:

$$
\frac{R_n}{\Omega} = \frac{1}{\Omega} 0.6 F_n [b_p - 2(d_b + 1.6)] t_p
$$
  
\n
$$
\frac{R_n}{\Omega} = \frac{1}{2} \times 0.6 \times 0.45 \times [248 - 2(22 + 1.6)] 16 = 431.99 kN
$$
  
\nDemand shear strength:

$$
V_a = \frac{Puffop}{2} = \frac{\max(476.08, 234.02, 166.09)}{2} = 238.04 \text{ kN}
$$

Demand/Capacity ratio:  $D/C = \frac{238.04}{100000} = 0.55$ 431.99  $D/C = \frac{256.04}{124.00} = 0.55$ : OK





# **4.4.4. Checking moment end plate (internal flange)**

### **Flexural yielding**

Calculate total length of yielding line of internal flange from Table 3-3 DG16:

$$
Y = \frac{b_p}{2} \left( \frac{b_1}{p_f} + \frac{b_2}{s} \right) + \frac{2}{g} \left[ b_1 \left( p_f + 0.75 p_b \right) + b_2 \left( s + 0.25 p_b \right) \right] + \frac{g}{2}
$$
  
= 
$$
\frac{248}{2} \left( \frac{730}{50} + \frac{630}{78.74} \right) + \frac{2}{100} \left[ 730 \left( 50 + 0.75 x 100 \right) + 630 \left( 78.74 + 0.25 x 100 \right) \right] + \frac{100}{2}
$$
  
= 5984.65mm

With: 
$$
s = \frac{1}{2} \sqrt{b_p g} = \frac{1}{2} \sqrt{248 \times 100} = 78.74
$$
 mm





## **No prying bolt moment strength**





# **Bolt rupture with prying moment strength**

$$
Q_{\text{max},j} = \frac{m^2 t_p^2}{4a_j} \sqrt{F_p^2 - 3\left(\frac{F_i^2}{m^2 t_p}\right)^2} = \frac{101 \times 16^2}{4 \times 45.72} \sqrt{3450^2 - 3\left(\frac{93.94}{101 \times 16}\right)^2}
$$
 47.10 kN  
\nWithin:  $w^2 = b_p / 2 - d_p = 248 / 2 - 22$  102 mm  
\nforce of inner bolts  $a_i = 25.4 \left(3.682 \left(\frac{t_p}{d_b}\right)^3 - 0.085\right) = 25.4 \left(3.682 \left(\frac{16}{20}\right)^3 - 0.085\right)$  45.72 mm  
\nforce of inner bolts  $a_i = 25.4 \left(3.682 \left(\frac{t_p}{d_b}\right)^3 - 0.085\right) = 25.4 \left(3.682 \left(\frac{16}{20}\right)^3 - 0.085\right)$  45.72 mm  
\n
$$
\frac{t_p F_y \left(0.85 \frac{1}{2} + 0.8 \times 102\right) + \frac{d_b F_w}{8}}{4 \times 50}
$$
 94.30 kN  
\n
$$
\frac{M_g}{\Omega} = \max \left\{ \frac{1}{\Omega} \left[ 2(P_r - Q_{\text{max},i}) (d_1 + d_2) \right] + \frac{20 \times 0.75}{8} \right\}
$$
 94.30 kN  
\n
$$
= \max \left\{ \frac{1}{\Omega} \left[ 2(235.62 - 47.10) (0.72 + 0.62) \right] + \frac{1}{\Omega} \left[ 2(235.62 - 47.10) (0.72 + 0.62) \right] \right\}
$$
 252.62 kN  
\nDemand/Capacity  $D/C = \frac{217.46}{252.62} = 0.86$  OK  
\nratio

## **Bolts shear**



## **Bolts bearing under shear load**





## **Shear yielding of plate**



## **Shear rupture of plate**



## **4.4.5. Weld check**

## **Web weld shear strength**



## **Web weld strength to reach yield stress**



## **Shear yielding of web**



## **Flange weld capacity (external flange)**





#### **Flange weld capacity (internal flange)**



## **4.4.6. Checking panel zone**

#### **Panel web shear**

Shear yield strength of the rafter:

 $P_e = 0.6 F_y A_g = 0.6 F_y (2 b_f t_f + b_w t_w) = 0.6 \times 34.5 \times (2 \times 212 \times 10 + 680 \times 5) / 100 = 1581.48$  kN

For: 
$$
P_r = \max (P \text{uftop}, P \text{uftot}) = \max (476.08, 421.35) = 476.08
$$

$$
P_r = 476.08 < 0.4 P_c = 0.4 \times 158148 = 632.59
$$

$$
-\frac{R_{n}}{\Omega} = \frac{0.6F_{y}bt_{w}}{1.67} = \frac{0.6 \times 34.5 \times 70 \times 0.5}{1.67} = 433.83 \text{ kN}
$$

Shear yield strength of the diagonal stiffeners (consider diagonal stiffeners as compresion elements): Inertia radius of stiffeners:  $r = t_{ps}/\sqrt{12} = 12/\sqrt{12} = 3.46$  mm

Slenderness of stiffeners:  $KL / r = 0.65 \times 1018.47 / 3.46 = 191.10$ 

$$
\frac{KL}{r} = 191.10 > 4.71 \sqrt{\frac{E}{F_y}} = 4.71 \sqrt{\frac{20000}{34.5}} = 113.4
$$

$$
F_{\sigma} = 0.877 F_{\epsilon} = 0.877 \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = 0.877 \frac{\pi^2 \times 20000}{191.10^2} = 4.74 \frac{kN}{cm^2}
$$

$$
\frac{R_{\eta_{ps}}}{\Omega} = 2 \frac{1}{\Omega} F_{\alpha} A_{g} \cos \theta = 2 \frac{1}{\Omega} F_{\alpha} (b_{ps} - \alpha) t_{ps} \frac{b_{e} - 2t_{fe}}{L_{ps}}
$$
  
Strength of diagonal stiffness: 
$$
= 2 \frac{1}{1.67} 4.74 x (109 - 10) x 12 x 10^{-2} \frac{800 - 2x 20}{1018.47}
$$

$$
= 50.32 kN
$$

kN

Total panel web shear resitance:  $\frac{R_n}{R_1} + \frac{R_{\text{mps}}}{R_2} = 433.83 + 50.32 = 484.15$  $\Omega$   $\Omega$ 

Demand shear strength:  $V_a = \max(Puttop, Putbot) = \max(476.08, 421.35) = 476.08$  kN

Demand/Capacity ratio: 
$$
D/C = \frac{476.08}{484.15} = 0.98
$$

 $\qquad \qquad \Longrightarrow$ OK

#### **4.4.7. Local web yielding**

Web resistance in local web yielding:

$$
\frac{1}{\Omega} = \frac{1}{\Omega} [0.5(6k + 2t_p) + N] F_{\text{gw}} t_w = \frac{1}{1.5} [0.5(6 \times 16 + 2 \times 16) + 38] 34.5 \times 10 \quad \text{xf} = 117.3 \text{ kN}
$$

With:  $k = t_p = 16$  mm

 $N = t_f + 2W_f = 10 + 2x9 = 38$  mm Stiffeners resistance in local web yielding:

$$
\frac{R_{ns}}{\Omega} = 2\frac{1}{\Omega}F_y A_s = 2\frac{1}{1.67}34.5 \times 10^{-2} \times 1338 = 552.83 \text{ kN}
$$

With:  $A_s = t_s (b_s - c lip) = 12(121.5 - 10) = 1338 mm^2$ Total resistance of web:

$$
\frac{R_n}{\Omega} + \frac{R_m}{\Omega} = 117.3 + 552.83 = 670.13 \text{ kN}
$$

Demand shear strength:

 $V_a = \max(P{\text{uftop}}, P{\text{ufbot}}) = \max(476.08, 421.35) = 476.08$  kN Demand/Capacity ratio:

$$
D/C = \frac{476.08}{670.13} = 0.71
$$
  
OK

## Yielding str*e*ngth due to axial load **4.4.8. Transverse stiffeners**

Gross section area of transverse stiffeners:  $A_g = 2 (b_s - c l \dot{q} p) t_s = 2x (121.5 - 10) x 12 = 2676 mm^2$ 

Tensile yielding strength of stiffeners:  $\frac{V}{Q} = \frac{1}{2} F_y A_z = \frac{1}{2} 34.5 \times 10^{-2} \times 2676 = 552.83$  $\frac{V}{\Omega} = \frac{1}{\Omega} F_y A_s = \frac{1}{1.67} 34.5 \times 10^{-2} \times 2676 = 552.83 \text{ kN}$ Demand axial strength is calculated by subtracting web resistance in local web yielding from maximum tension force at bottom flange:  $P_{tension} = \max(-421.35, 278.80, 37.89) - 117.3 = 161.5 \text{ kN}$ 

Demand/Capacity ratio: 
$$
D/C = \frac{161.5}{552.83} = 0.29
$$

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#### **Compression**

Inertia radius of stiffeners:

 $r = t$  / $\sqrt{12} = 12/\sqrt{12} = 3.46$  *mm* Slenderness of stiffeners:  $KL / r = 0.65 \times 400 / 3.46 = 75.14$ 

$$
\frac{KL}{r} = 75.14 < 4.71 \sqrt{\frac{E}{F_y}} = 4.71 \sqrt{\frac{20000}{34.5}} = 113.4
$$

$$
F_{\sigma} = \left[ 0.658^{-1} \right] F_{y} = \left[ 0.658^{\frac{34.5}{34.96}} \right] 34.5 = 22.83 \frac{kN}{cm}
$$

$$
F_{\sigma} = \left[ 0.658^{\frac{F_{y}}{F_{\epsilon}}} \right] F_{y} = \left[ 0.658^{\frac{34.5}{34.96}} \right] 34.5 = 22.83 \frac{kN}{cm^{2}}
$$

Compressive strength shall be determined based on the limit state of flexural buckling:

$$
\frac{P_n}{\Omega} = 2\frac{1}{\Omega}F_{\sigma}A_g = 2\frac{1}{1.67}22.83x13.38 = 365.83 \text{ kN}
$$

Demand axial strength is calculated by subtracting web resistance in local web yielding from maximum compressive force at bottom flange:

 $P_{\text{compression}} = \left| \min(-421.35, 278.80, 37.89) \right| - 117.3 = 304.05 \text{ kN}$ Demand/Capacity ratio:

$$
D/C = \frac{304.05}{365.83} = 0.83
$$
  

$$
\implies OK
$$

## **Welding stiffeners with rafter flange**

Area of flange weld:

$$
A_w = \frac{\sqrt{2}}{2}W_xL = \frac{\sqrt{2}}{2}W_y\left(b_x - \text{clip} - W_x\right) = \frac{\sqrt{2}}{2}7(121.5 - 10 - 7) = 517.25 \text{mm}^2
$$
  
Load angle factor when force perpendicular with well

 $\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5$ 

Weld strength of total flange weld:

$$
V_a = \max\left(P_{\text{tension}}, P_{\text{compression}}\right) = \max\left(161.5, 304.05\right) = 304.05 \, \text{kN}
$$

 $V_a = \max ( P_{\text{tension}}, P_{\text{compression}} ) = \max (161.5, 304.05 ) = 304.05 \,\text{kN}$ Demand/Capacity ratio:

$$
D/C = \frac{304.05}{385.27} = 0.79
$$
\n
$$
\longrightarrow \qquad \text{OK}
$$

#### **Welding stiffeners with rafter web**

Area of web weld:

$$
\frac{K_n}{\Omega} = 4 \left( \frac{1}{\Omega} 0.6 F_{EXX} \chi A_w \right) = 4 \left( \frac{1}{2} 0.6 \times 41.4 \times 1.0 \times 18.96 \right) = 941.93
$$

Load angle factor when force parallel with weld  $\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(0) = 1.0$ Weld strength of total web weld:

$$
\frac{R_n}{\Omega} = 4\left(\frac{1}{\Omega} 0.6 F_{EXX} \chi A_w\right) = 4\left(\frac{1}{2} 0.6x 41.4x 1.0x 18.96\right) = 941.93 \text{ kN}
$$

Demand shear strength:  $V_a = \max ( P_{\text{tension}}, P_{\text{compression}} ) = \max (161.5,304.05) = 304.05 \text{ kN}$ 

Demand/Capacity ratio:  $D/C = \frac{304.05}{0.11 \cdot 0.32} = 0.32$ 941.93  $D/C = \frac{304.05}{0.11.02}$ 

 $\qquad \qquad \Longrightarrow$ OK

#### **4.4.9. Checking diagonal stiffeners**

#### **Yielding strength due to axial load**

Section area of diagonal stiffeners:  $A_g = 2(h_{ps} - \frac{di}{p})t_{ps} = 2(109 - 10)12 = 2376 \text{ mm}^2$ 

Tensile yielding strength of stiffeners:  $\frac{R_n}{R_n} = \frac{1}{2} F_r A_s = \frac{1}{1} 34.5 \times 23.76 = 490.85$ 1.67  $\frac{R_n}{\Omega} = \frac{1}{\Omega} F_y A_g = \frac{1}{1.67} 34.5 \times 23.76 = 490.85 \text{ kN}$ 

Demand axial strength is calculated by subtracting web resistance in panel web shear from maximum of tensile force at bottom flange and compressive force at top flange:

$$
P_{\text{tension}} = \frac{\max[\left|\min(476.08, -234.02, 166.09)\right|, \max(-421.35, 278.80, 37.89)\right] - 433.83}{\cos\theta} = \frac{-155.03}{\cos\theta} < 0 \text{ kN}
$$

Because required strength is negative so it's not necessary to check tensile yielding strength of diagonal stiffeners.

**Checking compressive strength of diagonal stiffeners**

Inertia radius of stiffeners:  $r = t_{ps} / \sqrt{12} = 12 / \sqrt{12} = 3.46$  mm

Slenderness of stiffeners:  $\frac{KL}{m}$  = 191.33 > 4.71  $\frac{E}{m}$  = 4.71  $\frac{20000}{3.5}$  = 113.4 34.5 *<sup>y</sup> KL*  $_{101.22 \times 1.71}$   $E$  $r = \sqrt{F}$  $= 191.33 > 4.71$   $\frac{12}{1} = 4.71$ 

$$
\frac{KL}{r} = 191.33 > 4.71 \sqrt{\frac{E}{F_y}} = 4.71 \sqrt{\frac{20000}{34.5}} = 113.4
$$

$$
F_e = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = \frac{\pi^2 \times 20000}{191.33^2} = 5.39 \frac{kN}{cm^2}
$$

$$
F_{\text{av}} = 0.877 F_e = 0.877 \times 5.39 = 4.73 \frac{kN}{cm^2}
$$

Compressive strength shall be determined based on the limit state of flexural buckling:

$$
\frac{P_n}{\Omega} = 2\frac{1}{\Omega}F_a A_g = 2\frac{1}{1.67}4.74 \times 11.88 = 67.44 \text{ kN}
$$

Demand axial strength is calculated by subtracting web resistance in panel web shear from maximum of compressive force at bottom flange and tensile force at top flange:

$$
P_{\text{compression}} = \frac{\max \left[ \max \left( 476.08, -234.02, 166.09 \right), \left| \min \left( -421.35, 278.80, 37.89 \right) \right| \right] - 433.83}{\cos \theta} = \frac{42.25}{\left( b_c - 2t_f \right) / L_{\text{ps}}} = \frac{42.25}{\left( 800 - 2 \times 20 \right) / 1018.47} = 56.62 \text{ kN}
$$

Demand/Capacity ratio:  $D/C = \frac{56.62}{\sqrt{1.1}} = 0.84$  $D/C = \frac{30.02}{\sqrt{24}} =$ 67.44  $\qquad \qquad \blacksquare$ OK

#### **Welding stiffeners with rafter flange**

Area of flange weld:  $A_w = \frac{\sqrt{2}}{2} W_{ps} L = \frac{\sqrt{2}}{2} W_{ps} ( b_{ps} - \text{clip} - W_{ps} ) = \frac{\sqrt{2}}{2} 7 (109 - 10 - 7) = 455.38 \text{mm}^2$ Load angle factor when force perpendicular with weld:  $\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5$ 

Weld strength of total flange weld:  $\frac{R_n}{a} = 4\left(\frac{1}{2} 0.6 F_{EXX} \chi A_w\right) = 4\left(\frac{1}{2} 0.6 \times 41.4 \times 1.5 \times 4.55\right) = 339.07$ 2  $\frac{R_{n}}{\Omega}$  =  $4\left(\frac{1}{\Omega}0.6F_{\text{EXX}}\chi A_{w}\right)$  =  $4\left(\frac{1}{2}0.6x41.4x1.5x4.55\right)$  = 339.07 kN Demand shear strength:  $V_a = \max ( P_{\text{tension}}, P_{\text{compression}} ) = \max ( 0, 56.62 ) = 56.62 \text{ kN}$ 

Demand/Capacity ratio: 
$$
D/C = \frac{56.62}{339.07} = 0.17
$$

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#### **Welding stiffeners with rafter web**

Area of web weld:  $A_w = \frac{\sqrt{2}}{2} W_{ps} L = \frac{\sqrt{2}}{2} W_{ps} (L_{ps} - \text{clip} - W_s) = \frac{\sqrt{2}}{2} 7 (1018.47 - 10 - 7) = 4957.02 \text{mm}^2$ Load angle factor when force parallel with weld  $\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(0) = 1.0$ 

Weld strength of total web weld:  $V_a = \max ( P_{\text{tension}}, P_{\text{compression}} ) = \max ( 0, 56.62 ) = 56.62 \text{ kN}$ Demand shear strength:  $V_a = \max\left(P_{tension}, P_{compression}\right) = \max\left(0, 56.62\right) = 56.62 \text{ kN}$ 

Demand/Capacity ratio:  $D/C = \frac{56.62}{24.62 \times 10^{-10}} = 0.02$ 2462.64  $D/C = \frac{30.02}{24.02 \times 10^{-10}}$ 

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# **4.5. Single Plate Connection Design**

#### **4.5.1. Input**

#### *4.5.1.1 Loading*

Run the model and specify all beam having section B1. Select the **Display > Show Tables** command to show Choose Tables for Display form. Make sure that the Table: Element Forces - Frames item and ENVE combination in the **Select Load Cases** form are selected before click the OK button.



*Figure 4 118 Choose Tables for Display form*

In Element Forces tables, select File > Export Current Table > To Excel.

Units: As Noted					Element Forces - Frames			
Frame Text	<b>Station</b> m	<b>OutputCase</b> Text	<b>CaseType</b> Text	<b>StepType</b> Text	Р KN	V <sub>2</sub> KN	V <sub>3</sub> ΚN	KN-m
261	$\overline{0}$	<b>ENVE</b>	Combination	Max	3.01	$-0.603$	$\overline{0}$	0.0002593
261	0.5	<b>ENVE</b>	Combination	Max	3.01	$-0.528$	0	0.0002593
261	1	<b>ENVE</b>	Combination	Max	3.01	$-0.453$	0	0.0002593
261	1.5	<b>ENVE</b>	Combination	Max	3.01	$-0.377$	$\overline{0}$	0.0002593
261	$\overline{2}$	<b>ENVE</b>	Combination	Max	3.01	$-0.302$	0	0.0002593
261	2.5	<b>ENVE</b>	Combination	Max	3.01	$-0.226$	0	0.0002593
261	3	<b>ENVE</b>	Combination	Max	3.01	$-0.151$	0	0.0002593
261	3.5	<b>ENVE</b>	Combination	Max	3.01	$-0.075$	0	0.0002593
261	$\vert$	<b>ENVE</b>	Combination	Max	3.01	$\overline{0}$	0	0.0002593
261	4.5	<b>ENVE</b>	Combination	Max	3.01	0.126	0	0.0002593
261	5 <sup>1</sup>	<b>ENVE</b>	Combination	Max	3.01	0.251	0	0.0002593
261	5.5	<b>ENVE</b>	Combination	Max	3.01	0.377	$\vert 0 \vert$	0.0002593
261	6 <sup>1</sup>	<b>ENVE</b>	Combination	Max	3.01	0.503	$\overline{0}$	0.0002593
261	6.5	<b>ENVE</b>	Combination	Max	3.01	0.628	0	0.0002593
261	7	<b>ENVE</b>	Combination	Max	3.01	0.754	0	0.0002593
261	7.5	<b>ENVE</b>	Combination	Max	3.01	0.88	0	0.0002593
261	8	<b>ENVE</b>	Combination	Max	3.01	1.006	0	0.0002593
261	$\overline{0}$	<b>ENVE</b>	Combination	Min	$-0.141$	$-1.006$	0	$-0.0004561$
261	0.5	<b>ENVE</b>	Combination	Min	$-0.141$	$-0.88$	0	$-0.0004561$
261	$\blacksquare$	<b>ENVE</b>	Combination	Min	$-0.141$	$-0.754$	0	$-0.0004561$

*Figure 4 118 Choose Tables for Display form*



#### **NOTE**

Take absolute value for both shear and axial force to enter into "Bolt Connections" file. In excel file just exported, in Station column, select stations at two ends of beam. Filter the largest value in "V2" column and largest value in "P", enter these value to "Bolt Connections" file.

# *4.5.1.2 Material and geometry*

In order to illustrate a single plate connection with full of dangerous cases, we assume information as figure below.



#### **4.5.2. Plate (beam side)** *4.5.2.1 Bolt shear*

Shear strength of bolt group:

$$
\frac{R_n}{\Omega} = \frac{CF_{m}A_b n_r}{\Omega} = \frac{3.08 \times 45 \times \frac{\pi \times 2^2}{4} \times 1}{2} = 217.71 \,\text{kN}
$$

Where:

− C: coefficient for eccentrically loaded bolt groups (represents the number of bolts that are effective in resisting the eccentric shear force) is selected from table 7-8 AISC Steel Construction Manual 13<sup>th</sup>, with n=4, s=75mm (3in),  $e_x = 202$ mm (7.95in), using interpolation method we find out C value is approximately **3.08**.

 $7 - 38$ 

#### DESIGN CONSIDERATIONS FOR BOLTS



− n<sub>v</sub>: number of shear plane through bolt. In this case, n<sub>v</sub>=1.

Demand shear strength for bolt group:

$$
V_a = \sqrt{P^2 + V^2} = \sqrt{50^2 + 200^2} = 206.16 \text{ kN}
$$

Demand/Capacity ratio  $D$ 

$$
D/C = \frac{206.16}{217.71} = 0.95 : OK
$$

# *4.5.2.2 Shear yielding/buckling and flexure yielding*



# *4.5.2.3 Bolt bearing under shear load*



## *4.5.2.4 Shear yielding*



## *4.5.2.5 Shear rupture*



# *4.5.2.6 Block shear rupture*



# *4.5.2.7 Flexure rupture*





# *4.5.2.8 Bolt bearing under axial load*



# *4.5.2.9 Tension yielding*



# *4.5.2.10 Tension rupture*



#### *4.5.2.11 Tear out under axial load*



# **4.5.3. Beam checking**

Checking bolt bearing under shear load and tension load, shear yielding, tension rupture and tear out under axial load similarly for beam. The other limit states are not necessary.

# **5.1. Crane Types CHAPTER 5. CRANE SYSTEMS DESIGN**

The most common crane systems used in steel buildings are given in Table 5 1.

These categories of crane service classification have been established in Table 5 2.

This document refers only to Classes A through D.





#### *Table 5 2 CMAA Crane Service Classes5*



<sup>5</sup> Crane Manufacturers Association of America CMAA (2008), Specification No. 70, Charlotte, NC.



*Figure 5 1 Underhung Bridge Crane*

The monorail cranes are characterized by the hoist being suspended from the lower flange of a single supporting runway beam (See Figure 5 2). Monorails are used where the need to lift and move items can be confined to one direction.



*Figure 5 2 Monorail Crane*

Top Running Cranes as show in Figure 5 3 are generally used in workshops and warehouses where lifting capacity is required over a large span of the floor area.



*Figure 5 3 Overhead Crane in Design Example*

Top Running Cranes usually provide greater hook height and clearance below the crane girder than underhung cranes.



The JIB crane is a crane that has a rotating horizontal boom attached to a fixed.

Bridge cranes can be designed having either single girder, double girder or box girder.

Single girder is generally used on shorter spans and lower capacities or service classifications. Double girder cranes provide greater hook height, but are no more durable than single girder cranes. However, crane and accessories such as bridge cranes, hoist, and trolley, etc. are provided by the manufacturer, not by BMB&A.

*Figure 5 4 Jib Crane*

# **5.2. Design procedure**

The clients will usually show their basic requirements in the design brief. Then the designers need to establish various parameters that will influence the structural design of the building.

These may include like below:

- (1) Crane type (top running, underhung, etc.);
- (2) Capacity (rated in tons) and service classification (CMAA);
- (3) Power source: most use electric instead of hand geared.

For electric powered cranes, method of operation (pendant, cab, or radio);

- (4) Crane load:
- a. The self-weight: Bridge weight (CW) and weight of trolley with hoist (HT);
- b. Maximum wheel load without impact;

c. Special allowances for vertical impact, lateral force, longitudinal force, or the loads factored for dynamic effects and lateral loads, if required.

- (5) Dimensions:
- a. The building layouts

b. Number wheelbase (NWb), distance between cranes (LCr), end-truck length (N), distance wheelbase(W);

c. Level of the top of rail (TOR), the clearance above the top of the rail, Horizontal clearance, vertical clearance, and clearance beneath the runway beam or hook height.

(6) Deflection limits for the crane runway beam and portal frame.

Utilization and state of loading for fatigue assessment.



#### **NOTE**

There can be a significant difference in wheel loads and geometry between single and double girder cranes. If the designer cannot establish the make of the crane, then a contingency of 10% load could be added to the load provided by one manufacturer to allow for other make which might be adopted.

The speed of hand-geared cranes is low, and the impact forces which supporting structures may resist are low compared to the faster electric powered cranes.

Once the crane wheel loads and overall geometry have been established, the general design procedure is as given below:

- (1) Design the runway beams
- (2) Determine the crane load reactions on the bracket and load combinations
- (3) Design of main frames.

# **5.3. Design the runway beam**

It is assumed the crane runway beams are simple beams supported at the brackets that cantilever from the main portal columns.



*Figure 5 5 Schematics of Crame Runway Beam*

# **NOTE**

#### **Bracing**

When runway beam depth is less than **900mm**, top flange beam need to be braced by I-260x164x5x6, and V50x50x5 or larger if necessary.

When runway beam depth is not less than 900mm, need to use both top and bottom flange bracing, by I-260x164x5x6, and V50x50x5 or larger if necessary.

#### **Width of runway beam's top flange**

When length of building is greater than **50m**, the width of runway beam's top flange need to be greater than **200mm** to avoid errors in rail installation.

#### **5.3.1. Crane Wheel Load**

#### **Wheel Load**

The maximum wheel load ( $W_{\text{max}}$ ) and the minimum wheel load ( $W_{\text{min}}$ ) can be provided by the crane manufacturer or may be conservatively approximated from the crane loads as follows:



 $NWD = Number of end truck wheels at one end of the bridge$ 

#### **Vertical Impact Force**

The maximum wheel load  $(\mathbb{WL}_{\text{max}})$  used for the design of runway beams, including monorails, their connections and support brackets, shall be increased by the percentage given below to determine the induced vertical impact or vibration force. Vertical impact shall not be required for the design of frames, support columns, or the building foundation. **Crane Type** Monorail cranes (powered) Cab-operated or radio operated bridge cranes (powered) Pendant-operated bridge cranes (powered) Bridge cranes or monorail cranes with hand-geared bridge, trolley and hoist  $\frac{0}{0}$ 25 25 10 0

#### **Lateral Force**

The lateral force shall be assumed to act horizontally at the traction surface of a runway beam, in either direction perpendicular to the beam, and shall be distributed with due regard to the lateral stiffness of the runway beam and supporting structure.

The lateral force in each end-truck wheel on crane runway beams with electrically powered trolleys shall be calculated as below:

$$
T=20\% \frac{RC+HT}{NWb}
$$

#### **Longitudinal Force**

The longitudinal force, acting horizontally at the top of the rails and in each direction parallel to each runway beam on crane runway beams, shall be calculated as 10 percent of the maximum wheel loads excluding vertical impact:

$$
P = 10\% WL_{\text{max}}
$$

#### **5.3.2. Design runway beam using SAP 2000**

Step-by-step moving-load analysis is initiated through the following process:

- • Define Vehicles type through *Define > Moving Loads > Vehicles*: **VEH-X**: defines the maximum wheel load. **VEH-Y-T50**: defines 100% Lateral Forces. **VEH-Y-T100**: defines 50% Lateral Forces.
- • Define The Paths through *Define > Moving Loads > Paths*. **PATH-X**: defines the lane in X-axis. **PATH-Y**: defines the lane in Y-axis.
- • Define Moving-load Patterns and Cases through *Define > Load Patterns* and *Load Cases*: **DEAD**: defines self-weight of model. **WLmax**: defines moving-load **VEH-X** on **PATH-X** lane.

**T100**: defines moving-load **VEH-Y-T100** on **PATH-Y** lane.

**T50**: defines moving-load **VEH-Y-T50** on **PATH-Y** lane.

Set the load-case type to Moving Load, and then specify the vehicles and paths assigned to this moving load, as shown below:



*Figure 5 6 Load-case Data*

#### **Combinations**



#### **Analysis**

- • Set Analysis Options: *Plane Frame*.
- • Sum ratios of runway beam both axis should be less than 1.0.
- Check Deflection as table below:

#### Table 5 3 Types Deflection Limitations for Top Running Cranes<sup>6</sup>



#### **Output**

Using BMB's calculation sheet presents the images and data of the runway-beam design.

# **5.4. Design Steel Frame**

#### **5.4.1. General**

The general design main frame procedure is as given below:

- 1. Determine the load reactions on the bracket.
- 2. Add the crane runway beam dead load to the dead load and add the following new load case:
- a. Crane loads with the maximum wheel load at left column.
- Lateral crane loads with maximum at left column and acting from right to left.
- b. Crane loads with maximum load at right column.
- Lateral crane loads with maximum at right column and acting from left to right



*Figure 5 7 Crane Loading Conditions*

Crane buildings have single or multiple cranes acting in one or more aisles shall be designed with the crane or cranes located longitudinally in the aisle or aisles in the positions that produce the most unfavorable effect. Unbalanced loads shall be applied as induced by a single crane operating in a crane aisle, and by a crane or cranes operating in one crane aisle of a building with multiple crane aisles. See Table 5 4 for a summary of these provisions.

		Schematic	<b>Vertical Impact</b>	<b>Lateral Force</b>
Multiple aisles with	Any one crane in any aisle	CRANE FL FL. <b>RB</b> <b>RB</b>	$0\%$	$100\%$
multiple cranes	Any two adjacent cranes in any aisle	CRANE FL. <b>FL</b> CRANE <b>RB</b> <b>RB</b>	$0\%$ Both cranes	50% Both cranes, or 100% Either crane
	Any one crane in any two adjacent aisles	<b>RB</b> RB <b>RR</b> <b>RB</b>	$0\%$ Both cranes	50% Both cranes, or 100% Either crane
	Any two adjacent cranes in any aisle and one crane in any other nonadjacent aisle	CRANE CRANE $-FL$ $FL -$ CRANE <b>RB</b> <b>RB</b> <b>BB</b> <b>BB</b>	$0\%$ All cranes	50% All three cranes, or 100% Any one crane

*Table 5 4 Loading for Building Frames and Support Columns*

## **5.4.2. Bracket System**



*Figure 5 8 Loading on bracket*

#### **5.4.3. Bracing**

The lateral load on a crane runway beam is transmitted to the main frame column by the bracing systems in the minor axis.



*Figure 5 9 Bracing of minor axis* 

# **5.5. Design example - One Crane per Aisle**

## **5.5.1. Input**





*Figure 5 10 Building Layout* 

#### **5.5.2. Crane Load**

#### **Crane Wheel Load**



## **Loading for Runway Beam**



#### **Loading for Building Columns**





## **5.5.3. Runway Beam Modeling**





*Figure 5 11 Runway Beam in SAP2000 Modeling Section: RB (I – 766x184x8x10), U (U 120x52x4.8x7.8)*

#### **Define Vehicles type**



*(a) Define Vehicles*



*(b) Vehicles "VEH-X" defined for WLmax*

*(c) Vehicles "VEH- Y-T100" defined for T100*

*Figure 5 12 Define Vehicles*





*Figure 5 13 Define Paths of Moving Loads \* PATH-X defined the path on the Major-axis (beams labeled 1 and 2) PATH-Y defined the path on the Minor-axis (beams labeled 3 and 4)*



#### **NOTE**

The **discretization** is as same as output-station spacing of frame. The effect of refining path discretization is apparent in the influence lines which follow (Figure 5 14):



*Figure 5 14 Influence line*



#### *Define Load-Patterns*

#### *Define Load-Case*



*Load-case WLmax Data*

*Load-case T100 Data*

#### **Output**



*Figure 5 15 Steel P-M Interaction Ratios of Runway Beam\**



#### **NOTE**

This ratio is only used for reference, since the SAP model does not include the longitudinal force. The total ratio of the major and minor axis must be less than **1.00**.

#### **Check Runway Beam by Calculation Sheet**

#### **I. INPUT DATA**

1. Material



2. Section

Section of runway beam



Factor



#### **II. FORCE**







**Resultant Moment M33, kNm** 



Resultant Moment M22, kNm



Resultant Shear V22, kN



Resultant Shear V33, kN





#### **NOTE**

\*Length Lb: the largest un-bracing length of runway beam \*Factor Cb: get from Detail of Steel Stress Check Data

#### III. CHECK OF MENBERS FOR STRENGTH CAPACITY

1. CHECK OF RUNWAY BEAM FOR STRENGTH CAPACITY

#### a. Design of runway beam for Flexure

Design of runway beam for flexure - major axis



d. Design of runway beam for Combined Forces and Torsion

Design of runway beam for combined flexure and axial force



Design of runway beam for Shear

$$
\frac{V_r}{V_{cy}} = \frac{V_{22}}{V_{cy}} = 0.323 \qquad \text{[Satisfactory]}
$$

$$
\frac{V_r}{V_{\infty}} = \frac{V_{33}}{V_{\infty}} = \qquad \qquad 0.024 \qquad \text{[Satisfactory]}
$$

2. CHECK OF BRACING FOR STRENGTH CAPACITY Design of bracing for Compression

> $P<sub>c</sub>$  = 41.2 kN [Satisfactory] [Chapter E5]

#### **III. CHECK DEFORMATION**

1. Vertical Defection



6.100  $L/600 =$ 12.5 mm [Satisfactory]  $\Delta_{\rm v}$  $mm <$  $\Delta_{\text{limit}}$  =  $\equiv$ 

2. Horizontal Defection



#### **I. INPUT DATA**

1. Bolt



#### **II. CALCULATING**

**Bolt resisting tension** 

Tensile resistance of bolt

$$
\frac{R_{nt}}{\Omega} = \frac{F_{nt} \times A_b}{\Omega} = 169.65 \text{ kN}
$$

$$
\Omega = 2
$$



OK

Where: Max. Tension in outermost bolt

$$
N_{\text{bM}} = \frac{V_{\text{max}}}{m_r} = 37.9 \text{ kN}
$$

Where: mr is no. of bolt on 1 row Total of tension on 1 bolt

> $T_b = 37.9$  kN  $rac{R_{nt}}{\Omega}$  $\geq$  $T_{\rm b}$

Hence:

#### **Check Welding at Bracket**

#### **I. MATERIAL**

#### 1. Bracket

2. Weld



 $h_h =$  $7 \text{ mm}$  $I_v =$  $900$ <sub>mm</sub>  $800$ <sub>mm</sub>  $I_h =$ 

 $\mathbf{v}$ 

### **II. INTERNAL FORCE FOR CALCULATING**

Total length of horizontal weld

Total length of vertical weld

Shear



#### **IV. CALCULATING**



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#### **5.5.4. Main Frame**

#### **The load frame**

Columns supporting crane fixed about the major axes and pinned about minor axes.



*Typical Longitudinal Force for "CR1 Case"*

The crane runway beam and rail are to be included as dead loads, applied at the bracket. Note that for clarity, dead loads (DL), live roof loads (LL), wind loads (WL) are not shown in this example.

#### **The combinations**



#### **Deflection Checking**



*Figure 5 16 Crane Bracket Horizontal Deflection* 

 $[Y] = min[5.08cm; (575/240)] = 5.08cm$ 

#### **Steel stress check**



#### **NOTE**

When check steel column need to overwrite Factor "Unbraced Length Ratio (Major)" to 1.00 as Figure 5 17.

			Item Description
	<b>Item</b>	Value	
1	Current Design Section	Program Determined	
$\overline{c}$	<b>Framing Type</b>	Program Determined	
$\overline{3}$	OmegaO	Program Determined	
4	Consider Deflection?	No	
5	Deflection Check Type	Program Determined	
6	DL Limit, L /	Program Determined	
7	Super DL+LL Limit, L /	Program Determined	
8	Live Load Limit, L /	Program Determined	
9	Total Limit, L7	Program Determined	
10 <sup>10</sup>	Total-Camber Limit, L7	Program Determined	
11	DL Limit, abs	Program Determined	
12	Super DL+LL Limit, abs	Program Determined	
13	Live Load Limit, abs	Program Determined	
14	Total Limit, abs	Program Determined	
15	Total-Camber Limit, abs	Program Determined	
16	<b>Specified Camber</b>	Program Determined	
17	Net Area to Total Area Ratio	Program Determined	
18	Live Load Reduction Factor	Program Determined	
19	Unbraced Length Ratio (Major)	$\mathbf{1}$	
20	Unbraced Length Ratio (Minor)	Program Determined	
21	Unbraced Length Ratio (LTB)	Program Determined	
22	Effective Length Factor (K1 Major)	Program Determined	Explanation of Color Coding for Values-
23	Effective Length Factor (K1 Minor)	Program Determined	
24	Effective Length Factor (K2 Major)	Program Determined	Blue: All selected items are program ▼ determined
	Set To Prog Determined (Default) Values:	Reset To Previous Values:	<b>Black:</b> Some selected items are user defined
	All Items Selected Items	All Items Selected Items	Value that has changed during Red: . the current session
		OΚ Cancel	

*Figure 5 17 Define ratio for steel stress check*

# **6.1. General CHAPTER 6. MEZZANINE FLOOR DESIGN**

Mezzanines and platforms are often required in industrial buildings. The type of usage dictates design considerations. For proper design the designer needs to consider the following design parameters:

- 1. **Occupancy or Use**.
- 2. **Design Loads** (Uniform and Concentrated).

The dead load includes the weight of panel, concrete slab, finish floor and, wall on floor, and self-weight of joist.

The live load depends on the purpose of the floor. Refer Chapter 2 for more details.

3. **Type of slab**: composite slab, checkered plate, gratings, expanded metals, etc. Each type of floor requires different design of joists.

Mezzanine joists are analyzed and designed as simple span members.

Joist beam spacing needs being less than **2000mm** and depend on bay spacing for equal distances.

- 4. **Stair, Opening, Guard rail requirements**.
- 5. **Design Criteria** (if required): Deflection limitation, Vibration Control, Lateral Stability Requirements.
- 6. **Future Expansion**.

#### Design Procedure

- 1. Design of Joists:
- 2. Design of Flooring
- 3. Design of Main Frames

# **6.2. Composite Slab**

Composite Metal Deck Slabs – most commonly used today. Advantages:

- − Stay in place form.
- − Slab shoring typically not required.
- − Metal deck serves as positive reinforcement.
- Metal deck serves as construction platform.

Shear connector can use Steel Headed Stud or Steel Channel Anchors.

Steel headed stud anchors shall be welded through the deck to the steel cross section. Such anchorage shall be provided by steel headed stud anchors, a combination of steel headed stud anchors and arc spot (puddle) welds, or other devices specified by the contract documents.



*Figure 6 1 Composite Beam Dimensions with Ribs Perpendicular to Beam Span* 

## **6.2.1. Section Properties**

The section properties are reported with respect to the section local axes (2-3). Furthermore the section properties are reported assuming that the entire section is transformed into an equivalent area of the specified base material.



#### **Compact and Non-compact Requirements**

**For flexure**, sections are classified as compact, non-compact or slender-element sections.

Width-to-Thickness Ratios of I-shaped built-up sections are shown as below:

**Ratio:** *b/t Limiting Width-to-Thickness Ratios for Flanges<sup>1</sup>* Compact Section Limits:  $b / t \le \lambda_0 = 0.38 \sqrt{E / F_y}$  $\begin{array}{c|c}\n & \frac{D}{2} & t \\
 & \frac{D}{2} & \frac{D}{2} \\
 & - & - & - \\
\end{array}$ Noncompact Section Limits:  $b / t \leq \lambda_r = 0.95 \sqrt{k_e E / F_t}$ **Ratio:**  $h/t_w$  **Limiting Ratios for Webs of Doubly-symmetric I-shaped sections<sup>2</sup>** Compact Section Limits:  $h / tw \le \lambda_n = 3.76 \sqrt{E/Fy}$ *VIIII III III II* IX Noncompact Section Limits:  $b / tw \le \lambda_r = 5.70 \sqrt{E/Fy}$  $\int f_n$ **Ratio:** *<i>c k* **Limiting Ratios for Webs of Singly-symmetric I-shaped sections<sup>3</sup>** Compact Section Limits:  $\frac{b_c}{L}$ <sub> $\sqrt{E/F}$ </sub>  $\frac{c}{\sqrt{E/F_{\rm g}}}$ /  $\frac{h}{f} \le \frac{h_p}{f} \sqrt{1 + \frac{g_p}{f}} \le 5.70 \sqrt{E/F}$ *p*  $\leq$   $\frac{p}{\sqrt{p}}$   $\leq$  $5.70\sqrt{E/}$  $\left( 0.54 M_{_p} / M_{_p} - 0.09 \right)^2$ *y*  $t_w = \left(0.54 M_{\frac{h}{2}}/M\right)$ −  $0.54 M$  ,  $/M$  ,  $-0.09$ *w*  $(0.54M_p/M_p)$ Noncompact Section Limits:  $h / tw \le \lambda_r = 5.70 \sqrt{E / F_y}$ *where:*  $Fy = 345 MPa$  : Modulus of elasticity of steel  $F_y = 345 MPa$ : Specified minimum yield stress of ST345  $k_c$ : Coefficient for slender unstiffened elements  $k_c = \frac{4}{\sqrt{h/m}}$ ; 0.35  $\leq k_c \leq 0.76$  $F<sub>L</sub>$ : Magnitude of flexural stress in compression flange at which flange local buckling or lateral-torsional buckling is influenced by yielding For major axis bending of compact and noncompact web built-up I-shaped members, with:  $S_{\rm xt}/S_{\rm xc} \geq 0.7$ :  $F_{\rm L} = F_y \times 0.75$  $S_{\rm st}$  /  $S_{\rm st}$  < 0.7 :  $F_{\rm L}$  =  $F_y \times S_{\rm st}$  /  $S_{\rm st}$  ≥ 0.5 $F_y$  $S_{\alpha t} = I x / h_t$ : Section modulus about the x axis of the outside fiber of the tension flange, mm<sup>3</sup>.  $M_y = S_x F_y$ : Section modulus about the x axis of the outside fiber of the compression flange, mm<sup>3</sup>.  $M_y = S_x F_y$ : the yield moment of the extreme fiber. *(AISC F4-1)*  $M<sub>p</sub> = Z<sub>x</sub>F<sub>y</sub>$ : the plastic bending moment. *(AISC F4-2)* 

<sup>&</sup>lt;sup>8</sup>AISC 360-10, Table B4.1 Case 15.

<sup>&</sup>lt;sup>9</sup>AISC 360-10. Table B4.1 Case 16.

#### **Checking Dimension Requirements**

Concrete slabs on formed steel deck connected to steel beams need to be satisfied the following requirements<sup>10</sup>:

(1) The nominal rib height (hr) shall not be greater than 3 in. (75 mm).

(2) The average width of concrete rib or haunch (wr) shall be not less than 2 in. (50 mm), but shall not be taken in calculations as more than the minimum clear width near the top of the steel deck.

(3) The slab thickness above the steel deck shall be not less than 2 in. (50 mm).

(4) a. The concrete slab shall be connected to the steel beam with welded steel headed stud anchors, not larger than 19 mm in diameter.

Steel headed stud anchors, after installation, shall be extended not less than 38mm above the top of the steel deck and there shall be at least 13 mm of specified concrete cover above the top of the steel headed stud anchors.

 b. The diameter of a steel headed stud anchor shall not be greater than 2.5 times the thickness of the base metal to which it is welded, unless it is welded to a flange directly over a web.

 c. Steel anchors shall have at least 1 in. (25 mm) of lateral concrete cover in the direction perpendicular to the shear force, except for anchors installed in the ribs of formed steel decks. The minimum distance from the center of an anchor to a free edge in the direction of the shear force shall be 8 in. (203 mm) if normal weight concrete is used and 10 in. (250 mm) if lightweight concrete is used. (5) The Stud spacing

a. Steel deck shall be anchored to all supporting members at spacing not to exceed 460 mm.

b. The minimum center-to-center spacing of steel headed stud anchors shall be six diameters along the longitudinal axis of the supporting composite beam and four diameters transverse to the longitudinal axis of the supporting composite beam, except that within the ribs of formed steel decks oriented perpendicular to the steel beam the minimum center-to-center spacing shall be four diameters in any direction.

c. The maximum center-to-center spacing of steel anchors shall not exceed eight times the total slab thickness or 36 in. (900 mm).



*Figure 6 2 Dimension Requirements of Steel headed stud anchors*
# **6.2.2. Beam Bending Capacities for Fully Composite Beam**

## **Properties of Transformed Section**

To calculate the moment capacity with an elastic stress distribution, first need to calculate the location of the elastic neutral axis (ENA) and the transformed section moment of inertia as below:



#### **Check Bending Capacities**

Moment Capacity for Positive Bending

$$
f_b = \frac{M_{\text{max}}}{S_{tr}} \le \frac{F_y}{\Omega_b}; \Omega_b = 1.67
$$

Moment Capacity for Negative Bending

$$
f_c = \frac{M_{\text{max}}}{nS_t} \le 0.45 f_c
$$

# **6.2.3. Beam Shear Capacity**

#### **Shear Capacity**

The nominal shear strength of unstiffened or stiffened webs of all other doubly symmetric shapes and singly symmetric shapes and channels, is determined as follows:



#### **Checking the Beam Shear**

The beam shear at the ends of the beam is checked using the following equation.

*(AISC 360-10 G2-1) (AISC G2.1b, 1)* Safety factor for shear  $V_{\text{max}}$  The required shear strength Shear capacity *Vn* where:  $\Omega_{\nu} = 1.67$  $\frac{1}{\max} \leq \frac{1}{\Omega}$ *v*  $V_{\text{max}} \leq \frac{V}{2}$ Ω

# **6.2.4. Steel Anchors**

Shear connector capacities are defined for both shear studs and channel shear connectors. Next the equations used for determining the number of shear connectors on the beam are provided.

The nominal shear force between the steel beam and the concrete slab transferred by steel anchors for Positive Flexural Strength:

$$
V' = 0.5 \min \begin{cases} 0.85 f_c \, ^1A_c & (AISC 360-10 G2.1b) \\ F_y A_s & (AISC 360-10 G2.1b) \end{cases}
$$

where:

 $A_c$  *b<sub>eff</sub>*  $\times t_o \times n$ : area of concrete slab within effective width

area of steel cross section *As*

# **Steel Headed Stud Anchor**

The nominal shear strength of one steel headed stud anchor embedded in a solid concrete slab



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#### **Steel Channel Anchors**

The nominal shear strength of one hot-rolled channel anchor embedded in a solid concrete slab shall be determined as follows:

$$
Q_n = 0.3 \left( tf + 0.5 \, tw \right) l_a \sqrt{f_c E_c}
$$

where:

 $l_a$  length of channel anchor

*tf* thickness of flange of channel anchor

thickness of channel anchor web *tw*

#### **Check Shear Connector Capacity**

The number of shear connectors is given as follows:

$$
\sum Q_n = Q_n N_r \frac{L}{2s} \geq V'
$$

*(AISC 360-10 I3.1c)*

*(AISC 360-10 I8.2a)*

where:

- Number of Stud each position *Nr*
	- *s* Shear Connectors Spacing
- Length of joist beam *L*

## **6.2.5. Beam Deflection and Camber**

#### **Checking Stress when made camber**



#### **Checking Deflection of Joist Beam**



## **6.2.6. Check Decking**

## **Decking Properties**

It is necessary to determine decking's material each project information. Normally, yielding strength of decking is 235 MPa.

[width ]-[rid depth ]-[rib spacing ]-[ average width rib ]	Thickness			
	1.15 0.75 0.95			
$[1000]$ - $[50]$ - $[334]$ - $[166]$	455282 691840 574099 26975 22441 17842			
$[0870]$ -[75]-[287]-[148]	1168944 922823 1415089 23643 36178 29917			

*Table 6 1 Properties of decking*

## **Checking the decking**

Loading impact on decking:  $q_c = B_r \times (q_{cL} \times n + G_{\text{oncrete}} \times t_o)$ 



*Load assign decking (5 cases) Moment 3-3 Diagram*

*Figure 6 3 Schematics of Decking* 

*Shear 2-2 Diagram*

Checking capacity:

$$
\frac{M_{\rm max}}{\mathcal{S}_{\rm x}} \leq \frac{F\mathcal{Y}}{\Omega_{\rm b}}; \Omega_{\rm b} = 1.67
$$

Checking deflection :

$$
f_{\max} \leq [f] = \frac{B}{180}
$$

# **6.2.7. Design Example 8a - Composite Slab**

## **Input**



# *6.2.7.1 Design Composite Joist Beam using Calculation Sheet*

# **Slab Loading**



Dead load applied on each joist:  $q_{DL} = 0.334 + 3.6 + 1.5 \times 2.955 = 8.37(kN / m)$ Hence,



## **Section Properties**



**Classification of Filled Composite Sections for Local Buckling**



**Class:** Noncompact

# **Checking Dimension Requirements**



# **Determining Composite Properties for Plastic Design**



# **Check Bending Capacities**



# **Check Shear Capacities**



# **Check Shear Connector Capacity**



#### **Beam Deflection and Camber**



# **Checking Deflection of Joist Beam**



# **Design Data**



# *6.2.7.2 Design Composite Mezzanine Decking using Calculation Sheet*

Decking [1000]-[50]-[334]-[166] thickness 0.75mm:  $S_x = 1.78E4(mm^3)$ ;  $S_x = 1.78E4(mm^3)$ 

# **Checking the decking**



# **6.3. Checkered Plate Slab**

**NOTE**

# **6.3.1. General**

Allowable thicknesses of Checkered Plate are 4mm, 5mm, 6mm or 8mm (included ribbed plate). Checkered Plate Slab can be designed with or without Flat Bar.



Incase floor load is not less than 4 kN/m²; designers should use Checkered Plate Slab with Flat Bar under.

If floor load is less than 4 kN/m², Checkered Plate Slab without Flat Bar can be used with the joists having maximum spacing 1.2÷1.5m.

## **6.3.2. Detailing Requirement**



*Figure 6 4 Checkered Plate with Flat Bar*



*Figure 6 5 Bolt Connection of Checkered Plate and Joists*



*Figure 6 6 Weld connection at the intersection of Checkered Plates*

## **6.3.3. Checkered Plate with Flat Bar**

#### **Design Strip Properties**



*Figure 6 7 Design Strip of Checkered Plate with Flat Bar*



#### **Check Deflection**

Checkered Plate is considered as a uniform loaded long rectangular plate with longitudinal edges which are free to rotate but cannot move toward each other during bending. The design strip cut out this plate is in the condition of a uniformly loaded bar submitted to the action of an tensile force *H* , as show in Figure 6‑8. The magnitude of *H* is such as to prevent the ends of bar from moving along the x-axis.



*Figure 6 8 Uniformly Loaded Rectangular Plates with Simply Supported Edges*



#### **Check Bending**



#### **Check Weld Connection**



## **6.3.4. Checkered Plate without Flat Bar**

Formulae show ratio of design length and slab thickness(According to formulae of A.L.Teloian):

$$
\frac{l}{t} = \frac{4n_{o}}{15} \left( 1 + 75 \frac{E_{1}}{n_{o}^{4}q} \right) \Longrightarrow l = t \times \frac{4n_{o}}{15} \left( 1 + 75 \frac{E_{1}}{n_{o}^{4}q} \right)
$$

where  $\frac{1}{\epsilon} = \frac{f}{\epsilon} = \frac{1}{16}$  $_{o}$   $\lfloor L \rfloor$  180 *f*  $\frac{1}{n_a} = \left[ \frac{f}{L} \right] = \frac{1}{180}$ : Ratio of allowable deflection of slab. Hence, available joist spacing (distance edge flange to edge flange of joists)  $\lfloor l_{\rho} \rfloor \leq l$ .

# **Input 6.3.5. Design Example 8b - Checkered Plate with Flat Bar**



# **Properties Of Design Strip**



## **Check Deflection**



# **Check Bending**



#### **Check Weld Connection**



# **6.3.6. Design Example 8c - Checkered Plate without Flat Bar**

Input: as same as Design Example 8b, Checkered Plate without Flat Bar.

Specify Thickness: 4.00 mm.



Then, check Deflection and Bending as same as Example 8a.

#### **Conclusion**

The type of checkered plate (with or without flat bar) depends on the load and thickness (if required).

# **6.4. Grating Slab**

# **6.4.1. General**

Open steel rectangular pattern floor is constructed from mild steel and is constructed by a forge-welding process in which the load bearing and transverse bars are heated and joined under pressure.

Gratings is suitable for most floor walkways, gantries, platforms, etc. and could even be used upright as fencing.



*Figure 6 9 Standard Grating*

Standard Range – Readily available with either plain or serrated load bearings bars (5mm thick) at 40 pitch and twisted transverse bars at 50 or 100 centers.

## **6.4.2. Design Example 8c - Grating Slab**

#### **Input**



Consider 1m width of grating slab.



# **7.1. Purlins and Girts Design CHAPTER 7. ACCESSORIES**

# **7.1.1. General**

Purlins and girts are the immediate supporting member for roof and wall sheeting respectively. They act principally as beams, but also perform as struts and as compression braces in restraining rafters and columns laterally against buckling.

Purlins and girts are almost universally zed (Z), channel (C), or double channel section members.

**C-sections** have equal flanges and may be used in single spans and un-lapped continuous spans in multi-bay buildings. Their freestanding stable shape allows easy handling and storage and is easily adapted for use in small and medium sized buildings as structural framework.

**Z-sections** feature one broad and one narrow flange allowing the two sections to fit together snugly, making them suitable for lapping. Z sections of the same depth and different thickness' can be lapped in any combination. Purlins and Girts that are lapped form a structurally continuous line along the length of the building, a factor that contributes significantly to the reduction in building costs.



*Figure 7 1 Typical Purlins and Girts Details* 

## **7.1.2. Design procedure**

Purlins and girts can be calculated as simply supported spans, lapped continuous spans or increased thickness end spans in lapped continuous systems.

# *7.1.2.1 Purlin/Girt spacing*

Purlin spacing must be chosen to suit the type of roof sheeting and ceiling system if any. The use of translucent fiberglass roof sheeting will also restrict the purlin spacing. Some suspended ceiling systems require a maximum purlin spacing of 1200mm (DONGIL project). Purlin deflections must also be controlled.

Roof purlins are to be arranged according to the following guide lines as applicable:

- 700 mm between first roof purlin and the eave strut
- Intermediate spacing not exceeding 1600 mm.

## *7.1.2.2 Loading*

The maximum spans are determined not only from wind load considerations, but also from live load requirements. Refer Chapter 2 for more detail.

Collateral load applied on purlin such as Fire Sprinkler, MEP System, Plaster Ceiling System, etc. should be clearly specified as concentrated or uniformed load to identify the most dangerous case.







*Figure 7 2 Schematics of Purlin/Girt/Eave strut* 

The peak local pressures zones around the perimeter of the roof govern the purlin spacing these areas, and the purlins in end bays is usually adopted for the rest of the roof because of the difference in loads and bending moments between end bays and internal span purlins. It is therefore advantageous for economical design to consider:

- Increased wall thicknesses in end span purlins, or
- Reduced end bay spacing, or

Extra purlin spacing, extra purlins, or increase lapped length in end spans, provided this increases the design strength of the purlins.

It is necessary to provide bridging between purlins to reduce the effective lengths to control flexural-torsion buckling. BMB recommends at least one row or purlin bracing in every span, and that un-braced length be restricted to less than 20 times the section depth, or 4000 mm, whichever is less.





# *7.1.2.3 Lapped Length*

The lapped length for Z-section purlins is a minimum of 10 percent of the span.

# *7.1.2.4 Purlin bolts*

Standard purlins and girts cleats are generally used without analysis or design. Purlin cleats are subjected not only to axial load but also to bending moments. Angel cleats also provide greater robustness during transport and erection. One yardstick for robustness is that girt cleats should not yield when stood on by a heavy worker. This would equate to a 1.1 kN load applied to the tip of the cleat with a 1.5 load factor to allow for dynamic effects as the worker climbs the steel work.

The standard bolt is an M12 Type 4.6. Refer Figure 7 6 for more details.



## **Input**





**Checking ratio of slenderness: (compressive member with pin connection double end)**

$$
\frac{kL}{r} = \frac{1 \times 2.3}{0.019} = 121 < 200
$$

 $\overline{\Gamma}$ 

Checking capacity the compression



# **7.1.3. Detailing Requirement**







*Figure 7 4 Alternative Flange Brace at Expansion Joint Detail*



*Figure 7 5 Typical Purlin Bracing Details* 





*Figure 7 6 Typical Purlin Cleat Details* 



## **NOTE**

For purlin Z300, C300, use X sag rod bridging.



*Figure 7.7 Detail A – Sag rod at wall*



*Figure 7.8 Detail B – Sag rod at eave*



*Figure 7.9 Detail C – Sag rod at ridge*

*Figure 7.10 Detail D – Sag rod at roof*



*Figure 7.11 X - Sag rod*

# **7.2. Drainage**

The purpose of this section is to provide information for gutter capacity, the spacing of drainage downspouts (conductors), and secondary emergency overflow design for the roofs of metal buildings.



#### Emergency overflow

Where roof drains are required, secondary (emergency overflow) roof drains or scuppers shall be provided where the roof perimeter construction extends above the roof in such a manner that water will be entrapped if the primary drains allow buildup for any reason.

# **7.2.2. Example**

Building Dimensions: 80m wide x 80m long (10 @ 8m bays).

Symmetrical gable roof configuration. Roof slope: 5%.

Rainfall intensity: I=250 mm/h= 9.84 in./h

# **Solve**

Width of roof to be drained:  $L = 8m = 26.25 \text{ ft}$ Length of gutter to be drained:  $L = 8m = 26.25 \text{ ft}$ Factor:  $B = 1.00$ 

Gutter Size 720(width x height):  $0.18 \text{ m} \times 0.202 \text{ m} = 0.59 \text{ ft} \times 0.66 \text{ ft}$ 

Gutter capacity:

$$
\left[L\right] = \left[\frac{0.59^{\frac{3}{7}} \times 0.66^{\frac{3}{7}} \times \left(\frac{43200}{131.23 \times 9.84 \times B}\right)^{\frac{5}{14}}}{0.481}\right]^{\frac{28}{13}} \times 0.3048 = 9m \ge L = 8m \cdot OK
$$

Downspout capacity: Diameter 220mm thickness 6.6mm

$$
[A] = \frac{9.84 \times 131.23 \times 26.25 \times 1}{1200} \times 0.00064516 = 0.018 m^2 > A = \pi \frac{0.214^2}{4} = 0.036 m^2 OK
$$

# **APPENDIX A: HOT ROLLED SECTION**



*Table A.1 I Section*

*Table A.2 Channel Section*







## *Table A.4 Pipe Section*



*Table A.5 Tube in Square Section*

Section	$W(\text{kg}/\text{m})$	Section	$W(\text{kg}/\text{m})$	Section	$W(\text{kg}/\text{m})$
$-20x20x1.2$	0.7	$\Box$ - 80x80x4.5	10.3	$\Box$ - 250x250x9	66.5
$-20x20x1.6$	0.87	$\Box$ - 90x90x2.3	6.23	$\Box$ - 250x250x12	86.8
$-25x25x1.2$	0.87	$\Box$ - 90x90x3.2	8.51	$-250x250x16$	112.4
$-25x25x1.6$	1.12	$\Box$ - 90x90x4.5	11.7	$\Box$ - 300x300x6	54.7
$-25x25x2.3$	1.53	$-90x90x6$	15.1	$\Box$ - 300x300x9	80.6
$-25x25x3.2$	1.98	$\Box$ - 100x100x2.3	6.95	$\Box$ - 300x300x12	106
$-30x30x1.2$	1.06	$\Box$ - 100x100x3.2	9.52	$\Box$ - 300x300x16	138
$-30x30x1.6$	1.38	$\Box$ - 100x100x4	11.7	$\Box$ - 300x300x19	160
$-30x30x2.3$	1.73	$\Box$ - 100x100x4.5	13.1	$\Box$ - 350x350x6	64.1
$-30x30x3.2$	2.48	$\Box$ - 100x100x6	17	$\Box$ - 350x350x9	94.7
$-40x40x1.6$	1.88	$\Box$ - 100x100x9	24.1	$\Box$ - 350x350x12	124
$-40x40x2.3$	2.62	$\Box$ - 100x100x12	30.2	$\Box$ - 350x350x16	163
$-40x40x3.2$	3.49	$-125x125x3.2$	12	$\Box$ - 350x350x19	190
$-50x50x1.6$	2.38	$-125x125x4.5$	16.6	$-400x400x9$	109
$-50x50x2.3$	3.34	$\Box$ - 125x125x5	18.3	$\Box$ - 400x400x12	143
$\Box$ - 50x50x3.2	4.5	$\Box$ - 125x125x6	21.7	$\Box$ - 400x400x14	166
$\Box$ - 50x50x4.5		$60.2$   $\Box$ - 125x125x9	31.1	$\Box$ - 400x400x16	188
$\Box$ - 50x50x6	7.56	$\Box$ - 125x125x12	39.7	$\Box$ - 400x400x19	220
$-60x60x2.3$	4.06	$\Box$ - 150x150x4.5	20.1	$\Box$ - 400x400x22	251
$-60x60x3.2$	5.5	$\Box$ - 150x150x6	26.4	$\Box$ - 450x450x9	122
$-60x60x4.5$	7.43	$\Box$ - 150x150x9	38.2	$\Box$ - 450x450x12	160
$-60x60x6$	9.45	$\Box$ - 175x175x4.5	23.7	$\Box$ - 450x450x16	209
$-75x75x2.3$	5.14	$\Box$ - 175x175x6	31.1	$-450x450x19$	250
$-75x75x3.2$	7.01	$\Box$ - 200x200x4.5	27.2	$\Box$ - 450x450x22	286
$-75x75x4.5$	9.55	$\Box$ - 200x200x6	46.9	$\Box$ - 500x500x12	181
$-75x75x6$	12.3	$\Box$ - 200x200x9	52.3	$\Box$ - 500x500x16	238
$-80x80x2.3$	5.5	$-200x200x12$	67.9	$-500x500x19$	280
$-80x80x3.2$	7.51	$\Box$ - 250x250x6	45.2	$\Box$ - 500x500x22	320

# *Table A.6 Tube in Rectangular Section*



## *Table A.7 Downspout*



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