



# DESIGN GUIDELINES BMB&A

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# R&D TEAM DESIGN GUIDELINES BMB&A

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With lots of outstanding advantages compared with other materials, steel is increasingly used in civil and industrial buildings. Over the past hundreds of years, a vast majority of researches on steel structure conducted in order to improve safety for the structure as well as decrease cost of the building. To pursue the dream of becoming the largest steel company in Vietnam and over the world, design department made effort continuously to both finish all projects on schedule and study and apply newest achievements over the world to buildings. Standards and documents from the most developed country in steel field America used to guarantee the stability of the structure as well as strict requirements of architects and weight from customers.

From the first design guide published in 2011 in this latest design guide, there are more additional theories and guidelines for constructing SAP model and calculating connection, mezzanine floors, loads for building, etc.... according to the newest standards. In addition, many practical pictures, tables, serving for computing and calculating supplemented as well as mistakes are also updated and revised. The object of this editing is to make a standard of designing for all designers and support the other department to understand more about the processing of the design department.

In the compilation process, mistakes cannot be avoided so it would be so helpful to receive any feedback from readers to more complete in the next edition.

Thanks and Best regards,

Chief of Approval Design,

Mai Xuan Quang

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# **GENERAL INFORMATION**

#### Introduction

 The Design Department manual outlines the design process requirements of the BMB&A Standard in BMB&A J/S CO.
 The Design manual establishes rules and standards to ensure that parameters and figures are presented uniformly throughout in BMB&A J/S CO.

Scope of the design manual

(3) The design manual is binding for all Design Team, Estimator Team and Office in the BMB&A J/S CO.

# CHAPTER 1. MATERIALS 1.1. Materials

## Table 1 1 Material Specifications (unless indicated in drawings)

Materials	Specifications	Minimum Strength
	(or equivalent)	
Built-up Member	ASTM-A572 GR50 (Q345)	$Fy = 34.5 kN / cm^2$
	ASTM-A36 (Q235)	$Fy = 23.5 kN / cm^2$
Steel Plate For Connection	ASTM-A572 GR50 (Q345)	$Fy = 34.5 kN / cm^2$
Hot Rolled Member	JIS G3101/SS400	$Fy = 23.5 kN / cm^2$
Cold Formed Light Gauge Shapes, Purlin	JIS G3302 (G450, Z275)	$Fy = 45.0 \text{kN} / \text{cm}^2$
& Girt Bracing		
Roof Sheeting	ASTM A792M/755M	$Fy = 30.0 kN / cm^2$
Decking	JIS G3302	$Fy = 23.5 kN / cm^2$
Checker Plate	SS400	$Fy = 23.5 kN / cm^2$
X - Bracing Rod	JIS G3101/SS400	$Fu = 40 kN / cm^2$
X - Bracing Cable	ASTM A416	$Fu = 54 kN / cm^2$
Bracing (Steel Pipe or Steel Angle)	JIS G3101/SS400	$Fy = 23.5 kN / cm^2$
Anchor Bolt Grade 5.6	TCVN 1916-1995	$Fu = 50 kN / cm^2$
Machine Bolts for Purlin/Girt Connections	DIN 933 TYPE 4.6	$Fu = 40 kN / cm^2$
High Strength Bolt Grade 10.9	DIN 933 TYPE 10.9	$Fu = 100 kN / cm^2$
High Strength Bolt Grade 8.8	DIN 933 TYPE 8.8	$Fu = 80 \text{kN} / \text{cm}^2$
Minimum Flow Strength of Welding	AWS A5.17 & AWS A5.20 (E60XX)	$Fy \ge 34.5 kN / cm^2$

# **1.2. Plates**

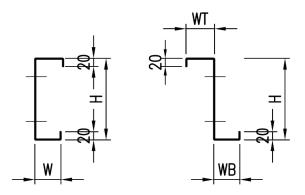
	Thickness (mm)		Order Size (m)	Specifications
3	14	35	W = 1m; 1.5m; 2m; 2.5m; 3m	ASTM-A572 Gr50 (Q345)
4	16	40	L = 6m; 9m; 12m	ASTM-A36 (Q235)
5	18	50		
6	20	52		
8	22	55		
10	25			
12	30			

## **1.3. Hot Roll Members**

## Refer Appendix A for more sections.

Type (mm)	Order size (mm)	Usage
I section	IPE 150x75x5x7 IPE 200x100x5.5x8 IPE 200x200x8x12 IPE 250x125x6x9 IPE 300x150x6.5x9 IPE 300x300x10x15 IPE 400x200x8x13 □ 100x100x2.5	Wind column, end wall rafter, mezzanine joist, & sub-structure
	□ 150x150x2.8/3.5 □ 200x100x2.8	EB bracing, man door/ window/ shutter door frames, sub-structure
Channel	U 100x50x5x7.5 U 125x65x6x8 U 150x75x6.5x10 U 200x73x6.5x8	Joist A Cap channel for crane beams, a stringer
	U 200x73x8.5x10 U 250x78x7x11	for the staircase
Angle	V 50x50x5/6 V 60x60x5 V 63x63x5/6 V 75x75x6/7 V 90x90x6/7 V 100x100x8/10	Flange braces, X bracing & open web joist members
Pipe	Ø 22x2.0 Ø 32x2.0 Ø 42x3.2	Steel handrail
	Ø 60x3.0 Ø 90x2.0/2.3/3.0/3.2/4.0 Ø 114x2.5/2.9/3.18/3.96 Ø 141x3.96 Ø 168x3.96 Ø 219x4.78 Ø 273x6.35	X bracing, Space frame, diagonal member

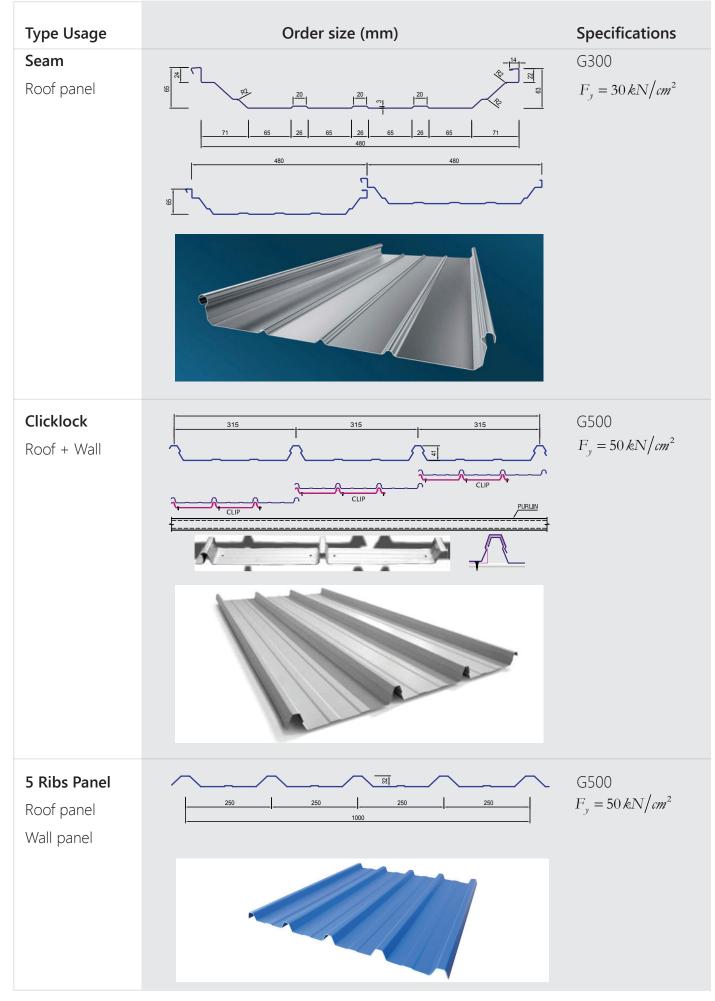
## **1.4. Cold Form Sections**

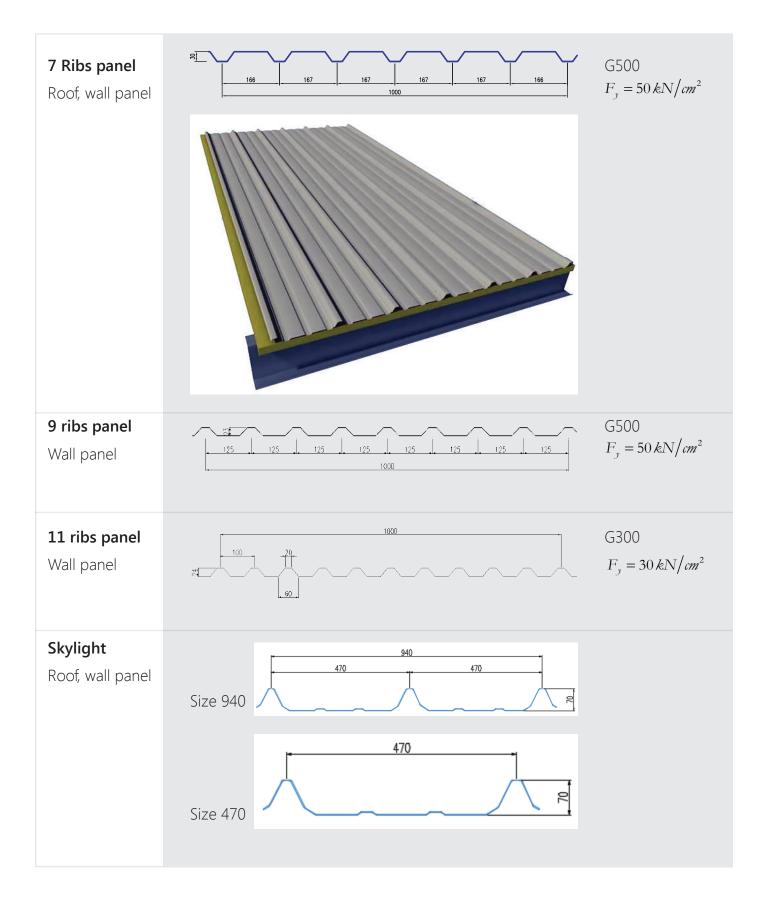


Sym.	WB-WT	Thickness/Quantity (kg/m)						
		1.5	1.8	2	2.3	2.5	2.8	3
C100	50*	2.593	3.111	3.457	3.975	4.321	-	-
C120	50*	2.833	3.399	3.777	4.344	4.721	-	-
C150	50*	3.193	3.831	4.257	4.896	5.321	-	-
C150	65	3.553	4.263	4.737	5.448	5.922	-	-
C150	75*	3.793	4.552	5.057	5.816	6.322	-	-
C180	50*	3.553	4.263	4.737	5.448	5.922	-	-
C180	65	3.913	4.696	5.217	6.000	6.522	-	-
C200	50*	3.793	4.552	5.057	5.816	6.322	7.080	7.586
C200	65	4.153	4.984	5.537	6.368	6.922	7.752	8.306
C200	70*	4.273	5.128	5.697	6.552	7.122	7.976	8.546
C200	100*	4.993	5.992	6.658	7.656	8.322	9.321	9.987
C250	65	4.753	5.704	6.338	7.288	7.922	8.873	9.506
C300	65	-	6.424	7.138	8.208	8.922	9.993	10.707
Z150	50-56	3.265	3.918	4.353	5.006	5.441	-	-
Z150	62-68*	3.553	4.263	4.737	5.448	5.922	-	-
Z175	50-56	3.553	4.263	4.737	5.448	5.922	-	-
Z175	62-68*	3.853	4.624	5.137	5.908	6.422	-	-
Z200	50-56*	3.853	4.624	5.137	5.908	6.422	7.192	7.706
Z200	62-68	4.153	4.984	5.537	6.368	6.922	7.752	8.306
Z200	72-78*	4.393	5.272	5.857	6.736	7.322	8.200	8.786
Z250	62-68	4.753	5.704	6.338	7.288	7.922	8.873	9.506
Z250	72-78*	4.993	5.992	6.658	7.656	8.322	9.321	9.987
Z300	62-68	-	6.424	7.138	8.208	8.922	9.993	10.707
Z300	72-78*	-	6.712	7.458	8.577	9.322	10.441	11.187

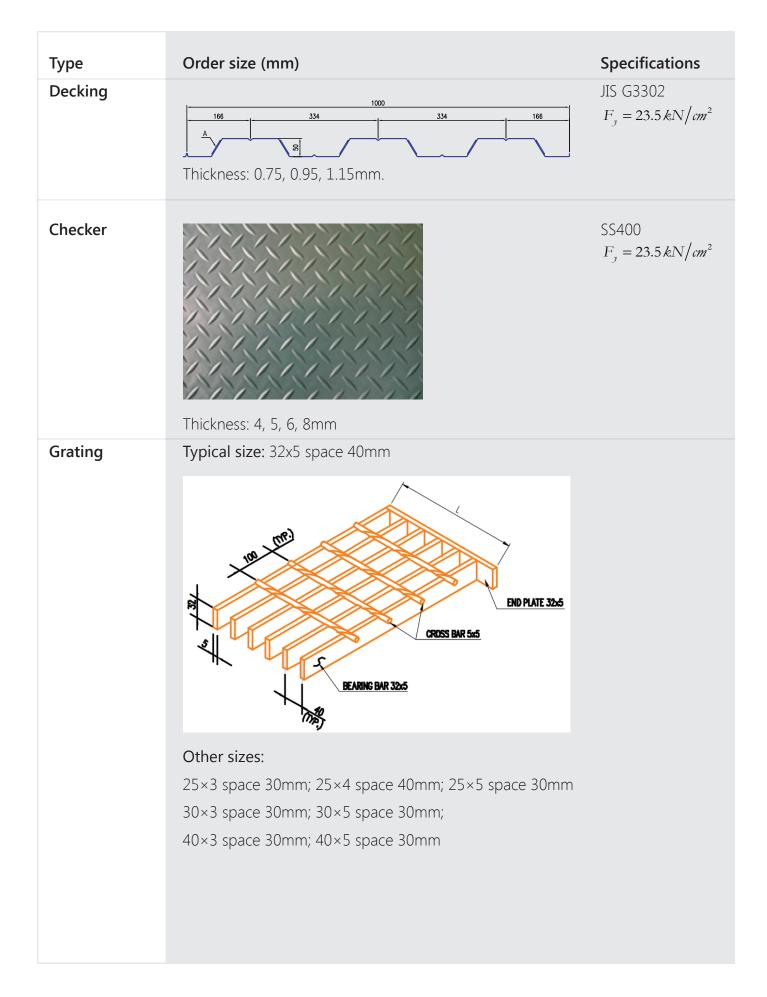
(\*) Not available at Hung Yen Factory.

# 1.5. Sheeting





## 1.6. Slab



# **1.7. Bolts**

	Туре	Bolt diameter (mm)	Order length (mm)	Usage	Specifications
Cast-in Anchor Bolt	Headed bolt			Anchor bolt for end wall frame & partitions, column bases	TCVN 1916-1995 (Grade 5.6) $F_{\mu} = 50  kN/cm^2$
	& Hooked bolt	M16 M20 M24 M27 M30 M36 M42 M48	400 500 600 700 800 900 1000 1200	Anchor bolt for mainframe & mezzanine column bases	
Post- installed Anchor Bolt	Chemical bolt	M16 M20 M24 M27 M30 M12 M16 M20	190 260 300 340 380 106 145 184	Anchor bolt for partitions & stair column bases Connect to RC column	(Grade 4.6) $F_{u} = 40  kN/cm^2$
Machine bolt High strength bolt		M12 M16 M20 M24 M27 M30 M36	60 70 90 90 100 100	Purlin/girt connection Mainframe connection	DIN 933 Grade 4.6 DIN 933 Grade 8.8 DIN 933 Grade 10.9

# 1.8. Bracing

<b>Type</b> (mm)	Order size (mm)	Tension strength (kN)	Usage	Specifications
Rod	Ø16 Ø20 2Ø16 Ø25 2Ø20	30.16 47.12 60.32 73.63 94.25	X-bracing in roof and wall	JIS G3101/SS400 $F_{\mu} = 40  kN/cm^2$
Cable	Ø16 Ø20 2Ø16 Ø25 2Ø20	40.72 63.62 81.43 99.4 127.23		ASTM A416 $F_{\mu} = 54  kN/cm^2$

# **CHAPTER 2. CODES AND LOADS**

## 2.1. Codes and Manuals

The Pre-Engineered Building described in these calculations was designed according to the latest U.S.A. Buildings and Design Codes that have been referred to in the design:

- 1. "Minimum Design Loads for Buildings and Other Structures", ASCE 7-10.
- 2. "International Buildings Code", IBC– 2012.
- 3. "Metal Building Systems Manual 2012" issued by MBMA.
- 4. "American Institute of Steel Construction" Allowable Stress Design, AISC 360-10.
- Cold formed components have been designed in accordance with:
   "American Iron and Steel Institute" Cold Formed Steel Design Manual, AISI S100-2007
- 6. Welding has been applied in accordance with:
- "American Welding Society", AWS-2010.
- Wind speed & live load for Vietnam Project has been applied in accordance with: "Load and Effects - Design Standard" TCVN 2737:1995.
- 8. Earthquake load has been considered in accordance with: "Uniform Building Code", UBC 1997 Edition.

The above codes are to be used for the design of buildings by BMB&A design engineers unless otherwise specified in the Contract Information Form.

# 2.2. Design Loads

BMB&A Pre-Engineered Buildings are designed according to ASCE 7-10.

## 2.2.1. Dead load (DL)

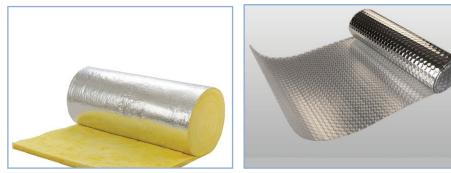
## Dead Load on Roof Sheeting

This includes the self-weight of rigid frames and imposed dead load due to secondary elements like roof sheeting, purlins, insulation...

Following are some standard dead loads (in kN/m2) as below:

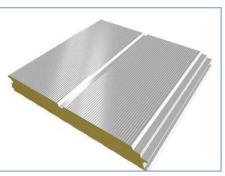
Purlin spacing	Insulation	$\mathbf{DL} \left( \mathrm{kN} / \mathrm{m}^2 \right)$
>1.2m	• 1 layer of (Fiberglass/Air bubble/PE/Sandwich)	0.10
>1.2 <i>m</i>	<ul> <li>2 layers of (<i>Fiberglass/Air bubble/PE/Sandwich</i>)</li> <li>PU (2 layers sheeting)</li> <li>Rock Wool</li> </ul>	0.15
≤1.2m	• 1 layer of (Fiberglass/Air bubble/PE/Sandwich)*	0.15

\* Calculate manually dead load when using 2 layers insulation and purlin spacing is smaller than 1.2 m





Air Bubble



PU (Poly Urethane)







PE (Poly Ethylene)

Sandwich Figure 2.1 Types of Insulation

Rock Wool

## Mezzanine dead load

Concrete slab (in kN/m2) Decking panel/ Checker plate/ Grating (in kN/m2) Finishing (in kN/m2) Wall load on beams, joists (in kN/m) Stair/ elevator (in kN) These values are determined depending on the specific weight & the sizes of material.

Following are some specific weight that BMB&A is used to calculate:

	Material	Uniform Load
Reinforced c	oncrete	25.00 kN/m <sup>3</sup>
Brick wall		18.00 kN/m <sup>3</sup>
Tempered gla	ass	25.00 kN/m <sup>3</sup>
Solar cell		0.23-0.27 kN/m <sup>2</sup>
Grating		$0.45-0.82 \text{ kN/m}^2$
	Fiberglass thickness 50mm/ 100mm	$0.12 \div 0.24 \text{ kN/m}^3$
	Air bubble P2/A2	$0.002 \div 0.004 \text{ kN/m}^2$
Insulation	PE foam thickness 5mm	$0.20 \div 0.30 \text{ kN/m}^3$
Insulation	Sandwich	0.16 kN/m <sup>3</sup>
	Rock wool	$0.27-0.35 \text{ kN/m}^3$
	PU thickness 50mm	0.38-0.40 kN/m <sup>3</sup>
Sheeting thic	kness 0.45mm	0.05 kN/m <sup>2</sup>

## 2.2.2. Collateral load (AL)

Collateral load here means superimposed dead load (in kN/m2) includes:

Ceiling (Gypsum board) HVAC duct Lighting fixtures

Sprinklers

Besides, some collateral systems (such as HVAC duct, sprinklers) have live load; we calculate them into live load (LL).

## 2.2.3. Live load

## Roof live load (LLr)

The roof live load is used to design purlin, depends on the tributary area of rigid frames. It includes the weight of labor & accessary to install, repair the roof.

Refer to ASCE 7-10, Table 4-1 Minimum Uniformly Distributed Live Loads,  $L_0$ , and Minimum Concentrated Live Loads. For built-up frames, minimum uniformly distributed live load on the roof is 1.0 kN/m<sup>2</sup>. ASCE 7-10 section 4.8.2 allows the use of 0.57kN/m<sup>2</sup> as live load for roof and purlins. Roof live loads as per other building codes should be verified before proceeding with your design. Some customers/consultants may require pattern loading in live load applications.

## Live load on frame (LL)

Refer to TCVN 2737:1995 (Load and Effects - Design Standard), section 4.3.1, minimum uniformly distributed live load on the frame is 0.3 kN/m<sup>2</sup>.

Country	Code	LL on roof (kN/m2)	LL on frame (kN/m2)
United States	ASCE 7-10	0.57 - 1	0.57 - 1
Vietnam	TCVN 2737:1995	0.57	0.3
Thailand	~ ASCE 7-10	0.57	0.3
Myanmar	MNBC 2016 ~ ASCE 7-10	0.57	0.3
Cambodia	~ ASCE 7-10	0.57	0.3
Indonesia	~ ASCE 7-10	0.57	0.3
Philippines	NSCP 2015 ~ ASCE 7-10	0.75-1.00	0.6

## Mezzanine live load (FL)

For floor loads of different occupancy, refer to CHAPTER 6. MEZZANINE FLOOR DESIGN.

## 2.2.4. Wind load (WL)

The wind load pressure is determined in accordance with *Chapter 26-Chapter30, ASCE 7-10*. Wind loads are governed by wind speed, roof slope, wave height and open wall conditions of the building. BMB&A's steel buildings are not designed for a wind speed less than 110 km/h.

Wind load on frame W depends on importance factor  $I_w$ .

$$W = q_z \times \left(GC_{pf} - GC_{pi}\right) \times B$$

Where: - W = wind design pressure on frame

 $-q_z$  = velocity pressure evaluated at height z above ground (N/m<sup>2</sup>);

 $q_z = 0.613 \times K_z \times K_z \times K_d \times V^2 (N/m^2)$ ; V in m/s. (27.3-1, ASCE 7-10)

-  $GC_{pf}$  = product of the equivalent external pressure coefficient and gust-effect factor to be used in determination of wind loads for MWFRS of low-rise buildings

-  $GC_{pi}$  = product of the internal pressure coefficient and gust-effect factor to be used in determination of wind loads for buildings

-B = bay spacing (m)

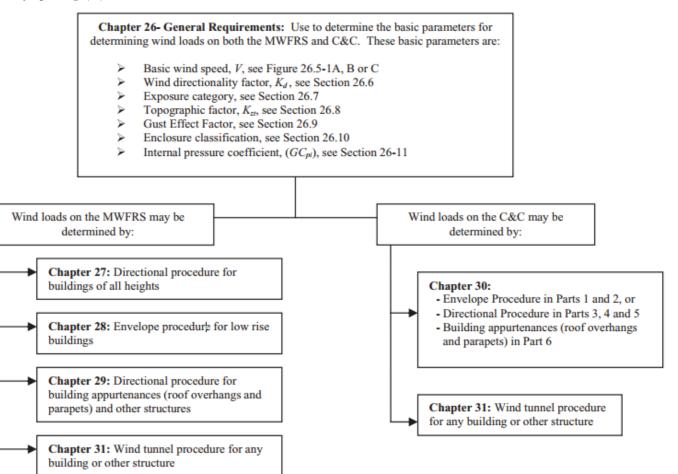


Figure 2.4 Outline of process for determining Wind load

## **Risk categorization**

Building and other structures shall be classified, based on the risk to human life, health, and welfare associated with their damage or failure by nature of their occupancy or use, refer to *Table 1.5-1, ASCE 7-10* for purposes of applying flood, wind, snow, earthquake, and ice provisions. Each building or other structure shall be assigned to the highest applicable risk category or categories.

## Table 2 1 Risk Category of Buildings and Other Structures in Flood, Wind, Snow, Earthquake, and Ice Loads (Table 1.5-1, ASCE 7-10)

Use or Occupancy of Buildings and Structures	Risk Category
Buildings and other structures that represent a low risk to human life in the event of failure	Ι
All buildings and other structures except those listed in Risk Categories I, III, and $\ensuremath{\mathrm{IV}}$	II
Buildings and other structures, the failure of which could pose a substantial risk to human life. Buildings and other structures, not included in Risk Category IV, with potential to cause a substantial economic impact and/or mass disruption of day-to-day civilian life in the event of failure. Buildings and other structures not included in Risk Category IV (including, but not limited to, facilities that manufacture, process, handle, store, use, or dispose of such substances as hazardous fuels, hazardous chemicals, hazardous waste, or explosives) containing toxic or explosive substances where their quantity exceeds a threshold quantity established by the authority having jurisdiction and is sufficient to pose a threat to the public if released.	Π
Buildings and other structures designated as essential facilities. Buildings and other structures, the failure of which could pose a substantial hazard to the community. Buildings and other structures (including, but not limited to, facilities that manufacture, process, handle, store, use, or dispose of such substances as hazardous fuels, hazardous chemicals, or hazardous waste) containing sufficient quantities of highly toxic substances where the quantity exceeds a threshold quantity established by the authority having jurisdiction to be dangerous to the public if released and is sufficient to pose a threat to the public if released. a Buildings and other structures required to maintain the functionality of other Risk Category IV structures.	IV

<sup>a</sup> Buildings and other structures containing toxic, highly toxic, or explosive substances shall be eligible for classification to a lower Risk Category if it can be demonstrated to the satisfaction of the authority having jurisdiction by a hazard assessment as described in Section 1.5.2 ASCE 7-10 that a release of the substances is commensurate with the risk associated with that Risk Category.

Country	Code		Risk category			
United States	ASCE 7-10 (Table 1.5-1)	Ι	II	III	IV	
Vietnam	TCVN 9386:2012	IV	III, II	Ι	Special	
	(Appendixes E, F)					
Thailand	~ ASCE 7-10	Ι	II	III	IV	
Myanmar	MNBC 2016 (Table 3.1.2)	Ι	II	III	IV	
Cambodia	~ ASCE 7-10 (Table 1.5-1)	Ι	II	III	IV	
Indonesia	~ ASCE 7-10	Ι	II	III	IV	
Philippines	NSCP 2015 (Table 103-1)	$\vee$	III, IV	II	Ι	

## Table 2 2 Risk Category of Buildings for different codes

Minimum design loads for structures shall incorporate the applicable importance factors given in *Table 1.5-2,* ASCE 7-10.

Table 2 3 Importance Factors by Risk Category of Buildings and Other Structures for Snow, Ice, and Earthquake Loads <sup>a</sup>

<b>Risk Category</b>	Wind Importance Factor, $I_w$	Seismic Importance Factor, Ie
Ι	1.00	1.00
Π	1.00	1.00
III	1.00	1.25
IV	1.00	1.50

## Exposing categories<sup>1</sup>

#### Table 2 4 Exposure Categories

Exposure	Description
В	Urban and suburban areas, wooded areas, or other terrain with numerous closely spaced obstructions having the size of single-family dwellings or larger.
C	Open terrain with scattered obstructions having heights generally less than 30ft (9.1 m). This category includes flat open country and grasslands.
D	Flat, unobstructed areas and water surfaces. This category includes smooth mud flats, salt flats, and unbroken ice.

For a site located in the transition zone between exposure categories, the category, resulting in the largest wind forces shall be used.

Table 2 5 Comparison of Exposure between TCVN2737 and ASCE7-10 Standard Codes

	Unobstructed areas	Rural areas	Urban areas
TCVN2737:1995	А	В	С
ASCE7-10	D	С	В

<sup>1</sup>ASCE 7-10, Section 26.7.3.

## Basic wind speed V<sub>50y,3s</sub>

Basic wind speed depends on every national standard. So we need to convert it into the value  $V_{50y,3s}$  in ASCE 7-10 with design 3-second gust wind speeds (m/s) at 10m above ground for Exposure C. In this manual, we will summarize by below table.

Country	Code	T years return period	Gust duration
United States	ASCE 7-10	50	3s
Viet Nam	TCVN 2737:1995	20	3s
Thailand	~ ASCE 7-10	50	1h
Myanmar	MNBC 2016 ~ ASCE 7-10	50	3s
Cambodia	~ ASCE 7-10	50	3s
Indonesia	~ ASCE 7-10	50	3s
Philippines	NSCP 2015 ~ ASCE 7-10	Search PHL map in NSC	P 2015

#### Table 2 6 Wind return period for different codes

## Example

Vietnam Location: converting 20 years return period to 50 years return period by following a formula:

$$\frac{V_T}{V_{50}} = 0.36 + 0.1 \times \ln(12T)$$
 (C26.5-2, ASCE 7-10)

$$\frac{V_{20}}{V_{50}} = 0.36 + 0.1 \times \ln(12T) = 0.36 + 0.1 \times \ln(12 \times 20) = 0.908 \rightarrow V_{50} = \frac{V_{20}}{0.908}$$

Thailand Location: If converting gust duration from 1h to 3s.

See Figure C26.5-1 (ASCE 7-10). Maximum speed average over ts to hourly wind speed

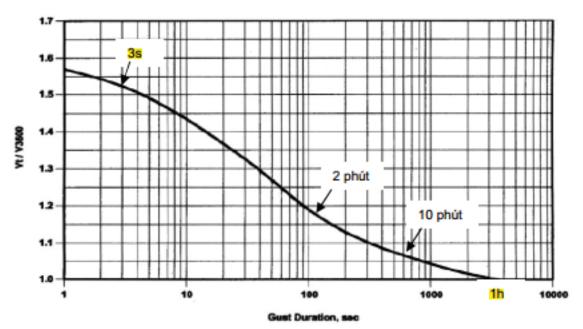


Figure 2.5 Converting gust duration diagram

According to the above diagram, we see

$$\frac{V_{3s}}{V_{1b}} = 1.52 \rightarrow V_{3s} = 1.52 V_{1b}$$

## Design wind speed $V_{Ty,3s}$

Using the Figure 26.5-1Å, 1B, 1C ASCE 7-10 to determine the design wind speed, it depends on risk category of building

#### Table 2 7 Design wind returns period

Risk category of building	T year return period (ASCE 7-10)	Design T year return period	Gust duration
Ι	50	300	3s
II	50	700	3s
III, IV	50	1700	3s

Velocity pressure conversion factor:

$$\frac{V_{Ty,3s}}{V_{50y,3s}} = 0.36 + 0.1 \times \ln(12T)$$

(C26.5-2, ASCE 7-10)

 $\rightarrow$  Design wind speed:  $V_{Ty,3s} = V_{50,y,3s} \times [0.36 + 0.1 \times \ln(12T)]$ 

#### Velocity pressure exposure coefficient Kz

According to Table 27.3-1, ASCE 7-10, Velocity pressure exposure coefficient Kz depends on exposure and height of building above the ground.

Height above ground level, z	Exposure		
(m)	В	С	D
0-4.6	0.57	0.85	1.03
6.1	0.62	0.90	1.08
7.6	0.66	0.94	1.12
9.1	0.70	0.98	1.16
12.2	0.76	1.04	1.22
15.2	0.81	1.09	1.27
18.0	0.85	1.13	1.31
21.3	0.89	1.17	1.34
24.4	0.93	1.21	1.38
27.4	0.96	1.24	1.40
30.5	0.99	1.26	1.43
36.6	1.04	1.31	1.48
42.7	1.09	1.36	1.52
48.8	1.13	1.39	1.55
54.9	1.17	1.43	1.58
61.0	1.20	1.46	1.61
76.2	1.28	1.53	1.68
91.4	1.35	1.59	1.73
106.7	1.41	1.64	1.78
121.9	1.47	1.69	1.82
137.2	1.52	1.73	1.86
152.4	1.56	1.77	1.89

Table 2 8 Velocity pressure exposure coefficients, Kh and Kz (Table 27.3-1, ASCE 7-10)



## NOTE

1. The velocity pressure exposure coefficient Kz may be determined from the following formula:

For  $15 ft. \le z < z_g$ ,  $K_z = 2.01 \left( z/z_g \right)^{2/\alpha}$ For z < 15 ft.,  $K_z = 2.01 \left( 15/z_g \right)^{2/\alpha}$ 

- 2.  $\alpha$  and  $z_g$  are tabulated in Table 26.9.1 ASCE 7-10.
- 3. Linear interpolation for intermediate values of height z is acceptable.
- 4. Exposing categories are defined in Table 2 4.

## Topographic factor Kzt

The topographic factor for the site  $K_{zt}$  is taken to be 1 in most cases. For more detail, see Section 26.8.1 ASCE 7-10.

## Wind directionality factor Kd

Wind directionality factor  $K_d$  depends on type & shape of the building, see table below.

Structure type	Directionality factor Kd
Buildings	
Main wind force resisting system	0.85
Components and cladding	0.85
Arched roofs	0.85
Chimneys, tanks, and similar structures	
Square	0.90
Hexagonal	0.95
Round	0.95
Solid freestanding walls and solid freestanding and attached signs	0.85
Open signs and lattice framework	0.85
Trussed towers	
Triangular, square, rectangular	0.85
All other cross sections	0.95

## Gust effect factor

The gust-effect factor for a rigid building or low-rise building is permitted to be taken as 0.85.

For flexible or dynamically sensitive building or high-rise building, calculate G value according to section 26.9 ASCE 7-10.

## **Enclosure classification**

For the purpose of determining internal pressure coefficients, all buildings shall be classified as enclosed, partially enclosed, or open

Enclosed Building	A building that does not comply with the requirements for open or partially enclosed buildings
Partially Enclosed Building	A building that complies with both of the following conditions: 1. The total area of openings in a wall that receives positive external pressure
	<ul> <li>exceeds the sum of the areas of openings in the balance of the building</li> <li>envelope (walls and roof) by more than 10%.</li> <li>2. The total area of openings in a wall that receives positive external pressure</li> <li>exceeds 4 ft<sup>2</sup> (0.37 m<sup>2</sup>)</li> </ul>
Open Building	A building having each wall at least 80% open. This condition is expressed for each wall by the equation $A_o > 0.8A_g$ , where: $A_o =$ total area of openings in a wall that receives positive external pressure. $A_g =$ the gross area of that wall in which $A_0$ is identified.

## Internal pressure coefficient (GCpi)

Internal pressure coefficients,  $(GC_{pi})$ , shall be determined from *Table 26.11-1 ASCE 7-10* based on building enclosure classifications.

Enclosure classification	GC pi
Open buildings	0.00
Partially Enclosed Buildings	+0.55
	-0.55
Enclosed Buildings	+0.18
	-0.18

Table 2 10 Internal pressure coefficients, GCpi (Table 26.11-1, ASCE 7-10)



### NOTE:

1. Plus and minus signs signify pressures acting toward and away from the internal surfaces, respectively.

2. The values of  $(GC_{pi})$  shall be used with  $q_z$  or  $q_b$  as specified.

3. Two cases shall be considered to determine the critical load requirements for the appropriate condition:

(i) A positive value of  $(GC_{pi})$  applied to all internal surfaces

(ii) A negative value of  $(GC_{pi})$  applied to all internal surfaces

## 2.2.5. Crane load (CR)

Crane loads are determined using the crane data available from the crane manufacturer. Crane data include wheel load, curb weight, crane weight, wheelbase, ends hook approach (used when two cranes operate in one aisle) and minimum vertical and horizontal clearances.

Refer CHAPTER 5. CRANE SYSTEMS DESIGN for more detail.

## 2.2.6. Earthquake

## 2.2.6.1 Seismic design concept

An effective seismic design generally includes:

1. Selecting an overall structural concept, including the layout of a lateral-force-resisting system that is appropriate to the anticipated level of ground shaking. This includes providing a redundant and continuous load path to ensure that a building responds as a unit when subjected to ground motion2. Determining code-prescribed forces and deformations generated by the ground motion, and distributing the forces vertically to the lateral-force-resisting system. The structural system, configuration, and site characteristics are all considered when determining these forces.

3. Analysis of the building of the combined effects of gravity and seismic loads to verify that adequate vertical and lateral strength and stiffness are achieved to satisfy the structural performance and acceptable deformation levels prescribed in the governing building code.

4. Providing details to assure that the structure has the sufficient inelastic deformability to undergo fairly large deformations when subjected to a major earthquake. Appropriately detailed members possess the necessary characteristics to dissipate energy by inelastic deformations.

## 2.2.6.2 Structural Response

If the base of a structure is suddenly moved, as in a seismic event, the upper part of the structure will not respond instantaneously but will lag because of the inertial resistance and flexibility of the structure. The resulting stresses and distortions in the building are the same as if the base of the structure was to remain stationary while timevarying horizontal forces are applied to the upper part of the building. These forces, called inertia forces, are equal to the product of the mass of the structure times acceleration, i.e., F = ma (the mass m is equal to weight divided by the acceleration of gravity, i.e., m = w/g). Because earthquake ground motion is three-dimensional (one vertical and two horizontal), the structure, in general, deforms in a three-dimensional manner. Generally, the inertia forces generated by the horizontal components of ground motion require greater consideration for seismic design since adequate resistance to vertical seismic loads is usually provided by the member capacities required for gravity load design. In the equivalent static procedure, the inertia forces are represented by equivalent static forces.

## 2.2.6.3 Load path

Buildings are generally composed of vertical and horizontal structural elements.

The vertical elements commonly used to transfer lateral forces on the ground are:

- 1) Shear walls;
- 2) Braced frames;
- 3) Moment-resisting frames.

The horizontal elements that distribute lateral forces to the vertical elements are:

- 1) Diaphragms, such as floor and roof slabs;
- 2) Horizontal bracing that transfers large shears from discontinuous walls or braces.

The seismic forces that are proportional to the mass of the building elements are considered to act as their centers of mass. All of the inertia forces originating from the masses on and off the structure must be transmitted to the lateral-force-resisting elements, and then to the base of the structure and into the ground.

A complete load path is a basic requirement for all buildings. There must be a complete lateral-forceresisting system that forms a continuous load path between the foundation, all diaphragm levels, and all portions of the building for proper seismic performance. The general load path is as follows. Seismic forces originating throughout the building, mostly in the heavier mass elements such as diaphragms, are delivered through connections to horizontal diaphragms; the diaphragms distribute these forces to vertical force-resisting elements such as shear walls and frames; the vertical elements transfer the forces into the foundation; and the foundation transfers the forces into the supporting soil.

If there is a discontinuity in the load path, the building is unable to resist seismic forces regardless of the strength of the elements. Interconnecting the elements needed to complete the load path is necessary to achieve good seismic performance. Examples of gaps in the load path would include a shear wall that does not extend to the foundation, a missing shear transfer connection between a diaphragm and vertical elements, a discontinuous chord at a diaphragm's notch, or a reentrant corner, or a missing collector.

A good way to remember this important design strategy is to ask yourself the question, "How does the inertia load get from here (meaning the point at which it is generated) to there (meaning the shear base of the structure, typically the foundations)?"

## 2.2.6.4 The design base shear according to UBC 97

The total design base shear in a given direction shall be determined following section 1630.2 UBC 97:

$$V = \frac{C_v \times I}{R \times T} \times W \quad (kN)$$

Besides, V value must be satisfied below equation:

$$0.11C_a \times I \times W \le V = \frac{C_v \times I}{R \times T} \times W \le \frac{2.5C_a \times I}{R} \times W$$

For seismic zone 4, the total design base shear shall also not be less than the following:

$$V \ge \frac{0.8Z \times N_v \times I}{R} \times W$$

### 2.2.6.5 Seismic coefficient (Cv, Ca)

The seismic coefficient is determined by Table 2 11 and Table 2 12, it depends on soil profile type (Table 2 13) and seismic zone factor Z (Table 2 15).

Soil profile type	Seismic zone factor, Z				
	Z=0.075	Z=0.15	Z=0.2	Z=0.3	Z=0.4
S <sub>A</sub>	0.06	0.12	0.16	0.24	0.32N <sub>v</sub>
S <sub>B</sub>	0.08	0.15	0.20	0.30	0.40N <sub>v</sub>
S <sub>c</sub>	0.13	0.25	0.32	0.45	0.56N <sub>v</sub>
S <sub>D</sub>	0.18	0.32	0.40	0.54	0.64N <sub>v</sub>
S <sub>E</sub>	0.26	0.50	0.64	0.84	0.96Nv
S <sub>F</sub>	Refer to site-specific Geotechnical investigation and dynamic site response				
	analysis to determine $C_v$				

#### Table 2 11 Seismic coefficients, C<sub>v</sub> (Table 16-R, UBC 97)

### Table 2 12 Seismic coefficients, C<sub>a</sub> (Table 16-Q, UBC 97)

Soil profile type	Seismic zone factor, Z				
	Z=0.075	Z=0.15	Z=0.2	Z=0.3	Z=0.4
S <sub>A</sub>	0.06	0.12	0.16	0.24	0.32N <sub>v</sub>
S <sub>B</sub>	0.08	0.15	0.20	0.30	0.40N <sub>v</sub>
S <sub>C</sub>	0.09	0.18	0.24	0.33	0.40N <sub>v</sub>
S <sub>D</sub>	0.12	0.22	0.28	0.36	0.44N <sub>v</sub>
S <sub>E</sub>	0.19	0.30	0.34	0.36	0.36Nv
S <sub>F</sub>	Refer to site-specific Geotechnical investigation and dynamic site response				
	analysis to determine $C_v$				

The seismic coefficients  $C_v$  and  $C_a$ , given in *Tables 16-R and 16-Q UBC 97*, are site-dependent ground motion coefficients that define the seismic response throughout the spectral range. They are measures of expected ground acceleration at a site.

For a given earthquake, a building on soft soil types such as  $S_C$  or  $S_D$  experiences a greater force than if the same building were located on a rock, type  $S_A$  or  $S_B$ . This is addressed in the UBC through the  $C_a$  and  $C_v$  coefficients, which are calibrated to soil type  $S_B$  with a value of unity. Instead of a single coefficient, two coefficients,  $C_a$  and  $C_v$  are used to distinguish the response characteristics of short-period and long-period buildings. Long period buildings are more affected by soft soils than short-period buildings.

In SAP software, a designer needs to choose Soil profile type & Seismic zone factor, the program will calculate  $C_a$ ,  $C_v$  automatically according to UBC 97.

1997 UBC Seismic Load Pattern	
Load Direction and Diaphragm Eccentricity         Image: Global X Direction         Image: Global Y Direction         Ecc. Ratio (All Diaph.)         Override Diaph. Eccen.	Seismic Coefficients         Per Code       User Defined         Soil Profile Type       SC         Seismic Zone Factor       0.40         User Defined Ca       0.4         User Defined Cv       0.56
Time Period         O Method A       Ct (ft) =         Image: Program Calc       Ct (ft) =         Image: User Defined       T =         Image: Lateral Load Elevation Range         Image: Program Calculated         Image: User Specified         Image: Max Z         Min Z	Near Source Factor Per Code User Defined Seismic Source Type B Dist. to Source (km) 15 User Defined Na 1 User Defined Nv 1
Factors Overstrength Factor, R 4.5	Other Factors Importance Factor, I 1.
OK	Cancel

## 2.2.6.6 Soil profile types

Soil profile type	Soil profile name/Generic description
S <sub>A</sub>	Hard rock
S <sub>B</sub>	Rock
S <sub>C</sub>	Very dense soil and soft rock
S <sub>D</sub>	Stiff soil profile
S <sub>E</sub>	Soft soil profile
$S_{\rm F}$	Soil requiring site – specific evaluation. See section 1629.3.1 UBC 97



## NOTE

Use Soil Types SD if there is not any requirement from customers.

## Table 2 14 Soil profile types for different codes

Country	Code Soil profile types						
United States	UBC 97 (Table 16-J)	S <sub>A</sub>	S <sub>B</sub>	S <sub>C</sub>	$S_D$	$S_E$	S <sub>F</sub>
Vietnam	TCVN 9386:2012 (Table 3.1)	-	А	В	С	D	E,S1,S2
Thailand	UBC 97 (Table 16-J)	S <sub>A</sub>	S <sub>B</sub>	S <sub>C</sub>	$S_D$	$S_E$	S <sub>F</sub>
Myanmar	MNBC 2016 (Table 3.4.2)	А	В	С	D	E	F
Cambodia	UBC 97 (Table 16-J)	S <sub>A</sub>	$S_B$	$S_{C}$	$S_D$	$S_E$	S <sub>F</sub>
Indonesia	UBC 97 (Table 16-J)	S <sub>A</sub>	$S_B$	S <sub>C</sub>	$S_D$	$S_E$	S <sub>F</sub>
Philippines	NSCP 2015	Sa	Sв	Sc	Sd	Se	Sf

## 2.2.6.7 Seismic zone & Seismic zone factor (Z)

Table 2 15 Seismic zone factor, Z (Table 16-I, UBC 97)

Zone	1	2A	2B	3	4
Z	0.075	0.15	0.20	0.30	0.40



## Note: The zone shall be determined from the seismic zone map in Figure 16-2 UBC 97

The map accounts for the geographical variations in the expected levels of earthquake ground shaking and gives the estimated peak horizontal acceleration on a rock having a 10% chance of being exceeded in a 50-year period (or 500-year return period).

The value of the seismic zone coefficient Z can be considered the peak ground acceleration of the percentage of gravity.

It means, the peak ground acceleration:  $a_g = Z \times g$ 

For example, Z = 0.4 indicates a peak ground acceleration of 0.4g equal to 40% of gravity. For the buildings are not located in the United States, refer to *Appendix Chapter 16 UBC 97* to determine seismic zone.

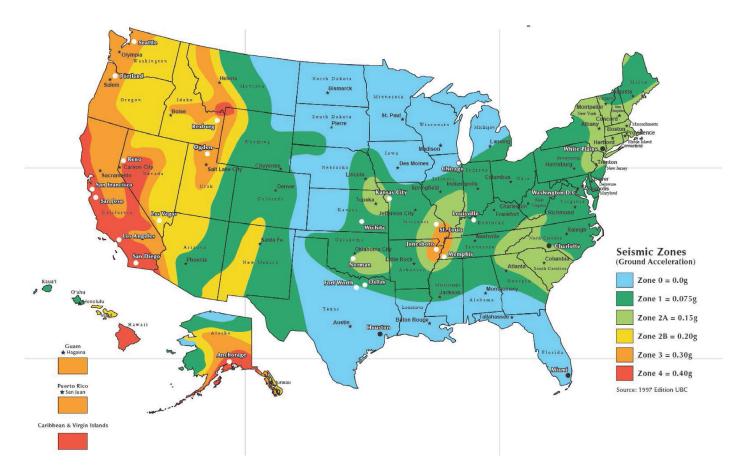


Figure 2.6 The United States Seismic zones map with 500-year return period

Country	Code	Seismic zones & peak ground acceleration						
United States	UBC 97	1	2A	2B	3	4		
	(Table 16-I)							
	a <sub>g</sub> =	0.075g	0.15g	0.2g	0.3g	0.4g		
Vietnam	TCVN 9386:2012	1	2A	2B	-	-		
	(Appendix G, H)							
Thailand	~ UBC 97	1	2A	2B	3	-		
Myanmar	MNBC 2016	Ι	II	III	IV	$\vee$		
	(Fig. 3.4.1.5)							
Cambodia	~ UBC 97	-	-	-	-	-		
Indonesia	~ UBC 97	1	2A	2B	3	4		
Philippines	NSCP 2015	-	-	2B	-	4		
	(Table 208-3)							

## Vietnam

Use Appendix H - TCVN 9386:2012 to determine ground acceleration for every area in Vietnam.

In this standard, the peak ground acceleration was surveyed considered on rock, 500-year return period. Thailand

According to Section 1653 UBC 97, we have a seismic zone for some areas in Thailand.

Table 2 17 Seismic zone in Thailand



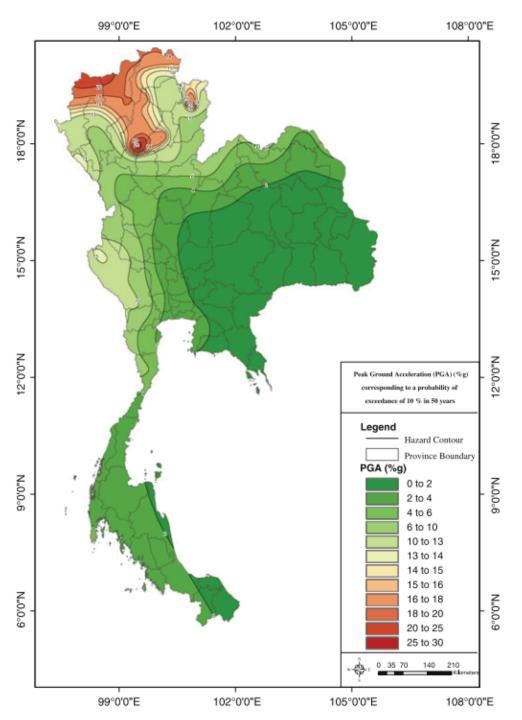


Fig. 8 Thailand hazard maps for PGA corresponding to a probability of exceedance of 10% in 50 years Figure 2.7 Thailand Seismic zones map with 500-year return period

### Myanmar

The seismic zone of Myanmar is divided into I, II, III, IV, V corresponding to seismic zone 2B, 3, 4 of UBC 97.

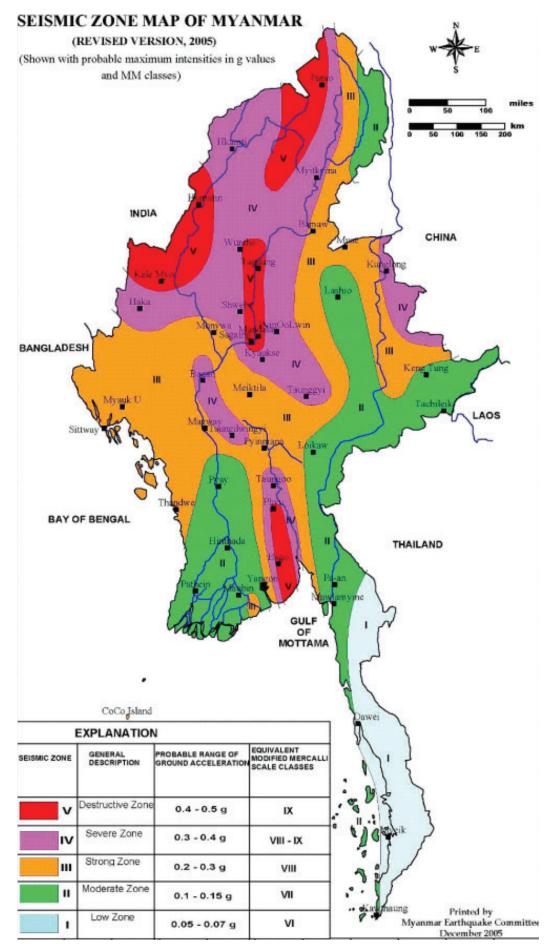


Figure 2.8 Myanmar Seismic zones map with 500-year return period (Figure 3.4.1.5 MNBC 2016)

## Philippines

The Philippines archipelago is divided into 2 seismic zones only. Zone 2 covers the provinces of Palawan (except Busuanga), Sulu and Tawi-Tawi while the rest of the country is under zone 4.

Table 2 18 Seismic zone factor, Z in Philippines (Table 208-3, NSCP 2015)

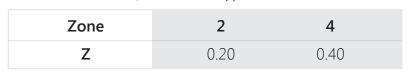




Figure 2.9 Philippines Seismic zones map with 500-year return period (Figure 208-1 NSCP 2015)

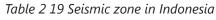
## Cambodia

Use UBC 97 to design a building in Cambodia. Almost buildings no need to design earthquake.

## Indonesia

According to Section 1653 UBC 97, we have a seismic zone for some areas in Indonesia.

Area	Zone
Bandung	4
Jakarta	4
Medan	3
Surabaya	4



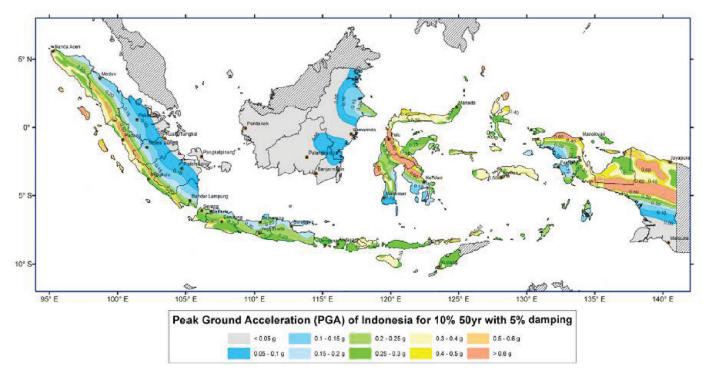


Figure 2.10 Indonesia Seismic zones map with 500-year return period

## 2.2.6.8 Important factor (I)

Occupancy category	Occupancy or functions of structure	Seismic importance factor, I	Seismic importance factor, Ip1	Wind importance factor, Iw				
1. Essential facilities 2	Group I, Division 1 Occupancies having surgery and emergency treatment areas Fire and police stations Garages and shelters for emergency vehicles and emergency aircraft Structures and shelters in emergency- preparedness centers Aviation control towers Structures and equipment in government communication centers and other facilities required for emergency response Standby power-generating equipment for Category 1 facility Tanks or other structures containing housing or supporting water or other fire-suppression material or equipment required for the protection of Category 1, 2 or 3 structures	1.25	1.50	1.15				
2. Hazardous facilities	Group H, Divisions 1, 2, 6 and 7 Occupancies and structures therein housing or supporting toxic or explosive chemicals or substances Non-building structures housing, supporting or containing quantities of toxic or explosive substances that, if contained within a building, would cause that building to be classified as a Group H, Division 1, 2 or 7 Occupancy	1.25	1.50	1.15				
3. Special occupancy structures3	Group A, Divisions 1, 2 and 2.1 Occupancies Buildings housing Group E, Divisions 1 and 3 Occupancies with a capacity greater than 300 students Buildings housing Group B Occupancies used for college or adult education with a capacity greater than 500 students Group I, Divisions 1 and 2 Occupancies with 50 or more residents incapacitated patients, but not included in Category 1 Group I, Division 3 Occupancies All structures with an occupancy greater than 5,000 persons Structures and equipment in power-generating stations, and other public utility facilities not included in Category 1 or Category 2 above, and required for continued operation	1.00	1.00	1.00				
4. Standard occupancy structures3	All structures housing occupancies or having functions not listed in Category 1, 2 or 3 and Group U Occupancy towers	1.00	1.00	1.00				
5. Miscellaneous structures	Group U Occupancies except for towers	1.00	1.00	1.00				

Table 2 20 Occupancy category (Table 16-K, UBC 97)

For steel building, we usually use important factor I = 1.

orad Direction and Diaphragm Eccentricity     Global X Direction     Global Y Direction     Ecc. Ratio (All Diaph.)	Seismic Coefficients Per Code C User Defined Soil Profile Type SC Seismic Zone Factor 0.40 User Defined Ca 0.4
Override Diaph. Eccen.     Override       Time Period     C       C     Method A     Ct (ft) =       Image: C     Program Calc     Ct (ft) =       Image: C     User Defined     T =         Image: C     Program Calculated       Image: C     Program Calculated       Image: C     Image: C         Image: C     Image: C	User Defined Cv 0.56 Near Source Factor Per Code C User Defined Seismic Source Type B Dist. to Source (km) 15 User Defined Na 1 User Defined Nv 1
Overstrength Factor, R 8.5	Other Factors Importance Factor, I 1.

## 2.2.6.9 The total seismic dead load (W)

According to section 1630.1.1 UBC 97, we can determine total seismic dead load W through mass source in SAP.

💢 Mass Source Data		_		×
Mass Source Name	MSSSRC1			
Mass Source				
Mass Multipliers for Load P Load Pattern AL DL LL CR1 AL FL6	atterns Multiplier 1. 1. 0.25 0.25 1. 0.25 1. 0.25		Add Modify Delete	
OK	Can	cel		

 $W = Mass \ source = DL + AL + 0.25 \ (LL + FL + CR) \ (kN)$ 



### NOTE

Load Pattern FL is fully applied floor load. Each Crane will contribute one Crane Load Pattern.

### 2.2.6.10 Numerical coefficient representative of the inherent over strength and global ductility capacity of lateral force-resisting systems (R)

The coefficient R shown in *Table 16-N UBC 97* is a measure of ductility and over strength of a structural system, based primarily on performance of similar systems in past earthquakes.

A higher value of R has the effect of reducing the design base shear. For example, for a steel special moment-resisting frame, the factor has a value of 8.5, whereas for ordinary moment-resisting frame, the value is 4.5. This reflects the fact that a special moment-resisting frame performs better during an earthquake.

Table 2 21 Structural systems (Table 16-N UBC 97)

Basic structural system	Lateral force resisting system description	R	$\Omega_{_0}$	Height limit for seismic zones 3 and 4 (feet) x 304.8 for mm
1. Bearing wall	1. Light-framed walls with shear panels			
system	a. Wood structural panel walls for structures			65
	three stories or less	5.5	2.8	65
	<ul><li>b. All other light-framed walls</li><li>2. Shear walls</li></ul>	4.5	2.8	65
	a. Concrete	4.5	2.8	160
	b. Masonry	4.5	2.8	160
	3. Light steel-framed bearing walls with tension- only bracing	2.8	2.2	65
	4. Braced frames where bracing carries gravity	4.4	2.2	160
	load	2.8	2.2	-
	a. Steel	2.8	2.2	65
	b. Concrete c. Heavy timber	2.0	2.2	03
2. Building frame system	<ol> <li>Steel eccentrically braced frame (EBF)</li> <li>Light-framed walls with shear panels</li> <li>Wood structural panel walls for structures</li> </ol>	7.0	2.8	240
	three stories or less	6.5	2.8	65
	<ul><li>b. All other light-framed walls</li><li>3. Shear walls</li></ul>	5.0	2.8	65
	a. Concrete	5.5	2.8	240
	<ul><li>b. Masonry</li><li>4. Ordinary braced frames</li></ul>	5.5	2.8	160
	a. Steel	5.6	2.2	160
	b. Concrete	5.6	2.2	-
	c. Heavy timber 5. Special concentrically braced frames	5.6	2.2	65
	a. Steel	6.4	2.2	240
3. Moment-	1. Special moment-resisting frame (SMRF)			
resisting frame	a. Steel	8.5	2.8	N.L.
system	b. Concrete	8.5	2.8	N.L.
	2. Masonry moment-resisting wall frame	6.5	2.8	160
	(MMRWF)	5.5	2.8	
	3. Concrete intermediate moment-resisting frame (IMRF)			
	4. Ordinary moment-resisting frame (OMRF)			
	a. Steel	4.5	2.8	160
	b. Concrete	3.5	2.8	
	5. Special truss moment frames of steel (STMF)	6.5	2.8	240
4. Dual systems	1. Shear walls a. Concrete with SMRF	8.5	2.8	N.L.

Basic structural system	Lateral force resisting system description	R	$\Omega_{_0}$	Height limit for seismic zones 3 and 4 (feet) x 304.8 for mm
	<ul> <li>b. Concrete with steel OMRF</li> <li>c. Concrete with concrete IMRF</li> <li>d. Masonry with SMRF</li> <li>e. Masonry with steel OMRF</li> <li>f. Masonry with concrete IMRF</li> <li>g. Masonry with masonry MMRWF</li> <li>2. Steel EBF</li> <li>a. With steel SMRF</li> <li>b. With steel OMRF</li> <li>3. Ordinary braced frames</li> <li>a. Steel with steel SMRF</li> <li>b. Steel with steel OMRF</li> <li>c. Concrete with concrete SMRF</li> <li>d. Concrete with concrete IMRF</li> <li>4. Special concentrically braced frames</li> <li>a. Steel with steel SMRF</li> <li>b. Steel with steel SMRF</li> <li>b. Steel with steel SMRF</li> </ul>	4.2 6.5 5.5 4.2 4.2 6.0 8.5 4.2 6.5 4.2 6.5 4.2 7.5 4.2	2.8 2.8 2.8 2.8 2.8 2.8 2.8 2.8 2.8 2.8	160 160 160 160 - 160 N.L. 160 N.L. 160 - - - N.L. 160
5. Cantilevered column building systems	1. Cantilevered column elements	2.2	2.0	35
6. Shear wall- frame interaction systems	1. Concrete	5.5	2.8	160
7. Undefined systems	See Sections 1629.6.7 and 1629.9.2 UBC 97	-	-	-

## Table 2 22 R and factors for nonbuilding structures (Table 16-P UBC 97)

Structure type	R	$\mathbf{\Omega}_{_0}$
1. Vessels, including tanks and pressurized spheres, on braced or unbraced legs.	2.2	2.0
2. Cast-in-place concrete silos and chimneys having walls continuous to the foundations.	3.6	2.0
3. Distributed mass cantilever structures such as stacks, chimneys, silos and skirt-supported	2.9	2.0
vertical vessels.		
4. Trussed towers (freestanding or guyed), guyed stacks and chimneys.	2.9	2.0
5. Cantilevered column-type structures.	2.2	2.0
6. Cooling towers.	3.6	2.0
7. Bins and hoppers on braced or unbraced legs.	2.9	2.0
8. Storage racks	3.6	2.0
9. Signs and billboards.	3.6	2.0
10. Amusement structures and monuments.	2.2	2.0
11. All other self-supporting structures not otherwise covered.	2.9	2.0

For steel building, we usually use R=4.5.

1997 UBC Seismic Load Pattern	
Load Direction and Diaphragm Eccentricity         Image: Global X Direction         Image: Global Y Direction         Ecc. Ratio (All Diaph.)         Override Diaph. Eccen.	Seismic Coefficients         Per Code       User Defined         Soil Profile Type       SC         Seismic Zone Factor       0.40         User Defined Ca       0.4         User Defined Cv       0.56
Time Period         O Method A       Ct (ft) =         Image: Program Calc       Ct (ft) =         Image: User Defined       T =         Image: Lateral Load Elevation Range         Image: Program Calculated         Image: Other Specified         Image: Max Z         Min Z	Near Source Factor         Per Code       User Defined         Seismic Source Type       B         Dist. to Source (km)       15         User Defined Na       1         User Defined Nv       1
Factors Overstrength Factor, R 4.5	Other Factors Importance Factor, I 1.

### 2.2.6.11 Structure period (T)

Structure period or elastic fundamental period of vibration, of the structure in the direction under consideration.

#### Method A

For all buildings, the value T may be determined from section 1630.2.2 UBC 97

$$T = T_{\mathcal{A}} = C_t \left( b_n \right)^{3/4}$$
(s)

Where:

-  $C_t$  = the structural coefficient depends on the material of the frame, is always input in English unit in SAP software.

- $C_t = 0.035(0.0853)$  for steel moment-resisting frames.
- $C_t = 0.030(0.0731)$  for reinforced concrete moment-resisting frames and eccentrically braced frames.
- $C_t = 0.020(0.0488)$  for all other buildings.

 $-h_n$  = height above the base to level that is uppermost in the main portion of the structure. In SAP software, it's measured from the elevation of the specified bottom story/minimum elevation level to the (top of the) specified top story/maximum elevation level. (m)

The designer can modify  $h_n$  value by choosing an option "User specified" in Lateral load elevation range & fill in Min Z/ Max Z.

1997 UBC Seismic Load Pattern	
- Load Direction and Diaphragm Eccentricity-	Seismic Coefficients
<ul> <li>Global × Direction</li> </ul>	Per Code O User Defined
C Global Y Direction	Soil Profile Type SC 💌
Ecc. Ratio (All Diaph.) 0.05	Seismic Zone Factor 0.40 💌
	User Defined Ca
Override Diaph. Eccen. Override	User Defined Cv 0.56
Time Period	Near Source Factor
C Method A Ct (ft) =	Per Code     O     User Defined
Program Calc Ct (ft) = 0.035	Seismic Source Type B
C User Defined T =	Dist. to Source (km) 15
Lateral Load Elevation Range	User Defined Na 1
O Program Calculated	User Defined Nv 1
User Specified     Reset Defaults	
Max Z 11.4	
Min Z 0.	
Factors	Other Factors
Overstrength Factor, R 4.5	Importance Factor, I 1.
<u> </u>	Cancel

**DESIGN GUIDELINES** 

#### Method B

The fundamental period T may be calculated using the structural properties and deformation characteristics of the resisting elements in a properly substantiated analysis. The analysis shall be in accordance with the requirements of *Section 1630.1.2 UBC 97.* The fundamental period T may be computed by using the following formula:

$$T_{B} = 2\pi \sqrt{\frac{m}{k}} = 2\pi \sqrt{\left(\sum_{i=1}^{n} w_{i} \delta_{i}^{2}\right)} \div \left(g \sum_{i=1}^{n} f_{i} \delta_{i}\right)$$
(s)

Where:

-  $f_i = lateral$  force at level i (kN)

-  $\delta_i$  = horizontal displacement at Level i relative to the base due to applied lateral force f (m)

- g = acceleration due to gravity,  $g = 9.81 (m/s^2)$ 

-  $w_i$  = that portion of the total seismic dead load assigned to level i. (kN)

SAP software use this method to calculate T if designer choose "Program calc" in Time period.

Global X Direction	Per Code     User Defined
O Global Y Direction	Soil Profile Type SC 💌
Ecc. Ratio (All Diaph.) 0.05	Seismic Zone Factor 0.40 💌
Override Diaph. Eccen. Override	User Defined Ca 0.4 User Defined Cv 0.56
Time Period O Method A Ct (ft) =	Near Source Factor     Per Code     O User Defined
Program Calc Ct (ft) = 0.035	Seismic Source Type B
O User Defined T =	Dist. to Source (km) 15
Lateral Load Elevation Range	User Defined Na  1
Program Calculated     User Specified     Max Z     Min Z	User Defined Nv  1
Factors Overstrength Factor, R 4.5	Other Factors Importance Factor, I 1.

- If the seismic zone is zone 4 then:

• If  $T_B \leq 1.3T_A \rightarrow T = T_B$ 

• If 
$$T_B > 1.3T_A \rightarrow T = T_A$$

- If the seismic zone is zone 1, 2A, 2B, 3 then:

- If  $T_B > 1.4T_A \rightarrow T = T_A$
- If  $T_B > 1.4T_A \rightarrow T = T_A$

#### 2.2.6.12 Near-source factor (Na, Nv)

For seismic zone 4 (Z=0.4), we need near-source factor  $N_a$ ,  $N_v$  to determine seismic coefficient  $C_a$ ,  $C_v$  corresponding. They depend on seismic source type and the distance from the site to seismic source.

Seismic source type	Closest distance to known seismic source		eismic source
	≤ 2 km	5 km	≥ 10 km
A	1.5	1.2	1.0
В	1.3	1.0	1.0
С	1.0	1.0	1.0

Table 2 23 Near-source factor Na (Table 16-S UBC 97)

#### Table 2 24 Near-source factor NV (Table 16-T UBC 97)

Seismic source type	Closest distance to known seismic source			
	≤ 2 km	5 km	10 km	≥ 15 km
А	2.0	1.6	1.2	1.0
В	1.6	1.2	1.0	1.0
С	1.0	1.0	1.0	1.0

The purpose of  $N_a$  and  $N_v$  is to increase the soil-modified ground motion parameters,  $C_a$  and  $C_v$ , when there are active faults capable of generating large-magnitude earthquakes within 15 km of a seismic zone 4 site. So that

#### $N_a, N_v \ge 1$ .

In SAP software, a designer needs to choose "Seismic source type" & "Dist. to source", the program will be calculated  $N_a$ ,  $N_v$  automatically according to UBC 97.

1997 UBC Seismic Load Pattern	
Load Direction and Diaphragm Eccentricity Global X Direction Global Y Direction Ecc. Ratio (All Diaph.) Override Diaph. Eccen. Override	Seismic Coefficients         Per Code       User Defined         Soil Profile Type       SC         Seismic Zone Factor       0.40         User Defined Ca       0.4         User Defined Cv       0.56
Time Period         O Method A       Ct (ft) =         Image: Program Calc       Ct (ft) =         Image: User Defined       T =         Image: Lateral Load Elevation Range       Image: Program Calculated         Image: Image: Program Calculated       Image: Program Calculated         Image: User Specified       Reset Defaults	Near Source Factor         Per Code       User Defined         Seismic Source Type       B         Dist. to Source (km)       15         User Defined Na       1         User Defined Nv       1
Max Z Min Z Factors Overstrength Factor, R OK	Other Factors Importance Factor, I 1.

#### 2.2.6.13 Seismic source types

Seismic	Seismic source description	Seismic source definition		
source type		Maximum moment magnitude, M	Slip rate, SR (mm/ year)	
A	Faults that are capable of producing large magnitude events and that have a high rate of seismic activity	$M \ge 7.0$	$SR \ge 5$	
В	All faults other than types A and C	$M \ge 7.0$ $M < 7.0$ $M \ge 6.5$	SR < 5 SR > 2 SR < 2	
С	Faults that are not capable of producing large magnitude earthquakes and that have a relatively low rate of seismic activity	M < 6.5	<i>S</i> R ≤ 2	

Table 2 25 Seismic source type (Table 16-U UBC 97)

The seismic source types labeled A, B, or C (*Table 16-U UBC 97*) is used to identify earthquake potential and activity of faults in the immediate vicinity of the structure.

They are defined in terms of the slip rate of the fault and the maximum magnitude of earthquake that may be generated at the fault. The highest seismic risk is posed by seismic source type A, which is defined by a maximum moment magnitude of 7.0 or greater and a slip rate of 5 mm/year or greater.

Moment magnitude (M) was introduced in 1979 by Hanks and Kanamori and has since become the most commonly used method of describing the size of a microseism. Moment magnitude measures the size of events in terms of how much energy is released.

The slip rate is how fast the two sides of a fault are slipping relative to one another, as determined from geodetic measurements, from offset man-made structures, or from offset geologic features whose age can be estimated. It is measured parallel to the predominant slip direction or estimated from the vertical or horizontal offset of geologic markers.

## 2.3. Load Combinations

Based on design method (LRFD or ASD), ASCE 7-10 define load combination coefficients for each method. BMB&A uses allowable stress design (ASD) method, see *section 2.4.1 ASCE 7-10*.

## 2.3.1. For Frame Structure

- 1. Dead Load (DL)
- 2. Dead Load (DL) + Live Load (LL)
- 3. Dead Load (DL) + Live Load (Floor/Crane)

4. Dead Load (DL) + 0.75 Live Load (LL) + 0.75 Live Load (Floor/Crane)

5. Dead Load (DL) + [0.6 Wind Load (WL) or 0.7 Earthquake load (EL)]

6a. Dead Load (DL) + 0.75 Live Load (LL) + 0.75 Live Load (Floor/Crane) + 0.45 Wind Load (WL)

6b. Dead Load (DL) + 0.75 Live Load (LL) + 0.525 Earthquake load (EL)

7. 0.6 Dead Load (DL) + 0.6 Wind Load (WL)

8. 0.6 Dead Load (DL) + 0.7 Earthquake load (EL)

## 2.3.2. For Cold-Formed Section

1. Dead Load (DL) + Roof Live Load (LLr)

2. Dead Load (DL) + 0.6 Wind Load (WL)

3. 0.6 Dead Load (DL) + 0.6 Wind Load (WL)

All of these are basic combination. For more loads (snow load, rain load, flood load..., etc.) see section 2.4 ASCE 7-10.

## 2.4. Serviceability Consideration

Table 2 26 Deflection Limitations

CONTRUCTION	LOAD & D	EFORMALOAD	REMARK
1/ Rafter (Vertical deflection) <sup>2</sup>	LL	DL + LL	
a/ Supporting plaster ceiling	L/360	L/240	
b/ Supporting non-plaster ceiling	L/240	L/180	
c/ Not supporting ceiling	L/180	L/120	
2/ Column (Horizontal deflection) <sup>3</sup>	DL + WL 10 y	vr. H/60	10 Yr. Wind
II/ Purlin1	DL + LL	DL + WL 10 yr.	10 Yr. Wind
1/ Purlin (Vertical deflection)	L/150	L/150	
2/ Girt (Horizontal deflection)	L/120	L/120	
<b>III/</b> Floor members (Vertical deflection)	LL	DL + LL	
Beam, Joist, Decking, Checker Plate	L/360	L/240	
IV/ Top Running Cranes <sup>4</sup>	DL		
1/ Runway Beam (Vertical deflection)	L		
2/ Runway Beam (Horizontal deflection)	L	./400	
V/ Crane Bracket (Horizontal deflection) <sup>3</sup>	DL + CR	or WL 10 yr.	Crane lateral
1/ Cab or Radio - Operator cranes	H/240 c	or ≤ 5.08cm	or 10 Yr.
2/ Pendant - Operator cranes	F	1/100	Wind

<sup>2</sup> International Building Code - IBC 2012, Table 1604.3, page 335.

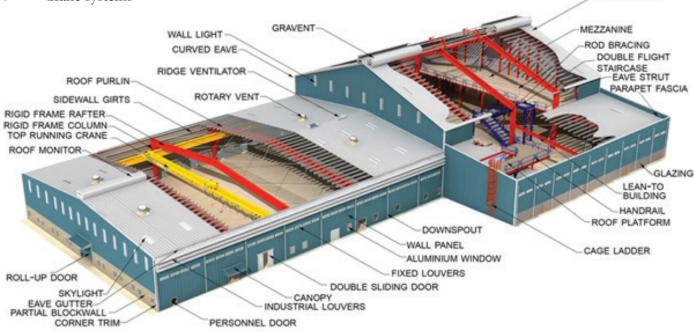
<sup>3</sup> Metal Building Systems Manual 2012, Table 3.3, page 331.

<sup>4</sup> Metal Building Systems Manual 2012, Table 3.5, page 333.

## **CHAPTER 3. PRE-ENGINEERED BUILDING (PEB) SYSTEM**

Planning of the pre-engineered buildings (low rise metal buildings) and arranging different building components is a very important step for the designer before proceeding with the design of each component. The Following building configurations are significantly affecting the building Stability and Cost:

- 1. Main Frame configuration (orientation, type, roof slope, eave height)
- 2. Roof purlins spacing
- 3. Wall girts (connection & spacing)
- 4. End wall system
- 5. Expansion joints
- 6. Bay spacing
- 7. Bracing systems arrangement
- 8. Mezzanine floor beams/columns (orientation & spacing)
- 9. Crane systems



ROOF PANEL

Figure 3.1 Pre-engineered building system

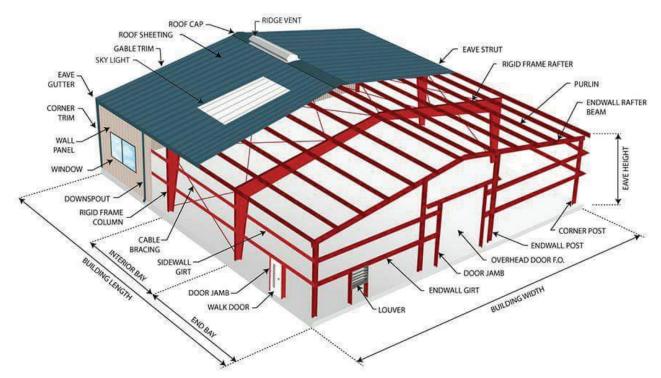


Figure 3.2 Pre-engineered building system

## **3.1. Main Frame Configuration**

Main frame is the basic supporting component in the PEB systems; main frames provide the vertical support for the whole building plus providing the lateral stability for the building in its direction while lateral stability in the other direction is usually achieved by a bracing system.

The width of the building is defined as the out-to-out dimensions of girts/eave struts and these extents define the sidewall steel lines. Eave height is the height measured from bottom of the column base plate to top of the eave strut. Rigid frame members are tapered using built-up sections following the shape of the bending moment diagram. Columns with fixed base are straight. Also the interior columns are always maintained straight.

### 3.1.1. Main frame orientation

Building should be oriented in such a way that the length is greater than the width. This will result in more number of lighter frames rather than less number of heavy frames. This also will reduce the wind bracing forces results in lighter bracing systems.

## 3.1.2. Main frame types

### 3.1.2.1 Clear span

Clear Span rigid frames are single gable frames and offer full-width clear space inside the building without interior columns. This type of frame is extensively used anywhere an unobstructed working area is desired in diverse applications such as auditoriums, gymnasiums, aircraft hangars, showrooms and recreation facilities.

The deepest part of the frame is the knee, the joint between the rafter and the column, which is generally designed as horizontal knee connection. An alternate design of knee joint is as vertical knee connection that is employed for flush side-wall construction. Clear Span rigid frames are appropriate and economical when:

i) Frame width is in the range 24m-30m.

ii) Headroom at the exterior walls is not critical.

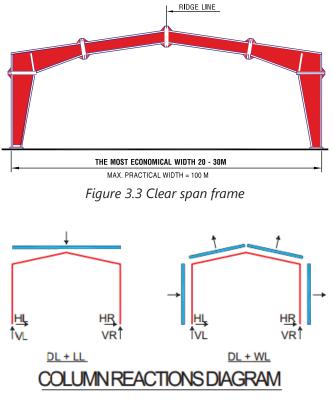


Figure 3.4 Column reaction of clear span frame

#### 3.1.2.2 Multi – span

When clear space inside the building is not the crucial requirement then Multi-Span rigid frames offer greater economy and theoretically unlimited building size. Buildings wider than around 90m experience a build-up of temperature stresses and require temperature load analysis and design. Multi-span rigid frames have straight interior columns, generally hot-rolled tube sections pin connected at the top with the rafter. When lateral sway is critical, the interior columns may be moment connected at the top with the rafter, and in such a situation built-up straight columns are more viable than hot-rolled tube columns.

Multi-Span rigid frame with an interior column located at ridge requires the rafter at ridge to have a horizontal bottom flange in order to accommodate horizontal cap plate. Multi-Span rigid frame is the most economical solution for wider buildings (width > 24m) and is used for buildings such as warehouses, distribution centers and factories. The most economical modular width in multi-span buildings is in the range 18m-24m.

The disadvantages of such a framing system include:

- The susceptibility to differential settlement of column supports.
- Locations of the interior columns are difficult to change in future.
- Longer un-braced interior columns especially for wider buildings.
- Horizontal sway may be critical and governing the design in case of internal columns pined with rafter.

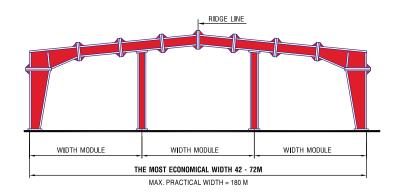


Figure 3.5 Multi - span frame

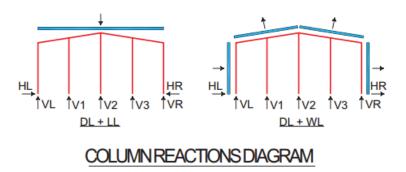


Figure 3.6 Column reaction of multi - span frame

### 3.1.2.3 Lean – to

Lean-To is not a self-contained and stable framing system rather an add-on to the existing building with a single slope. This type of frame achieves stability when it is connected to an existing rigid framing. Usually column rafter connection at knee is pinned type, which results in lighter columns. In general, columns and rafters are straight except that rafters are tapered for larger widths (> 12m). For clear widths larger than 18m, tapered columns with moment resisting connections at the knee are more economical. Lean-To framing is typically used for building additions, equipment rooms and storage.

For larger widths "Multi-Span-Lean-To" framing can be adopted with exterior column tapered and moment connected at the knee.

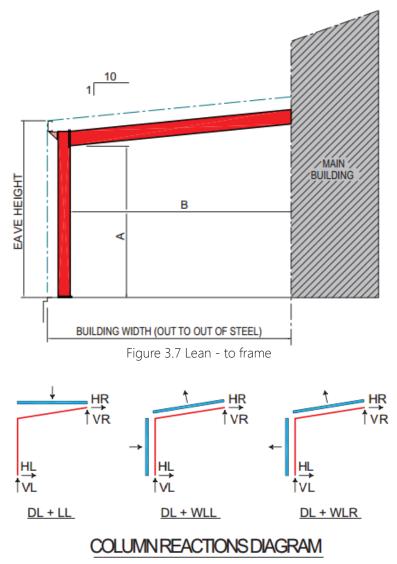


Figure 3.8 Column reaction of lean - to frame

### 3.1.2.4 Mono-slope

Mono-slope or single-slope framing system is an alternative to gable type of frame that may be either Clear Span or multi-span. Mono-Slope configuration results in more expensive framing than the gable type. Mono-slope framing system is frequently adopted where:

- Rainwater needs to be drained away from the parking areas or from the adjacent buildings
- Larger headroom is required at one sidewall
- A new building is added directly adjacent to an existing building and it is required to avoid:
- The creation of a valley condition along the connection of both buildings.
- The imposition of additional loads on the columns and foundations of the existing building.

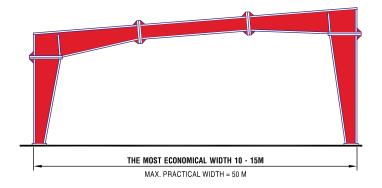


Figure 3.9 Mono-slope frame

For larger widths "mono-slope-multi-span" framing will be more economical when column free area inside the building is not an essential requirement.

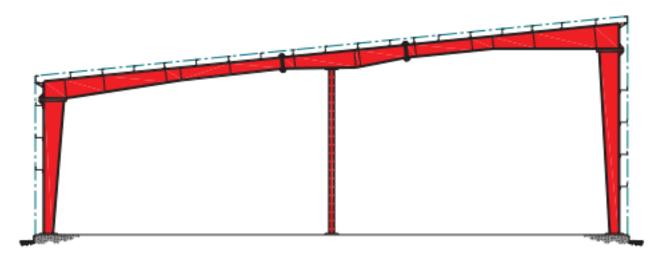


Figure 3.12 Mono-slope frame with 2 spans

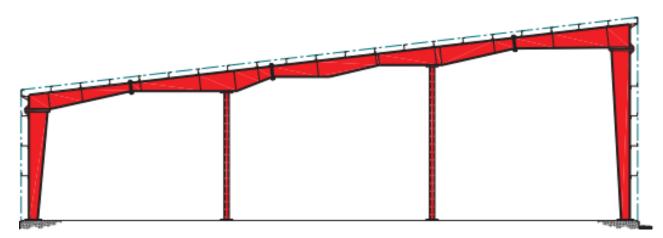


Figure 3.13 Mono-slope frame with 3 spans

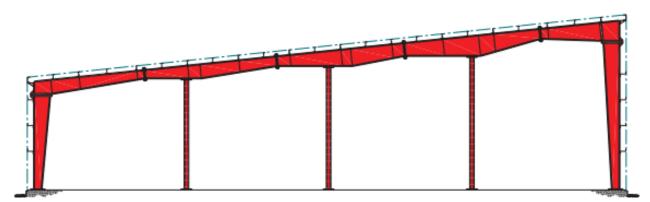


Figure 3.14 Mono-slope frame with 4 spans

#### 3.1.2.5 Space saver

Space Saver framing system offers straight columns, keeping the rafter bottom flange horizontal for ceiling applications with rigid knee connection. Selection of Space Saver is appropriate when:

- The frame width is between 6m to 18m and eave height does not exceed 6m.
- Straight columns are desired.
- Roof slope  $\leq 5\%$  are acceptable.
- Customer requires minimum air volume inside the building especially in cold storage warehouses

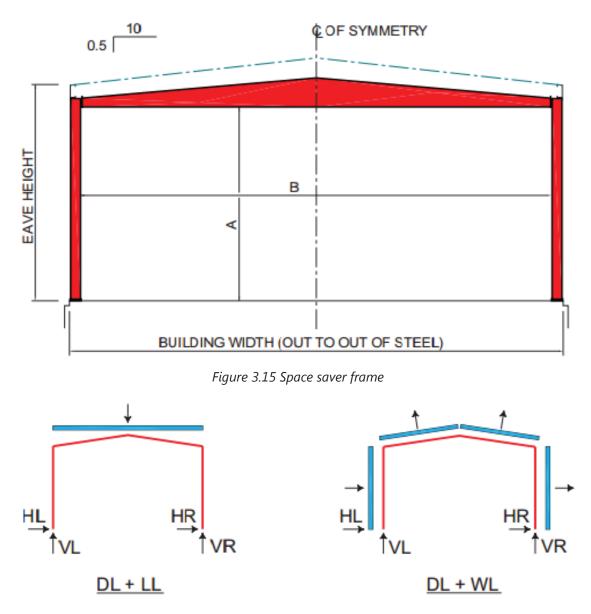
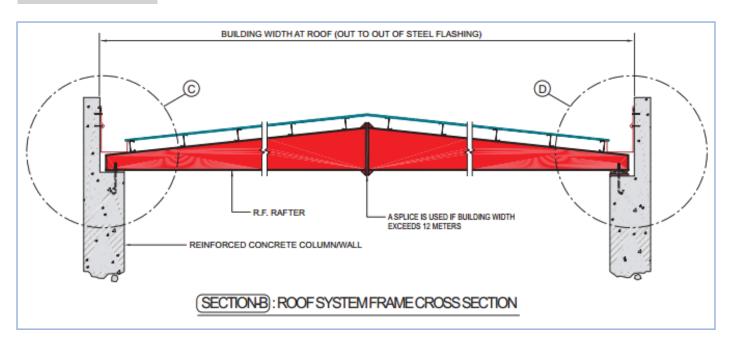


Figure 3.16 Column reaction of space saver frame

#### 3.1.2.6 Roof system

A Roof System framing consists of beam (rafter) resting onto a planned or an existing substructure. The substructure is normally made of concrete or masonry. The rafter is designed in such a way to result in only vertical reaction (no horizontal reaction) by prescribing a roller support condition at one end. The roller supports are provided at one end by means of roller rods.





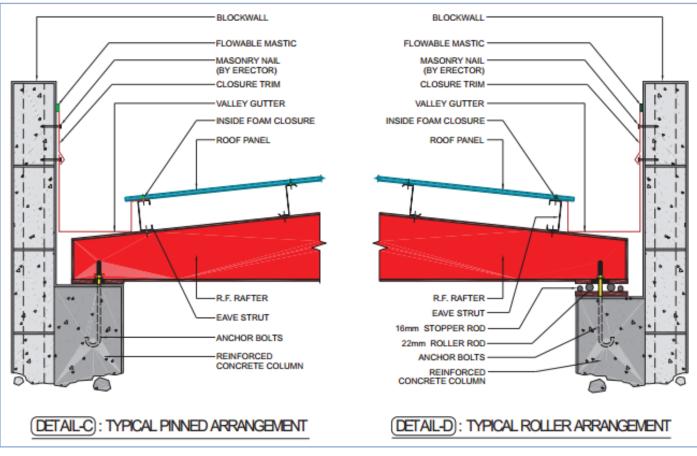


Figure 3.18 Detail C – typical pinned arrangement

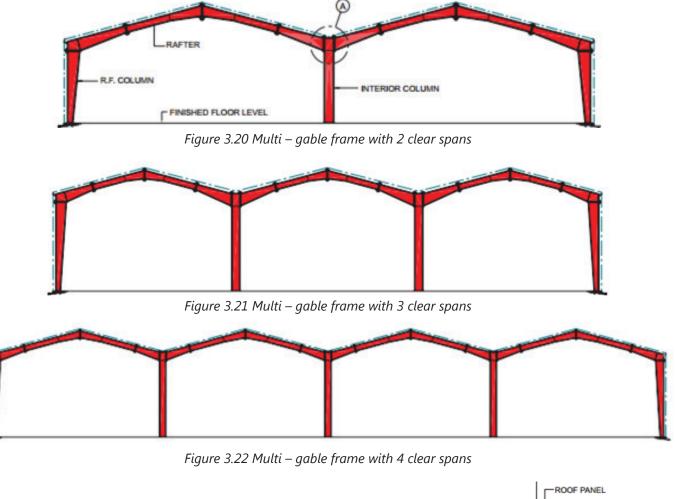
Figure 3.19 Detail D – typical rolled arrangement

A Roof System is generally not economical for spans greater than 12m although it can span as large as 36m. This is due to fact that the Roof System stresses are concentrated at mid-span rather than at the knees.

## 3.1.2.7 Multi – gable

Multi-Gable buildings are not recommended due to maintenance requirement of valley region, internal drainage and bracing requirement inside the building at columns located at valley. Especially in snow areas, Multi-Gable framing should be discouraged. However, for very wide buildings this type of framing offers a viable solution due to:

- Reduced height of ridge and thus the reduced height of interior columns
- Temperature effects can be controlled by dividing the frame into separate structural segments



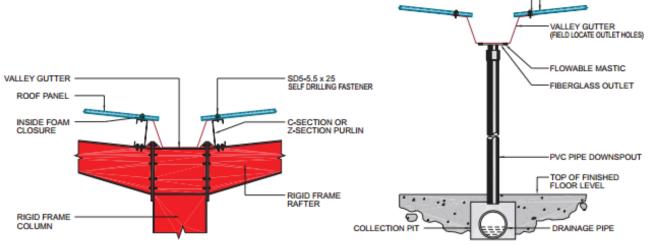




Figure 3.24 Recommended interior drainage arrangement

Thus, Multi-Gable buildings are more economical than Multi-Span buildings for very wide buildings. Multi-Gable frames may be either Clear Spans or Multi-Spans. The columns at the valley location should be designed as rigidly connected to rafters on either side using a vertical type of connection.

## 3.1.3. Roof slope

Optimum roof slope:

Type of building	Recommended slope (%)
Multi - span buildings	5
Clear span, width up to 45m	10
Clear span, width up to 60m	15
Clear span, width > 60m	20

### 3.1.4. Eave height

Eave height is governed by:

- Clear height at eave (head clearance)
- Mezzanine clear heights below beam and above joist
- Crane beam/ Crane hook heights

Minimize eave height to the bare minimum requirement since the eave height affects the price of the building by adding to the price of sheeting, girts and columns. If columns are unbraced eave height affects the frame weight significantly. Also higher eave heights increase the wind loads on the building.

If eave height to width ratio becomes more than 0.8 then the frame may have a fixed based design in order to control the lateral deflection.

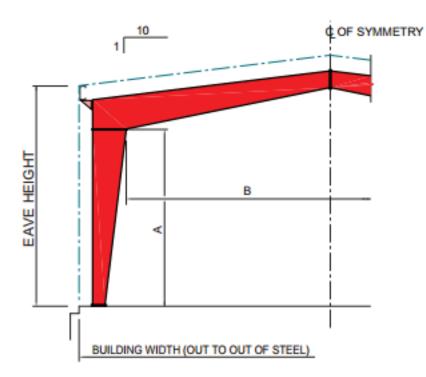


Figure 3.25 Eave height in the building

## 3.2. End Wall Systems

## 3.2.1. Generals

The standard end wall is designed as post & beam (all connections are pinned) the lateral stability is provided by the diaphragm action, in the absence of this shear diaphragm wind bracing are required.

End rigid frame are used in case of:

- Future extension is intended; in this case only wind posts are required.
- Crane running to the end wall
- Open for access condition prevails at the end wall
- X-bracing is not allowed at end wall in the case of by-framed end wall.

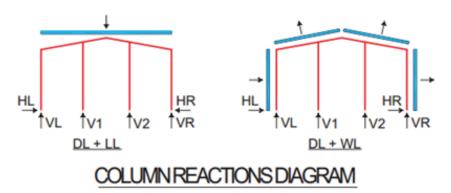
### 3.2.2. Post & Beam End Wall Rafter

All end wall rafters are designed as simple beams over the end wall posts. They are comprised of built-up sections or hot rolled sections.

#### Design concept

**Under gravity load (Dead load + Live Load):** The top flange of the end wall rafters is under compression and it is braced against lateral torsional buckling at every purlin location.

**Under uplift loads (Dead + Wind Load):** The bottom flange of the end wall rafters is under compression and it is unbraced against lateral torsional buckling. The buckling length is the distance between end wall columns, this may significantly reduce the section bending capacity or flange braces are to be used.



### 3.2.3. End Wall Post

All end wall posts are flush with the end wall structural line in end post & beam gable type and they are supporting end wall rafters. In all cases end wall posts are oriented so that end wall wind pressure is producing bending moments about the column major axis. End wall posts are comprised of built-up sections or hot rolled sections.

#### Design concept

End wall posts are the supporting elements for end wall girts/block walls for wind loads (pressure or suction) which produce bending moment about posts major axes, end wall posts are designed as simple beam supported at foundation level (base plate) and are connected with Spanner at roof purlins level.

For Post & Beam end walls additional vertical loads from end wall rafters are transmitted to posts producing axial loads (compression or tension).

End wall post buckling length about major axis is the column length.

Flanges unsupported length is depending on the end wall type and position.

## 3.3. Expansion Joints

The purpose to add the expansion joint is separating the building into many areas in order that every area has allowable temperature deformation. When separating the foundation into 2 parts, expansion joint become settlement joint.

*Section L7 AISC 360-10* recommends that the length of building needs to be added expansion joint or research NRC 1974 for more.

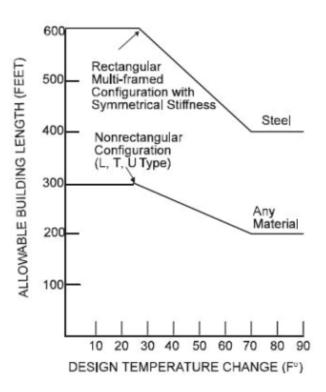


Fig. 1: Maximum allowable building length without expansion joints for various design temperature changes.

Unless more specifc site information is available, most engineers assume a range of 50° to 70° F (10° to 21° C) for continuously heated and air-conditioned buildings. Using that assumption, most steel, rectangular, framed confguration buildings with symmetrical stiffness can tolerate 460 ft (140 m) between expansion joints. Or the designer can refer to section 11.1.2 TCVN 5575:2012 to know the length & width of building with expansion joint to be added.

Type of building	Maximum distance (m)					
	Between 2 exp	pansion joints				
	Longitudinal Lateral From expa		From expansion joist or the			
	direction	direction	end of building to axis of			
			nearest wall bracing system			
Building with insulation	230	150	150			
Building without insulation	200	120	120			
& heating factory						
Viaduct	130	-	-			
Note: If there are 2 wall bracing systems in every expansion area, the distance of these not exceeding						
from 40m to 50m for building; from 25m to 30m for viaduct.						

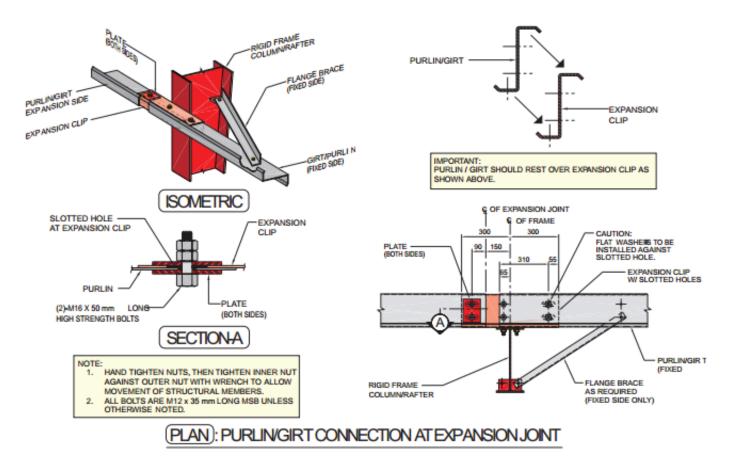


Figure 3.26 Purlin/girt connection at expansion joint

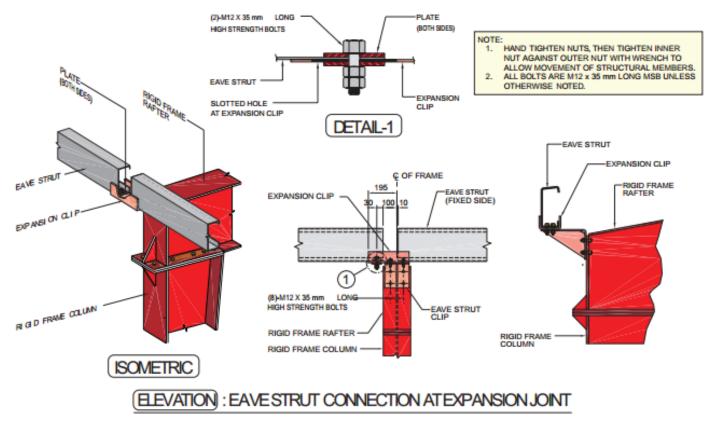
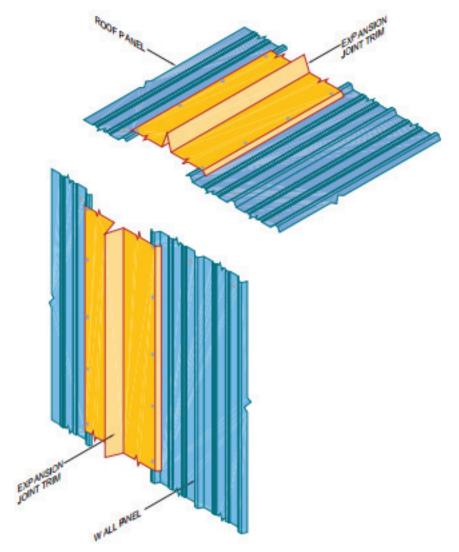


Figure 3.27 Eave strut connection at expansion joint



*Figure 3.28 Roof & wall panels at expansion joint* 

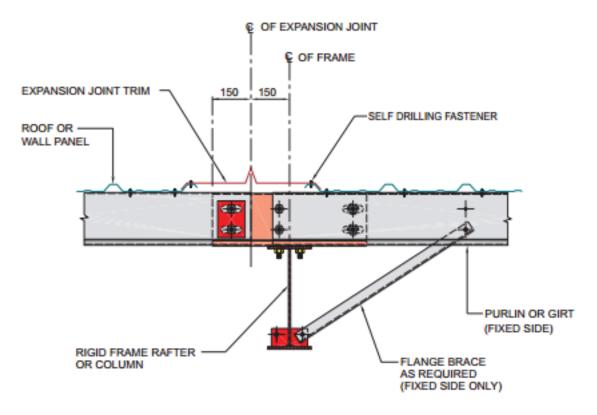
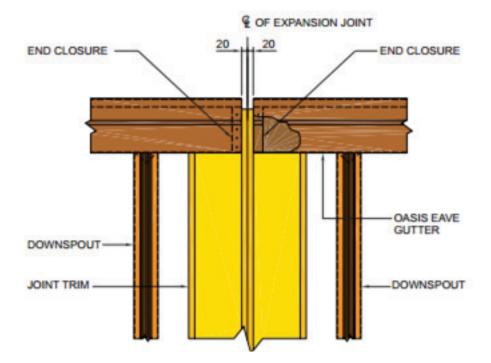
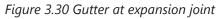


Figure 3.29 Panel connection at expansion joint





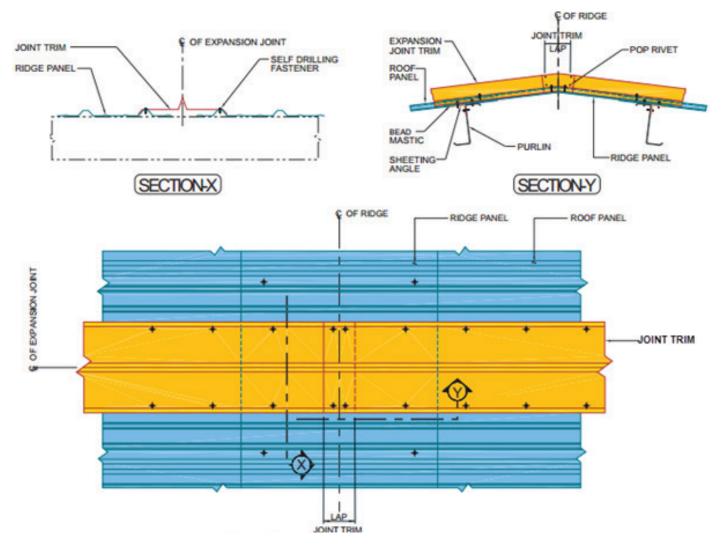


Figure 3.31 Expansion joint at ridge

## 3.4. Bay Spacing

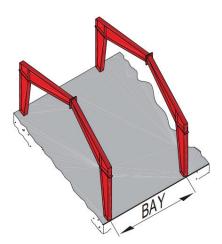


Figure 3.32 Bay spacing

The most economical bay spacing is around 8m for the standard loads as following:

Live Loads on roof and frame (kN/m2)	Wind Speed (km/h)
0.57	130

For greater loads than standard loads the economical bay spacing tends to decrease.

For buildings with heavy cranes (crane capacity > 10 Tons) the economical bay spacing ranges between 6m and 7m.

Smaller end bays than interior bays will taper off the effect of higher deflection and bending moment in end bays as compared to interior bays and help reduce the weights of purlins/girts in the end bays. This will avoid the need of nested purlins/girts in the end bays and result in uniform size of purlin/girt sizes.

Some buildings require bay spacing more than 10m in order to have a greater clear space at the interior of the building in Multi-Span buildings. Such a situation can be handled by providing jack beams that support the intermediate frames without interior columns. Thus the exterior columns will have bay spacing of say 6m while the interior columns are spaced at 12m. Intermediate frames allow the purlin to span for 6m.

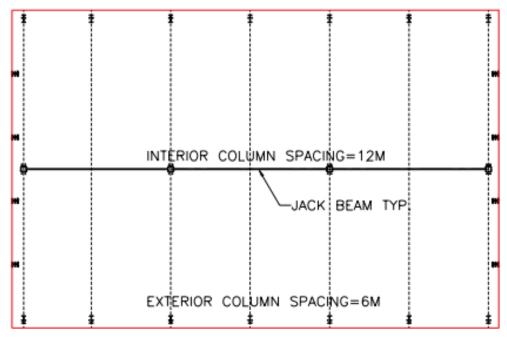


Figure 3.33 Jack beam plan with bay Spacing 12m

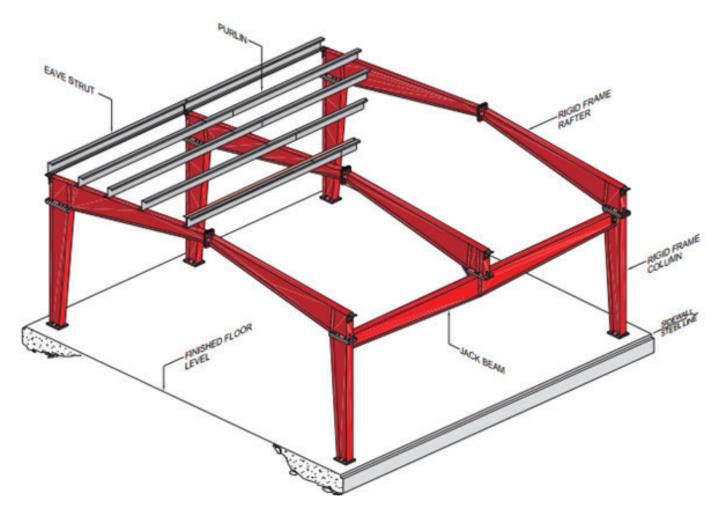


Figure 3.34 Jack beam at side wall

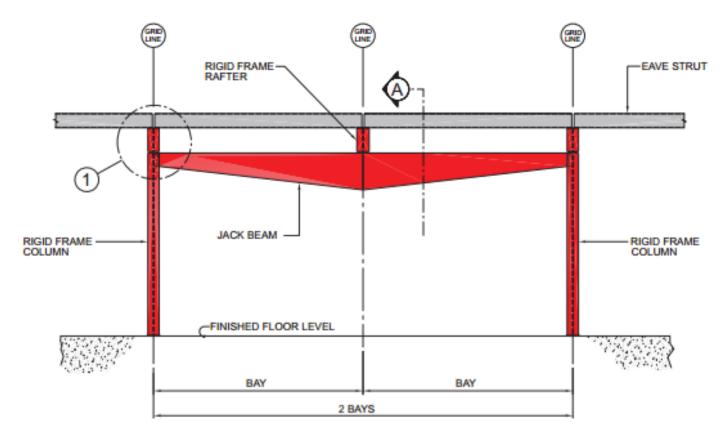


Figure 3.35 Elevation of jack beam at side wall

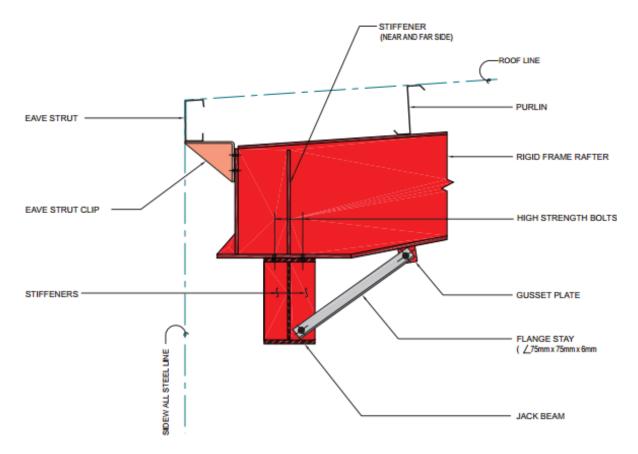


Figure 3.36 Section A - Jack beam at side wall

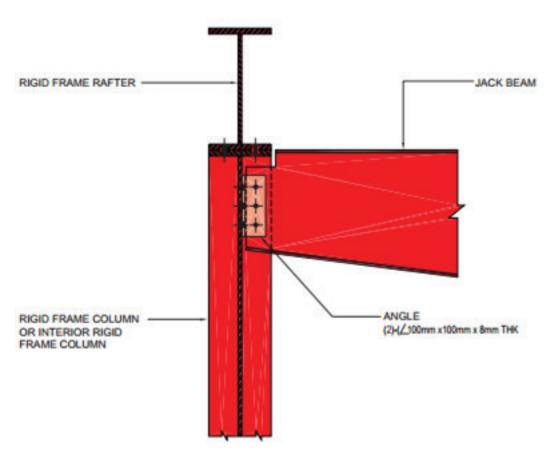
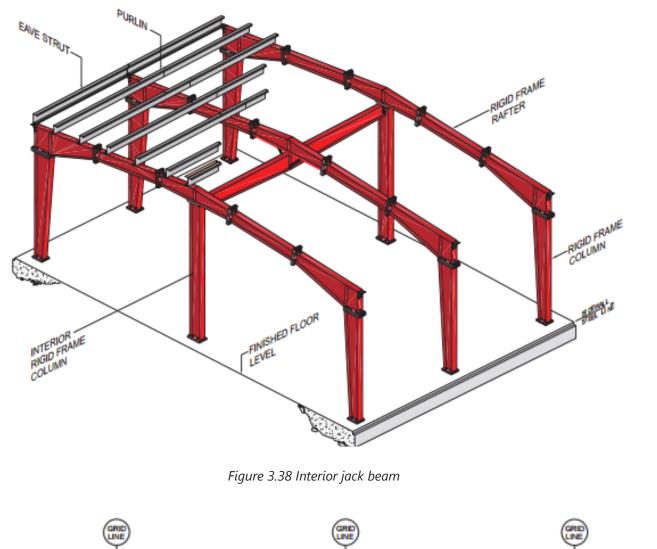


Figure 3.37 Detail 1 - Jack beam at side wall



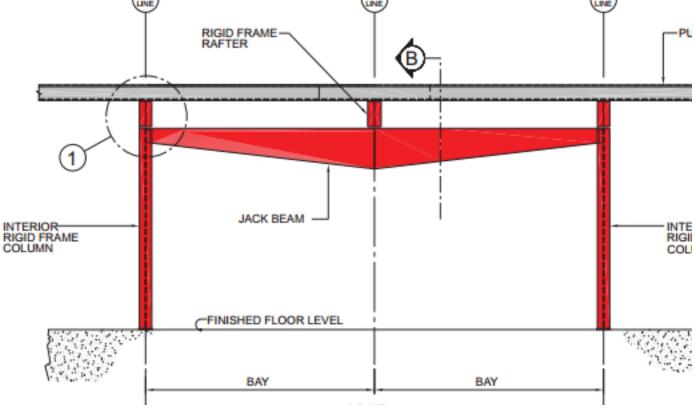


Figure 3.39 Elevation - Interior jack beam

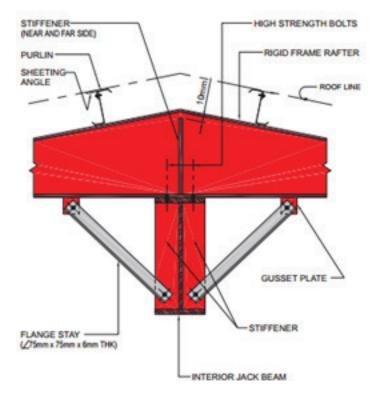


Figure 3.40 Jack beam at middle span of rafter

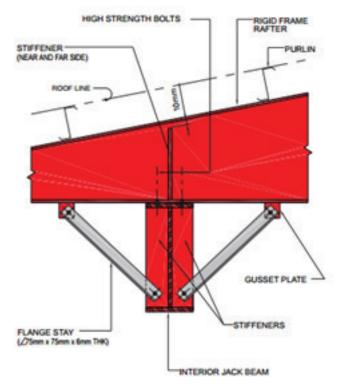


Figure 3.41 Jack beam at intermediate span of rafter

## 3.5. Bracing Systems Arrangement

Bracing is a structural system used to provide stability in a structure in a direction where applied forces on that structure would otherwise make it unstable. Whether it is a force due to wind, crane or seismic applications, the bracing system will always eventually transmit that load down to the column base and then to the foundations. The rules of arranging different types of bracing systems are as follows:

## 3.5.1. Bracing for wind and seismic loads in the longitudinal direction

1. In long buildings, braced bays shall be provided in intervals not to exceed 5 bays.

2. Sidewall bracing shall be generally placed in the same bays of roof bracing. This may not be possible at times due to openings in the sidewalls. In such cases, sidewall bracing shall be placed in bays adjacent to those containing the roof bracing with a consideration that load transfers to the adjacent bays.

3. Roof rod bracing shall not cross the ridgeline.

4. Cables/rods braces shall not exceed 15m in length. If a cross bracing contains rods longer than 15m, then the bracing should be broken to two sets of bracings with a strut member between them so that the rod/cable lengths shall not exceed 15 m.

5. Sidewall bracing shall be comprised of any one of the following types:

- Cables
- Rods or angles.
- Portal frame with/without rods or angles.

6. There shall be only one type of bracing in the same sidewall. Do not mix different types/materials in the same sidewall.

7. It is preferable to use only one type of wall bracing in the whole building otherwise the lateral loads (especially seismic loads) will not be divided equally between bracing lines. For cases when this will result in excessive weight for bracing system advanced calculation is to be done to determine the force that will be carried by each type depending on its stiffness and location.

8. Do not use rod/cable Ø25mm for roof bracing because its weight make a large vertical deflection.

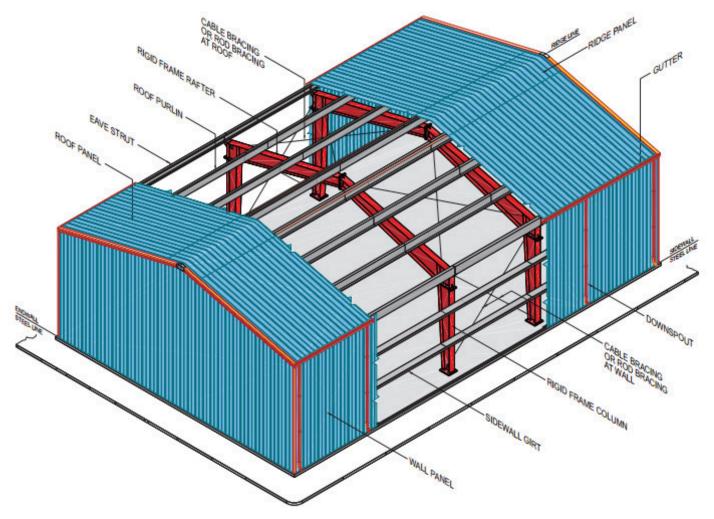
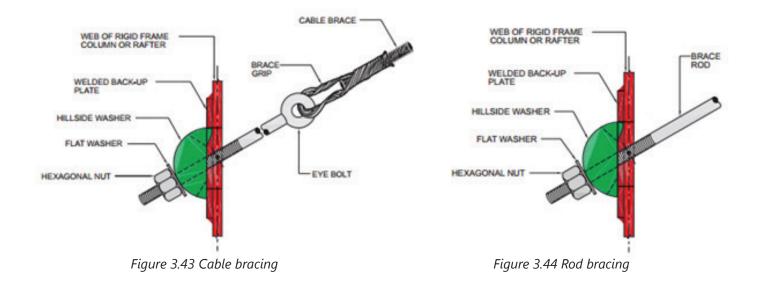
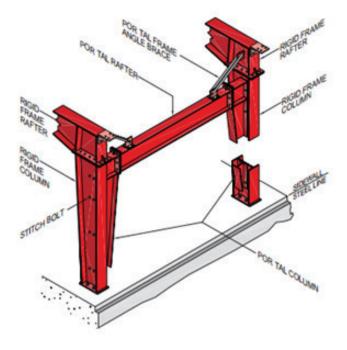


Figure 3.42 Cable or rod bracing at roof & wall of a braced bay





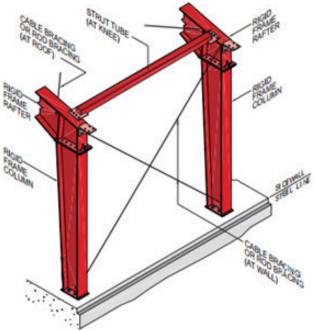


Figure 3.45 Portal frame

Figure 3.46 Cable or rod bracing with strut tube

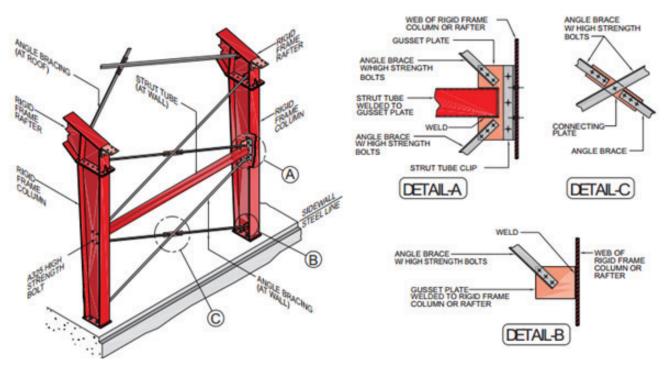


Figure 3.47 Angle bracing with strut tube

### 3.5.2. Wind and seismic bracing in P&B end wall

1. End wall bracing is not required for a fully sheeted P&B end wall with flush girt construction. If P&B end walls have by-framed girts then this end wall needs bracing.

2. If required, bracing in P&B end walls shall comprise cables or rods, unless otherwise specified by the customer. In such a case the end wall members shall be either built-up or hot-rolled members.

3. If an end wall requires bracing and the customer requests that no bracing to be placed in the plane of the end wall, then it is recommended that the load in the plane of the end wall is transferred back to the first rigid frame through additional roof bracing in the end bay.

### 3.5.3. Crane Bracing

1. In crane buildings, bracing has to be designed for longitudinal crane loads for top running or underhung cranes. The bracing shall be placed in intervals not to exceed 5 bays.

2. Longitudinal bracing for top running cranes shall be comprised of any one of the following types.

- Angles or pipes;
- Portal frame with rods (or angles);
- Portal frame without rods (or angles).

3. Longitudinal bracing for top running cranes shall be of only one type in the same longitudinal plane of a building.

4. Longitudinal bracing for underhung cranes shall consist of either rods or angles.

5. Lateral bracing for underhung cranes (attached to crane brackets), if any shall consist of either rods or angles.

6. Whenever a brace rod is used for crane bracing, the minimum diameter of that rod shall be 20mm.

7. A brace rod shall not exceed 15m in length. If angles are used the critical slenderness ratio of a bracing angle shall not exceed 300.

## **CHAPTER 4. MAIN FRAME AND CONNECTION 4.1. Main frame design procedure and constraints**

4.1.1. Design procedure 4.1.1.1 Stress Unity Checks

Combined Stress Unity Check

$$\frac{f_a}{F_a} + \frac{f_{bx}}{F_{bx}} + \frac{f_{by}}{F_{by}} \le 1.0$$

Where fa, fbx and fby are actual axial, major axis bending and minor axis bending stresses respectively. Fa, Fbx and Fby are corresponding allowable stresses. If section fails in combined unity check then check the allowable stresses:

(1) If Mn/Omega Capacity or Pnc/Omega Capacity are much lower than Mn/Omega No LTB or Pnt/Omega Capacity then it implies that the member is not properly braced then try one of the following:

	Pr	Pnc/Omega	Pnt/Omega
	Force	Capacity	Capacity
Axial	1252.377	2184.504	3250.850
	Mr	Mn/Omega	Mn/Omega
	Moment	Capacity	No LTB
Major Moment	3.762	327.434	327.434
Minor Moment	0.000	187.005	

Figure 4 1 Stress check in SAP

• For rafters and exterior columns (with sheeted side walls) adding flange braces with roof purlins or wall girts will adjust the allowable stresses for the unbraced flange.

• For exterior columns (without sheeted side walls) then providing EB (strut tubes) adequately connected to bracing system at an appropriate height would reduce the unbraced length and adjust the allowable stress.

• For interior I-section columns they can also be braced by means of EB if allowed and adequately connected to bracing system.

• For interior I-section columns that brace points cannot be added in the design then stress ratios can be improved by increasing flanges width or by minor adjustment in the flange thickness.

• For columns connected with mezzanine beams/joists columns are considered braced at mezzanine level.

• For columns supporting top running crane beam the columns are considered laterally braced at the level of carne beam top flange.

(2) If allowable stresses are sufficiently high and still the section is failing in unity check, then unity check ratio can be improved by increasing the following in the given order:

- Increasing the web depth
- Increasing the flange width
- Increasing the flange thickness
- (3) If Shear stress unit ratio fv/Fv > 1.0 increase web thickness.

### 4.1.1.2 Controlling Deflections

Refer Table 2 26 Deflection Limitations for Deflection Limitations.

If lateral deflection exceeds the prescribed limit (normally H/60) then check the H/Width ratio. If H/B > 0.75 then fixing the base would result in more economical frame. If H/B<0.75 then increase the web depth at knee of both column and rafter (difference between knee depth of column and rafter < 200mm)

In multi-span frames before going for the option of fixing the exterior column at base, check whether fixing the tops of interior columns control the lateral sway. If not then fix the exterior column bases.

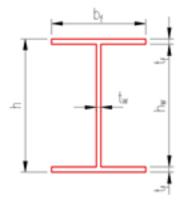
If Vertical deflection  $\Delta v$  exceeds the prescribed limit (normally Span/180), increase the web depth at knee of both column and rafter. A slight increase in the rafter depth at ridge will also help control the vertical deflection.

# 4.1.2. Design constraints *4.1.2.1 Standard*

$$\frac{b_{w}}{t_{w}} < 180 ; \frac{b_{f}}{t_{f}} < 31$$
$$\frac{b}{b_{f}} < 5; \frac{t_{f}}{t_{w}} < 2.5$$

Compression element:  $\frac{kL}{r} < 200$ 

Tension element:  $\frac{kL}{r} < 300$ 



## 4.1.2.2 Fabrication limitation for built-up section

Web thickness	Minimum	4mm: beam, rafter
		5mm: column
	Maximum	50mm but give priority to 12mm
Flange thickness	Minimum	5mm: beam, rafter
		6mm: column
	Maximum	50mm but give priority to 12mm
Web depth	Minimum	200mm
	Maximum	1500mm
Flange width	Minimum	134mm
_	Maximum	746mm



### NOTE

Width of continuous flange should be constant along the one welded piece.

• Variation of thickness at any butt weld splice of continuous flange/web within the one welded piece should be limited to maximum 6mm.

Width flange/web having constant width, use steel plate gouge like below table.

Width								÷						
(mm)	2	3	4	5	6	7	8	9	10	11	12	13	14	15
1200	600	400	300	240	200	171	150	133						
	596	397	298	238	198	169	148	131						
1500	750	500	375	300	250	214	188	167	150	136				
	746	496	372	298	248	212	184	164	148	134				
2000	1000	667	500	400	333	286	250	222	200	182	167	154	143	133
	996	663	496	397	330	284	248	220	198	180	165	152	141	131

## 4.1.2.3 Shipping limitation

Maximum fabricated out-to-out length of the piece is **14m** for transportation by truck (in Vietnam), and **11.7m** for transportation by dry cargo container (foreign) except **13.5m** for Cambodia.

## 4.1.2.4 Other guidelines

- (1) At knee connection, maximum difference between column depth and rafter depth is 200mm.
- (2) In a tapered section, the minimum difference in web depth at start and end should be 100mm.
- (3) Minimum base plate thickness = 14 mm.
- (4) Minimum base plate width = 164 mm.
- (5) Minimum splice plate thickness = 12 mm.
- (6) Minimum splice plate width = 164 mm.
- (7) Minimum anchor bolt diameter = M20 (except end-wall post M16).
- (8) Minimum splice bolt = M16.

## 4.1.2.5 Optimization

To produce the most economical frame profiles, let apply the following rules:

(1) Minimize number of splices in the columns and rafters by providing maximum possible lengths regardless of the material savings that can be produced otherwise. Section lengths should be multiple of 3m i.e., 3m, 6m, 9m and 12m in order to reduce the scrap.

(2) In case of different bay spacing avoid using more than 3 frames.

(3) Different frame should be adopted if saving of 5% on all frames with a minimum of 1.0ton is ascertained.

(4) When different frames have to be used due to different bay spacing, maintain the same web cuts for all such frames.

(5) Minimize the number of different flange widths in a frame. Maximum different widths of flanges in all the frames should preferably be less than three.

(6) As much as possible maintain uniformity in the base plate detail and anchor bolt sizes for all the frames.

(7) Try to locate the splices at the locations where the bending moment is least and/or where the depth is least in a frame.

(8) Try to follow the shape of bending moment diagram for the controlling load combination in the configuration of the frame by maintaining the stress unity check ratios closer to 1.

## 4.1.3. Instruction to build Sap2000 Model

This section provides step-by-step instructions for building a basic SAP2000 model.

### Defining

Units and geometry Material and member section properties Load patterns, load cases and load combinations

#### Modeling

Draw/Edit Frame Objects and Bracing System Assigning/Edit Section Properties, Releases & Restraints. Assign Load Assign Unbraced Length Ratio (ULR)

#### Analyzing

Set Analysis Option Set Load Cases to Run form

**Display the result** 

Design Steel Frame Check Deflection Limitation

### 4.1.3.1 Defining

In this Step, the basic grid that will serve as a template for developing the model will be defined. Then a material will be defined and sections will be selected.

a. Setting Default Units (KN, m, C) through: File > New Model Form.

New Model					
	ation del from Defaults u del from an Existir		N, m, C 💌	Project Info Modify	mation /Show Info
Select Template					
Blank	Grid Only	<u>∧ ∧</u> ∬⊸. Beam	2D Trusses	3D Trusses	2D Frames
				Z	I
3D Frames	Wall	Flat Slab	Shells	Staircases	Storage Structures
Underground Concrete	Solid Models	Pipes and Plates			

Figure 4 2 New model form

# b. Setting up geometry in 2 ways:

(1) Creating Generally Grid System: **New Model form > Grid Only** 

(2) Define Grid System Data: Define>Coordinate Systems/Grids>GLOBAL>Modify/Show System

Number of Grid Lines         1         A           X direction         11         2         B           Y direction         11         3         C           Y direction         11         5         E           6         F         7         G	GLOBAL Ordinate Line Type 0. Primary 5. Primary 10. Primary 10. Primary 20. Primary 25. Primary 30. Primary	Visibility Show Show Show Show Show Show	Units KN, m, C Bubble Loc. Grid Colo End End End End End End	Grid Lines Quick Start
Coordinate System Name         System Name           GLOBAL         X Girid Data           Number of Grid Lines         1           X direction         11           Y direction         11           Z direction         5           7 direction         5	Ordinate Line Type 0. Primary 5. Primary 10. Primary 15. Primary 20. Primary 25. Primary 30. Primary	Show Show Show Show Show Show	KN, m, C Bubble Loc. Grid Colo End End End End End End	Quick Start
GLOBAL         X Grid Data           Number of Grid Lines         1         A           X direction         11         2         B           Y direction         11         4         D           5         E         6         F           7         G         7         G	0.         Primary           5.         Primary           10.         Primary           15.         Primary           20.         Primary           25.         Primary           30.         Primary	Show Show Show Show Show Show	End End End End End	
Grid ID         G           Number of Grid Lines         1         A           X direction         11         3         C           Y direction         11         5         E           6         F         7         G	0.         Primary           5.         Primary           10.         Primary           15.         Primary           20.         Primary           25.         Primary           30.         Primary	Show Show Show Show Show Show	End End End End End	
Number of Grid Lines         1         A           X direction         11         2         B           Y direction         11         3         C           Y direction         11         5         E           6         F         7         G	0.         Primary           5.         Primary           10.         Primary           15.         Primary           20.         Primary           25.         Primary           30.         Primary	Show Show Show Show Show Show	End End End End End	
X direction         11         2         B           Y direction         11         3         C         4         D           Z direction         11         5         E         6         F           7         G         7         G	5.         Primary           10.         Primary           15.         Primary           20.         Primary           25.         Primary           30.         Primary	Show Show Show Show	End End End End	
X direction         11         3         C           Y direction         11         4         D         5         E         6         F         6         F         7         G	10.         Primary           15.         Primary           20.         Primary           25.         Primary           30.         Primary	Show Show Show	End End End	
Y direction         11         4         D           Z direction         11         5         E           6         F         7         G	15.         Primary           20.         Primary           25.         Primary           30.         Primary	Show Show	End End	
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	<ol> <li>Primary</li> <li>Primary</li> </ol>	Show	Start	Hide All Grid Lines
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		Show	Start Start	Glue to Grid Lines
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	40. Primary 48. Primary	Show	Start	Bubble Size 3.
First Grid Line Location	48. Primary 56. Primary	Show	Start	
	56. Primary	Show	Start	<b>•</b>
X direction 0.				Reset to Default Color
Grid ID C	Ordinate Line Type	Visibility	Bubble Loc.	
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2 Z2	.2 Primary	Show	End	Reorder Ordinates
Z direction 0. 3 Z3	5 Primary	Show	End	
4 Z4	8 Primary	Show	End	
5 Z5	11.75 Primary	Show	End	
6				
OK Cancel 7				OK Cancel
				▼

Figure 4 3 Quick Grid Lines form

Figure 4 4 Define Grid System Date form

# c. Define Materials through: **Define > Materials**

Material Property Data	Material Property Data	Material Property Data
General Data       Material Name and Display Color       Steel       Material Type       Steel       Material Notes       Weight and Mass       Weight per Unit Volume       76.9729       Mass per Unit Volume       Tostropic Property Data       Modulus of Elasticity, E	General Data       Material Name and Display Color       STEEL2350       Material Type       Steel       Material Notes       Weight and Mass       Weight per Unit Volume       75.9729       Mass per Unit Volume       Isotropic Property Data       Modulus of Elesticity, E	General Data       Material Name and Display Color       Material Type       Tendon       Material Notes       Weight and Mass       Weight Property Data       Uniaxial Property Data       Modulus of Elasticity, E
Poisson's Ratio, U 0.3 Coefficient of Thermal Expansion, A 1.170E-05 Shear Modulus, G 76903069	Poisson's Ratio, U 0.3 Coefficient of Thermal Expansion, A 11.170E-05 Shear Modulus, G 76903069	Poisson's Ratio, U 0. Coefficient of Thermal Expansion, A 1.170E-05 Shear Modulus, G 98250300
Other Properties for Steel Materials Minimum Yield Stress, Fy 345000 Minimum Tensile Stress, Fu 450000 Effective Yield Stress, Fye 380000 Effective Tensile Stress, Fue 500000	Other Properties for Steel Materials       Minimum Yield Stress, Fy     235000       Minimum Tensile Stress, Fu     400000       Effective Yield Stress, Fye     370000       Effective Tensile Stress, Fue     432000	Other Properties for Tendon Materials Minimum Yield Stress, Fy [235000 Minimum Tensile Stress, Fu 400000]
STEEL3450	Switch To Advanced Property Display	Switch To Advanced Property Display DK Cancel A36 (Tendon)

Figure 4 5 Material Property Data

### d. Define Frame Sections

#### Define Frame Section through: Define/Section Properties/Frame Sections

I / Wide Flange Section			Non	prismatic Secti	on Definition				
Section Name Section Notes Dimensions	hC-250212 Modify/Show Notes	Display Color		Nonprismatic Section Notes	Section Name	1C Modify	/Show Notes	Dis	play Color
Outside height (13) Top flange width (12) Top flange thickness (1f) Web thickness (1w) Bottom flange width (12b) Bottom flange thickness (1fb) Material	0.212 8.000E-03 5.000E-03 0.212 8.000E-03 Properties Section	Properties		Start Sectio 1C-250212 1C-250212	n End Secti TC-600212	on Length	Length Type Variable Variable	EI33 Variation Parabolic Parabolic	El22 Variation
		ident Properties			Add	Insert OK	Modify	Delete	

Figure 4 6 I/Wide Flange Section form

Figure 4 7 Nonprismatic Section Definition form

### Define Tendon Section through: Define/Section Properties/Tendon Sections

Tendon Section Data						
Tendon Section Name Section Notes	BR1 Modify/Show					
Tendon Modeling Options For Analysis Model Model Tendon as Loads  Model Tendon as Elements						
Tendon Parameters Prestress Type Material Property +	Prestress Type					
Tendon Properties Specify Tendon Diameter Specify Tendon Area Torsional Constant Moment of Inertia Shear Area	0.016 2.011E-04 6.434E-09 3.217E-09 1.810E-04					
Units KN, m, C	Display Color					

Figure 4 8 Tendon Section Data form (Diameter 16mm)

### e. Define Load patterns, load cases and load combinations

- Define Load Patterns through: **Define > Load Patterns**
- Define Load Cases through: Define > Load Cases
- Define Load Combinations through: **Define > Load Combinations** or using **\*.s2k** text file

In "**1A - LOAD APPLICATION AND PURLIN - Under 18m**" file, go to sheet "2" and enter the number for each load pattern, click **COMBINATION** then click **EXPORT TO .TXT** which will display a window to specify position for saving file. Click the Save button to save file. 74

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1					Date								
	BMB Project	ect PROJEC		ST #1A	Designed by								
MB&/	A J/S CO. Description	DESI	GN LOADING	COMBINATION	Checked by								
SYME	BOLS AND NOTATIO	ON											
					-								
	Load	Symbols	No.	Case Type			-						
	Dead load Collateral load	DL	1	Linear Static Linear Static				COMBI	NATIC	ON			
	Roof live load	AL	1	Linear Static	-		L	7.7.0805					
	Wind load	WL	3	Linear Static	-								
	Floor load	FL		Linear Static			ſ	EVRORT		THE			
	Crane load	CR		Linear Static				EXPORT	10.	191			
	Earthquake load	EL		Linear Static									
COM	BINING NOMINAL L	OAD USIN	G ALLOWA	BLE STRESS DE	SIGN (ASCE	7-10, 2.4)							
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	COMBO 2	DL + LL DL + 0.6W	11										
	COMBO 3	DL + 0.6W		Nhap ten file	text can luu								×
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	COMBO 5				« BMB DESIGN	GUIDE > SAP		•	◆• Se	earch SAP	_	_	
	COMBO 5 COMBO 6 COMBO 7	DL + 0.75L	L + 0.45WL2 L + 0.45WL3	Organize 🗸	Wew folder	GUIDE • SAP		*	* <del>7</del> Se	earch SAP	8	e •	0
	COMBO 6 COMBO 7 COMBO 8	DL + 0.75L DL + 0.75L 0.6DL + 0.	L + 0.45WL2 L + 0.45WL3 6WL1		New folder			•	◆• Se	earch SAP	8	= <b>•</b>	-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize 🕶	New folder	ne		•	◆• Se	earch SAP	8	= <b>•</b>	-
	COMBO 6 COMBO 7 COMBO 8	DL + 0.75L DL + 0.75L 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize 🕶	New folder Drive A Nan	ne combination.tx	t	•	* <del>y</del> Se	earch SAP		= <b>•</b>	-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize 🗸 Coogle Desktog Libraries	New folder Drive ANan	ne	t	•	*7 Se	earch SAP	*		-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize   Organize  Google  Desktop  Libraries  Docum	New folder Drive ANan	ne combination.tx	t	· •	*• Se	earch SAP		≡ •	-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize   Google  Desktop  Libraries  Docum  Music	New folder Drive Nan o ents	ne combination.tx	t	·	Se Se	CITCH SAP		= •	-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize V Coganize V Cogani	New folder Drive Nan o ents	ne combination.tx	t	·	See	eren SAP		∃ ▼	-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize   Google  Desktop  Libraries  Docum  Music	New folder Drive Nan o ents	ne combination.tx	t	•	See		*	≡ •	-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize ▼ ▲ Google ■ Desktop ⇒ Libraries ◎ Docum ▲ Music ■ Pictures ♥ Videos	New folder Drive A Nan Pents E	e combination.tx combination1.t	đ bđ	•	Se Se		*	≡ ▼	-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize V Coganize V Cogani	New folder	ne combination.tx	đ bđ	•	See	COTCO SAP	*	₩ .	-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize ▼ Google Desktop Libraries Docum Music Picture Videos Homegra	New folder Drive A Nan Pents E	ne combination.tx combination1.1	đ bđ	•	See		*	₩ •	0
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize ▼ Corganize ♥ Corganize ♥ Corga	New folder Drive Nan ents E	e combination.tx combination1.t	đ bđ	· ·	See		*	<b>₩</b>	-
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize ▼ Corganize ♥ Corganize ♥ Corga	New folder Drive Nan ents E sents E	e combination.tx combination1.t	đ bđ	·	Se Se		*	= •	0
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize ▼ Corganize ♥ Corganize ♥ Corga	New folder Drive A Nan ents E ame: combinat type: Unicode 1	e combination.tx combination1.t	đ bđ		• • Se	Save		Cance	
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize ▼ Coganize ▼ Cocal	New folder Drive A Nan ents E ame: combinat type: Unicode 1	e combination.tx combination1.t	đ bđ				<sup>89</sup>		
	COMBO 6 COMBO 7 COMBO 8 COMBO 9	DL + 0.75L DL + 0.75L 0.6DL + 0. 0.6DL + 0.	11 + 0 45WL2 11 + 0 45WL3 6WL1 6WL2	Organize ▼ Coganize ▼ Cocal	New folder Drive A Nan ents E ame: combinat type: Unicode 1	e combination.tx combination1.t	đ bđ						

Figure 4 9 Design loading combination sheet

In SAP2000, you have to define at least one load combination before export .s2k text file. Click the **Define > Load Combinations** command, in the **Define Load Combinations** form which has just appeared, click the **Add New Combo** button and create an combination as shown in figure below.

Load Combination Name	COMB1			
Notes	Modify/Show Notes			
Load Combination Type		Linear Add	•	
Options				
Convert to User Load Con		near Load Case from L	.oad Combo	
Define Combination of Load Ca				
Load Case Name	Load Case Type Linear Static	Scale Factor		
	Linear Static	1.		
			Add	
			Modify	
			Delete	

Figure 4 10 Create load combination COMB1

After create any combination, click the File > Export > Sap2000 .s2k Text File command to display the Choose Tables for Export to Text File form shown in figure below.

MODEL DEFINITION (44 of 44 tables selected)	Load Patterns (Model Def.)
🗄 🖾 System Data	Select Load Patterns
🗄 🖾 Property Definitions	5 of 5 Selected
🖩 🖾 Load Pattern Definitions	
🗄 🖾 Other Definitions	Options
🖶 🖾 Load Case Definitions	E Selection Only
🖶 🛛 Connectivity Data	Open File After Export
🗄 🗹 Frame Assignments	Use Text Editor
🖶 🛛 Options/Preferences Data	O Use Microsoft Wor
🗄 🛛 Miscellaneous Data	
	Expose All Input Tables
	Named Sets
	Save Named Set
	Show Named Set
	Delete Named Set
	OK Cancel
	UN Lancel

Figure 4 11 Choose Tables for Export to Text File form

Make sure that all tables, load patterns, and Open File After Export option are selected before click the **OK** button. Copy all data in combination.txt text file and paste to .s2k text file at highlight text shown in figure below.

TABLE: "LOAD PATTERN DEFINITIONS" LoadPat=DL DesignType=DEAD SelfWtMult=1 LoadPat=LL DesignType=LIVE SelfWtMult=0 LoadPat=WL1 DesignType=WIND SelfWtMult=0 AutoLoad=None LoadPat=WL2 DesignType=WIND SelfWtMult=0 AutoLoad=None LoadPat=WL3 DesignType=WIND SelfWtMult=0 AutoLoad=None
TABLE: "AUTO WAVE 3 - WAVE CHARACTERISTICS - GENERAL" WaveChar=Default WaveType="From Theory" KinFactor=1 SWaterDepth=45 WaveHeight=18 WavePeriod=12 WaveTheory=Linear
TABLE: "COMBINATION DEFINITIONS" ComboName=COMB1 ComboType="Linear Add" AutoDesign=No CaseType="Linear Static" CaseName=DL ScaleFactor=1 SteelDesign=None ConcDesign=None AlumDesign=None ColdDesign=None
TABLE: "FUNCTION - RESPONSE SPECTRUM - USER" Name=UNIFRS Period=0 Accel=1 FuncDamp=0.05 Name=UNIFRS Period=1 Accel=1
TABLE: "FUNCTION - TIME HISTORY - USER" Name=RAMPTH Time=0 Value=0 Name=RAMPTH Time=1 Value=1 Name=UNIFTH Time=4 Value=1 Name=UNIFTH Time=1 Value=1
TABLE: "FUNCTION - POWER SPECTRAL DENSITY - USER" Name=UNIFPSD Frequency=0 Value=1 Name=UNIFPSD Frequency=1 Value=1
TABLE: "FUNCTION - STEADY STATE - USER" Name=UNIFSS Frequency=0 Value=1 Name=UNIFSS Frequency=1 Value=1
TABLE: "GROUPS 1 - DEFINITIONS" GroupName=ALL Selection=Yes SectionCut=Yes Steel=Yes Concrete=Yes Aluminum=Yes ColdFormed=Yes Stage=Yes Bridge=Yes AutoSeismic=No AutoWind=No SelDesSteel=No SelDesAlum=No SelDesCold=No MassWeight=Yes Color=Red
TABLE: "JOINT PATTERN DEFINITIONS" Pattern=Default
TABLE: "MASS SOURCE" MassSource=MSSSRC1 Elements=Yes Masses=Yes Loads=No IsDefault=Yes

Figure 4 11 Choose Tables for Export to Text File form

Save and close file. Return to SAP and import this file. Click the **File > Import > SAP2000 .s2k Text File** command to display **Import Tabular Database** shown in figure below. Click the OK button, specify created .s2k text file in appeared window and click the Done button to import file.

Impo	nt Type	
œ	New model	
C	Add to existing model	
	Advanced Options	

Figure 4 13 Import Tabular Database form

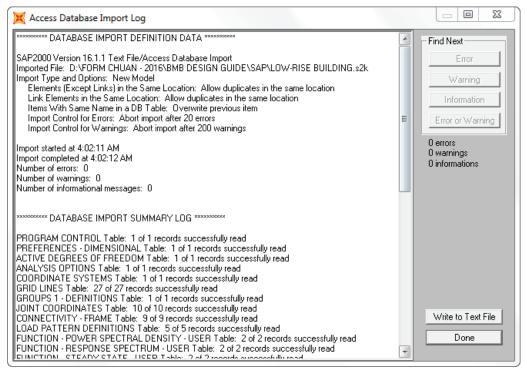


Figure 4 14 Click the Done button to finish importing .s2k text file

# 4.1.3.2 Modeling

### a. Draw Frame Objects

In this step, frame objects with the associated sections previously defined are drawn using the grids and snap-to options, and generated using Edit>Draw Frame/Cable.

The Properties of Object pop-up form for frames will appear as shown in figure below.

Properties of Object	×
Line Object Type	Straight Frame
Section	1C-250212
Moment Releases	Continuous
XY Plane Offset Normal	0.
Drawing Control Type	None <space bar=""></space>

Figure 4 15 Properties of Object form

Use default values as seen in figure above to draw because we will assign information to frames after dividing them with appropriate length.

### b. Draw Bracing System

To draw EB and Tendon/Cable system, using Draw Frame/Cable button.

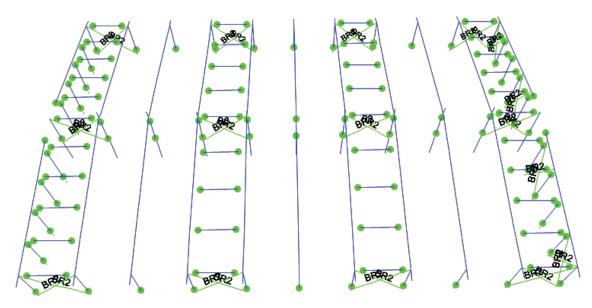


Figure 4 16 Bracing System

### c. Edit Modeling

### Divide frame objects

SAP2000 allows members to be sub-divided into multiple objects after they are drawn to accommodate changes in geometry (this differs from the internal meshing done during analysis where the number of objects remains the same).

Select rafters and vertical lines, then click the **Edit > Edit Lines > Divide Frames** command to show Divide Selected Frames form. In this form, make sure that Break at intersection with selected Joints, Frames, Area Edges and Solid Edges is selected before click the OK button.

Divide Selected Frames		
Divide Selected Straight Frame Objects		Units
O Divide into Specified Number of Frames		KN, m, C 💌
Number of Frames		
Last/First Length Ratio		
Break at intersections with selected Joints, Frames, Area Edg	es and Solid Edges	
O Divide at Specified Distance from I-end of Frame		
Distance Type		
Distance		
O Divide at Intersection with a Coordinate Plane in the Current C	Coordinate System	
Coordinate Plane		
Intersection with Plane at		OK
C Divide at Intersection with Visible Grid Planes in the Current C	oordinate System	Cancel
Grid Plane		

Figure 4 17 Divide Selected Frames form

Re	eplicate		
	Linear	Radial	Mirror
	Increments		Replicate Options
	dx 0.		Modify/Show Replicate Options
	dy 8		12 of 12 active boxes are selected
	dz 0.		Delete Original Objects
	Increment Data		
	Number 8		
		04	Cancel

Figure 4 18 Replicate form

### Assign member end releases and restraint

- Assign Frame Releases through: Assign > Frame > Releases/Partial Fixity
- Assign Joint Restraints through: Assign > Joint > Restraints
- Assign Frame Sections through: Assign > Frame > Frame Sections
- Assign Unbraced Length Ratio (ULR):

ULR is specified as a fraction of the frame object length. Multiplying these factor times the frame object length gives the unbraced length for the object. There are three types of ULR include ULR in major axis, minor axis, and lateral torsional buckling.

(1) ULR (major) is ULR for buckling about the frame object major axis (occur when object is compressed). This item is taken to be the maximum distance between two major bracing divided by actual length of object and it is usually specified as Program Determined.

(2) ULR (minor) is ULR for buckling about the frame object minor axis (occur when object is compressed). This item is taken to be the maximum distance between two bracing points (bracing member frame into web or flange of the braced member) divided by actual length of object.

(3) ULR (LTB) is ULR for lateral-torsional buckling for the frame object (occur when object is flexed). This item is taken to be the maximum distance between two bracing points (bracing member frame into flange of the braced member) divided by actual length of object.



### NOTE

Assume purlin spacing and girt spacing is **1.5m**.

With rafters, bracing members are purlins and flange braces. To conservative, just locations that have both purlin and flange brace are considered as bracing point.

Therefore, unbraced length is spacing between flange braces.

With outer columns, bracing members are girts and flange braces.

Therefore, unbraced length is the greater value of spacing between flange brace or distance from ground to the first flange brace.

With inner columns, bracing members are longitudinal steel members and portal frame. Unbraced length is the largest of unbraced lengths.

2 Fr 3 Or 4 Co	Item urrent Design Section raming Type	Value Program Determined		Unbraced length factor for lateral-		
2 Fr 3 Or 4 Co	-	Program Determined			1	
2 Fr 3 Or 4 Co	-	r rogram blocommed		torsional buckling for the frame		
3 0) 4 Co	annig 1360	Program Determined		object. This item is specified as a fraction of the frame object		
	mega0	Program Determined	-	length.Multiplying this factor times the		
5 D	onsider Deflection?	No	_	frame object length gives the unbraced		
	eflection Check Type	Program Determined	_	length for the object. Specifying 0 means the value is program determined.		
6 DI	L Limit, L /	Program Determined	_	Thearis the value is program determined.		
7 St	uper DL+LL Limit, L /	Program Determined	_			
8 Li	ive Load Limit, L /	Program Determined				
9 To	otal Limit, L/	Program Determined				
10 To	otalCamber Limit, L/	Program Determined				
11 DI	L Limit, abs	Program Determined				
12 <mark>S</mark> t	uper DL+LL Limit, abs	Program Determined				
13 Li	ive Load Limit, abs	Program Determined				
14 To	otal Limit, abs	Program Determined				
	otalCamber Limit, abs	Program Determined				
16 <mark>S</mark> p	pecified Camber	Program Determined				
17 No	et Area to Total Area Ratio	Program Determined				
18 Li	ive Load Reduction Factor	Program Determined				
	nbraced Length Ratio (Major)	Program Determined				
	nbraced Length Ratio (Minor)	0.25				
	nbraced Length Ratio (LTB)	0.25		1		
	ffective Length Factor (K1 Major)	Program Determined		Explanation of Color Coding for Values-		
	ffective Length Factor (K1 Minor)	Program Determined				
24 Ef	ffective Length Factor (K2 Major)	Program Determined	-	Blue: All selected items are program determined		
t To Pr	rog Determined (Default) Values	Reset To Previous Values		Black: Some selected items are user defined		
Alti	tems Selected Items	All Items Selected Ite	ms	Red: Value that has changed during the current session	1	

Figure 4 19 Steel frame design overwrites form

# d. Assign loads through Assign/Frame Loads.

In this step, the dead, live and wind loads will be applied to the model.

Calculate frame loads which will be assigned to frame using "1A - LOAD APPLICATION AND PURLIN - Under 18m" file.



### NOTE

It will be easier to assign frame load when frames are not be divided into different length members.

# 4.1.3.3 Analyzing

In this part SAP2000 will assemble and solve the global matrix. The following steps are needed: Setting Analysis Option through: **Analysis > Set Option:** check the available DOFs. If you are analyzing a plane truss, check UX and UY, leave the UZ, RX, RY and RZ blank.

From the Analysis menu, select Run.Click the Run Now button to start running.

				Click to:
Case Name	Туре	Status	Action	Run/Do Not Run Case
DL LL	Linear Static Linear Static	Not Run Not Run	Run Bun	Show Case
WL1 WL2 WL3	Linear Static Linear Static Linear Static	Not Run Not Run Not Run	Run Run Run	Delete Results for Case
				Run/Do Not Run All
				Delete All Results
				Show Load Case Tree
nalysis Monitor O	ptions			Model-Alive
🔿 Always Show				Run Now
Never Show				

Figure 4 20 Set Load Cases to Run form

When the analysis is finished, the program automatically displays a deformed shape view of the model, and the model is locked. Check the general stability of model by clicking the Start Animation button at the lower right hand of monitor. If any member is deformed unusually, you should unlock model and revise it.

### 4.1.3.4 Display the result

- Displaying the deformed shape through: Display > Show Deformed Shape
- Return normal model through: F4 or the **Display > Show Undeformed Shape**
- Show Model Definition or Analysis Result through: Ctrl+T or the **Display > Show Table**

# 4.1.3.5 Design Steel Frame

Click the **Design menu > Steel Frame Design > View/Revise Preferences** command.

The Steel Frame Design Preferences form shown in figure below displays.

Item       Value         1       Design Code       AISC 360-10         2       Multi-Response Case Design       Envelopes         3       Framing Type       OMF         4       Seismic Design Category       D         5       Importance Factor       1.         6       Design System Rho       1.         7       Design System Cd       5.5         10       Design System Cd       5.5         11       Design System Cd       5.5         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stiffness Reduction Method       Tau-b Fixed         15       Omega[Rending]       1.67         16       Omega[Compression]       1.67         17       Omega[Shear]       1.67         18       Omega[Consion				Item Description
2       Multi-Response Case Design       Envelopes         3       Framing Type       OMF         4       Seismic Design Category       D         5       Importance Factor       1.         6       Design System Rho       1.         7       Design System Sds       0.5         8       Design System Rho       3.         10       Design System Cd       5.5         11       Design System Cd       5.5         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stiffness Reduction Method       Tau-b Fixed         15       Omega[Compression)       1.67         17       Omega[Compression)       1.67         18       Omega[Compression)       1.67         20       Omega[Shear')       1.67         21       Omega[Shear'short Webed Rolled I)       1.5         21       Ignore Special Seismic Load?       No         23       Ignore Special Seismic Load?       No         24       Is Doubler Plate Plug-Welded?       No       V		ltem	Value 🔺	
3       Framing Type       OMF         4       Seismic Design Category       D         5       Importance Factor       1.         6       Design System Rho       1.         7       Design System Rho       0.5         8       Design System Omega0       3.         10       Design System Cd       5.5         11       Design Provision       ASD         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stiffness Reduction Method       Tau-b Fixed         15       Omega[Rending]       1.67         16       Omega[Compression]       1.67         17       Omega[Compression]       1.67         18       Omega[Shear]       1.67         20       Omega[Shear]       1.67         21       Ignore Secial Seismic Load?       No         23       Ignore Special Seismic Load?       No         24       Is Doubler Plate Plug-Welded?       No	1	Design Code	AISC 360-10	
4       Seismic Design Category       D         5       Importance Factor       1.         6       Design System Rho       1.         7       Design System Rho       0.5         8       Design System Omega0       3.         10       Design System Cd       5.5         11       Design Provision       ASD         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stiffness Reduction Method       Taub Fixed         15       Omega(Tension-Vielding)       1.67         16       Omega(Tension-Fracture)       2.         19       Omega(Tension)       1.67         21       Ignore Seismic Code?       No         23       Ignore Seismic Code?       No         24       Is Doubler Plate Plug-Welded?       No       V	2	Multi-Response Case Design	Envelopes	
5       Importance Factor       1.         6       Design System Rho       1.         7       Design System Rho       0.5         8       Design System Omega0       3.         10       Design System Cd       5.5         11       Design System Cd       5.5         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stiffness Reduction Method       Tau-b Fixed         15       Omega(Compression)       1.67         16       Omega(Tension-Fracture)       2.         19       Omega(Shear-Short Webed Rolled I)       1.5         21       Ignore Seismic Code?       No         23       Ignore Special Seismic Load?       No         24       Is Doubler Plate Plug-Welded?       No         24       Is Doubler Plate Plug-Welded?       No       V	3	Framing Type	OMF	
6       Design System Rho       1.         7       Design System Sds       0.5         8       Design System R       8.         9       Design System Cd       3.         10       Design System Cd       5.5         11       Design System Cd       5.5         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stifness Reduction Method       Taub Fixed         15       Omega(Compression)       1.67         16       Omega(Tension/Yielding)       1.67         19       Omega(Shear)       1.67         20       Omega(Shear)       1.67         20       Omega(Shear)       1.67         21       Ignore Sismic Code?       No         23       Ignore Special Seismic Load?       No         24       Is Doubler Plate Plug-Weided?       No       V	4	Seismic Design Category	D	
7       Design System Sds       0.5         8       Design System R       8.         9       Design System Cherga0       3.         10       Design System Cd       5.5         11       Design Provision       ASD         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stiffness Reduction Method       Tau-b Fixed         15       Omega[Compression]       1.67         16       Omega[Compression]       1.67         17       Omega[Tension-Yielding]       1.67         18       Omega[Tension-Yielding]       1.67         20       Omega[Shear-Short Webed Rolled I]       1.57         21       Omega[Torsion)       1.67         22       Ignore Special Seismic Load?       No         23       Ignore Special Seismic Load?       No         24       Is Doubler Plate Plug-Weided?       No       V         8       Design Seismic Load?       No       Elack: Not a Default Value	5	Importance Factor	1.	
8       Design System R       8.         9       Design System Omega0       3.         10       Design System Cd       5.5         11       Design Provision       ASD         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stiffness Reduction Method       Taub Fixed         15       Omega(Compression)       1.67         16       Omega(Compression)       1.67         17       Dmega(Tension-Fracture)       2.         19       Omega(Tension-Fracture)       2.         20       Omega(Torsion)       1.67         21       Dargea(Shear-Short Webed Rolled I)       1.5         22       Ignore Seismic Code?       No         23       Ignore Special Seismic Load?       No         24       Is Doubler Plate Plug/Welded?       No       V         8       Blue:       Default Value       Black: Not a Default Value	6	Design System Rho	1.	
9       Design System Omega0       3.         10       Design System Cd       5.5         11       Design System Cd       5.5         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stiffness Reduction Method       Tau-b Fixed         15       Omega(Compression)       1.67         16       Omega(Compression)       1.67         17       Omega(Shear)       1.67         20       Omega(Shear)       1.67         21       Omega(Consion)       1.67         22       Ignore Seismic Code?       No         23       Ignore Special Seismic Load?       No         24       Is Doubler Plate Plug-Welded?       No         24       Is Doubler Plate Plug-Welded?       No       Isonal Value	7	Design System Sds	0.5	
10       Design System Cd       5.5         11       Design System Cd       5.5         11       Design Provision       ASD         12       Analysis Method       Direct Analysis         13       Second Order Method       General 2nd Order         14       Stiffness Reduction Method       Taub Fixed         15       Omega(Bending)       1.67         16       Omega(Compression)       1.67         17       Omega(Tension-Yrelding)       1.67         19       Omega(Shear)       1.67         20       Omega(Shear)       1.67         21       Jonore Sismic Code?       No         22       Ignore Sismic Code?       No         23       Ignore Sismic Load?       No         24       Is Doubler Plate Plug-Welded?       No       V	8	Design System R	8.	
11     Design Provision     ASD       12     Analysis Method     Direct Analysis       13     Second Order Method     General 2nd Order       14     Stiffness Reduction Method     Tau-b Fixed       15     Omega[Rending]     1.67       16     Omega[Compression]     1.67       17     Omega[Tension-Yielding]     1.67       20     Omega[Shear-Short Webed Rolled I]     1.57       21     Omega[Shear-Short Webed Rolled I]     1.67       22     Ignore Special Seismic Load?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No	9	Design System Omega0	3.	
12     Analysis     Method     Direct Analysis       13     Second Order     General 2nd Order       14     Stiffness Reduction Method     Tau-b Fixed       15     Omega(Bending)     1.67       16     Omega(Compression)     1.67       17     Omega(Tension-Fracture)     2.       19     Omega(Shear)     1.67       20     Omega(Shear)     1.67       21     Dargea(Shear-Short Webed Rolled I)     1.5       22     Ignore Seismic Code?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No	10	Design System Cd	5.5	
13     Second Order Method     General 2nd Order       14     Stiffness Reduction Method     Tau-b Fixed       15     Omega(Bending)     1.67       16     Omega(Compression)     1.67       17     Omega(Tension-Fracture)     2.       19     Omega(Shear-Short Webed Rolled I)     1.5       21     Omega(Torsion)     1.67       22     Ignore Seismic Code?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No	11	Design Provision	ASD	
14     Stiffness Reduction Method     Taub Fixed       15     Omega(Bending)     1.67       16     Omega(Compression)     1.67       17     Omega(Tension-Fracture)     2.       18     Omega(Tension-Fracture)     2.       19     Omega(Tension-Fracture)     2.       20     Omega(Tension)     1.67       21     Ignore Special Seismic Load?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No	12	Analysis Method	Direct Analysis	
15       Omega(Bending)       1.67         16       Omega(Compression)       1.67         17       Omega(Tension-Yrelding)       1.67         18       Omega(Tension-Fracture)       2.         19       Omega(Shear-Short Webed Rolled I)       1.67         20       Omega(Shear-Short Webed Rolled I)       1.57         21       Omega(Torsion)       1.67         22       Ignore Seismic Code?       No         23       Ignore Special Seismic Load?       No         24       Is Doubler Plate Plug-Welded?       No       V	13	Second Order Method	General 2nd Order	
16     Omega(Compression)     1.67       17     Omega(Tension-Yielding)     1.67       18     Omega(Shear)     1.67       20     Omega(Shear-Short Webed Rolled I)     1.67       21     Omega(Torsion)     1.67       22     Ignore Seismic Code?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No	14	Stiffness Reduction Method	Tau-b Fixed	
17     Omega[Tension-Yielding]     1.67       18     Omega[Tension-Fracture]     2.       19     Omega[Shear]     1.67       20     Omega[Shear-Short Webed Rolled I)     1.5       21     Omega[Tonsion]     1.67       22     Ignore Seismic Code?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No	15	Omega(Bending)	1.67	
18     Omega[Tension-Fracture)     2.       19     Omega[Shear]     1.67       20     Omega[Shear-Short Webed Rolled I)     1.5       21     Omega[Consion]     1.67       22     Ignore Seismic Load?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No	16	Omega(Compression)	1.67	
19       Omega[Shear]       1.67         20       Omega[Shear-Shott Webed Rolled I)       1.5         21       Omega[Torsion)       1.67         22       Ignore Seismic Code?       No         23       Ignore Special Seismic Load?       No         24       Is Doubler Plate Plug-Welded?       No         Blue: Default Value	17	Omega(Tension-Yielding)	1.67	
20     Omega[Shear-Short Webed Rolled I)     1.5       21     Omega[Torsion)     1.67       22     Ignore Seismic Code?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No	18	Omega(Tension-Fracture)		
21     Omega(Torsion)     1.67       22     Ignore Seismic Code?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No    Blue: Default Value Black: Not a Default Value				
22     Ignore Seismic Code?     No       23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No	20	Omega(Shear-Short Webed Rolled I)		
23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No         Blue: Default Value   Black: Not a Default Value				
23     Ignore Special Seismic Load?     No       24     Is Doubler Plate Plug-Welded?     No         Blue:     Default Value   Black: Not a Default Value	22	Ignore Seismic Code?	No	Evolution of Color Coding for Values
24 is Doubler Plate Hug-weided? No Black: Not a Default Value				
et To Default Values Black: Not a Default Value Black: Not a Default Value	24	Is Doubler Plate Plug-Welded?	No 👻	Blue: Derault Value
All Items Selected Items All Items Selected Items the current session				Red: Value that has changed durin

Figure 4 21 Steel Frame Design Preferences form

Make sure that the Design Code is set to **AISC 360-10**, the Framing Type is set to **OMF**(ordinary-moment-frame) and the Design Provision is set to **ASD**.

Review the information contained in the other items and then click OK to accept the selections.

Click the **Design menu > Steel Frame Design > Select Design Combos** command to access the Design Load Combinations Selection form. Uncheck **Automatically Generate Code-Based Design Load Combinations**, then click OK to accept the changes.

Des	ign Load Combinations Selection	
	Load Combinations for Design	1
	Select Type of Design Load Combination	
	Load Combination Type Strength	
	Select Load Combinations	
	List of Load Combinations Design Load Combinations	
	CV         CV1           CV2         Add ->           CV3         COMB1           ENV         <-Remove	
	Automatic Design Load Combinations	
	Automatically Generate Code-Based Design Load Combinations	
	Set Automatic Design Load Combination Data	
	OK Cancel	

Figure 4 22 Design Load Combinations Selection form

Click the **Design menu > Steel Frame Design > Start Design / Check of Structure** command to start the steel frame design process. When the design is complete, stress ratio are displayed on the model as a color pallet.

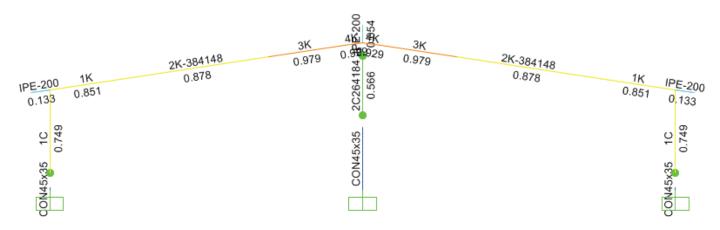
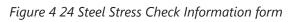


Figure 4 23 Model after checking

Click the **Design menu > Steel Frame Design > Display Design Info** command to show **Display Steel Design Results (AISC 360-10)** form. In the **Design Output** drop-down list, select P-M Ratio Colors & Values, then click OK to accept.

To see detail information of checking strength, right click on the **1K** member to show the **Steel Stress Check Information** form.

Frame ID Design Code	148 AISC 360-1	0			lysis Sectio ign Sectio		1К  1К			
COMBO ID	STATION /· LOC	MOMEN RATIO			ION CHE B-MAJ			AJ-SHR RATIO	-MIN-SHR- RATIO	1
										_
COMB1	0.00	0.851(C)							0.000	^
COMB1	3.03	0.386(C)								
COMB1		0.064(C)								
COMB1	5.06	0.075(C)	= 0.060	) +	0.015	+ 0.	000	0.126	0.000	
COMB1	6.07	0.260(C)	= 0.068	3 +	0.192	+ 0.	000	0.098	0.000	
COMB2	0.00	0.247(C)	= 0.013	3 +	0.234	+ 0.	000	0.102	0.000	Ŧ
-	ow Overwrites Inwrites	- Display D	etails for S	eleci Deta					omplete Deta	ils-



ISC 360-10 STEEL	SECTION CHECK	(Summaru	For Combo	and Station)			Units KN, m, C	-
Jnits : KN, m, 0		(Summary		and station)				
rame : 148	X Mid: 3.000	Combo:			ype: Brace			
ength: 6.067 .oc : 1.517	Y Mid: 8.000 Z Mid: 8.450	Shape:	1K Non-Compa	Frame Typ	pe: UMF Rot: 0.000 (	dagwaac		
.00 . 1.517	2 HIU. 0.490	61855.	non-compa	re reinchi i	NUC. 0.000 (	uegrees		
rovision: ASD	Analysis: Dir							
/C Limit=1.000	2nd Order: Ge			Reduction: Tau				
1phaPr/Py=0.024	AlphaPr/Pe=0.	057 Tau_b=	1.000	EA factor=0.80	0 EI facto	or=0.800		
)megaB=1.670	OmegaC=1.670	OmegaT	Y=1.670	OmegaTF=2.000				
)megaV=1.670	OmegaV-RI=1.5		T=1.670					
1=0.005 1=0.000	I 33=2.402E-04 I 22=5.152E-06			S33=8.135E-04 S22=6.283E-05	Av3=0.0			
= 0.000	fu=345000.000			233=0.001	Cw=0.00			
RLLF=1.000	Fu=448000.000			z22=9.753E-05				
TRESS CHECK FOR	ES & MOMENTS	Combo COMP4						
Location	Pr	Mr33	/ Mr22	Ur2	Ur3	Tr		
1.517	-26.092	-160.279	3.700E-05	-40.525 9	.069E-04 -	1.846E-05		
MM DEMAND/CAPACI D/C Ratio:	TY RATIO (H1 0.851 = 0.037	-1b)	000					
D/G NALIU:								
	- (1/2/(	11/10/ . (1	r33/MC33)	+ (Mr22/Mc22)				
XIAL FORCE & BIA		SIGN (H1-	1b)					
Factor	NXIAL MOMENT DE	SIGN (H1- K1	1b) K2	B1	B2	Cm		
Factor Major Bending	IXIAL MOMENT DE	SIGN (H1- K1 1.000	1b) K2 1.000	B1 1.000	1.000	1.000		
Factor	IXIAL MOMENT DE	SIGN (H1- K1	1b) K2	B1				
Factor Major Bending	NXIAL MOMENT DE 4.167 0.250 L1tb	SIGN (H1- K1 1.000 1.000 K1tb	1b) K2 1.000 1.000 Cb	B1 1.000	1.000	1.000		
Factor Major Bending	NXIAL MOMENT DE 1 4.167 1 0.250	SIGN (H1- K1 1.000 1.000	1b) K2 1.000 1.000	B1 1.000	1.000	1.000		
Factor Major Bending Minor Bending	XIAL MOMENT DE L 1 4.167 3 0.250 L1tb 0.250	SIGN (H1- K1 1.000 1.000 K1tb 1.000	1b) K2 1.000 1.000 Cb 1.343	B1 1.000	1.000	1.000		
Factor Major Bending Minor Bending	NXIAL MOMENT DE L J 4.167 J 0.250 Lltb 0.250 Pr	SIGN (H1- K1 1.000 1.000 Kltb 1.000 Pnc/Omega	1b) K2 1.000 1.000 1.343 Pnt/Omega	B1 1.000	1.000	1.000		
Factor Major Bending Minor Bending	XIAL MOMENT DE L 1 4.167 3 0.250 L1tb 0.250	SIGN (H1- K1 1.000 1.000 K1tb 1.000	1b) K2 1.000 1.000 Cb 1.343	B1 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB	XXIAL MOMENT DE L J 4.167 J 0.250 L1tb 0.250 Pr Force -26.092	SIGN (H1- K1 1.000 1.000 K1tb 1.000 Pnc/Omega Capacity 353.527	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000	B1 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB	NXIAL MOMENT DE L 1 4.167 9.250 L1tb 9.250 Pr Force -26.092	SIGN (H1- K1 1.000 K1tb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000 Mn/Omega	B1 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB Axial	NXIAL MOMENT DE L 4.167 0.250 L1tb 0.250 -26.092 -26.092 Moment	SIGN (H1- 1.000 KItb 1.000 RItb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega Capacity	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000 Mn/Omega No LTB	B1 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB	NXIAL MOMENT DE L 1 4.167 9.250 L1tb 9.250 Pr Force -26.092	SIGN (H1- K1 1.000 K1tb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000 Mn/Omega	B1 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB Axial Major Moment Minor Moment	NXIAL MOMENT DE L 4.167 9 0.250 L1tb 0.250 Pr Pr -26.092 Moment -160.279	SIGN (H1- 1.000 KItb 1.000 KItb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega Capacity 196.790	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000 Mn/Omega No LTB	B1 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB Axial Major Moment	NXIAL MOMENT DE L 4.167 0.250 L1tb 0.250 -26.092 -26.092 Moment -160.279 3.700E-05	SIGN (H1- K1 1.000 K1tb 1.000 K1tb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega Capacity 196.790 13.717	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000 Mn/Omega No LTB 196.790	B1 1.000 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB Axial Major Moment Minor Moment	NXIAL MOMENT DE L 4.167 9.250 L1tb 9.250 Pr -26.092 Moment -160.279 3.700E-05	SIGN (H1- 1.000 KLtb 1.000 KLtb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega Capacity 196.790 13.717 Un/Omega	1b) K2 1.000 1.000 1.343 Pnt/Onega Capacity 1035.000 Mn/Onega No LTB 196.790 Stress	B1 1.000 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB Axial Major Moment Minor Moment	NXIAL MOMENT DE L 4.167 0.250 L1tb 0.250 -26.092 -26.092 Moment -160.279 3.700E-05	SIGN (H1- K1 1.000 K1tb 1.000 K1tb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega Capacity 196.790 13.717	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000 Mn/Omega No LTB 196.790	B1 1.000 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB Axial Major Moment Minor Moment	NXIAL     MOMENT     DE       1     4.167       0.250     167       0.250     9.250       0.250     9.250       Force     9.250       -26.092     9.279       3.700E     0.279       3.700E     0.279	SIGN (H1- K1 1.000 KItb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega Capacity 196.790 13.717 Un/Omega Capacity	1b) K2 1.000 1.000 Cb 1.343 Pnt/Onega Capacity 1035.000 Mn/Onega No LTB 196.790 Stress Ratio	B1 1.000 1.000	1.000	1.000		
Factor Major Bending Minor Bending LTB Axial Major Moment Minor Moment SHEAR CHECK Major Shear Minor Shear	XIAL MOMENT DE L A 167 0.250 L1tb 0.250 Pr Pr Pr Pr Pr Pr Pr Pr Pr Pr	SIGN (H1- K1 1.000 1.000 K1tb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega Capacity 196.790 13.717 Un/Omega Capacity 128.123	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000 Mn/Omega No LTB 196.790 Stress Ratio 0.316	B1 1.000 1.000 	1.000	1.000		
Factor Major Bending Minor Bending LTB Axial Major Moment Minor Moment HEAR CHECK Major Shear Minor Shear	NXIAL     MOMENT     DE       L     L       J     0.250       L     0.250       Force       -26.092       Moment       Moment       Moment       Moment       Moment       0.250       Pr       Force       -26.092       Moment       -160.279       3.700E       0.2525       9.060E       0.000E       9.060E       0.000E	SIGN (H1- K1 1.000 KItb 1.000 Pnc/0mega Capacity 353.527 Mn/0mega Capacity 196.790 13.717 Un/0mega Capacity 128.123 284.594	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000 Mn/Omega No LTB 196.790 Stress Ratio 0.316	B1 1.000 1.000 	1.000	1.000		
Factor Major Bending Minor Bending LTB Axial Major Moment Minor Moment SHEAR CHECK Major Shear	XIAL MOMENT DE L A 167 0.250 L1tb 0.250 Pr Pr Pr Pr Pr Pr Pr Pr Pr Pr	SIGN (H1- K1 1.000 1.000 K1tb 1.000 Pnc/Omega Capacity 353.527 Mn/Omega Capacity 196.790 13.717 Un/Omega Capacity 128.123	1b) K2 1.000 1.000 1.343 Pnt/Omega Capacity 1035.000 Mn/Omega No LTB 196.790 Stress Ratio 0.316	B1 1.000 1.000 	1.000	1.000		

Figure 4 25 Steel Stress Check Data form

Click the Details button to show more detail information.



### NOTE

Unbraced length ratio for major bending is 4.167 which is taken to be distance between two bracing point in major axis (25m) divided by actual length of member (6m). You should carefully review this ratio to ensure that the design process is consistent with your expectations.

To check whether all steel frames passed the stress check, click the **Design menu > Steel Frame Design > Verify All Members Passed** command to display as figure below.

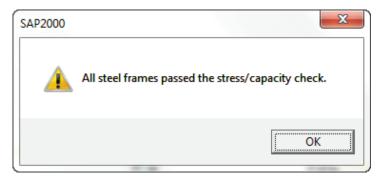


Figure 4 26 Checking capacity of all steel objects

### 4.1.3.6 Check Deflection Limitt

Refer Section 2.4 Serviceability Consideration for more details.

To optimize weight of building, unlock model, modify section and irritate above steps until total weight is smallest.

# 4.1.4. Export Joint Reaction

# 4.1.4.1 Transform from wind load with return period wind speed of 700 years to wind load with those of 50 years

In Load Case Data – Linear Static form, change Scale Factor for all Windload Case from 1 to 0.625 as Figure 4 27.

$$\frac{V_T}{V_{50}} = 0.36 + 0.1 \ln(12T)$$
  
$$\Rightarrow V_{50} = \frac{V_{700}}{0.36 + 0.1 \ln(12 \times 700)} = 0.791 V_{700} \Rightarrow WL_{50} = 0.791^2 \times WL_{700}^2 = 0.626 \times WL_{700}^2$$

Load Case Name         Notes           WL1         Set Def Name         Modify/Show	Static Design
Stiffness to Use     Zero Initial Conditions - Unstressed State     Stiffness at End of Nonlinear Case     Important Note: Loads from the Nonlinear Case are NOT included     in the current case	Analysis Type C Linear C Nonlinear C Nonlinear Staged Construction
Load SApplied Load Type Load Name Scale Factor Load Patterr ▼ WL1 ▼ .625 Load Pattern WL1 .625 Add [Modify] Delete	Mass Source MSSSRC1 DK Cancel

Figure 4 27 Load Case Data – Linear Static form

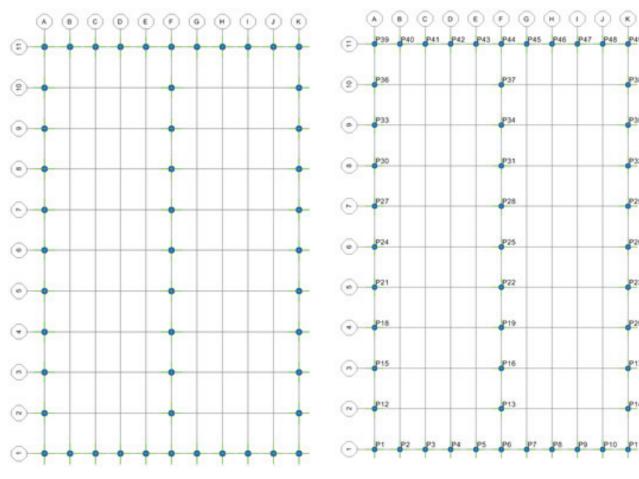
# 4.1.4.2 Changing label name of joints

Select the Edit > Change Labels command to access Interactive Name Change form. After enter all the information as figure below, select the Edit > Auto Relabel > All In List command to change label name.

[ Lho	ose A Nam	ed Item Type				
Ite	m Type	Element La	abels - J	oint		-
•	List Name	s of Selected Elem	nents Or	nly		
Auto	o Relabel C	ontrol				
Pre	efix	P	Fir	st Relabel Order	Y	-
Ne	ext Number	1	Se	econd Relabel Order	×	•
Inc	crement	1	Mi	inimum Number Digits	0	
Nam	ne List for E	lement Labels - Joi	int —			
Nam	ne List for E	lement Labels - Joi Current Name	int —	New Na	me	<b>_</b>
- Nam		Current Name 2	int	New Na	me	<b>^</b>
	1	Current Name	int —		me	<b>^</b>
	2	Current Name 2	int —	P1	me	<b>^</b>
1	2	Current Name 2 5	int	P1 P2	me	<b>^</b>
1	1 2 3 4	Current Name 2 5 8	int	P1 P2 P3	me	<b>^</b>
1	2 3 5	Current Name 2 5 8 11	int	P1 P2 P3 P4	me	
1	1 2 3 4 5	Current Name 2 5 8 11 14	int	P1 P2 P3 P4 P5	me	<b>•</b>
1 2 3 4 6	2 3 4 5 7	Current Name 2 5 8 11 14 20	int	P1 P2 P3 P4 P5 P6	me	
1 2 3 4 6 7	2 3 4 5 5 7 3	Current Name 2 5 8 11 14 20 23	int	P1 P2 P3 P4 P5 P6 P7	me	
1 2 3 4 5 6 7 8	1           2           3           4           5           6           7           3           9	Current Name 2 5 8 11 14 20 23 26	int	P1 P2 P3 P4 P5 P6 P7 P8	me	

Figure 4 28 Change Label Form

P41 P42 P43 P44 P45 P46 P47 P48 P49



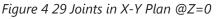


Figure 4 30 Label name after changing

P6

P3

P37

P34

P31

P28

P25

P22

P19

P16

P13

P38

P35

P32

P29

P26

P23

P20

P17

P14

P10 P11

**P7** 

### 4.1.4.3 Displaying the Joint Forces/Joint Reactions

### **Export Joint Reactions to Excel**

• When steel columns locate on concrete columns, specify all steel columns

and show table Element Joint Forces – Frames.

• When steel columns locate on foundations, specify all base joints

### and show table Joint Reactions.

### Create JOINT REACTIONS Results

Open file "7 - JOINT REACTIONS" and move to sheet "SteelColumn", and then enter all the information about the project and Joint Labels Plan.

In Excel file that has just appeared, filter all base joints. Then insert one column on the left of column "E". Copy all data from column "B" to column "K".

In sheet "InputS" in file "7 – JOINT REACTIONS":

- Delete all data in area from columns A to J and from rows 4 to the end of sheet.
- Select cell "A4" and paste data into this. Select cell "X6" and change content in this cell to "Wind load Y".
- Move to sheet "SteelColumn" and click the HIDE button.
- Set Print Area before print PDF file.

- Recheck all information; make sure that Vertical Reactions Vz for Dead load and Live load are always positive.

### 4.1.5. Design Example 4a: Low-Rise Building Sap Model

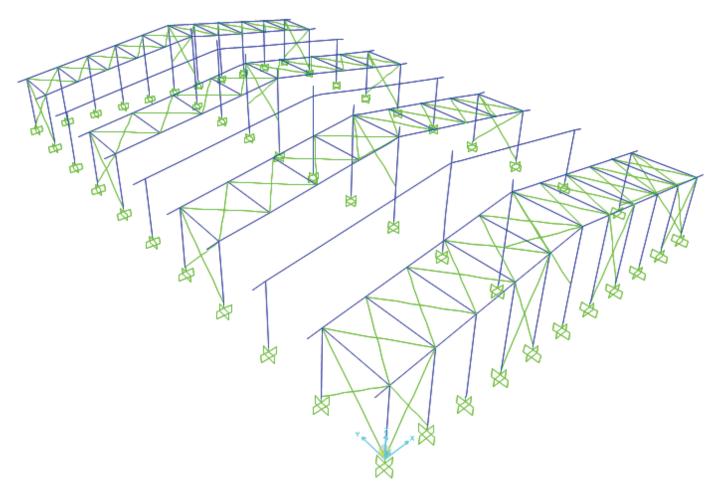


Figure 4 31 Low-rise building model

					Code of do	OC.	BM-7.2.1-0	2
		<u>PROJECT II</u>			Date of iss		24/10/2016	
BM8.		<u>FOR</u>	<u>M (PIF</u>	<u>)</u>	Times of is		04 Priority //	
Best Metal Building BMB & A J/SC	js.	(THÔNG TIN TF		AI DỰ AN)	Project sta Breakdow		· · · · · · · · · · · · · · · · · · ·	/ Customer
		PRO	<b>DJECT I</b>	NFORMATIO			Binbart	
Project name:	Zho	ong Yi Warehouse						
Location	Phn	om Penh, Camobodia						
	Royal Haskon	ing 🔲 Architype	П Ме	inhardt 🗌 🛛	3.I.S.T 🗌	Others	□	
Quotation no.: 148-1	7	Start date: 11/03/2	2017 Fin	ish date:16/0	3/2017	Buildi	ng no.:01	Area no.:4000
Requirements: 1.Nev	v Est.	/ 2.Arch.drwgs	/ 3.App	orl.drwgs 🗌 / 4	4.Design ca	alculatio	n 🗌 / 5.Sho	p drwgs 🗌
				DESCRIPTIO	N			
		MB Standard 🔳 / 2.						2
Material		t-up member: fy= 34				Galvaniz	ed 📕 fy= 4	5 kN/cm <sup>2</sup>
Specifications		chor bolt: 5.6 🔳 / 8		4. Connection				
_	Pair	nt/ Finishing: Alkyd		anized [] / Ep		ckness:	80 micron	
Exposure		B						
Qty. of identical Bldir	igs	01	Usage		Warehous	se		
Type of building		Multi span 1		lope (%)	15%			
Width (M)			10@5.0m O-O of brick wall					
Length (M)		80.0		oacing (M)	<u>10@8.0m</u>			
Height (M)	8.0		e Height			Clear	Height	
Type of End	Post & Beam				One, A	xis		Both
Frames	Main	Rigid Frame 🗌						Both 🗌
	Conci	ete Columns and rafters (by others)			One, Axis DBoth			Both 🗌
Type of Wall Bracing	Diago	onal Rod 🔳 / Cable	/ Por	tal frame 🗌 /	None 🗌			
Eave Condition	Gutte	r & Downspout 🔳 /	RC Gut	ter (by other)	/ Gutter	& Dowr	ispout (by o	ther) 🗌
Notes: Use Chinese	steel.	1						
		are installed at Lev			-			
		ns are installed or ors 5mx4m are be		-				
	-	luct 0.45mm; Dow				nn pass	1 column.	(See DWG)
	GN LO			BMB's design				
Live Load on Roof (k	N/m <sup>2</sup> )	0.57 (58.1 KG/r		-			-	
Live Load on Frame	/ (kN/m²	) 0.30 (30.6 KG/r	n <sup>2</sup> ) 📕 / C	Others				
Wind Speed (KM/h)	<b>`</b>	, 110 (30.6 M/s)						M/s) 🗌
Collateral Load (KG/	m²)		1	Concentrate				,
Collateral Load hang	,	rlin 🗌		Collateral Lo		,	burlin	
Notes: (if collateral load	is plast	er ceiling $\rightarrow$ describe h	ere also)					
		ROO	OF, WAL	L SHEETING	S			
Roof panel: Yes	Bare 2	Zinc-Alum AZ…		Pre-Painted	Zinc-Alum	AZ 50	7 ribs	] / 5 ribs 🔳
None		ainted Zinc-Alum Az	Z	Thickness: 0				] / Clip-lock 🗌
Wall panel: Yes 🔳		Zinc-Alum 🗌		Pre-Painted	Zinc-Alum	AZ150		- <u>-</u> –
None	Pre-P	ainted Zinc-Alum Az	Z	Thickness: 0	).40mm		5 ribs pa	
Notes: Use Local pa	anel; a	II trims and gutter	s are 0.	45mm thk.				

			OPE	N WALL CONDIT	ION	NS	
Both Side wall			Open 4.0 M for blo	ock wall, sheeted ι	ıp to	to roof.	
Both End wall			Open 4.0 M for blo	ock wall, sheeted ι	up to	to roof.	
<u>Notes:</u>							
			RO	OF MONITOR		None 🗌	
Туре		Jac	k roof 📕	Ridge vent		Other 🗌	
Overall width (	M)	2.0		Length (M)	(M) 64.0		
Eave Condition Curved Eave			ved Eave 🔳	Open Eave		Other 🗌	
Roof panel	Roof panel Same as roof panel (nominal)			_		thick mm	
Wall panel			ne as wall panel 📕 minal)	/ Other []		thick mm	
<u>Notes:</u>							
ROOF EXTER				OF EXTENSIONS	i	None 🗌	
Panel	Sam	e roof p	anel 📕/ Other 🗌			thick mm (nominal)	
Location	Side wall				End wall		
Soffit panel			Yes 🗌 / None 🛛			Yes 🗌 / None 🔳	
Quantity	02				2		
Width (M)	1.5			1.5		.5	
Length (M)	80.0			50.0		0.0	
<u>Notes:</u>							
	T			CANOPY None			
Panel		Same a (nomina		her 🗌		mm	
Eave Condition	ו	Curved	Eave	Open Eave		Other 🗌	
Location		E	Both Side wall	End wall (G	L.1 <sup>-</sup>	11) End wall (GL.1)	
Soffit panel		Ye	es 🗌 / None 🔳	Yes 🗌 / No	ne	Yes 🗌 / None 🔳	
Quantity		04		01		02	
Width (M)		2.5		2.5		2.5	
Length (M)		4.0		4.0		5.0	
Gutter, Downs	pout		None	None		None	
<u>Notes:</u>							
Project Eng.	ros		Notes for using this i 1. Must fill in Exact PIF 2. Must make sketch	<u>Country</u> of Project of		Name of <u>Consultancy/ Design Company</u> at top c rchitectural drawing.	

# 4.1.5.1 Defining

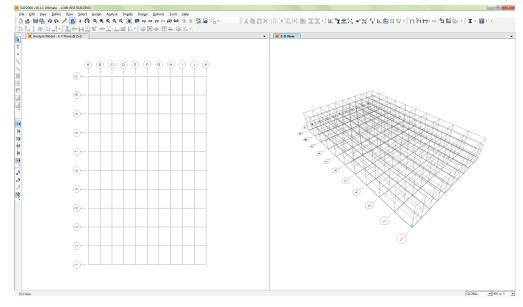


Figure 4 33 Setting up geometry and default units

# Define Section Properties as table below:

### Table 4 2 Frame Section Properties

TABLE: Frame	Section Prope	erties 01 - General						
SectionName	Material	Shape	d	bf1	tfl	tw	bf2	tf2
Text	Text	Text	m	m	m	m	m	m
1C		Nonprismatic						
1C-250212	STEEL3450	I/Wide Flange	0.25	0.212	0.008	0.005	0.212	0.008
1C-600212	STEEL3450	I/Wide Flange	0.6	0.212	0.008	0.005	0.212	0.008
1K		Nonprismatic						
1K-384164	STEEL3450	I/Wide Flange	0.384	0.164	0.006	0.005	0.164	0.008
1K-600164	STEEL3450	I/Wide Flange	0.6	0.164	0.006	0.005	0.164	0.008
2C264184	STEEL3450	I/Wide Flange	0.264	0.184	0.008	0.005	0.184	0.008
2K-384148	STEEL3450	I/Wide Flange	0.384	0.148	0.006	0.004	0.148	0.006
3K		Nonprismatic						
3K-384164	STEEL3450	I/Wide Flange	0.384	0.164	0.006	0.005	0.164	0.008
3K-600164	STEEL3450	I/Wide Flange	0.6	0.164	0.006	0.005	0.164	0.008
4K		Nonprismatic						
4K-600184	STEEL3450	I/Wide Flange	0.6	0.184	0.008	0.005	0.184	0.008
4K-825184	STEEL3450	I/Wide Flange	0.8	0.184	0.008	0.005	0.184	0.008
CDH	STEEL3450	I/Wide Flange	0.26	0.164	0.006	0.004	0.164	0.006
CON45x35	CON25	Rectangular	0.45	0.35				
EB	STEEL2350	Box/Tube	0.15	0.15	0.0028	0.0028		
IPE-200	STEEL2350	I/Wide Flange	0.2	0.1	0.008	0.0055	0.1	0.008
KDH	STEEL3450	I/Wide Flange	0.222	0.148	0.005	0.004	0.148	0.005

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# **4.1.5.2 Modeling** Create general model

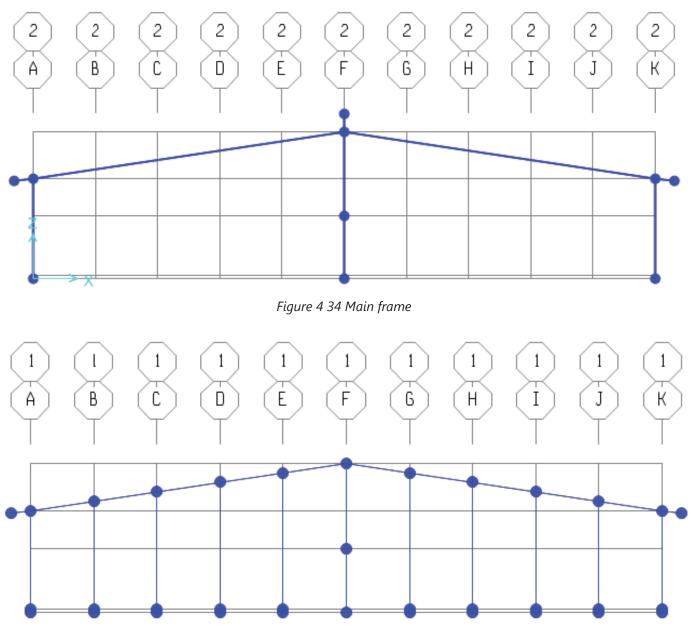
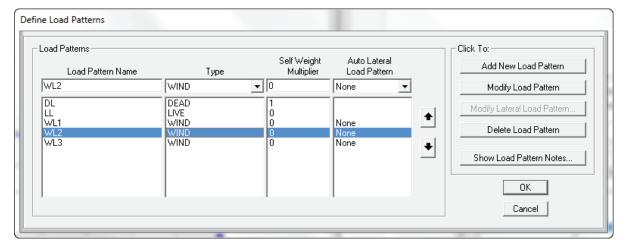


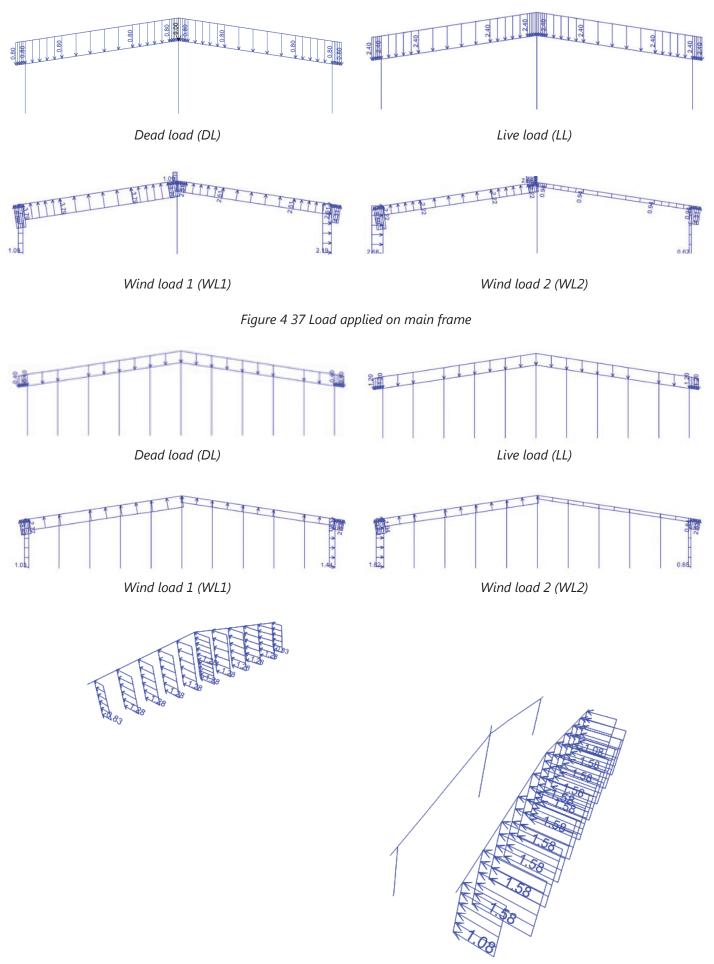
Figure 4 35 End wall frame



# Define load patterns and load combinations

Figure 4 36 Define Load Patterns form

# Assign load



Wind load 3 (WL3) Figure 4 38 Load applied on end-wall frame

### Divide frame objects

Draw vertical lines that are located at x=6m, x=17.5m, x=23.5m, x=26.5m, x=32.5m and x=44m as shown in figure below.

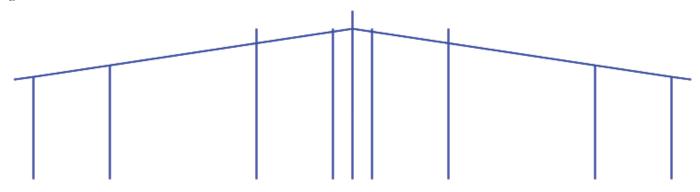


Figure 4 39 Vertical lines to divide rafter

Select rafters and vertical lines, then click the Edit > Edit Lines > Divide Frames command to show Divide Selected Frames form. In this form, make sure that Break at intersection with selected Joints, Frames, Area Edges and Solid Edges is selected before click the OK button.

ivide Selected Straight Frame Objects-			Units
C Divide into Specified Number of Fr	ames		KN, m, C 💌
Number of Frames			,
Last/First Length Ratio			
<ul> <li>Break at intersections with selecte</li> </ul>	d Joints, Frames, Area Edg	ges and Solid Edges	
O Divide at Specified Distance from I	-end of Frame		
Distance Type			
Distance			
O Divide at Intersection with a Coord	inate Plane in the Current	Coordinate System	
Coordinate Plane			
Intersection with Plane at			OK
O Divide at Intersection with Visible 0	arid Planes in the Current (	Coordinate System	Cancel
Grid Plane			

Figure 4 40 Divide Selected Frames form



Figure 4 41 Main frame after dividing

Assign member section, end releases and restraint

Figure 4 42 End wall frame

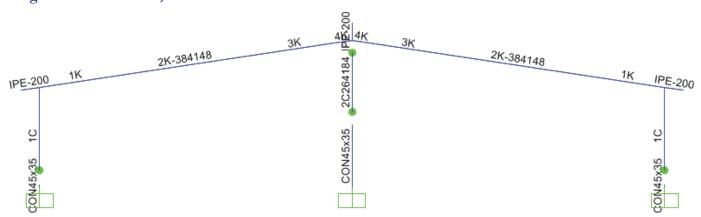


Figure 4 43 Main frame sections

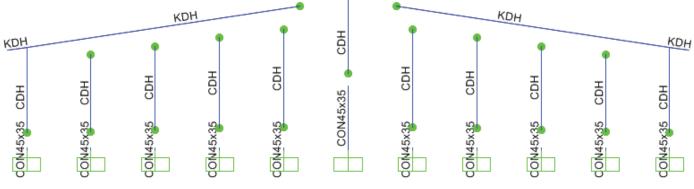


Figure 4 44 End wall frame sections

### Calculate ULR for each member in frame

- Assume purlin spacing and girt spacing are 1.5m.

- With rafters: unbraced length for 1K, 3K and 4K are 1.5m, for 2K-384148 are 3m and for KDH, IPE-200 are their length (no bracing)

- With outer columns: unbraced length for 1C and CDH are 5m (base on architectural drawing).

- With inner columns (2C264184): there are not any ST or portal frame, so unbraced length for this column is taken to be its length or ULR is 1.

Member	Length (m)	Unbraced Length (m)	ULR
1K	6.00	1.50	0.25
2K-384148	11.50	3.00	0.26
ЗК	6.00	1.50	0.25
4K	1.50	1.30	0.87
1C	7.80	5.00	0.64
2C264184	6.75	6.75	1.00
IPE-200	1.50	1.50	1.00
CDH	7.80	5.00	0.64
KDH	25.28	25.28	1.00

#### Table 4 3 Unbraced length ratio

Select 1K section members and click the **Design > Steel Frame Design > View/Revise Overwrites** command to show **the Steel Frame Design Overwrites** for AISC 360-10 form.

At the Value column, type 0.25 for both Unbraced Length Ratio (Minor) and Unbraced Length Ratio (LTB).

			Item Description
	Item	Value 🔺	Unbraced length factor for lateral-
1	Current Design Section	Program Determined	torsional buckling for the frame object. This item is specified as a
	Framing Type	Program Determined	fraction of the frame object
	Omega0	Program Determined	length.Multiplying this factor times the
4	Consider Deflection?	No	frame object length gives the unbraced
5	Deflection Check Type	Program Determined	length for the object.Specifying 0 means the value is program determined.
6	DL Limit, L /	Program Determined	inicans the value is program determined.
7	Super DL+LL Limit, L /	Program Determined	
8	Live Load Limit, L /	Program Determined	
9	Total Limit, L/	Program Determined	
10	TotalCamber Limit, L/	Program Determined	
11	DL Limit, abs	Program Determined	
12	Super DL+LL Limit, abs	Program Determined	
13	Live Load Limit, abs	Program Determined	
14	Total Limit, abs	Program Determined	
15	TotalCamber Limit, abs	Program Determined	
16	Specified Camber	Program Determined	
17	Net Area to Total Area Ratio	Program Determined	
18	Live Load Reduction Factor	Program Determined	
19	Unbraced Length Ratio (Major)	Program Determined	
20	Unbraced Length Ratio (Minor)	0.25	
	Unbraced Length Ratio (LTB)	0.25	
	Effective Length Factor (K1 Major)	Program Determined	Explanation of Color Coding for Values
	Effective Length Factor (K1 Minor)	Program Determined	
24	Effective Length Factor (K2 Major)	Program Determined 🗸	Blue: All selected items are program determined
et To	Prog Determined (Default) Values	Reset To Previous Values	Black: Some selected items are user defined
A	Il Items Selected Items	All Items Selected Items	Red: Value that has changed during the current session

Figure 4 45 Steel frame design overwrites form

Repeat these steps to assign ULR for the other member. **Replicate Frame** 

Select frame at X-Z Plan @ Y=8 and then select the Edit > Replicate.

To generate 8 additional frames with 8m bay spacing, enter like below:

Replicate	
Linear Radial	Mirror
Increments	Replicate Options
dx 0.	Modify/Show Replicate Options
dy 8	12 of 12 active boxes are selected
dz 0.	🗖 Delete Original Objects
Increment Data	
Number 8	
	Cancel

Figure 4 46 Replicate form

# Draw Bracing System

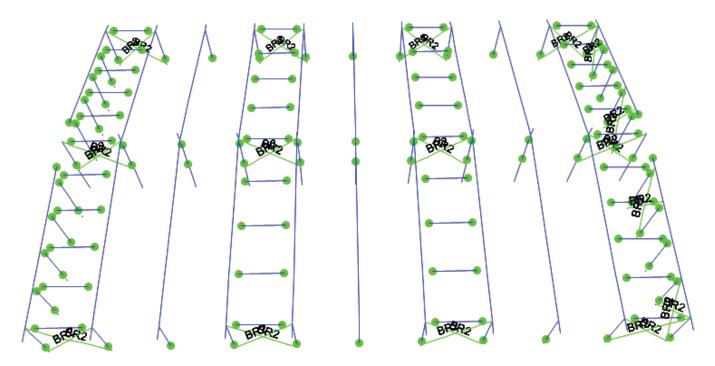


Figure 4 47 Column brace system

# 4.1.5.3 Analyzing

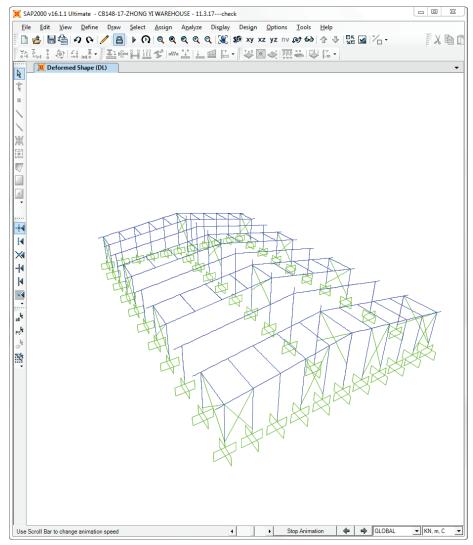


Figure 4 48 Model after running analysis

DESIGN GUIDELINES

# 4.1.5.4 Design Steel Frame

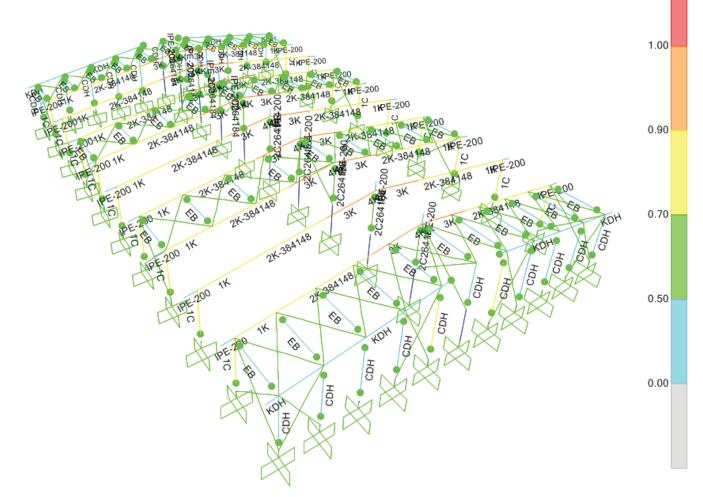


Figure 4 49 Model after checking

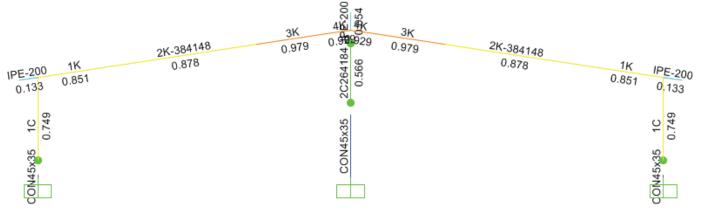


Figure 4 50 Main frame at plane Y=8

To check whether all steel frames passed the stress check, click the **Design menu > Steel Frame Design > Verify All Members Passed** command to display as figure below.

SAP2000	×
<u>^</u>	All steel frames passed the stress/capacity check.
	ОК

Figure 4 51 Checking capacity of all steel objects

# 4.1.5.5 Check Deflection Limitation

Create combos for deflection checking:

Name	LCombination
CV1	DL + 0.4224WL1
CV2	DL + 0.4224WL2
CV3	DL + 0.4224WL3
CV	CV1 + CV2 + CV3

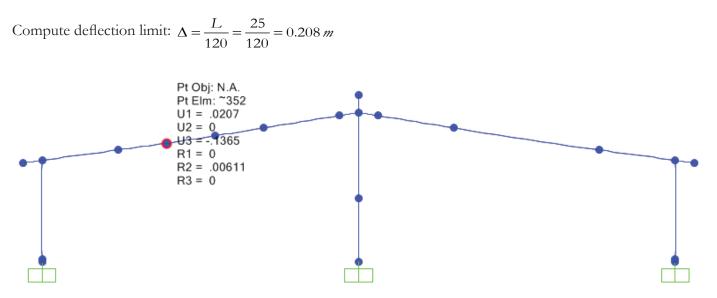
#### **Vertical Deflection**

Click the **Display > Show Deformed Shape** command which will show a Deformed Shape form. Select **COMB1** (**DL + LL**) from **Case/Combo Name** drop-down list and click OK to accept.

Deformed Shape	
Case/Combo	
Case/Combo Name	COMB1
Multivalued Options	
C Envelope (Max or Min)	
Step	
Scaling	
<ul> <li>Auto</li> </ul>	
C Scale Factor	
Dptions	
🔲 Wire Shadow	OK
Cubic Curve	Cancel

Figure 4 52 Deformed Shape form

Checking if vertical deflection (U3) of mid points of rafter are excess deflection limit. Put the cursor at mid points of rafter to see if deflection of that point is excess the limit or not.



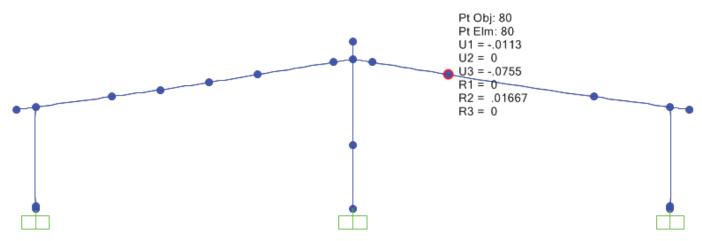


Figure 4 54 Deflection of point 113 – 0.0755m

### **Horizontal Deflection**

Checking if horizontal deflection of combination  $\mathbf{CV}$  of top of columns are excess deflection limit. Put the cursor at points located at the top of column to see if the horizontal deflection (U1 or U2) of that point is excess the limit or not.

Compute horizontal deflection limit:

$$\Delta = \frac{h}{60} = \frac{8}{60} = 0.133 \, m$$

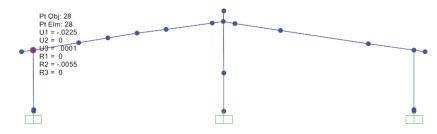


Figure 4 55 Deflection of point 28 – 0.0225m

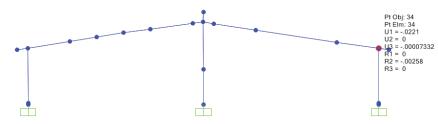


Figure 4 56 Deflection of point 34 – 0.0221m

### 4.1.5.6 Export joint reaction

Transform from wind load with return period wind speed of 700 years to wind load with those of 50 years

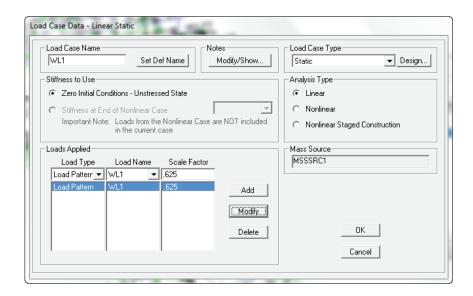
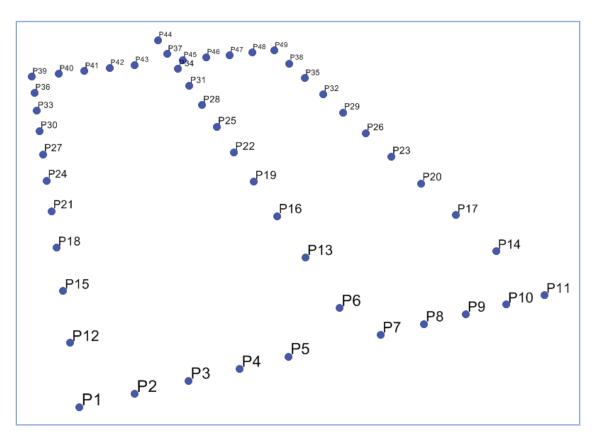


Figure 4 57 Load Case Data – Linear Static form



#### Figure 4 58 Labels after change

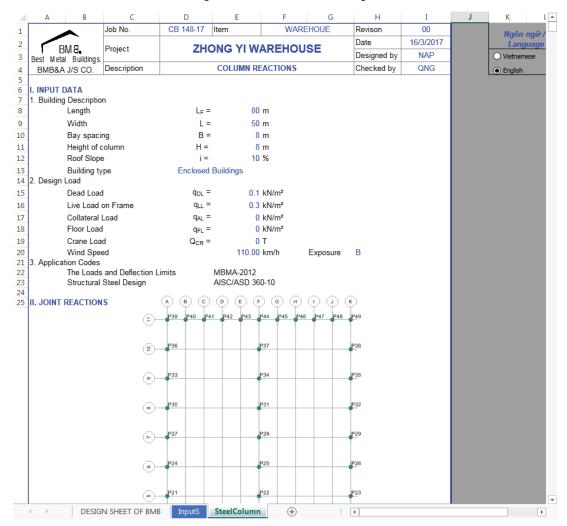


Figure 4 59 Column reactions file

: As Note		Filter-Sort	Select Options		Element	Joint Forces - F	rames		
	_				,				
	ame ext	Joint Text	OutputCase Text	CaseType Text	F1 KN	F2 KN	F3 KN	M1 KN-m	M2 KN-m
í í	24	P15	DL	LinStatic	6.049	0	18.24	0	0
í.	24	65	DL	LinStatic	-6.049	0	-14.976	0	47.1819
2	24	P15	LL	LinStatic	13.418	0	32.929	0	0
í.	24	65	LL	LinStatic	-13.418	0	-32.929	0	104.6626
í.	24	P15	WL1	LinStatic	-16.6	0	-35.755	0	0
2	24	65	WL1	LinStatic	11.281	0	35.755	0	-108.7342
í.	24	P15	WL2	LinStatic	-15.227	0	-23.277	0	0
2	24	65	WL2	LinStatic	2.269	0	23.277	0	-68.237
	24	P15	WL3	LinStatic	0	0	0	0	0
- i	24	65	WL3	LinStatic	0	0	0	0	0
	28	P17	DL	LinStatic	-6.049	0	18.24	0	0
	28	78	DL	LinStatic	6.049	0	-14.976	0	-47.1819
2	28	P17	LL	LinStatic	-13.418	0	32.929	0	0
	28	78	LL	LinStatic	13.418	0	-32.929	0	-104.6626
	28	P17	WL1	LinStatic	2.835	0	-22.773	0	0
2	28	78	WL1	LinStatic	-13.506	0	22.773	0	63.7311
	28	P17	WL2	LinStatic	0.091	0	-9.624	0	0
2	28	78	WL2	LinStatic	-3.124	0	9.624	0	12.5392
í.	28	P17	WL3	LinStatic	0	0	0	0	0
	28	78	WL3	LinStatic	0	0	0	0	0
									•

Figure 4 60 Element Joints Forces – Frame table

Α	В	С	D	E	F	G	н	I	J	K	L
TABLE: EI	ement Join	t Forces - Fra	ames								
Frame	Joint 📧	OutputCa 🔻	CaseTy 💌	-	F1 💌	F2 💌	F3 💌	M1 💌	M2 💌	M3 💌	FrameEle
39	P1	DL	LinStatic		0.046	4.454E-08	4.103	0	0	0.000004456	i 39-1
39	P1	LL	LinStatic		0.094	-1.381E-07	3.52	0	0	0.00001021	39-1
39	P1	WL1	LinStatic		-2.29	3.974E-07	-5.041	0	0	-0.00001329	39-1
39	P1	WL2	LinStatic		-3.892	1.578E-07	-4.642	0	0	-0.000006482	39-1
39	P1	WL3	LinStatic		-0.00693	-2.62	-9.232	0	0	-0.000945	39-1
45	P2	DL	LinStatic		8.674E-18	6.306E-18	5.427	0	0	0.000003164	45-1
45	P2	LL	LinStatic		1.995E-17	3.163E-18	5.867	0	0	0.000007239	45-1
45	P2	WL1	LinStatic		-3.816E-17	-1.155E-17	-9.038	0	0	-0.000009329	45-1
45	P2	WL2	LinStatic		1.18E-16	-6.739E-18	-6.419	0	0	-0.000004362	45-1
45	P2	WL3	LinStatic		3.459E-16	-4.216	1.606	0	0	-0.0006892	45-1
47	P3	DL	LinStatic		-4.337E-19	-5.117E-17	5.843	0	0	0.000001241	47-1
47	P3	LL	LinStatic		-8.674E-19	-2.341E-16	6.353	0	0	0.000003099	47-1
47	P3	WL1	LinStatic		0	2.878E-16	-8.825	0	0	-0.000004082	47-1
47	P3	WL2	LinStatic		-3.469E-18	1.209E-16	-6.087	0	0	-0.000001614	47-1
47	P3	WL3	LinStatic		-1.49E-17	-4.586	0.004948	0	0	-0.000467	47-1
49	P4	DL	LinStatic		2.212E-17	-6.696E-17	5.531	0	0	-1.967E-07	49-1
49	P4	LL	LinStatic		-4.77E-18	-5.815E-16	5.444	0	0	5.73E-08	3 49-1
49	P4	WL1	LinStatic		8.674E-18	8.165E-16	-7.848	0	0	-2.771E-07	49-1
49	P4	WL2	LinStatic		-3.123E-17	2.8E-16	-6.093	0	0	0.0000022	
49	P4	WL3	LinStatic		-5.658E-17	-4.956	-0.807	0	0	-0.0002868	3 49-1
51	P5	DL	LinStatic		8.413E-17	-4.799E-17	6.055	0	0		
51	P5	LL	LinStatic		-1.301E-18	4.268E-17	6.149	0	0	-0.00000165	51-1
51	P5	WL1	LinStatic		1.084E-19	-1.043E-16	-8,805	0	0	0.000001725	51-1
51	P5	WL2	LinStatic		-6.939E-18	-4.808E-18	-5.959	0	0	0.000001136	
51	P5	WL3	LinStatic		-4.479E-18	-5.326	0.304	0	0	-0.0001277	7 51-1
155	P6	DL	LinStatic		-4.561E-16	-4.359E-07	4.337	0	0		
155	P6	LL	LinStatic		-5.556E-15	-5.115E-07	4.313	0	0	-1.627E-18	
155	P6	WL1	LinStatic		0.077	3.508E-07	-3.54	0	0	-1.536E-07	
155	P6	WL2	LinStatic		0.078	2.328E-07	-2.75	0	0	-1.081E-07	
155	P6	WL3	LinStatic		-9.178E-17	-3.329	-9.442	0	0		
55	P7	DL	LinStatic		4.337E-19	-1.793E-17	6.055	0	0		
55	P7	LL	LinStatic		4.337E-19	1.63E-17	6.149	0	0	0.00000165	
55	P7	WL1	LinStatic		-3.469E-18	-2.704E-17	-5.605	0	0	-0.000001635	
55	P7	WL2	LinStatic		-3.469E-18	2.121E-17	-3.205	0	0	-2.239E-07	
55	P7	WL3	LinStatic		-1.009E-18	-5.326	0.304	0	0		
57	P8	DL	LinStatic		-1.262E-16	-7.876E-17	5.531	0	0	1.967E-07	
57	P8		LinStatic		1.735E-18	-6.953E-16	5.444	0	0		
57	P8	WI1	LinStatic		-6.939E-18	8.729E-16	-4.005	0	0	-4.853E-07	
57		VVL1	Linstatic		-0.939E-18	8.729E-10	-4.005	0	0	-4.853E-07	

Figure 4 61 Data of element joint forces after filter and sort

	A	В	C	D	E	F	G	н	I	J	K	- E	AD
L		Joint Read											
2	Joint		s(CaseType		F1	F2	F3	M1	M2	M3			
3	Text	Text	Text	Text	KN	KN	KN	KN-m	KN-m	KN-m			
4	P1	DL	LinStatic		0.046	4.454E-08	4.103	0	0	4.456E-06			
5	P1	LL	LinStatic		0.094	-1.381E-07	3.52	0	0	0.00001021			
6	P1	WL1	LinStatic		-2.29	3.974E-07	-5.041	0	0	-0.00001329			
7	P1	WL2	LinStatic		-3.892	1.578E-07	-4.642	0	0	-6.482E-06			
8	P1	WL3	LinStatic		-0.00693	-2.62	-9.232	0	0	-0.000945			
9	P2	DL	LinStatic		8.674E-18	6.306E-18	5.427	0	0	3.164E-06			
10	P2	LL	LinStatic		1.995E-17	3.163E-18	5.867	0	0	7.239E-06			
11	P2	WL1	LinStatic		-3.816E-17	-1.155E-17	-9.038	0	0	-9.329E-06			
12	P2	WL2	LinStatic		1.18E-16	-6.739E-18	-6.419	0	0	-4.362E-06			
13	P2	WL3	LinStatic		3.459E-16	-4.216	1.606	0	0	-0.0006892			
14	P3	DL	LinStatic		-4.337E-19	-5.117E-17	5.843	0	0	1.241E-06			
15	P3	LL	LinStatic		-8.674E-19	-2.341E-16	6.353	0	0	3.099E-06			
16	P3	WL1	LinStatic		0	2.878E-16	-8.825	0	0	-4.082E-06			
17	P3	WL2	LinStatic		-3.469E-18	1.209E-16	-6.087	0	0	-1.614E-06			
18	P3	WL3	LinStatic		-1.49E-17	-4.586	0.004948	0	0	-0.000467			
۱9	P4	DL	LinStatic		2.212E-17	-6.696E-17	5.531	0	0	-1.967E-07			
20	P4	LL	LinStatic		-4.77E-18	-5.815E-16	5.444	0	0	5.73E-08			
21	P4	WL1	LinStatic		8.674E-18	8.165E-16	-7.848	0	0	-2.771E-07			
22	P4	WL2	LinStatic		-3.123E-17	2.8E-16	-6.093	0	0	0.0000022			
23	P4	WL3	LinStatic		-5.658E-17	-4.956	-0.807	0	0	-0.0002868			
24	P5	DL	LinStatic		8.413E-17	-4.799E-17	6.055	0	0	-9.518E-07			
25	P5	LL	LinStatic		-1.301E-18	4.268E-17	6.149	0	0	-0.00000165			
26	P5	WL1	LinStatic		1.084E-19	-1.043E-16	-8.805	0	0	1.725E-06			
27	P5	WL2	LinStatic		-6.939E-18	-4.808E-18	-5.959	0	0	1.136E-06			
28	P5	WL3	LinStatic		-4.479E-18	-5.326	0.304	0	0	-0.0001277			
29	P6	DL	LinStatic		-4.561E-16	-4.359E-07	4.337	0	0	5.767E-19			
30	P6	LL	LinStatic		-5.556E-15	-5.115E-07	4.313	0	0	-1.627E-18			
31	P6	WL1	LinStatic		0.077	3.508E-07	-3.54	0	0	-1.536E-07			
32	P6	WL2	LinStatic		0.078	2.328E-07	-2.75	0	0	-1.081E-07			
33	P6	WL3	LinStatic		-9.178E-17	-3.329	-9.442	0	0	-4.686E-18			
34	P7	DL	LinStatic		4.337E-19	-1.793E-17	6.055	0	0	9.518E-07			
35	P7	LL	LinStatic		4.337E-19	1.63E-17	6.149	0	0	0.00000165			
36	P7	WL1	LinStatic		-3.469E-18	-2.704E-17	-5.605	0	0	-1.635E-06			
37	P7	WL2	LinStatic		-3.469E-18	2.121E-17	-3.205	0	0	-2.239E-07			
38	P7	WL3	LinStatic		-1.009E-18	-5.326	0.304	0	0	0.0001277			
39	P8	DL	LinStatic		-1.262E-16	-7.876E-17	5.531	0	0	1.967E-07			
10	P8	LL	LinStatic		1.735E-18	-6.953E-16	5.444	0	0	-5.73E-08			
41	P8	WL1	LinStatic		-6.939E-18	8.729E-16	-4.005	0	0	-4.853E-07			
1		*VLL	anotacit		0.0001-10	0.7250-10	-4.000	v	v	-+.0001-07			

# Figure 4 62 Data after paste

	<u>1. Sign Co</u> 2. Reactio	Z My Joint	x	1	72 74 1	-t_				
9	Joint	OutputCase	Horizontal Reaction Hx	Horizontal Reaction Hy	Vertical Reaction Vz	Moment Mx	Moment My	Moment Mz	HIRE	
0	Text	Text	KN	KN	KN	KN-m	KN-m	KN-m		
1	Point-P1	Dead Load	0.0	0.0	4.1	0.0	0.0	0.0		
2	Point-P1	Live Load	0.1	0.0	3.5	0.0	0.0	0.0		
	Point-P1	Left Windload Case1	-2.3	0.0	-5.0	0.0	0.0	0.0		
	Point-P1	Left Windload Case2	-3.9	0.0	-4.6	0.0	0.0	0.0		
	Point-P1	Wind load Y	0.0	-2.6	-9.2	0.0	0.0	0.0		
	Point-P2	Dead Load	0.0	0.0	5.4	0.0	0.0	0.0		
	Point-P2	Live Load	0.0	0.0	5.9	0.0	0.0	0.0		
	Point-P2	Left Windload Case1	0.0	0.0	-9.0	0.0	0.0	0.0		
	Point-P2	Left Windload Case2	0.0	0.0	-6.4	0.0	0.0	0.0		
	Point-P2	Wind load Y	0.0	-4.2	1.6	0.0	0.0	0.0		
_	Point-P3	Dead Load	0.0	0.0	5.8	0.0	0.0	0.0		
_	Point-P3	Live Load	0.0	0.0	6.4	0.0	0.0	0.0		
	Point-P3	Left Windload Case1	0.0	0.0	-8.8	0.0	0.0	0.0		
	Point-P3	Left Windload Case2	0.0	0.0	-6.1	0.0	0.0	0.0		
	Point-P3	Wind load Y	0.0	-4.6	0.0	0.0	0.0	0.0		
	Point-P4	Dead Load	0.0	0.0	5.5	0.0	0.0	0.0		
-	Point-P4	Live Load	0.0			0.0	0.0	0.0		
	Point-P4	Left Windload Case1	0.0			0.0	0.0	0.0		
	Point-P4	Left Windload Case2	0.0		-6.1		0.0	0.0		
	Point-P4	Wind load Y	0.0	-5.0	-0.8	0.0	0.0	0.0		
_	Point-P5	Dead Load	0.0	0.0	6.1	0.0	0.0	0.0		
_	Point-P5	Live Load	0.0	0.0	6.1	0.0	0.0	0.0		
	Point-P5	Left Windload Case1	0.0	0.0	-8.8	0.0	0.0	0.0		
	Point-P5	Left Windload Case2	0.0	0.0	-6.0 0.3	0.0	0.0	0.0		
	Point-P5	Wind load Y	0.0	-5.3	4.3	0.0	0.0	0.0		
	Point-P6	Dead Load Live Load				0.0		0.0		
	Point-P6 Point-P6	Live Load Left Windload Case1	0.0	0.0	4.3	0.0	0.0	0.0		
	Point-P6 Point-P6	Left Windload Case1	0.1	0.0	-3.5	0.0	0.0	0.0		
	Point-P6 Point-P6	Wind load Y	0.0	-3.3	-2.0	0.0	0.0	0.0		
_	Point-P6 Point-P7	Dead Load	0.0	-3.3	-9.4	0.0	0.0	0.0		
_	Point-P7 Point-P7	Live Load	0.0	0.0	6.1	0.0	0.0	0.0		
	Point-P7 Point-P7	Left Windload Case1	0.0		-5.6	0.0	0.0	0.0		
~1	i viner i	DESIGN SHEET OF BME		SteelColumn	+	: 4		0.0		

Figure 4 63 Joint reactions after hide rows

# 4.1.6. Design Example 4b: High-Rise Building Sap Model

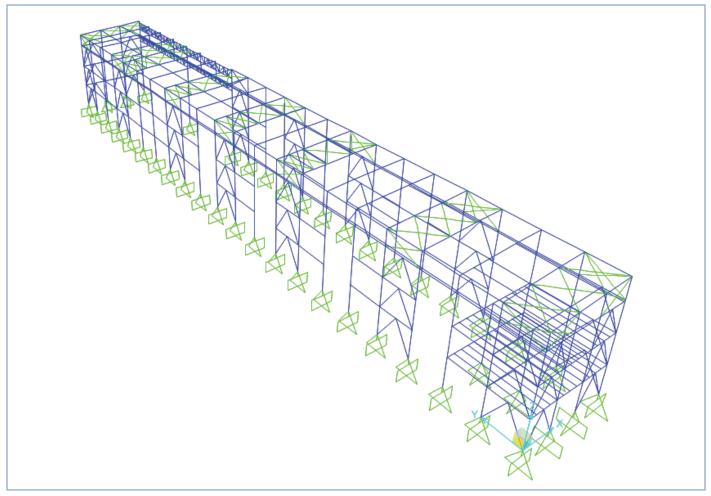


Figure 4 64 High-rise building model

# 4.1.6.1 Defining

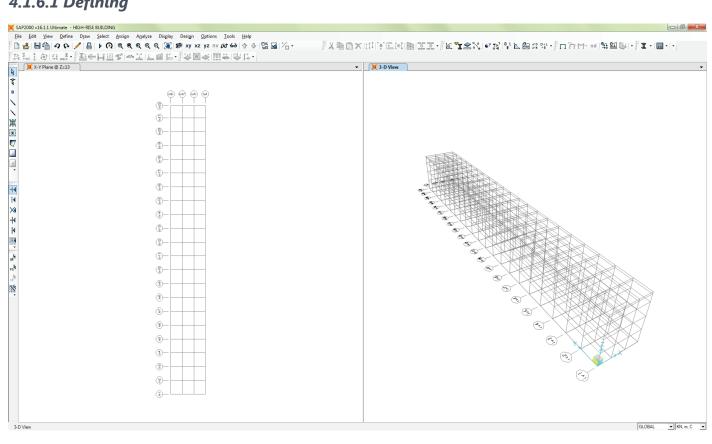


Figure 4 65 Setting up geometry and default units

							BM-7.2.1-02	
		PROJECT IN			Date of iss		24/10/2016	;
BM8	~~		<u>M (PIF</u>		Times of is Project sta		04 Priority [] /	Normal
Best Metal Building BMB & A J/SC	JS.	(THÔNG TIN TR		IAI DỰ AN)	Breakdowr			/ Customer 🗌
		PRC	<b>JECT</b>	INFORMATIC				
Project name:	Rub	y Project						
Location	Thila	awa Zone B						
Consultancy/ F Design Company	Royal Haskoni	ng 🗌 Architype	□ Me	einhardt 🗌 I	B.I.S.T 🔲	Others	□	
Quotation no.:MM72	7/17	Start date:01/12/2017	Fir	nish date:03/1	2/2017	Buildi	ng no.:1	Area no.:1
Requirements: 1.Nev	v Est.	/ 2.Arch.drwgs	/ 3.Ap	orl.drwgs 🗌 /	4.Design ca	lculatio	n 🗌 / 5.Sho	p drwgs 🗌
		BUI	LDING	DESCRIPTIC	N			
		/IB Standard 📕/ 2.T						
Material		-up member: fy= 24	r		Painted 🗌 / (	Galvaniz	ed 📕 fy=45	5 kN/cm <sup>2</sup>
Specifications			or bolt: 5.6 / 8.8 / 4. Connection bolt: 10.9 Finishing: Alkyd / Galvanized / Epoxy / Thickness: 140 micron					
	Pain			anized 🗌 / Epo		ckness:	140 micron	
Exposure		B	C				(0.5	
Qty. of identical Bldir	ngs	1	Usage	)	GRIDLIN	-	•	L Building)
Type of building		Single slope	Roof s	slope (%)	5 %			
Width (M)	20 (GL iii21 to middle column)	Width	Module (M)	(M) 1@20 (center to center ste			eel column)	
Length (M)		167 (GL iv1 to iv22)	Bay Spacing			@7 center s	steel columr	ר)
Height (M)	23 – a	t GL ivB Eave	e Height			Clear	Height 🗌	
	Post &	Beam			One, Ax			Both
Type of End Frames	Main F	Rigid Frame 🗌			One, Axis DB			Both 🗌
Frames	Concr	ete Columns and r	afters (k	oy others)	One, Ax	<is< td=""><td></td><td>Both 🗌</td></is<>		Both 🗌
Type of Wall Bracing	Diago	nal Rod 📕 / Cable	/ Po	rtal frame [] /	None 🗌			
Eave Condition	Gutter	& Downspout	RC Gut	ter (by other)	/ Gutter &	& Down	spout (by ot	her) 🗌
Notes: DO NOT	but ar	ny column in (	GL ivA	A : iv16 : iv	17 : iv19	; iv20	); iv21	
						,	,	
	GN LOA	-		BMB's design	_			
Live Load on Roof (k	N/m²)	0.57 (58.1 KG/n	n²) 🗾/ C	Others				
Live Load on Frame	(kN/m²)	0.30 (30.6 KG/n	n²) 🗾/ C	Others				
Wind Speed (KM/h)		110 (30.6 M/s)[	_/ 130	(36.1 M/s)	/ 160 (44 N	l/s) 🗾 /	Other (	. M/s) 🗌
Collateral Load (KG/	m²)			Concentrate	ed load (KG	/m)		
Collateral Load hang	on pur	in 🗌		Collateral L	oad don't ha	ang on p	burlin 🗌	
Notes: Earthquake	Zone G	lobal Seismic Ha	zard Zo	ne IIB				
		ROC	DF, WA		3S			
Roof panel: Yes	Bare Z	Zinc-Alum AZ		Pre-Painted	Zinc-Alum A	Z	7 ribs	] / 5 ribs 🗌
None	Pre-Pa	ainted Zinc-Alum AZ	Z	Thickness: (	).50mm		Seam	/ Clip-lock 🗌
Wall panel: Yes	Bare Z	Zinc-Alum 🗌		Pre-Painted	Zinc-Alum A	Z	7 ribs pa	anel
None	Pre-Pa	ainted Zinc-Alum AZ	Z	Thickness: (	Thickness: 0.50mm     5 ribs pane			

Notes:											
			INSULA	TION	Roof [	_ / V	Vall 📕 / No	ne 🗌			
Fiberglass thick 50m	m	Air Bu	ubble P2		Rockwall		] Sandw	rich		Other	
Fiberglass thick 100	nm 🗌	Air Bu	ubble A2		PU		] Panel	[			
Fiberglass density: 1	0-12 KG	G/m <sup>3</sup>	/ Or other	kind [	Density:				K	G/m <sup>3</sup>	
Other insulation:											
Notes:											
OPEN WALL CONDITIONS											
Both Side wall			1m for br	ick w	all - sheet to	roof					
End wall GL ivB	End wall GL ivB 1m for brick wall - sheet to roof										
End wall GL ivA – iv	l to iv15	1	1m for br	1m for brick wall - sheet to roof							
End wall GL ivA – iv	5 to iv2	2	Open 20.	5 M -	- sheet to roof						
Notes:											
			RC	OF I	MONITOR			None [			
Туре	Jack	roof		Rid	lge vent 🗌	(	Other 🗌				
Overall width (M)	3.5			Ler	ngth (M)	=	= 6 <mark>7 (pcs</mark>	) x 16.	<mark>880 = 1</mark>	<mark>18.16</mark>	
Eave Condition	Curv	ed Eav	e	Ор	en Eave 🗌	(	Other 🗌				
Roof panel		e as ro ninal)	of panel	panel 📕 / Other 🗌 mm							
Wall panel		e as wa ninal)	all panel	/ Oth	ner 🗌			t	thick	m	m
<u>Notes:</u>											
			ME	ZZA	NINE SLAB			None			
Туре	Con	crete sla	ab 🗌 thic	k	mm / De	cking	g panel 🗌 t	hick	n	nm	
	Mez	lezzanine with Checker Plate 🗌 thickmm									
Quantity		Mezzanine slab 1					Mezzanine slab 2				
Location											
Area (m <sup>2</sup> )											
Live load (KG/m <sup>2</sup> )											
Dead load (KG/m <sup>2</sup> )											
Others load (KG/m <sup>2</sup> )											
Height up to top of Concrete Slab (M)											
Type of support shear load				S	hear stud D20	)	Angle				
Notes:											
			RC	OOF E	EXTENSIONS	1		None			
Panel Sam	e roof pa	anel 🗌	Other					thick		. mm (non	ninal)
Location			Side wall						d wall		
Soffit panel		Yes	/ None					Yes 🗌	/ None		
Quantity											
Width (M)											
Length (M)											
<u>Notes:</u>											

				CANOPY	None					
Panel	Same	e as ro	of panel 🗌 / Other [	]		thick		mm (nominal)		
Eave Condition	Curve					roof panel   / Other				
Location			Side wall		End wall			1		
Soffit panel		Curved Eave       Open Eave       Other         Side wall       End wal         Yes       / None         Yes       / None         Yes       / None         Yes				ne 🗌				
Quantity										
Width (M)										
Length (M)										
Gutter, Downspout			Yes 🗌		None 🗌					
Notes:										
FASCIA   None										
Panel		Same	e as wall panel 🗌 / Other 🗌			thick.		mm (nominal)		
Location			Side wall			E	End w	all		
Soffit panel			Yes 🗌 / No	Yes 🗌 / None 🗌			/ N	one		
Туре										
High (M)										
Length (M)										
Fascia projecti (M)	ion									
Notes:										
	r		PAF	RTITIONS		None [				
Quantity	Partition 1			Partition 2						
Location										
Length (M)										
High (M)										
Туре						zontal of house				
Wall										
Condition	Fully Sheeted 🗌 / Block wall 🗌									
	Open up to M 🗌 / Open up to access 🗌					Open up to M 🗌 / Open up to access 🗌				
Wall Panel	Same	e as wa	all panel 🗌 / Other 🗌	]	•••••	thick		mm (nominal)		
Notes:					_					
	RANES	SYSTE	,		on		)	_		
Quantity			Crane No.2		Crane No.3		Crane No.4			
Location										
Crane capacity (TON)		<mark>10T</mark>								
Crane Span (M)		20								
Length of Crane Run (M)		<mark>167</mark>								
Bracket Height		+19.5m								
<u>Notes:</u> Only su	ıpply, il	nstall r	unning beam 📕 / O	ur supply does no	t ind	clude the Crane Sy	stem a	and Crane Rails 🗌		
Entire supply o	of Cran	e Syst	em include Crane, r	unning beam, for	crai	ne rails and electric	al sys	tems		
Notes:										

AC	CESSOF	RIES - U		EMS OR (		IENTS (If /	Applicab	le)			
Accessories	Length	Width	Height	Qty.	Accessories	Length Width Height			Qty.		
Rolling door					Rlg.door (motor)						
Door frame	0.85		2	10	Steel stair				1		
Double sliding dr					Stair handrail						
Window (Steel)					Opening frame						
Window (Almn)					Wall light	1		20.2	10		
Steel Louver (+ mesh)	5		1.5	10	Ladder						
Notes: Only supply	, frame fo	or doors,	sliding doc	ors,rolling	doors,windows	•			•		
Project Eng. (Sign & write down full name)		Notes for using this PIF 1. Must fill in Exact <u>Country</u> of Project & Name of <u>Consultancy/ Design Company</u> at top of PIF 2. Must make sketch drawing or provide architectural drawing.									

# Define Section Properties as table below:

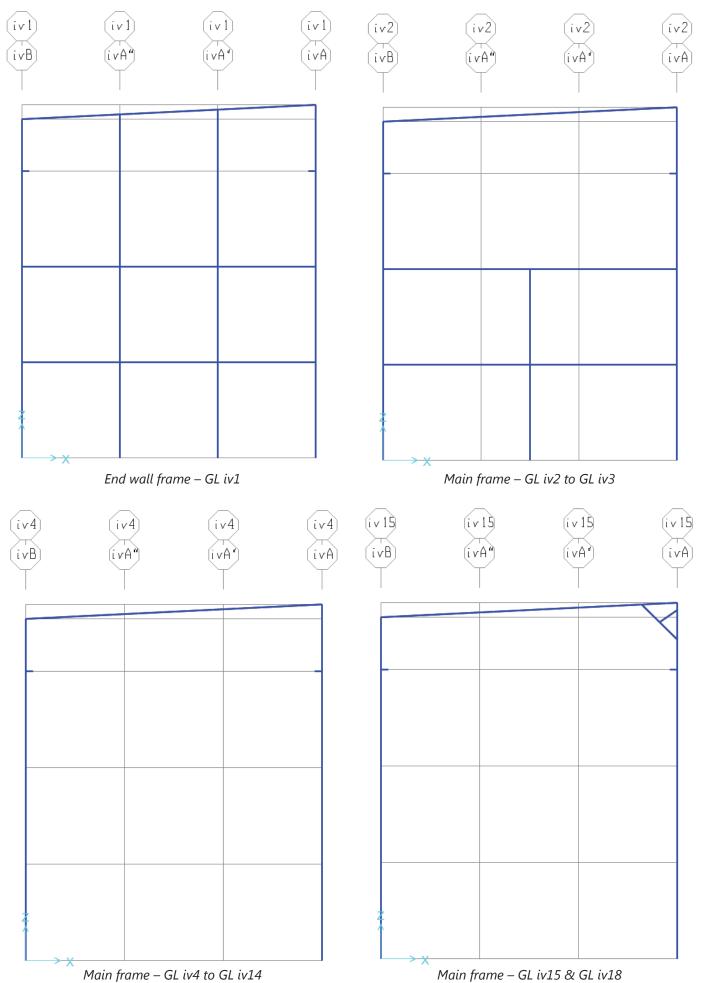
### Table 4 4 Frame Section Properties

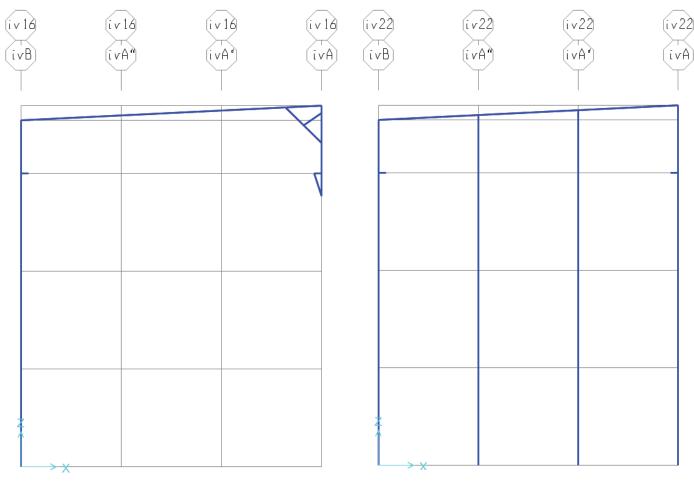
SectionName	Material	Shape	d	bf1	tf1	tw	bf2	tf2
Text	Text	Text	m	m	m	m	m	m
1K		Nonprismatic						
1K516248	STEEL3450	I/Wide Flange	0.516	0.248	0.012	0.006	0.248	0.01
1K900248	STEEL3450	I/Wide Flange	0.9	0.248	0.012	0.006	0.248	0.01
1KA		Nonprismatic						
2К	STEEL3450	I/Wide Flange	0.516	0.184	0.01	0.005	0.184	0.01
3К		Nonprismatic						
3K350184	STEEL3450	I/Wide Flange	0.35	0.184	0.01	0.005	0.184	0.01
3K766184	STEEL3450	I/Wide Flange	0.766	0.184	0.01	0.005	0.184	0.01
4K766184	STEEL3450	I/Wide Flange	0.766	0.184	0.01	0.005	0.184	0.01
5K		Nonprismatic						
5K-766	STEEL3450	I/Wide Flange	0.766	0.212	0.01	0.006	0.212	0.01
5K-900	STEEL3450	I/Wide Flange	0.9	0.212	0.01	0.006	0.212	0.01
B1	STEEL3450	I/Wide Flange	0.31	0.148	0.006	0.005	0.148	0.006
B2	STEEL3450	I/Wide Flange	0.512	0.164	0.008	0.005	0.164	0.008
BR	STEEL3450	I/Wide Flange	0.516	0.212	0.01	0.008	0.212	0.01
C1	STEEL3450	I/Wide Flange	0.8	0.496	0.018	0.012	0.496	0.018
C2		Nonprismatic						
C2-D	STEEL3450	I/Wide Flange	0.8	0.372	0.016	0.008	0.372	0.016
C2-T	STEEL3450	I/Wide Flange	0.8	0.372	0.014	0.008	0.372	0.014
C3	STEEL3450	I/Wide Flange	0.8	0.372	0.016	0.01	0.372	0.016
CD	STEEL3450	I/Wide Flange	0.524	0.248	0.014	0.01	0.248	0.014
CDH	STEEL3450	I/Wide Flange	0.52	0.372	0.012	0.006	0.372	0.012
CR (500~800~500)		Nonprismatic						
CR(800~800~500)		Nonprismatic						
CR2(800~1500~800)		Nonprismatic						
CR2-1500	STEEL3450	I/Wide Flange	1.5	0.372	0.014	0.01	0.372	0.014
CR2-800	STEEL3450	I/Wide Flange	0.8	0.372	0.014	0.01	0.372	0.014
CR3(800~1800~800)		Nonprismatic						
CR3-1800	STEEL3450	I/Wide Flange	1.8	0.372	0.016	0.012	0.372	0.016
CR3-800	STEEL3450	I/Wide Flange	0.8	0.372	0.016	0.012	0.372	0.016
CR500184	STEEL3450	I/Wide Flange	0.5	0.184	0.008	0.006	0.184	0.008
CR800184	STEEL3450	I/Wide Flange	0.8	0.184	0.008	0.006	0.184	0.008
D310164	STEEL3450	I/Wide Flange	0.31	0.164	0.006	0.005	0.164	0.006
EB	STEEL2350	Box/Tube	0.15	0.15	0.0028	0.0028		
G1	STEEL3450	I/Wide Flange	0.77	0.184	0.012	0.008	0.184	0.012
I-200	STEEL3450	I/Wide Flange	0.2	0.164	0.008	0.005	0.164	0.008
I-314	STEEL3450	I/Wide Flange	0.314	0.184	0.008	0.005	0.184	0.008
I-392212	STEEL3450	I/Wide Flange	0.392	0.248	0.01	0.006	0.248	0.01
IPE200	STEEL2350	I/Wide Flange	0.2	0.1	0.008	0.0055	0.1	0.008
JB1	STEEL3450	I/Wide Flange	0.392	0.212	0.01	0.006	0.212	0.01
KDH	STEEL3450	I/Wide Flange	0.314	0.164	0.008	0.005	0.164	0.008
P141	STEEL2350	Pipe	0.1413		-	0.00396		2.000
P168	STEEL2350	Pipe	0.1683			0.00396		
ST264164	STEEL3450	I/Wide Flange	0.264	0.164	0.008	0.005	0.164	0.008
V100x10	STEEL2350	Angle	0.1	0.1	0.01	0.01		0.000

**DESIGN GUIDELINES** 

# 4.1.6.2 Modeling

# Create general model





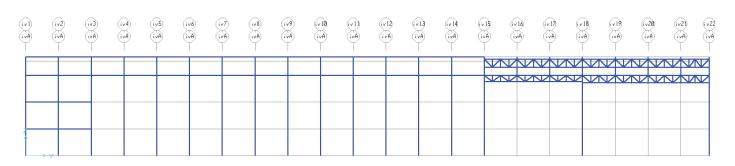
Main frame – GL iv16 to GL iv21 except GL iv18

Main frame – GL iv22



iv 1 ivB	iv2 ivB		r4 (1	v5) ( vB) (	ivð ( ivB) (	iv7 ( iv8 (	iv8 ivB	iv9 ivB	iv 10 ivB	iv1] iv8	iv12 iv8	-13 (i 28 (i	v L4 (i ivB (	vis ivit i	BIV BVI	iv 18 ivB	iv 19 iv	20) () VB) ()	ivB	iv8
_	_							_		_						_	_			_
<u> </u>	_	_						_		+						_	-			_
	~ ~																			

Figure 4 67 Elevation – GL ivB



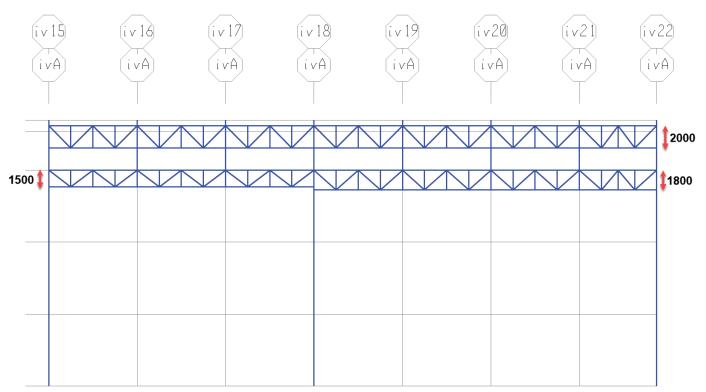


Figure 4 69 Jack beam and crane bracing

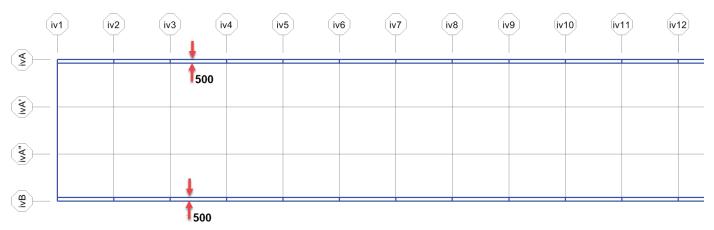
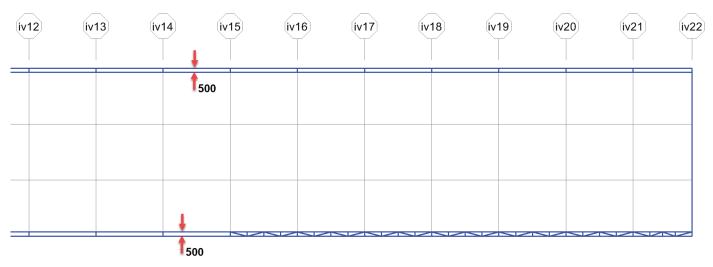
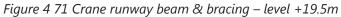


Figure 4 70 Crane runway beam & bracing – level +19.5m





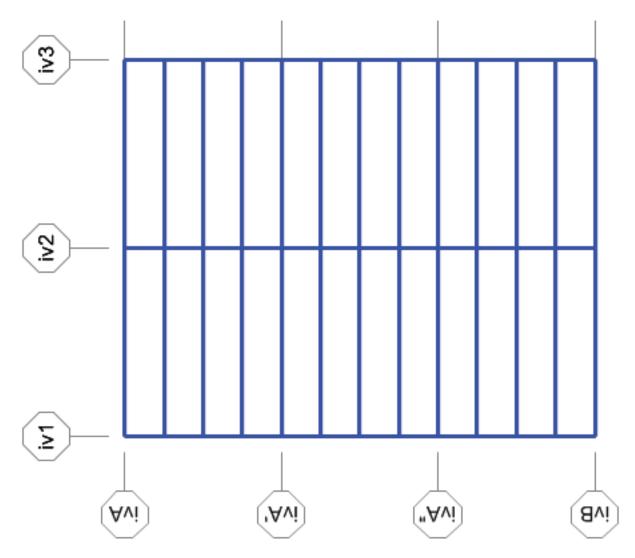


Figure 4 72 Mezzanine floor – level +6.5m, +13m

## Define load patterns and load combinations

Click the **Define > Load Patterns** command to access the **Define Load Patterns** form. In **Load Patterns** area, create load patterns including DL, LL, FL1, FL2, FL3, WL1, WL2, WL3, WL4, WL5, WL6, WL7, WL8, EL1, EL2, CR1, CR2 as shown in figure below.

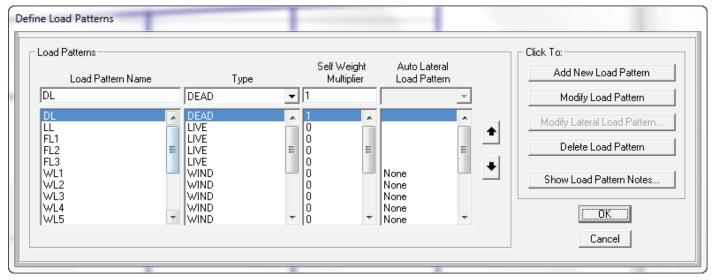


Figure 4 73 Define Load Patterns form



Figure 4 74 Define Load Patterns form

Select EL1 in Load Patterns area and then click the **Modify Lateral Load Pattern** button in Click to area to display the **1997 UBC Seismic Load Pattern** form. Enter all information as figure below.

1997 UBC Seismic Load Pattern	
Load Direction and Diaphragm Eccentricity     Global X Direction     Global Y Direction     Ecc. Ratio (All Diaph.)     0.05	Seismic Coefficients Per Code C User Defined Soil Profile Type SD Seismic Zone Factor 0.20
Override Diaph. Eccen. Override	User Defined Ca 0.28 User Defined Cv 0.4
Time Period         C       Method A       Ct (ft) =         Image: Program Calc       Ct (ft) =       0.035         C       User Defined       T =         Lateral Load Elevation Range       Image: Program Calculated       Reset Defaults         Image: Wax Z       Image: Program Calculated       Image: Program Calculated	Near Source Factor         Image: Per Code       Image: Defined         Seismic Source Type       Image: Defined         Dist. to Source (km)       Image: Defined         User Defined Na       Image: Defined         User Defined Nv       Image: Defined
Factors Overstrength Factor, R 4.5	Other Factors Importance Factor, I 1.
ОК	Cancel

Figure 4 75 EL1 definition

Select EL2 in Load Patterns area and then click the Modify Lateral Load Pattern button in Click to area to display the 1997 UBC Seismic Load Pattern form. Enter all information as figure below.

Note: Based on **UBC 97** standard, we have all the parameters for seismic zone 2B as above.

1997 UBC Seismic Load Pattern	-
Load Direction and Diaphragm Eccentricity         C       Global X Direction         Image: Global Y Direction         Ecc. Ratio (All Diaph.)         Override Diaph. Eccen.	Seismic Coefficients Per Code C User Defined Soil Profile Type SD Seismic Zone Factor 0.20 User Defined Ca 0.28 User Defined Cv 0.4
Time Period         C       Method A       Ct (ft) =         Image: Program Calc       Ct (ft) =       0.035         Image: User Defined       T =       0.035         Image: Lateral Load Elevation Range       Image: Program Calculated       Reset Defaults         Image: Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Program Calculated       Image: Program Calculated       Image: Program Calculated         Image: Program Calculated       Image: Program Calculated       Image: Program Calculated<	Near Source Factor  Per Code User Defined Seismic Source Type Dist. to Source (km) User Defined Na User Defined Nv
Factors Overstrength Factor, R 4.5	Other Factors

Figure 4 76 EL2 definition

Define mass source: click the **Define > Mass** Source command to show the Mass Source form.

Mass Source	
Mass Sources	Click to:
MSSSRC1	Add New Mass Source
	Add Copy of Mass Source
	Modify/Show Mass Source
	Delete Mass Source
	Default Mass Source
	OK Cancel

Figure 4 77 Mass Source form

Click the **Modify/Show Mass source** button to access the **Mass Source Data** form. Enter all information like figure below:

💢 Mass Source Data		
Mass Source Name	MSSSRC1	
Mass Source Element Self Mas Specified Load P	s and Additional Mass atterns	
Mass Multipliers for Loa Load Pattern DL LL FL1 CR1		Add Modify Delete
	OK Cance	I

Figure 4 78 Mass Source Data form

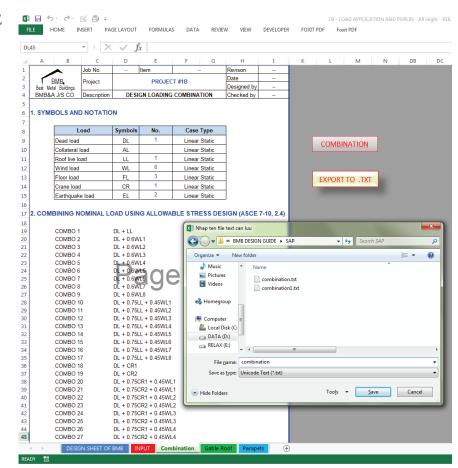


Figure 4 79 Design loading combination sheet

Define load combinations follow ASCE 7-10 standard using .s2k text file.

• In SAP2000, you have to define at least one load combination before export .s2k text file. Click the **Define > Load Combinations** command, in the **Define Load Combinations** form which has just appeared, click the **Add New Combo** button and create an combination as shown in figure below.

	(User-Generated)	COMB1		
Notes	Modify/Show Notes			
Load Combination Type		Linear Add	<b>.</b>	
Dptions				
Convert to User Load Con	nbo Create Nonlin	near Load Case from L	.oad Combo	
Define Combination of Load Ca	ase Results			
Load Case Name	Load Case Type	Scale Factor		
DL 💌	Linear Static	1.		
DL	Linear Static	1.		
02			Add	
			Modify	
			Modify Delete	

Figure 4 80 Create load combination COMB1

• After create any combination, click the File > Export > Sap2000 .s2k Text File command to display the Choose Tables for Export to Text File form shown in figure below.

Edit	
B · ⊠ MODEL DEFINITION (44 of 44 tables selected)	Load Patterns (Model Def.) Select Load Patterns 5 of 5 Selected
	Coptions Selection Only ✓ Open File After Exports ← Use Text Editor ← Use Microsoft Word Expose All Input Tables
	Named Sets Save Named Set Show Named Set Delete Named Set
	OK Cancel

Figure 4 81 Choose Tables for Export to Text File form

• Make sure that all tables, load patterns, and Open File After Export option are selected before click the **OK** button.

• Copy all data in combination.txt text file and paste to .s2k text file at highlight text shown in figure below.

```
TABLE: "LOAD PATTERN DEFINITIONS"
  LoadPat=DL DesignType=DEAD SelfWtMult=1
   LoadPat=LL DesignType=LIVE SelfWtMult=0
   LoadPat=WL1 DesignType=WIND SelfWtMult=0
                                                   AutoLoad=None
   LoadPat=WL2 DesignType=WIND SelfWtMult=0 AutoLoad=None
LoadPat=WL3 DesignType=WIND SelfWtMult=0 AutoLoad=None
TABLE: "AUTO WAVE 3 - WAVE CHARACTERISTICS - GENERAL"
   WaveChar=Default WaveType="From Theory" KinFactor=1
SWaterDepth=45 WaveHeight=18
                                 WavePeriod=12
WaveTheory=Linear
       "COMBINATION DEFINITIONS"
TABLE:
                                              AutoDesign=No
SteelDesign=None<u>ConcDesign=None</u>
TABLE: "FUNCTION - RESPONSE SPECTRUM - USER"
   Name=UNIFRS Period=0 Accel=1 FuncDamp=0.05
   Name=UNIFRS
                Period=1
                           Accel=1
TABLE: "FUNCTION - TIME HISTORY - USER"
  Name=RAMPTH Time=0 Value=0
   Name=RAMPTH Time=1 Value=1
   Name=RAMPTH Time=4
                          Value=1
   Name=UNIFTH
                Time=0
                          Value=1
   Name=UNIFTH Time=1 Value=1
TABLE: "FUNCTION - POWER SPECTRAL DENSITY - USER"
  Name=UNIFPSD Frequency=0 Value=1
   Name=UNIFPSD Frequency=1 Value=1
TABLE: "FUNCTION - STEADY STATE - USER"
  Name=UNIFSS Frequency=0 Value=1
   Name=UNIFSS Frequency=1 Value=1
TABLE: "GROUPS 1 - DEFINITIONS"
GroupName=ALL Selection=Yes
                                  SectionCut=Yes Steel=Yes
Concrete=Yes Aluminum=Yes ColdFormed=Yes Stage=Yes
Bridge=Yes AutoSeismic=No AutoWind=No SelDesSteel=No
SelDesAlum=No SelDesCold=No MassWeight=Yes Color=Red
TABLE: "JOINT PATTERN DEFINITIONS"
   Pattern=Default
TABLE: "MASS SOURCE"
  MassSource=MSSSRC1
                       Elements=Yes
                                      Masses=Yes Loads=No
IsDefault=Yes
```

Figure 4 82 Substitute combination defined by SAP with combination exported from excel.

• Save and close file. Return to SAP and import this file. Click the **File > Import > SAP2000 .s2k Text File** command to display **Import Tabular Database** shown in figure below. Click the OK button, specify created .s2k text file in appeared window and click the Done button to import file.

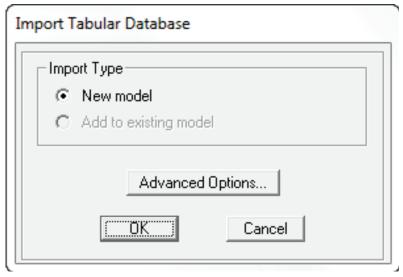
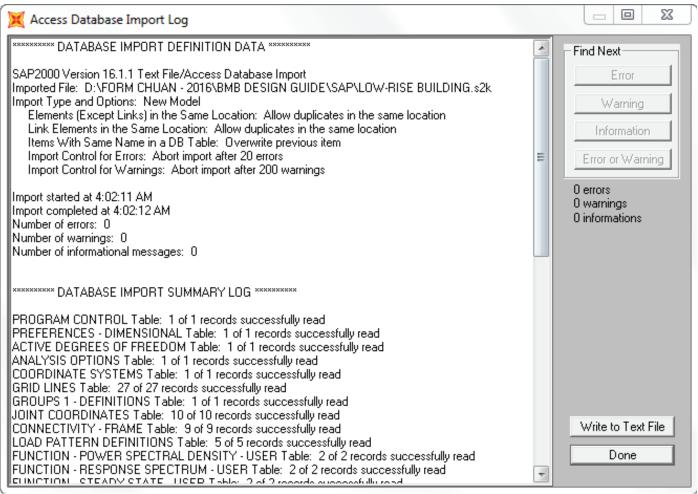
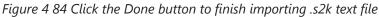


Figure 4 83 Import Tabular Database form





### Assign loads

In this step, the dead, live and wind loads will be applied to the model.

Calculate wind loads which will be assigned to frame using "1A - LOAD APPLICATION AND PURLIN – All Height" file.

In sheet "Input", enter all the information as below:

 IIIIIIIIIIIIIIIIIIIIIIIIIIIIIII	REVIEW VIEW DEVELOPER FOXIT PDF Foxit	WIND- ALL HIGHT(18m) MM727.xlsm - Excel		? 🖻 — 🗗 🗙 Sign in 🎑
BE31 $\checkmark$ : $\times \checkmark f_x$				· · · · · · · · · · · · · · · · · · ·
	F G H I BB	BC BD BE BF BG	BH BI BJ BK BL BM	BN BO BP BQ BR A
5	F G H I BB	BC BD BE BF BG	BH BI BJ BK BL BM	BN BO BP BQ BR
1.     Factory       22     1. Factory       23     Occupancy category       25     Total length of factory       27     2. Interior frame       28     Height of fover column (Eave height)       30     Bay spacing (Interior)       31     Span       32     Roof slope	Monoslope Roof           II $I_T =$			
35 3. Endwall frame 37 Bay spacing (endwall) 38 Bay spacing of endwall frame	$B_* = 8.00 \text{ m}$ $L_m = 6.67 \text{ m}$			
39         End spacing of endwall frame           40         4. Topographic           41         Exposure           42         Wind speed           43         Return period	L <sub>z</sub> = 6.67 m B 160 km/h 50 Year - 700			
44         5. Load           45         Dead load           46         Collateral load           47         Live load           48	$\begin{array}{llllllllllllllllllllllllllllllllllll$			
DESIGN SHEET OF BMB INPUT Left Monoslope i READY	Roof Right Monoslope Roof 🕀			: • • • • • • • • • • • • • • • • • • •

Figure 4 85 Input sheet

• Go to sheet "**Left Monoslope Roof**" and then click the Calculation button to compute wind load blowing from left to right.

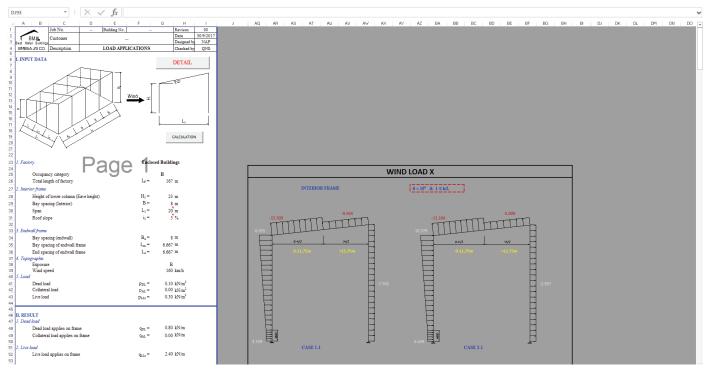


Figure 4 86 Left Monoslope Roof sheet

• Go to sheet "**Right Monoslope Roof**" and then click the Calculation button to compute wind load blowing from right to left.

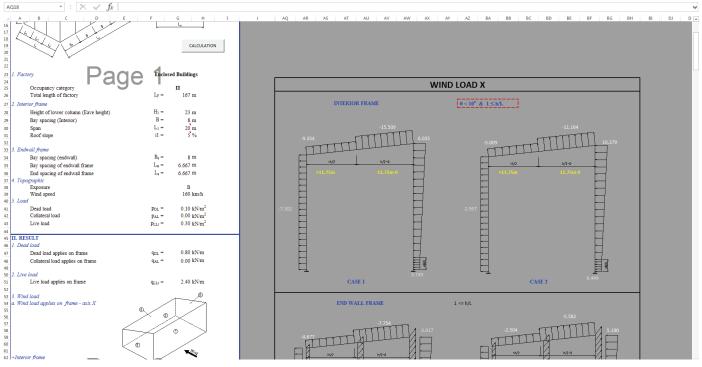


Figure 4 87 Right Monoslope Roof sheet

Calculate dead load and floor load using "2 - COMPOSITE BEAM & DECKING" file.

### In sheet "1", enter all the information as figure below:

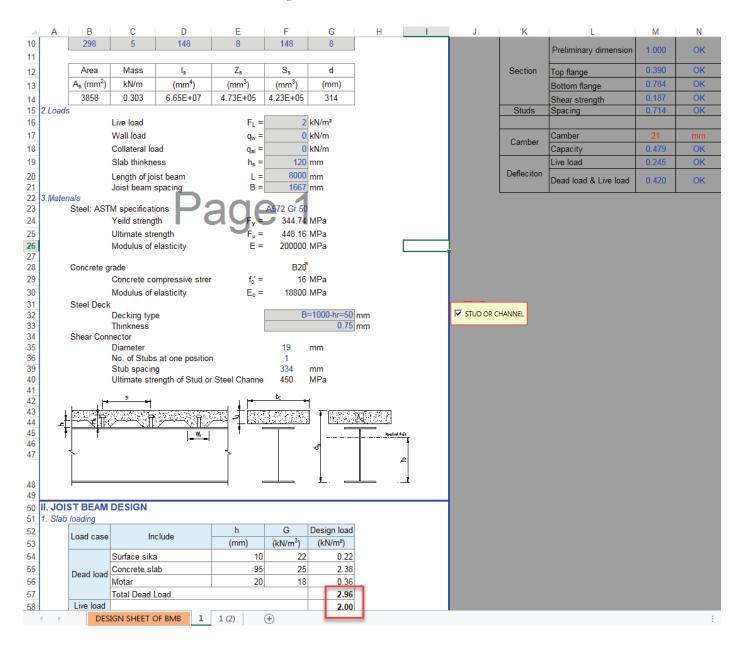


Figure 4 88 Calculate load for mezzanine floor

Use 2.96 kN/m2 for dead load and 2 kN/m2 for live load to apply on mezzanine floor. Calculate crane load using "**3** - **CRANE BEAM DESIGN**" file. Enter all the information as figure below:

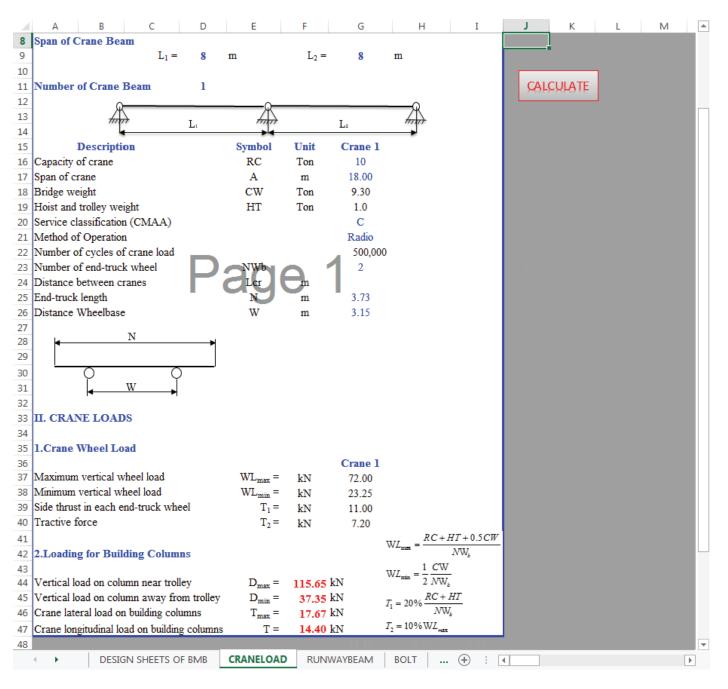


Figure 4 89 Crane load for bay spacing 8m

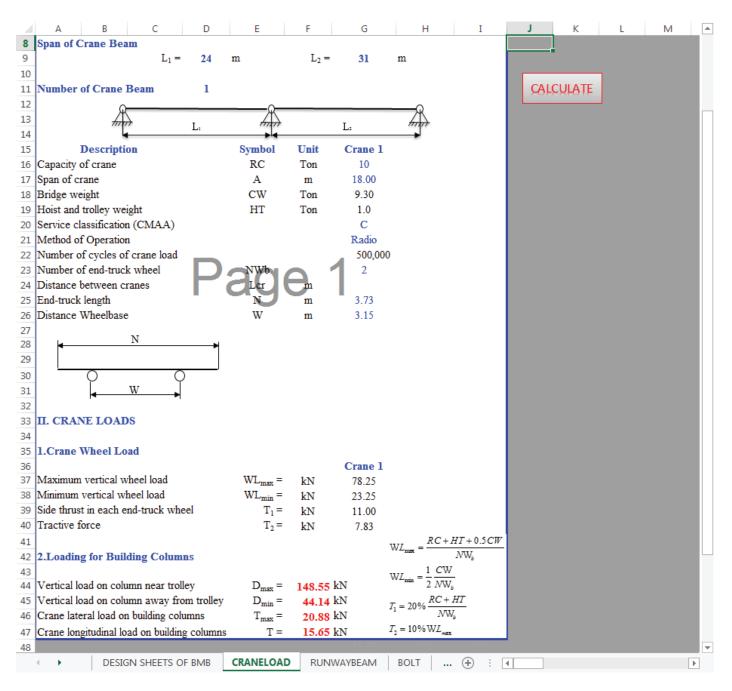
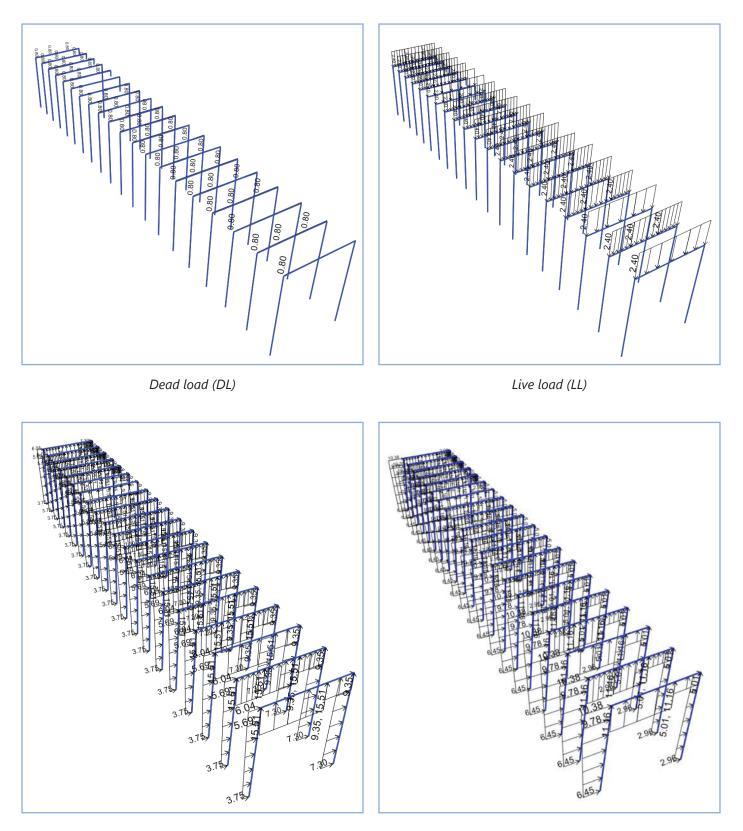
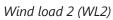


Figure 4 90 Crane load for bay spacing 24m and 31m

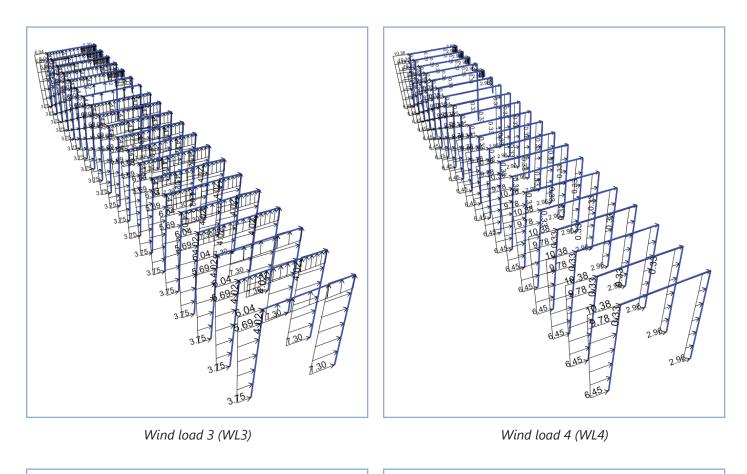
## Assign Load

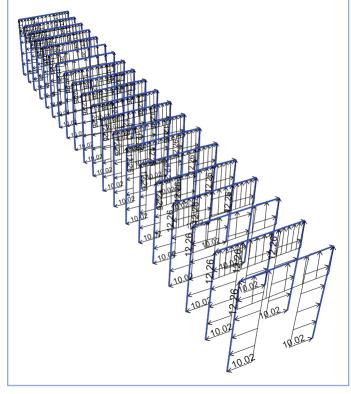
After calculate load applied on frame, we assign them to frame as shown in these figure below.

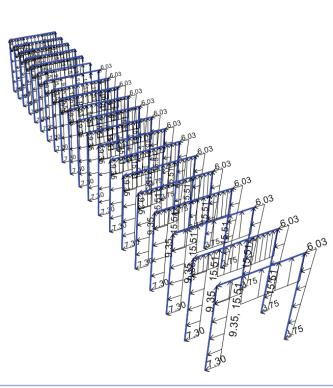




Wind load 1 (WL1)

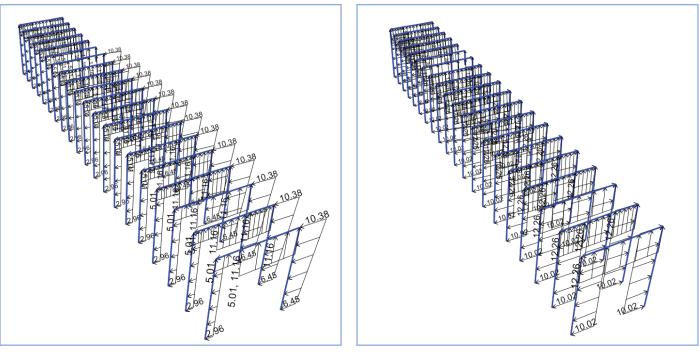






Wind load 5 (WL5)

Wind load 6 (WL6)

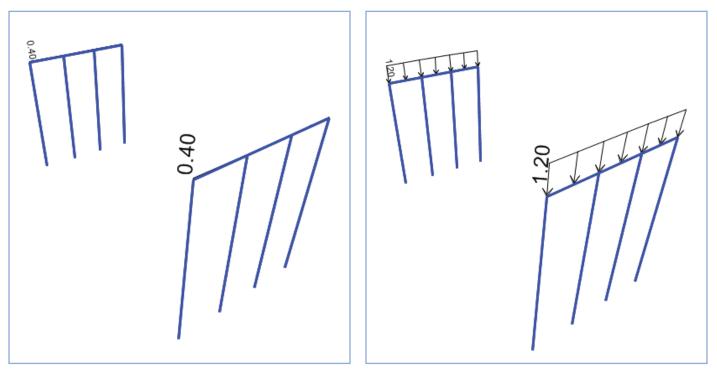


Wind load 7 (WL7)

Wind load 8 (WL8)

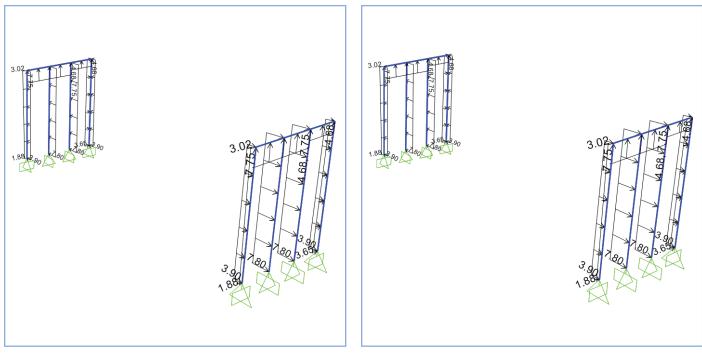


End wall frame



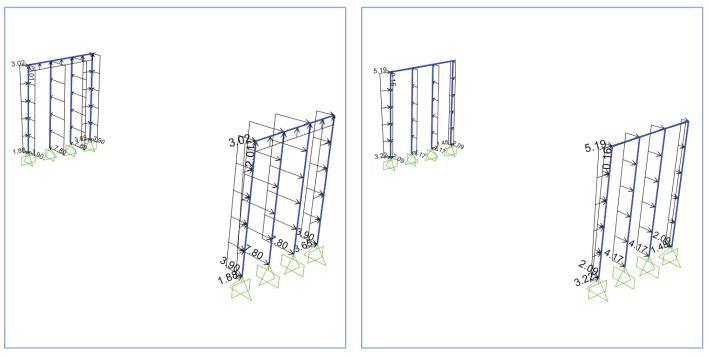
Dead load (DL)





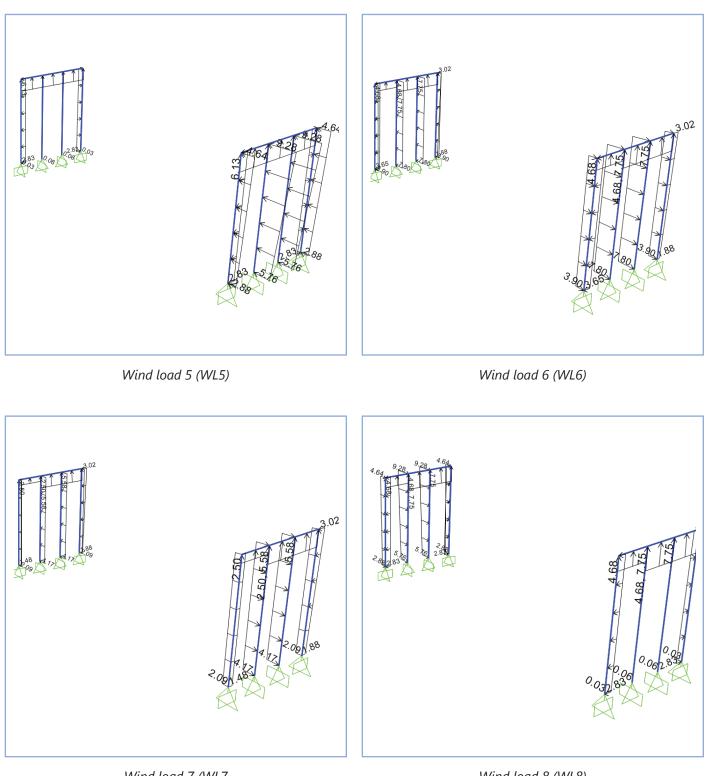
Wind load 1 (WL1)

Wind load 2 (WL2)



Wind load 3 (WL3)

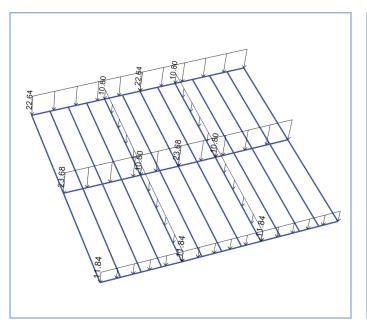
Wind load 4 (WL4)



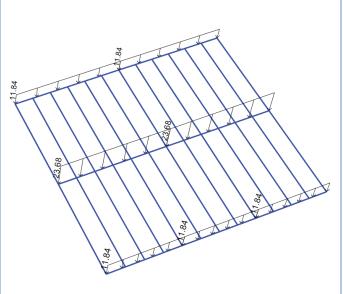
Wind load 7 (WL7

Wind load 8 (WL8)

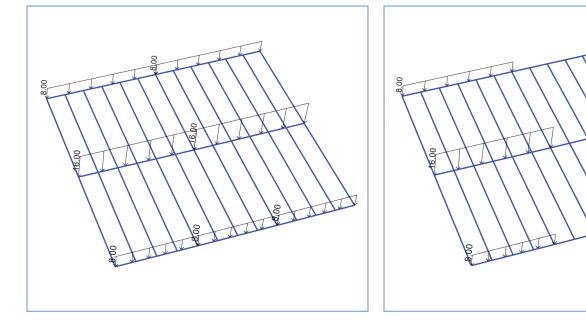
Figure 4 92 Load applied End-wall Frame



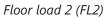
Deal load (DL) – level +6.5m



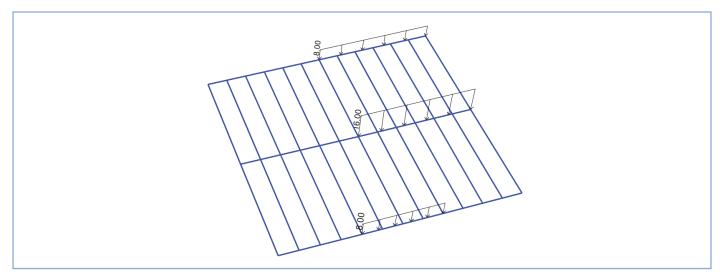
Deal load (DL) – level +13m



Floor load 1 (FL1)



00



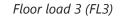


Figure 4 93 Mezzanine floor

Crane bracket

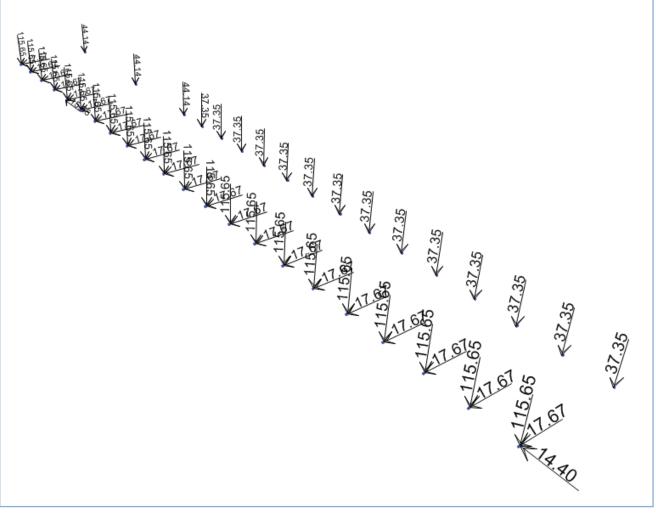


Figure 4 94 Crane load 1 (CR1)

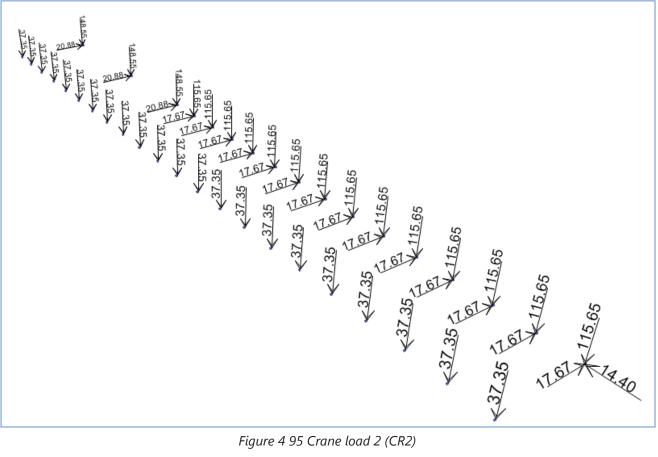
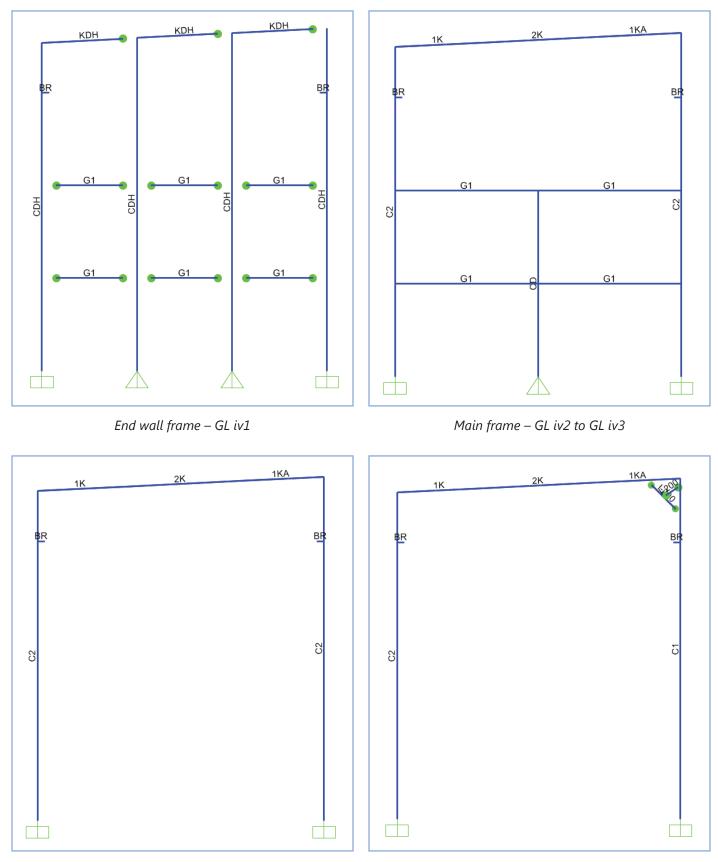


Figure 4 95 Crane load 2 (CR2)

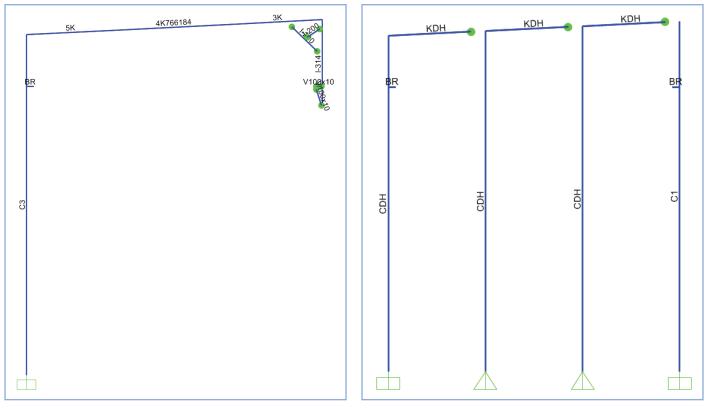
## Divide frame objects and assign member section, end releases and restraint

By the same manner in modeling low-rise building, we have model as shown in figure below:



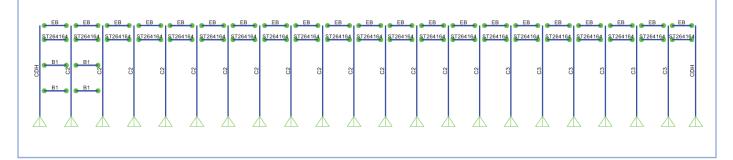
Main frame – GL iv4 to GL iv14

Main frame – GL iv15 & GL iv18

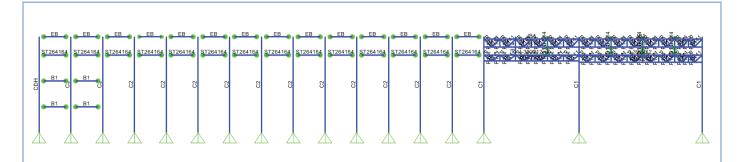


Main frame – GL iv16 to GL iv21 except GL iv18

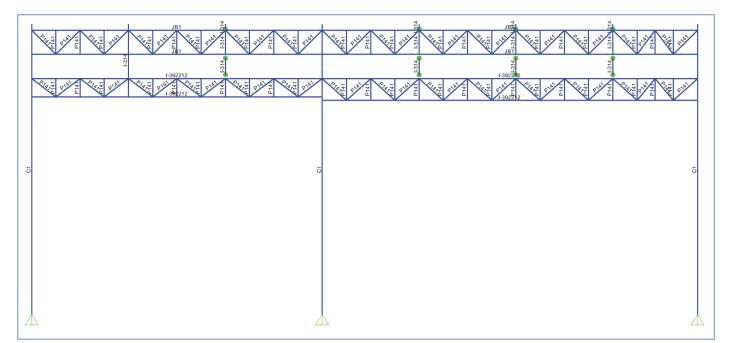
Main frame – GL iv22



Elevation – GL ivB



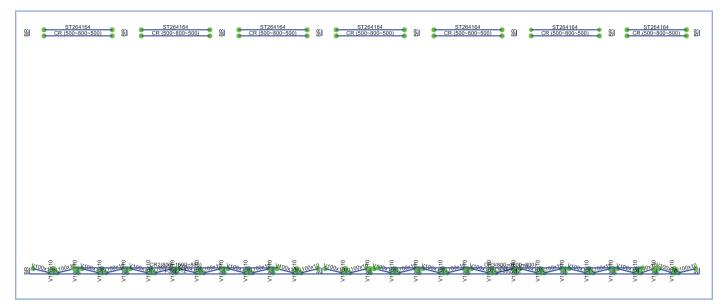
Elevation – GL ivA



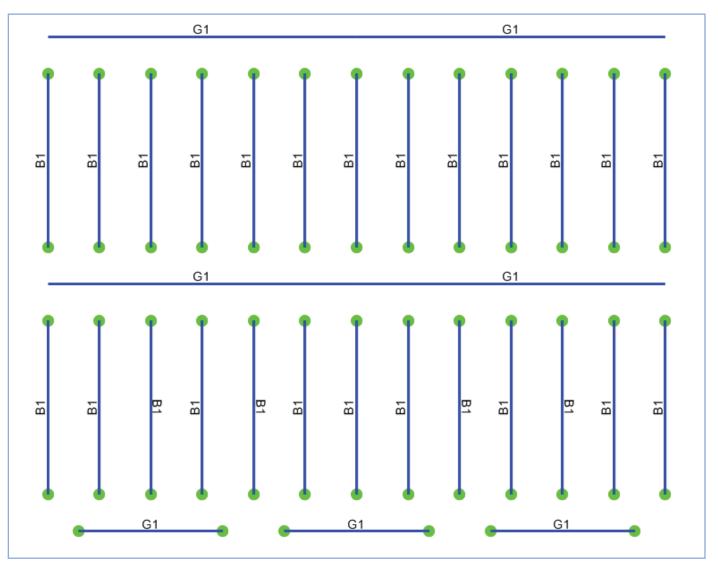
Jack beam and crane bracing

띪	ST264164 CR (500-800-500)	ST264164 CR (500-800-500)	ST264164 CR (500-800-500)	ST264164 CR (500~800~500)	ST264164 CR (500-800-500)	ST264164 CR (500-800-500)
쎪	CR (500~800~500) ST264164					

Crane runway beam & bracing - level +19.5m



Crane runway beam & bracing – level +19.5m



Mezzanine floor – level +6.5m, +13m

### Calculate ULR for each member in frame

- Assume purlin spacing and girt spacing are 1.5m.

- With rafters: unbraced length for 1K, 1KA, 5K, 3K are 1.5m, for 2K and 4k766184 are 3m and for KDH is their length (no bracing).

- With outer columns: unbraced length for C1, C2, C3 and CDH are 2.5m (based on architectural drawing).

- With the other members, set ULR as Program Determined.

#### Table 4 5 Unbraced length ratio

Member	Length (m)	Unbraced Length (m)	ULR
1K	6	1.5	0.25
2К	8	3	0.38
1KA	6	1.5	0.25
5K	6	1.5	0.25
4K766184	8	3	0.38
3К	6	1.5	0.25
KDH	6.667	6.667	1.00
C1	23	2.5	0.11
C2	23	2.5	0.11
C3	23	2.5	0.11
CDH	23	2.5	0.11

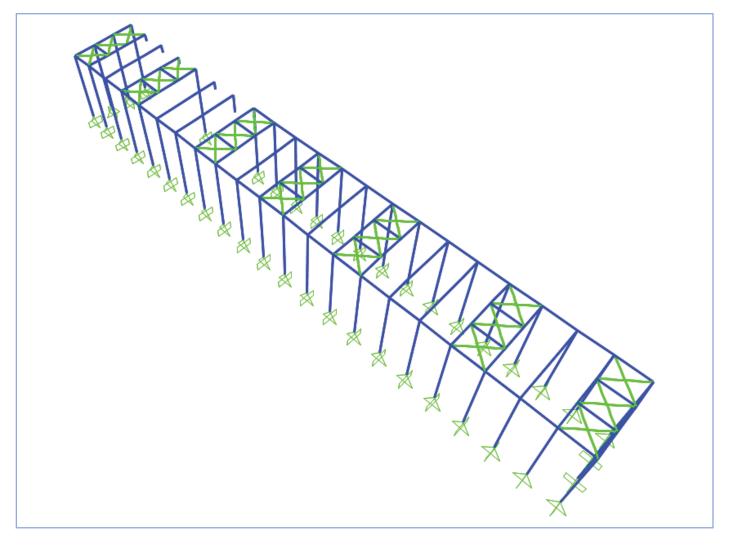
Select 1K section members and click the **Design > Steel Frame Design > View/Revise Overwrites** command to show the **Steel Frame Design Overwrites** for AISC 360-10 form. At the Value column, type **0.25** for both **Unbraced Length Ratio (Minor)** and **Unbraced Length Ratio (LTB)**.

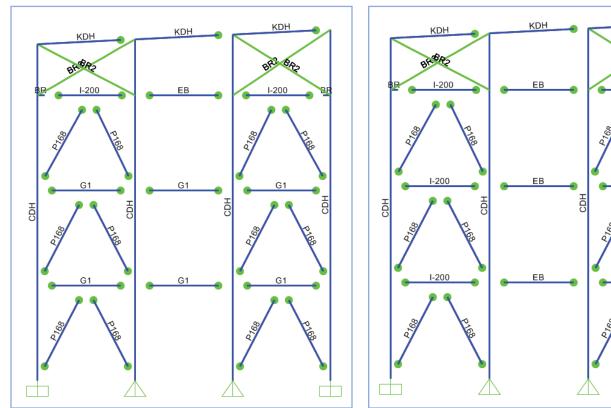
Item Design Section Type Type Type To Deflection? Type The Content of the Content	Value           Program Determined           Program Determined           Program Determined           No           Program Determined           Program Determined		Unbraced length factor for lateral- torsional buckling for the frame object. This item is specified as a fraction of the frame object length.Multiplying this factor times the frame object length gives the unbraced length for the object. Specifying 0 means the value is program determined
Type ) r Deflection? on Check Type ; L / iL+LL Limit, L / ad Limit, L / mit, L/ amber Limit, L/ ; abs iL+LL Limit, abs ad Limit, abs mit, abs	Program Determined Program Determined No Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined		object. This item is specified as a fraction of the frame object length.Multiplying this factor times the frame object length gives the unbraced length for the object.Specifying 0
) r Deflection? on Check Type ;; L / iL+LL Limit, L / ad Limit, L / mit, L/ amber Limit, L/ ;, abs iL+LL Limit, abs ad Limit, abs mit, abs	Program Determined No Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined		fraction of the frame object length.Multiplying this factor times the frame object length gives the unbraced length for the object.Specifying 0
r Deflection? on Check Type ;; L / iL+LL Limit, L / ad Limit, L / mit, L/ amber Limit, L/ ;, abs iL+LL Limit, abs ad Limit, abs mit, abs	No           Program Determined		length.Multiplying this factor times the frame object length gives the unbraced length for the object.Specifying 0
on Check Type ;; L / iL+LL Limit; L / ad Limit; L / mit; L/ amber Limit; L/ ;; abs iL+LL Limit; abs ad Limit; abs mit; abs	Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined		length for the object.Specifying 0
;, L / iL+LL Limit, L / ad Limit, L / mit, L/ amber Limit, L/ ;, abs iL+LL Limit, abs ad Limit, abs mit, abs	Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined		
L+LL Limit, L / ad Limit, L / mit, L/ amber Limit, L/ t, abs L+LL Limit, abs ad Limit, abs mit, abs	Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined		nneans the value is program determined
ad Limit, L / mit, L/ amber Limit, L/ t, abs L+LL Limit, abs ad Limit, abs mit, abs	Program Determined Program Determined Program Determined Program Determined Program Determined Program Determined		
mit, L/ amber Limit, L/ ;, abs :L+LL Limit, abs ad Limit, abs mit, abs	Program Determined Program Determined Program Determined Program Determined Program Determined		
amber Limit, L/ ;, abs IL+LL Limit, abs ad Limit, abs mit, abs	Program Determined Program Determined Program Determined Program Determined		
;, abs :L+LL Limit, abs ad Limit, abs mit, abs	Program Determined Program Determined Program Determined		
IL+LL Limit, abs ad Limit, abs mit, abs	Program Determined Program Determined		
ad Limit, abs mit, abs	Program Determined		
mit, abs			
	Description Distance in a l		
and the first state of the	Program Determined		
amber Limit, abs	Program Determined		
d Camber	Program Determined		
a to Total Area Ratio	Program Determined		
ad Reduction Factor	Program Determined		
ed Length Ratio (Major)	Program Determined		
ed Length Ratio (Minor)	0.25		
			1
			Explanation of Color Coding for Values
	Program Determined		
e Length Factor (K2 Major)	Program Determined		Blue: All selected items are program determined
termined (Default) Values	Reset To Previous Values		Black: Some selected items are user defined
Selected Items	All Items Selected	Items	Red: Value that has changed durin the current session
	ed Length Ratio (Minor) ed Length Ratio (LTB) e Length Factor (K1 Major) e Length Factor (K1 Minor) e Length Factor (K2 Major) etermined (Default) Values	ed Length Ratio (LTB) 0.25 e Length Factor (K1 Major) Program Determined e Length Factor (K1 Minor) Program Determined e Length Factor (K2 Major) Program Determined etermined (Default) Values Reset To Previous Values	ed Length Ratio (LTB)  e Length Factor (K1 Major)  e Length Factor (K1 Minor)  e Length Factor (K2 Major)  retermined  e Length Factor (K2 Major)  retermined (Default) Values  Reset To Previous Values

Figure 4 96 Steel frame design overwrites form

Repeat these steps to assign ULR for the other member.

# Draw Bracing System





End wall GL iv1

End wall GL iv22

KDH

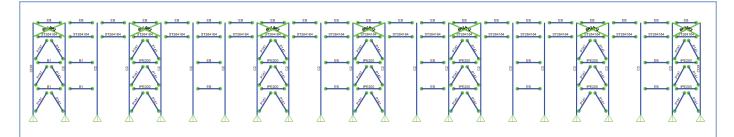
882 82

I-200

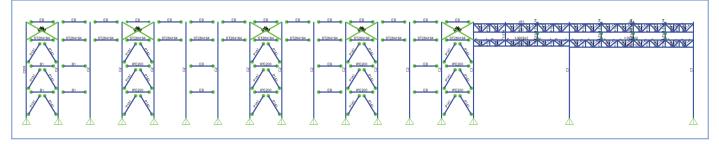
I-200

I-200

5



Elevation GL ivB



Elevation GL ivA

# 4.1.6.3 Analyzing

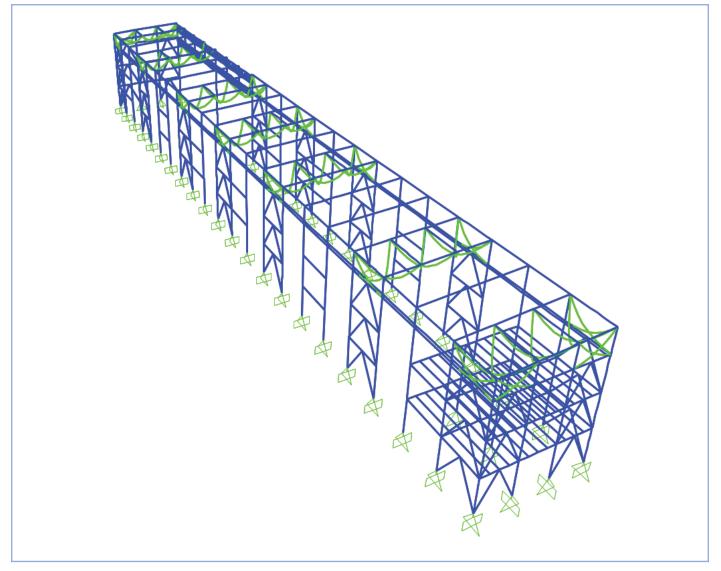


Figure 4 98 Model after running analysis

**DESIGN GUIDELINES** 

# 4.1.6.4 Design Steel Frame

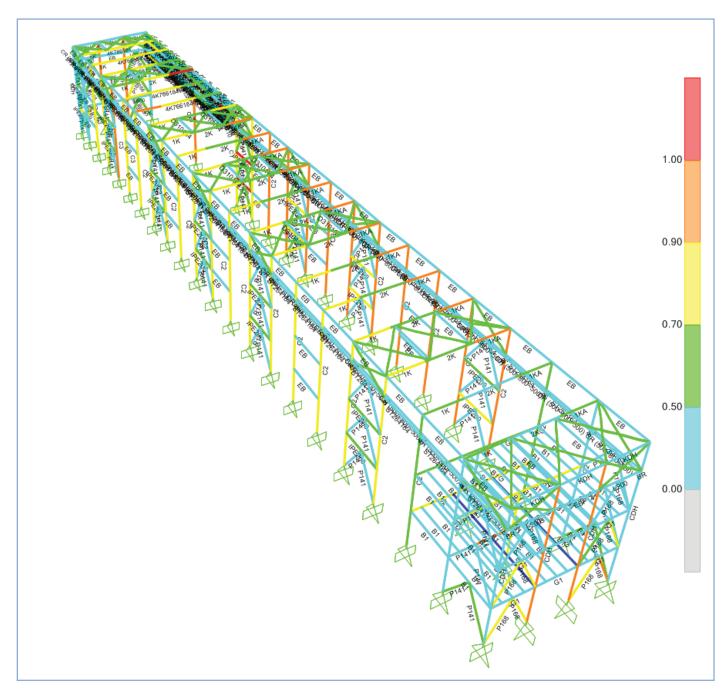


Figure 4 99 Model after checking

To see detail information of checking strength, right click on the 1K member to show the Steel Stress Check Information form.

Frame ID	624			Ana	lysis Sect	ion	1K		
Design Code	AISC 360-1	10		Des	ign Sectio	n	1K		
COMBO	STATION /	MOMEN	I IN	TERACT	ION CHI	СК	//-	MAJ-SHR	-MIN-SHR-/
ID	LOC	RATIO	=	AXL +	B-MAJ	+	B-MIN	RATIO	RATIO
COMB272	3.00	0.125(C)	= 0	.005 +	0.054	+	0.067	0.060	0.001
COMB272	6.00	0.122(C)	= 0	.006 +	0.003	+	0.113	0.025	0.001
COMB273	0.00	0.102(C)	= 0	.004 +	0.086	+	0.012	0.100	0.001
COMB273	3.00	0.093(C)	= 0	.004 +	0.025	+	0.064	0.050	0.001
COMB273	6.00	0.141(C)	= 0	.005 +	0.025	+	0.111	0.018	0.001
COMB274	0.00	0.604(T)		.009 +	0.594		0.000	0.648	0.000
-Modify/Show		- Display D	etails	for Select					omplete Detai oular Data
									neet: Default

Figure 4 100 Steel Stress Check Information form

Click the **Details** button to show more detail information.

AISC 360-10 STEEL	SECTION CHECK	(Summaru Fr	r Combo a	(noitet2 bo			Units KI	V, m, C 💌
Units : KN, m, O		, (Samuary re		na scaciony				
Frame : 624	X Mid: 2.996				pe: Brace			
Length: 6.000	Y Mid: 16.00			Frame Typ				
Loc : 0.000	Z Mid: 23.15	0 Class: S	tender	рылсрі к	lot: 0.000 de	grees		
Provision: ASD	Analysis: Dir	ect Analusis						+-
D/C Limit=1.000		eneral 2nd Ord	ler	Reduction: No M	lodification			
AlphaPr/Py=0.017		.009 Tau_b=1.		EA factor=1.000	EI factor	=1.000		
· · ·								
OmegaB=1.670	OmegaC=1.670			OmegaTF=2.000				
OmegaV=1.670	OmegaV-RI=1.	500 OmegaVT=	1.070					
A=0.011	133-0.001	r33=0.36	3	\$33=0.003	Av3=0.005			
J=0.000	122=2.798E-0			S22=2.256E-04	Av2=0.005			
E=199947978.8	fy=345000.00	0 Ry=1.072		z33=0.004	Cw=5.525E			
RLLF=1.000	Fu=448000.00	3		z22=3.462E-04				
STRESS CHECK FOR	ES & MOMENTS	Combo COMR27/	0					
Location	Pr	Mr33	Mr22	Ur2	Ur3	Tr		
0.000	39.372	368.641	0.053	88.682	-0.002 1.	802E-04		
PMM DEMAND/CAPAC		1.2,H1-1b)						
D/C Ratio:	0.604 = 0.009	+ 0.594 + 0.0 (Pr/Pc) + (Mr3		(Mulaa (Mulaa)				
	- (1/2)	(Pr/FC) + (Pra	537FIC35) +	(11/22/11/22)				
AXIAL FORCE & BIA	IXIAL MOMENT D	SIGN (H1.2	H1-1b)					
Factor	L	K1	K2	B1	B2	Cm		
Major Bending		1.000	1.000	1.000	1.000	1.000		
Minor Bending	0.250	1.000	1.000	1.000	1.000	1.000		
	Litb	Kitb	Cb					
LTB	0.250	1.000	1.282					
	Pr		nt/Omega					
	Force	Capacity	Capacity					
		1220.199	2215.437					
Axial	39.372							
Axial		Mn/Omera	Mn/Ameria					
Axial	Mr Moment	Mn/Omega Capacity	Mn/Omega No LTB					
Major Moment	Mr	Capacity 620.908						
	Mr Moment	Capacity	No LTB					
Major Moment Minor Moment	Mr Moment 368.641	Capacity 620.908	No LTB					
Major Moment	Moment 368-641 0.053	Capacity 620.908 53.512	No LTB 620.908	Status				
Major Moment Minor Moment	Mr Moment 368.641 0.053	Capacity 620.908 53.512 Vn/Omega	No LTB 620.908 Stress	Status				
Major Moment Minor Moment	Moment 368-641 0.053	Capacity 620.908 53.512	No LTB 620.908					
Major Moment Minor Moment SHEAR CHECK	Mr Moment 368.641 0.053 Ur Force	Capacity 620.908 53.512 Un/Omega Capacity 136.775	No LTB 620.908 Stress Ratio	Check				
Major Moment Minor Moment SHEAR CHECK Major Shear Minor Shear	Mr Moment 368.641 0.053 Ur Force 88.682 0.002	Capacity 620.908 53.512 Un/Omega Capacity 136.775	No LTB 620.908 Stress Ratio 0.648	Check OK				
Major Moment Minor Moment SHEAR CHECK Major Shear	Mr Noment 368.641 0.053 Ur Force 88.682 0.082 AL LOADS	Capacity 620.908 53.512 Un/Omega Capacity 136.775 676.283	No LTB 620.908 Stress Ratio 0.648	Check OK				
Major Moment Minor Moment SHEAR CHECK Major Shear Minor Shear	Mr Moment 368.641 0.053 Ur Force 88.682 0.002	Capacity 620.908 53.512 Un/Omega Capacity 136.775	No LTB 620.908 Stress Ratio 0.648	Check OK				

Figure 4 101 Steel Stress Check Data form



## NOTE

Unbrace length ratio for major bending is 3.337 which is taken to be distance between two bracing point in major axis (20m) divided by actual length of member (6m). You should carefully review this ratio to ensure that the design process is consistent with your expectations.

To check whether all steel frames passed the stress check, click the **Design menu > Steel Frame Design > Verify All Members Passed** command to display as figure below.

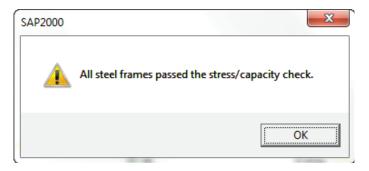


Figure 4 102 Checking capacity of all steel objects

## 4.1.6.5 Check Deflection Limitation

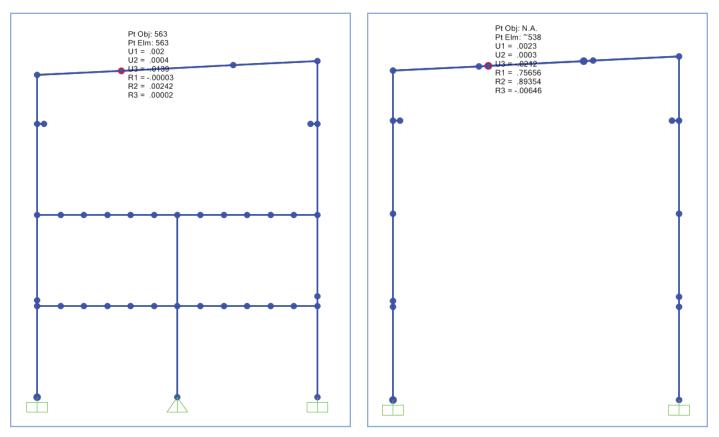
Create combos for deflection checking:

Name	Combination
CV1	DL + 0.4224WL1
CV2	DL + 0.4224WL2
CV3	DL + 0.4224WL3
CV4	DL + 0.4224WL4
CV5	DL + 0.4224WL5
CV6	DL + 0.4224WL6
CV7	DL + 0.4224WL7
CV8	DL + 0.4224WL8
CVN1	DL + 0.75CR1 + 0.3168WL1
CVN2	DL + 0.75CR1 + 0.3168WL2
CVN3	DL + 0.75CR1 + 0.3168WL3
CVN4	DL + 0.75CR1 + 0.3168WL4
CVN5	DL + 0.75CR2 + 0.3168WL1
CVN6	DL + 0.75CR2 + 0.3168WL2
CVN7	DL + 0.75CR2 + 0.3168WL3
CVN8	DL + 0.75CR2 + 0.3168WL4
CVN9	DL + 0.75CR1 + 0.3168WL6
CVN10	DL + 0.75CR1 + 0.3168WL7
CVN11	DL + 0.75CR2 + 0.3168WL6
CVN12	DL + 0.75CR2 + 0.3168WL7
CV	CV1 + CV2 + + CVN12

### Checking if vertical deflection (U3) of mid points of rafter are excess deflection limit.

Compute deflection limit:  $\Delta = \frac{L}{120} = \frac{20}{120} = 0.167 \text{ m}$ 

Put the cursor at mid points of rafter to see if deflection of that point is excess the limit or not.



Deflection of point 563 - 0.0109m

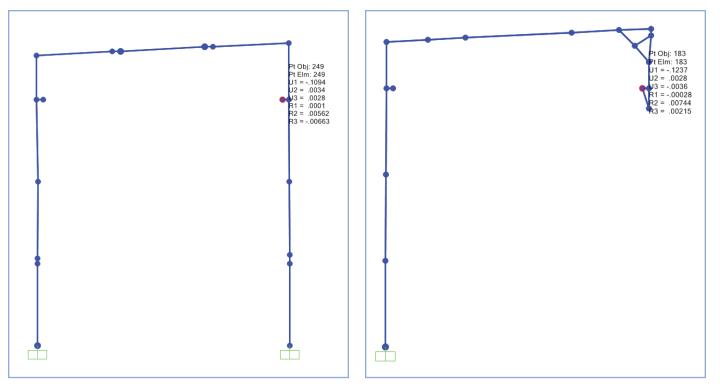
Deflection of point 538 – 0.0212m

Click the Display > Show Deformed Shape command which will show the **Deformed Shape** form. Select **CV** from **Case/Combo Name** drop-down list and click OK to accept.

Checking if horizontal deflection of crane bracket are excess deflection limit.

Compute horizontal deflection limit:  $\Delta = \frac{h}{100} = \frac{19.5}{100} = 0.195m$ 

Put the cursor at points located at the crane bracket to see if the horizontal deflection (U1 or U2) of that point is excess the limit or not.



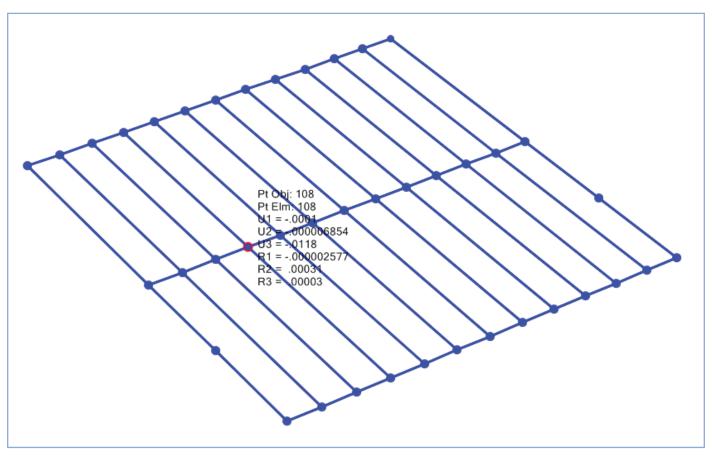
Deflection of point 249 - 0.1094m

Deflection of point 183 - 0.1237m

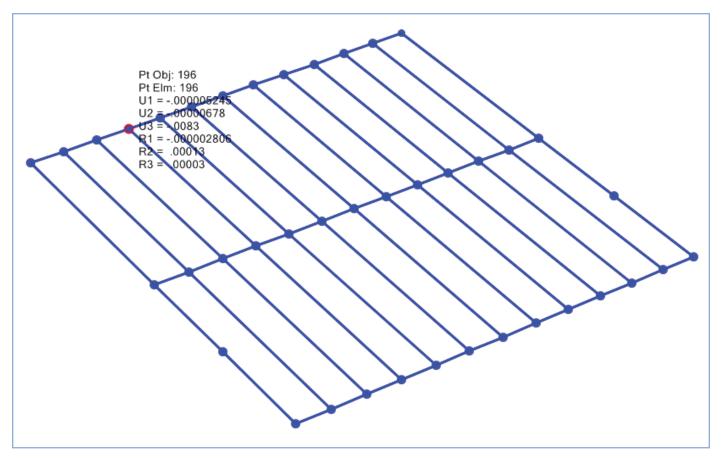
## Checking if vertical deflection of girder are excess deflection limit.

Compute vertical deflection limit:  $\Delta = \frac{L}{240} = \frac{10}{240} = 0.042 \, m$ 

Put the cursor at points located at the middle of girder to see if the vertical deflection (U1 or U2) of that point is excess the limit or not.



Deflection of point 249 - 0.1094m



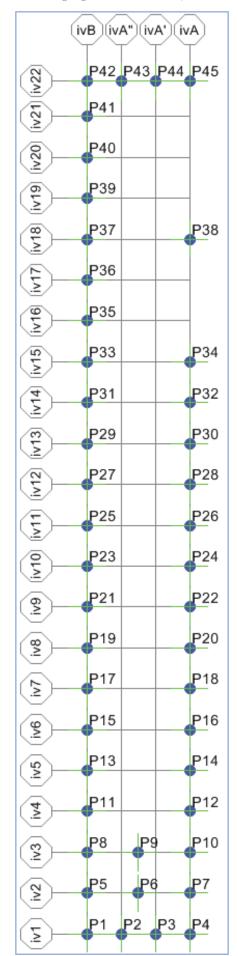
Deflection of point 196 - 0.0083m

# 4.1.6.6 Export joint reaction

Transform from wind load with return period wind speed of 700 years to wind load with those of 50 years

Load Case Data - Linear Static	
Load Case Name Notes WL1 Set Def Name Modify/Show	Load Case Type Static    Design
Stiffness to Use     Zero Initial Conditions - Unstressed State     Stiffness at End of Nonlinear Case     Important Note: Loads from the Nonlinear Case are NOT included     in the current case	Analysis Type C Linear Nonlinear Nonlinear Staged Construction
Loads Applied Load Type Load Name Scale Factor Load Patterr VL1 V.625	Mass Source MSSSRC1
Load Pattern WL1 .625 Add Modify Delete	ок
	Cancel

Figure 4 103 Load Case Data – Linear Static form



Changing label name of joints

Figure 4 104 Labels after change

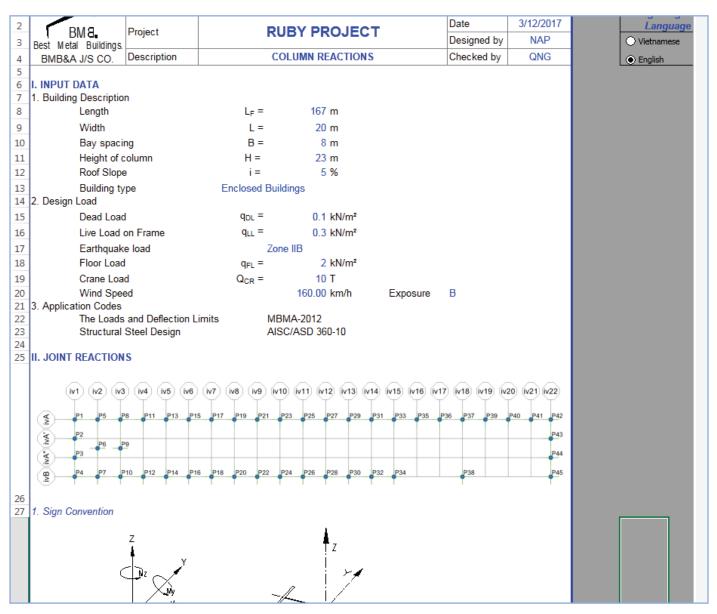


Figure 4 105 Column reactions file



## NOTE

Change Collateral load to Earthquake load in Design load area. Do not insert any more rows or column in this sheet.

Run the model and specify all joints in active plan. Select the **Display > Show Tables** command to show Choose Tables for Display form. Make sure that the Table: Joint Reactions item and DL, LL, FL1, FL2, FL3, CR1, CR2, WL1, WL2, WL3, WL4, WL5, WL6, WL7, WL8, EL1, EL2 in the Select Load Cases form are selected before click the OK button.

As Noted				Joint Reactions						
Joint Text	OutputCase Text	CaseType Text	F1 KN	F2 KN	F3 KN	M1 KN-m	M2 KN-m	M KN-i		
P1	DL	LinStatic	13.56	2.351	145.032	0	0.615	0.000488		
P1	LL	LinStatic	-0.016	0.308	5.022	0	-0.0543	0.00020		
P1	WL1	LinStatic	-34.426	21.371	-115.853	0	-26.1868	-0.007		
P1	WL2	LinStatic	-36.311	11.487	-128.174	0	-28.7126	-0.002		
P1	WL3	LinStatic	-48.534	22.579	-156.256	0	-36.5107	-0.006		
P1	WL4	LinStatic	-38.564	12.268	-128.847	0	-30.3766	-0.001		
P1	WL5	LinStatic	11.909	-25.65	-77.189	0	13.1126	0.003		
P1	WL6	LinStatic	43.611	21.896	217.576	0	33.9604	-0.014		
P1	WL7	LinStatic	36.851	12.104	193.237	0	27.2765	-0.008		
P1	WL8	LinStatic	12.394	0.399	2.517	0	13.1879	0.001		
P1	EL1	LinStatic	-21.793	-0.258	-99.633	0	-15.5845	0.001		
P1	EL2	LinStatic	-0.553	-23.909	-89.724	0	-0.0973	-0.177		
P1	CR1	LinStatic	7.419	-1.959	130.355	0	4.925	-0.029		
P1	CR2	LinStatic	-6.089	-0.002696	-7.305	0	-4.1668	0.000370		
P1	FL1	LinStatic	7.622	0.864	63.127	0	0.3299	0.000219		
P1	FL2	LinStatic	5.923	0.837	48.585	0	-0.6089	0.000440		
P1	FL3	LinStatic	1.698	0.027	14.541	0	0.9387	-0.000220		
P2	DL	LinStatic	-13.02	-0.002652	240.528	0.1646	0	-0.00000659		
P2	LL	LinStatic	-0.217	0.001389	9.284	0.0007146	0	-0.00000652		
P2	WL1	LinStatic	-25.872	71.896	104.692	-340.8842	0	0.000878		
P2	WL2	LinStatic	-25.105	38.505	111.136	-182.5022	0	0.000520		

Figure 4 106 Element Joints Forces – Frame table

_	A	В	C	D	E	F	G	Н	I	J	K	
1	TABLE: J	oint Reactions										Τ
2	Joint	OutputCase	CaseType	1	1	F2	F3	M1	M2	M3		
3	Text	Text	Text	k	N	KN	KN	KN-m	KN-m	KN-m		
4	P1	DL	LinStatic		13.56	2.351	145.032	0	0.615	0.0004886		Τ
5	P1	LL	LinStatic		-0.016	0.308	5.022	0	-0.0543	0.000203		
6	P1	WL1	LinStatic		-34.426	21.371	-115.853	0	-26.1868	-0.0077		
7	P1	WL2	LinStatic		-36.311	11.487	-128.174	0	-28.7126	-0.0025		
8	P1	WL3	LinStatic		-48.534	22.579	-156.256	0	-36.5107	-0.0067		
9	P1	WL4	LinStatic		-38.564	12.268	-128.847	0	-30.3766	-0.0016		
10	P1	WL5	LinStatic		11.909	-25.65	-77.189	0	13.1126	0.0035		
11	P1	WL6	LinStatic		43.611	21.896	217.576	0	33.9604	-0.0141		
12	P1	WL7	LinStatic		36.851	12.104	193.237	0	27.2765	-0.0086		
13	P1	WL8	LinStatic		12.394	0.399	2.517	0	13.1879	0.0017		
14	P1	EL1	LinStatic		-21.793	-0.258	-99.633	0	-15.5845	0.0015		
15	P1	EL2	LinStatic		-0.553	-23.909	-89.724	0	-0.0973	-0.1776		
16	P1	CR1	LinStatic		7.419	-1.959	130.355	0	4.925	-0.0291		
17	P1	CR2	LinStatic		-6.089	-0.002696	-7.305	0	-4.1668	0.0003708		
18	P1	FL1	LinStatic		7.622	0.864	63.127	0	0.3299	0.0002197		
19	P1	FL2	LinStatic		5.923	0.837	48.585	0	-0.6089	0.0004404		
20	P1	FL3	LinStatic		1.698	0.027	14.541	0	0.9387	-0.0002208		
21	P2	DL	LinStatic		-13.02	-0.002652	240.528	0.1646	0	-0.000006594		
22	P2	LL	LinStatic		-0.217	0.001389	9.284	0.0007146	0	-0.000006527		
23	P2	WL1	LinStatic		-25.872	71.896	104.692	-340.8842	0	0.0008783		
24	P2	WL2	LinStatic		-25.105	38.505	111.136	-182.5022	0	0.0005207		
25	P2	WL3	LinStatic		-38.691	71.907	188.878	-340.9069	0	0.0008838		
26	P2	WL4	LinStatic		-27.666	38.511	151.135	-182.4868	0	0.0005309		
27	P2	WL5	LinStatic		2.47	-65.006	-41.431	326.4516	0	-0.0006454		
28	P2	WL6	LinStatic		31.745	71.869	-203.171	-341.1807	0	0.0004952		T
29	P2	WL7	LinStatic		29.694	38.479	-184.574	-182.7928	0	0.0001412		Ť
30	P2	WL8	LinStatic		2.31	0.571	-39.448	-3.3364	0	0.0001707		t
31	P2	EL1	LinStatic		-18.885	0.0062	95.678	0.0953	0	0.0001161		Ť
32	P2	EL2	LinStatic		0.251	-45.597	-5.425	259.1744	0	-0.0005969		T
33	P2	CR1	LinStatic		4.892	-0.054	-24.413	0.4486	0	0.00000749		t
34	P2	CR2	LinStatic		-6.019	0.003564	43.775	0.045	0	0.00001829		t
35	P2	FL1	LinStatic		-7.292	-0.001562	101.217	0.0874	0	0.00000921		t
36	P2	FL2	LinStatic		-8.073	0.009305	61.738	0.017	0	0.00004952		t
37	P2	FL3	LinStatic		0.782	-0.011	39.479	0.0705	0	-0.0000486		t
38	P3	DL	LinStatic		12.685	-0.007224	236.806	0.206	0	-0.00001481		t
39	P3	LL	LinStatic		-0.051	-0.003801	5.423	0.0475	0	-0.00001405		t
40	P3	WL1	LinStatic		-26.321	72.072	-155.079	-345.5814	0	-0.0013		t
41	P3	WL2	LinStatic		-25.027	38.573	-144.142	-184.9689	0	-0.0006286		t

Figure 4 107 Data of element joint forces after filter and sort

1	TABLE:	Joint Read	tione									
2	Joint		s(CaseType	StenTyne	F1	F2	F3	M1	M2	M3		
3	Text	Text	Text	Text	KN	KN	KN	KN-m	KN-m	KN-m		
4	P1	DL	LinStatic		13.56	2.351	145.032	0	0.615	0.0004886		
5	P1	LL	LinStatic		-0.016	0.308	5.022	0	-0.0543	0.000203		
5	P1	WL1	LinStatic		-34.426	21.371	-115.853	0	-26.1868	-0.0077		
7	P1	WL2	LinStatic		-36.311	11.487	-128.174	0	-28.7126	-0.0025		
3	P1	WL3	LinStatic		-48.534	22.579	-156.256	0	-36.5107	-0.0067		
9	P1	WL4	LinStatic		-38.564	12.268	-128.847	0	-30.3766	-0.0016		
0	P1	WL5	LinStatic		11.909	-25.65	-77.189	0	13.1126	0.0035		
1	P1	WL6	LinStatic		43.611	21.896	217.576	0	33.9604	-0.0141		
2	P1	WL7	LinStatic		36.851	12.104	193.237	0	27.2765	-0.0086		
3	P1	WL8	LinStatic		12.394	0.399	2.517	0	13.1879	0.0017		
	P1	EL1	LinStatic		-21.793	-0.258	-99.633	0	-15.5845	0.0015		
15	P1	EL2	LinStatic		-0.553	-23.909	-89.724	0	-0.0973	-0.1776		
	P1	CR1	LinStatic		7.419	-1.959	130.355	0	4.925	-0.0291		
	P1	CR2	LinStatic		-6.089	-0.002696	-7.305	0	-4.1668	0.0003708		
18		FL1	LinStatic		7.622	0.864	63.127	0	0.3299	0.0002197		
_	P1	FL2	LinStatic		5.923	0.837	48.585	0	-0.6089	0.0004404	 	•
20	P1	FL3	LinStatic		1.698	0.027	14.541	0	0.9387	-0.0002208		
1	P2	DL	LinStatic		-13.02	-0.002652	240.528	0.1646	0	-6.594E-06		
22	P2	LL	LinStatic		-0.217	0.001389	9.284	0.0007146	0	-6.527E-06		
23	P2	WL1	LinStatic		-25.872	71.896	104.692	-340.8842	0	0.0008783		
24	P2	WL2	LinStatic		-25.105	38.505	111.136	-182.5022	0	0.0005207		
25	P2	WL3	LinStatic		-38.691	71.907	188.878	-340.9069	0	0.0008838		
26	P2	WL4	LinStatic		-27.666	38.511	151.135	-182.4868	0	0.0005309		
27	P2	WL5	LinStatic		2.47	-65.006	-41.431	326.4516	0	-0.0006454		
28	P2	WL6	LinStatic		31.745	71.869	-203.171	-341.1807	0	0.0004952		
9	P2	WL7	LinStatic		29.694	38.479	-184.574	-182.7928	0	0.0001412		
30	P2	WL8	LinStatic		2.31	0.571	-39.448	-3.3364	0	0.0001707		
31	P2	EL1	LinStatic		-18.885	0.0062	95.678	0.0953	0	0.0001161		
32	P2	EL2	LinStatic		0.251	-45.597	-5.425	259.1744	0	-0.0005969		
33	P2	CR1	LinStatic		4.892	-0.054	-24.413	0.4486	0	0.00000749		
34	P2	CR2	LinStatic		-6.019	0.003564	43.775	0.045	0	0.00001829		
35	P2	FL1	LinStatic		-7.292	-0.001562	101.217	0.0874	0	9.21E-07		
86	P2	FL2	LinStatic		-8.073	0.009305	61.738	0.017	0	0.00004952		
37	P2	FL3	LinStatic		0.782	-0.011	39.479	0.0705	0	-0.0000486		
38	P3	DL	LinStatic		12.685	-0.007224	236.806	0.206	0	-0.00001481		
39	P3	LL	LinStatic		-0.051	-0.003801	5.423	0.0475	0	-0.00001405		
10	P3	WL1	LinStatic		-26.321	72.072	-155.079	-345.5814	0	-0.0013		
1	P3	WL2	LinStatic		-25.027	38.573	-144.142	-184.9689	0	-0.0006286		

Figure 4 108 Data after paste

Select area from cell "W11" to cell "Z11" and click the Home > Insert command to insert more cells above active area. Revise load case name as shown in figure below.

W	Х	Y	Z
DL	Dead Load		
AL	Collateral I	Load	
LL	Live Load		
WL1	Left Windle	oad Case1	
WL2	Left Windle	oad Case2	
WL3	Left Windle	bad Case3	
WL4	Left Windle	bad Case4	
WL5	Left Windle	oad-Y	
WL6	Right Wind	lload Case	1
WL7	Right Wind	lload Case	2
WL8	Right Wind	lload-Y	

Move to sheet "SteelColumn" and click the HIDE button.

DESIGN GUIDELINES

A 1. Sign Co	B C	D	E	F	G	Н	I	J	К
2. Reactio	Z Joint V Joint	<b>-</b> X	1	2 74	- *				
Joint	OutputCase	Horizontal Reaction Hx	Horizontal Reaction Hy	Vertical Reaction Vz	Moment Mx	Moment My	Moment Mz	HIRE	
Text	Text	KN	KN	KN	KN-m	KN-m	KN-m		
Point-P1	Dead Load	0.0	0.0	4.1	0.0	0.0	0.0		
Point-P1	Live Load	0.1	0.0	3.5	0.0	0.0	0.0		
Point-P1	Left Windload Case1	-2.3	0.0	-5.0	0.0	0.0	0.0		
Point-P1	Left Windload Case2	-3.9	0.0	-4.6	0.0	0.0	0.0		
Point-P1	Wind load Y	0.0	-2.6	-9.2	0.0	0.0	0.0		
Point-P2	Dead Load	0.0	0.0	5.4	0.0	0.0	0.0		
Point-P2	Live Load	0.0	0.0	5.9	0.0	0.0	0.0		
Point-P2	Left Windload Case1	0.0	0.0	-9.0	0.0	0.0	0.0		
Point-P2	Left Windload Case2	0.0	0.0	-6.4	0.0	0.0	0.0		
Point-P2	Wind load Y	0.0	-4.2	1.6	0.0	0.0	0.0		
Point-P3	Dead Load	0.0	0.0	5.8	0.0	0.0	0.0		
Point-P3	Live Load	0.0	0.0	6.4	0.0	0.0	0.0		
Point-P3	Left Windload Case1	0.0	0.0	-8.8	0.0	0.0	0.0		
Point-P3	Left Windload Case2	0.0	0.0	-6.1	0.0	0.0	0.0		
Point-P3	Wind load Y	0.0	-4.6	0.0	0.0	0.0	0.0		
Point-P4	Dead Load	0.0	0.0	5.5	0.0	0.0	0.0		
Point-P4	Live Load	0.0			0.0	0.0	0.0		
Point-P4	Left Windload Case1	0.0		-	0.0	0.0	0.0		
Point-P4	Left Windload Case2	0.0			0.0	0.0	0.0		
Point-P4	Wind load Y	0.0	-5.0	-0.8	0.0	0.0	0.0		
Point-P5	Dead Load	0.0	0.0	6.1	0.0	0.0	0.0		
Point-P5	Live Load	0.0	0.0	6.1	0.0	0.0	0.0		
Point-P5	Left Windload Case1	0.0	0.0	-8.8	0.0	0.0	0.0		
Point-P5	Left Windload Case2	0.0	0.0	-6.0	0.0	0.0	0.0		
Point-P5	Wind load Y	0.0	-5.3	0.3	0.0	0.0	0.0		
Point-P6	Dead Load	0.0	0.0	4.3	0.0	0.0	0.0		
Point-P6	Live Load	0.0	0.0	4.3	0.0	0.0	0.0		
Point-P6	Left Windload Case1	0.1	0.0	-3.5	0.0	0.0	0.0		
Point-P6	Left Windload Case2	0.1	0.0	-2.8	0.0	0.0	0.0		
Point-P6	Wind load Y	0.0	-3.3	-9.4	0.0	0.0	0.0		
Point-P7	Dead Load	0.0	0.0	6.1	0.0	0.0	0.0		
Point-P7	Live Load	0.0	0.0	6.1	0.0	0.0	0.0		
	Left Windload Case1	0.0	0.0		0.0	0.0	0.0		
Point-P7	Leit Windiodu Odber	0.0	0.01	-0.01	0.01	0.01	0.0		

#### Figure 4 109 Joint reactions after hide rows

Select area from cell "A1" to cell "I795" and select the Page Layout > Print Area > Set Print Area command. Recheck all information; make sure that Vertical Reactions Vz for Dead load and Live load are always positive.

# 4.2. Pinned Base Plate Design

# 4.2.1. Input 4.2.1.1 Loading

Run the model and specify all joints that connecting as pin to foundation and have the same column section. Select the **Display** > Show Tables command to show Choose Tables for Display form. Make sure that the Table: Joint Reactions item and all combinations in the Select Load Cases form are selected before click the OK button.

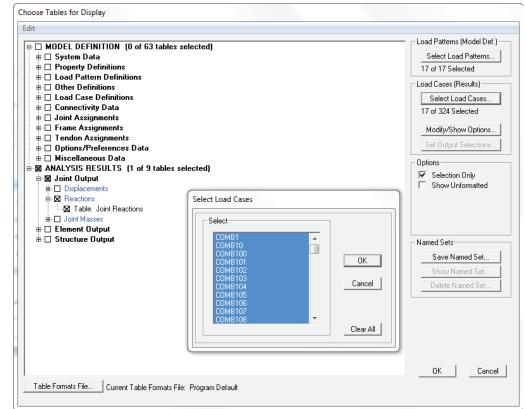


Figure 4 110 Choose Tables for Display form

Units: A	\s Noted				Joint F	Reactions			
	Joint	OutputCase	CaseType	F1  F2  F3  M1  M2					
	Text	Text	Text	KN	KN	KN	KN-m	KN-m	KN-m
	P6	COMB1	Combination	0.094	0.001481	664.18	0	0	0
	P6	COMB2	Combination	-4.435	0.053	669.033	0	0	0
	P6	COMB3	Combination	-4.083	0.029	668.896	0	0	0
	P6	COMB4	Combination	-4.399	0.053	666.395	0	0	0
	P6	COMB5	Combination	-3.979	0.03	666.402	0	0	0
	P6	COMB6	Combination	-0.107	0.009685	667.222	0	0	0
	P6	COMB7	Combination	4.411	0.053	669.472	0	0	0
	P6	COMB8	Combination	4.177	0.029	669.31	0	0	0
	P6	COMB9	Combination	-0.128	-0.005679	666.773	0	0	0
	P6	COMB10	Combination	-3.278	0.04	666.989	0	0	0
	P6	COMB11	Combination	-3.014	0.022	666.887	0	0	0
	P6	COMB12	Combination	-3.251	0.04	665.011	0	0	0
	P6	COMB13	Combination	-2.935	0.023	665.016	0	0	0
	P6	COMB14	Combination	-0.031	0.007679	665.631	0	0	0
	P6	COMB15	Combination	3.357	0.04	667.318	0	0	0
	P6	COMB16	Combination	3.182	0.022	667.197	0	0	0
	P6	COMB17	Combination	-0.047	-0.003844	665.294	0	0	0
	P6	COMB18	Combination	0.7	0.002414	666.219	0	0	0
	P6	COMB19	Combination	-0.634	0.002213	666.093	0	0	0
	P6	COMB20	Combination	-2.823	0.041	668.518	0	0	0
	P6	COMB21	Combination	-3.824	0.041	668.423	0	0	0

Figure 4 111 Joint Reactions table

In excel file just exported, filter largest value in "F3" column and enter this value and correspondence shear to Pmax and Vp in "Base Plate Calculation" file.

In Joint Reactions tables, select File > Export Current Table > To Excel.



## NOTE

All value in Joint Reactions file are counter sign with value in "Base Plate Calculation" file, so we have to change the sign when enter axial force into calculation file. Shear forces will be taken as absolute values in all cases.

Do as above manner, filter the smallest value in "F3" column and enter this value and correspondence shear to Tmax and VT, filter largest absolute value in "F1" and "F2" column and take them to  $V_{max}$  and  $T_{V}$ .

1. LOADS		
Pmax =	-992.10	kΝ
V <sub>p</sub> =	0.05	kΝ
T <sub>max</sub> =	1.00	kΝ
V <sub>T</sub> =	0.16	kΝ
V <sub>max</sub> =	7.10	kΝ
T <sub>V</sub> =	1.00	kΝ

Maximum compression Correspondence shear Maximum tension Correspondence shear Maximum Shear Correspondence tension



## NOTE

Compressive force will have negative sign "-", and tensile force will have positive sign "+" or no sign. If there is not tensile force at base, value for it will be taken as 1.

# 4.2.1.2 Material, geometry parameter

In order to illustrate a pinned base plate connection with full of dangerous cases, we assume information as figure below.

	A B	С	D E	F	G	Н	0	
4	BMB&A J/S CO.	Description	PINNED BASE PI		Checked			Engli:
5								
	I. INPUT							
	1. LOADS							Cor
7	Pmax = -200.00	1 PM	Maximum compression		A T			
9	$V_p = 20.00$	-	Correspondence shear		<b>•</b>			Ba
10	T <sub>max</sub> = 150.00	-	Maximum tension			,		
11	$V_{\rm T} = 20.00$		Correspondence shear					N
12	V <sub>max</sub> = 150.00	_	Maximum Shear			-		
13	T <sub>V</sub> = 50.00	) kN	Correspondence tension					
14		_			1			
15	2. MATERIAL				血 血			Ма
16	Anchor Bolt Properties					<u> </u>		
17	Type: Cast-ir	ı	Hooked Bolt	-				
18	d <sub>a</sub> = M30	)	Anchor Bolt Diameter	1				
19	5.6	5	Anchor Bolt Material	i.		1.1.1		M
20	F <sub>u</sub> = 50.00	) kN/cm <sup>2</sup>	Ultimate strength of bolt			44 . 4		
21		) kN/cm²	Nominal tensile strength		₄, Ų, Ų,			
22		5 kN/cm²	Nominal shear strength			- <u>A</u>		
	ASTM specifications for		A572 G50		Ť			
24	, , , , , , , , , , , , , , , , , , , ,	-	Yield strength					
25	E43x		Welding wire					
26		5 mm	Flange weld size					
27		3 mm	Web weld size					
28	Concrete Properties	) khl/om2	B25					
29 32	f' <sub>c</sub> = 2	2 kN/cm²	The compressive strength		N			
	3. GEOMETRY & PARA	METED			, S <sub>x</sub>	_		
33 34	Column section					· · · · ·		
35		) mm	Height 1	- 3.	4			
36		1 mm	Width		1	1 <sup>4</sup> .		
37		5 mm	Web thickness		<b>\$</b>	1		
38		3 mm	Flange thickness	ທີ່				
39	Base Plate Dimension	_	-		* *	4		
40	N = 320	) mm	Height					
41	B = 212	2 mm	Width	- 4	4 2			
42	TK = 20	) mm	Thickness					
43	Anchor Bolt Geometry				_			
44	n = 4	1	Number of inside bolts:		2 Bolts/1 inside row			
45	s <sub>x</sub> = 100	) mm	Spacing between anchors x di	rection				
46	s <sub>y</sub> = 100	) mm	Spacing between anchors y di	irection				
47	h <sub>ef</sub> = 650	) mm	Effective Embedment length					
	→ BT StartSh	eet <b>PINNE</b>	D FIXED PARTIAL FIXED	<b>(+)</b>				Þ

# 4.2.2. Checking base plate 4.2.2.1 Base plate for concentric axial compressive load

Calculate base plate area	$A_1 = NB = 320 \times 212 = 67840  mm^2$	678.4 cm <sup>2</sup>
Assume spacing between outermost	anchor bolt and concrete edge is 200 mm	
Length of concrete foundation	$N_{c} = \left(\frac{n}{2} - 1\right) s_{x} + 2 * 200 = \left(\frac{4}{2} - 1\right) 100 + 2 * 200$	500 mm
Width of concrete foundation	$B_c = s_y + 2x200 = 100 + 2x200$	500 mm

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Calculate effective concrete area	$\mathcal{A}_{2} = \min\left(N_{c} \frac{B}{N} N_{c}, N_{c} B_{c}\right)$ $\mathcal{A}_{2} = \min\left(500 \times \frac{212}{320} \times 500, 500 \times 500\right)$	1656.25 cm²
Maxlimum bearing stress between plate and concrete	$f_{p(\max)} = \frac{1}{\Omega} 0.85 f_c' \sqrt{\frac{A_2}{A_1}}$ $f_{p(\max)} = \frac{1}{2.31} 0.85 \times 20000 \sqrt{\frac{1656.25}{678.4}}$	11498.92 kN/m²
Allowable bearing force	$\frac{P_p}{\Omega} = f_{p(\text{max})} \mathcal{A}_1 = 11498.92 \times 678.4 \times 10^{-4}$	780.1 kN
Critical base plate cantilever dimension, I, is the larger of m, n	$m = \frac{N - 0.95d}{2} = \frac{320 - 0.95x300}{2}$	17.5 mm
and λn':	$n = \frac{B - 0.8b_f}{2} = \frac{212 - 0.8\times 184}{2}$	32.4 mm
	$X = \frac{4db_f}{\left(d + b_f\right)^2} \frac{\Omega P_{\text{max}}}{P_p} = \frac{4x300x184}{\left(300 + 184\right)^2} \frac{200}{780.1}$	0.24
	$\lambda = \frac{2\sqrt{X}}{1 + \sqrt{1 - X}} = \frac{2x\sqrt{0.24}}{1 + \sqrt{1 - 0.24}}$	0.53
	$\lambda n' = \frac{\lambda \sqrt{db_f}}{4} = 0.53 \frac{\sqrt{300 \times 184}}{4}$	30.87 mm
	$l = \max(m, n, \lambda n') = \max(17.5, 32.4, 30.87)$	32.4 mm
For the yielding limit state, the required minimum thickness of the base plate	$\frac{P_{\max}}{BN}\frac{l^2}{2} = \frac{1}{\Omega}F_y\frac{t_p^2}{4} \Rightarrow t_p = l\frac{\sqrt{2\Omega}P_{\max}}{F_yBN}$ $t_p = 32.4x\sqrt{\frac{2x1.67x200}{34.5x212x320}}$	5.47 mm

# 4.2.2.2 Base plate for uplift load

Tension for each anchor bolt	$\frac{T_{\max}}{n} = \frac{150}{4}$	37.5 mm
Distance from bolt rows to column web	$a = \frac{s_y - t_w}{2} = \frac{100 - 5}{2}$	47.5 mm
Moment caused by tension on bolt	$M_a = \frac{T_{\text{max}}}{n}a = 37.5 \times 47.5$	1781.25 kNmm
Width of effective area to assist	$b_{eff} = 2a = 2x47.5$	95 mm

bending moment		
Required minimum thickness of base plate	$\frac{1}{\Omega}F_{y}\frac{b_{eff}t_{p}^{2}}{4} = M_{a} \Longrightarrow t_{p} = \sqrt{\frac{4\Omega M_{a}}{F_{y}b_{eff}}} = \sqrt{\frac{4\times1.67\times1781.25}{0.345\times95}}$	19.05 mm
Demand/capacity thickness of base plate	$\frac{\max(5.47, 19.05)}{20} = 0.953 < 1$	OK

# 4.2.2.3 Design welding

# Weld capacity due to anchor tension

Load angle factor when force perpendicular with welding	$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90)$	1.5
Width of effective area resisting tension force of one bolt	$b_{eff} = 2a = 2x47.5$	95 mm
Allowable bearing capacity of column web weld	$\frac{R_{w}}{\Omega} = \frac{1}{\Omega} \chi 0.6F_{EXX} \frac{\sqrt{2}}{2} h_{f_{-}web} = \frac{1}{2} \times 1.5 \times 0.6 \times 43 \times \frac{\sqrt{2}}{2} \times 0.3$	4.10 kN/cm
Force apply on column web weld	$\frac{T_{\max}}{nb_{eff}} = \frac{150}{4x9.5}$	3.95 kN/cm
Demand/capacity welding	$\frac{3.96}{4.1} = 0.962 < 1$	OK

# Elastic method weld shear capacity

Load angle factor when force parallel with welding	$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(0)$	1.0
Total web weld length	$L_{shear} = 2d = 2x300 = 600$	600 mm
Allowable bearing capacity of column web weld	$\frac{R_{w}}{\Omega} = \frac{1}{\Omega} \chi 0.6F_{EXX} \frac{\sqrt{2}}{2} b_{f_{-web}} = \frac{1}{2} \times 1.0 \times 0.6 \times 43 \times \frac{\sqrt{2}}{2} \times 0.3$	2.74 kN/cm
Force apply on column web weld	$\frac{V_{\max}}{L_{sbear}} = \frac{150}{60}$	2.5 kN/cm
Demand/capacity welding	$\frac{2.5}{2.74} = 0.914 < 1$	OK

# Elastic method weld axial capacity

Load angle factor when force perpendicular with welding	$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90)$	1.5
Total flange weld length	$L = 4b_f = 4x184$	736 mm
Allowable bearing capacity of column web weld	$\frac{R_{\mu\nu}}{\Omega} = \frac{1}{\Omega} \chi 0.6F_{EXX} \frac{\sqrt{2}}{2} b_{f\_\text{flange}} = \frac{1}{2} \times 1.5 \times 0.6 \times 43 \times \frac{\sqrt{2}}{2} \times 0.5$	6.84 kN/cm

Force apply on column flange weld	$\frac{T_{\text{max}}}{L} = \frac{150}{73.6}$	2.04 kN/cm
Demand/capacity welding	$\frac{2.04}{6.84} = 0.298 < 1$	OK

# 4.2.3. Anchor bolt checking

# **4.2.3.1 Checking for combination maximum tension** Steel strength of a single anchor in tension

Allowable tensile capacity of each bolt	$\frac{R_n}{\Omega} = \frac{1}{\Omega} F_{nt} \frac{\pi d_b^2}{4} = \frac{1}{2} 37.5 \times \frac{\pi 3^2}{4}$	132.54 kN
Required tension force for each bolt	$\frac{T}{n} = \frac{150}{4}$	37.5 kN
Demand/Capacity	$\frac{37.5}{132.54} = 0.283 < 1$	OK

# Pullout of anchor in tension

The pullout strength in tension of a single hooked bolt	$\frac{N_{pn}}{\Omega} = \psi_{c,P} \frac{0.9 f'_c e_b d_a}{\Omega} = 1.4 \frac{0.9 \times 2 \times 4.5 \times 3 \times 3}{2.14}$	47.69 kN
The pullout strength in tension of a s	ingle hooked bolt:	
Assume there is no cracking in concr	rete so modification factor for pullout shall be taken as 1.	.4
Effective dimension to compute pulle	out strength of hooked bolt eh shall be taken as $4.5d_a$	
Demand/Capacity	$\frac{37.5}{47.69} = 0.786 < 1$	ОК

# Steel strength of a single anchor in tension

Allowable shear capacity of each bolt	$\frac{V_{sa}}{\Omega} = \frac{0.8F_{m}\pi d_b^2 / 4}{\Omega} = \frac{0.8 \times 22.5\pi \times 3^2 / 4}{2}$	63.62 kN
Demand/Capacity	$\frac{20}{4\times63.62} = 0.079 < 1$	OK

# Interaction of tensile and shear forces of anchor

Demand/Capacity	$\frac{0.786 + 0.079}{1.2} = 0.721 < 1$	ОК	
	1.2		

# 4.2.3.2 Checking for combination maximum shear

# Steel strength of a single anchor in tension

Allowable tensile capacity of each bolt	$\frac{R_{n}}{\Omega} = \frac{1}{\Omega} F_{nt} \frac{\pi d_{b}^{2}}{4} = \frac{1}{2} 37.5 \times \frac{\pi 3^{2}}{4}$	132.54 kN
Required tension force for each bolt	$\frac{T}{n} = \frac{50}{4}$	12.5 kN
Demand/Capacity	$\frac{12.5}{132.54} = 0.094 < 1$	OK

## Pullout of anchor in tension

The pullout strength in tension of a single hooked bolt	$\frac{N_{pn}}{\Omega} = \psi_{c,P} \frac{0.9 f_c' e_b d_a}{\Omega} = 1.4 \frac{0.9 \times 2 \times 4.5 \times 3 \times 3}{2.14}$	47.69 kN
, j	rete so modification factor for pullout shall be taken a out strength of hooked bolt eh shall be taken as 4.50	
Demand/Capacity	$\frac{12.5}{47.69} = 0.262 < 1$	OK

# Steel strength of anchors in shear

Allowable shear capacity of each bolt	$\frac{V_{sa}}{\Omega} = \frac{0.8F_{m}\pi d_{b}^{2}/4}{\Omega} = \frac{0.8\times22.5\times\pi\times3^{2}/4}{2}$	63.62 kN
Demand/Capacity	$\frac{150}{4x63.62} = 0.589 < 1$	OK

# Interaction of tensile and shear forces of anchor

Demand/Capacity	$\frac{0.262 + 0.589}{1.2} = 0.710 < 1$	OK	

# 4.3. Fixed Base Plate Design

# 4.3.1. Loading

Run the model and specify all joints that connecting as fixed to foundation and have the same column section. Select the **Display > Show Tables** command to show **Choose Tables for Display** form. Make sure that the Table: Joint Reactions item and **all combinations** in the **Select Load Cases** form are selected before click the OK button.

COMB102	Show Named Set
COMB103	Show Named Set Delete Named Set

Figure 4 113 Choose Tables for Display form

In Joint Reactions tables, select File > Export Current Table > To Excel.

	iew For <u>m</u> a As Noted	t-Filter-Sort <u>S</u> e	lect <u>O</u> ptions		Joint F	leactions			
	Joint Text	OutputCase Text	CaseType Text	F1 KN	F2 KN	F3 KN	M1 KN-m	M2 KN-m	M3 KN-m
	P5	COMB1	Combination	26.953	1.022	349.839	0	53.6876	-0.0003906
<u> </u>	P5	COMB2	Combination	-9.806	10.255	227.301	0	-103.2834	0.0471
	P5	COMB3	Combination	-11.15	5.887	257.215	0	-97,5585	0.0251
	P5	COMB4	Combination	-9.549	10.97	265.522	0	-105.5586	0.047
	P5	COMB5	Combination	-10.216	6.336	295.753	0	-95.646	0.025
	P5	COMB6	Combination	38.446	-12.021	324.417	0	65.7453	-0.046
	P5	COMB7	Combination	65.461	10.438	263.595	0	214.627	0.0472
	P5	COMB8	Combination	58.186	6.129	292.15	0	199.6791	0.0251
	P5	COMB9	Combination	38.301	0.969	276.608	0	64.8485	0.0001157
	P5	COMB10	Combination	-0.306	8.093	269.815	0	-63.3422	0.0352
	P5	COMB11	Combination	-1.314	4.817	292.251	0	-59.0485	0.0187
	P5	COMB12	Combination	-0.113	8.629	298.481	0	-65.0486	0.0351
	P5	COMB13	Combination	-0.613	5.153	321.154	0	-57.6141	0.0186
	P5	COMB14	Combination	35.883	-8.614	342.653	0	63.4294	-0.0346
	P5	COMB15	Combination	56.144	8.23	297.037	0	175.0907	0.0353
	P5	COMB16	Combination	50.688	4.999	318.452	0	163.8797	0.0187
	P5	COMB17	Combination	35.774	1.129	306.796	0	62.7567	-0.00004078
	P5	COMB18	Combination	31.476	-1.394	455.604	0	80.5156	-0.00005959
	P5	COMB19	Combination	21.24	0.709	359.368	0	22.9615	-0.0003486
	P5	COMB20	Combination	3.086	6.281	349.139	0	-43.2212	0.0355
	P5	COMB21	Combination	-4.59	7.858	276.962	0	-86.3868	0.0353

#### Figure 4 114 Joint Reactions table

In excel file just exported, filter largest value in "F3" column and enter this value, correspondence shear and moment M2 to  $P_{max}$ ,  $V_p$  and  $M_p$  in "Base Plate Calculation" file.

# NOTE



All values in Joint Reactions file are countered sign with value in "Base Plate Calculation" file, so we have to change the sign when enter axial force into calculation file. Shear forces and moment will be taken as absolute values in all cases.

Do as above manner, filter the smallest value in "F3" column and enter this value and correspondence shear and moment M2 to  $T_{max}$ ,  $V_T$  and  $M_T$ 

Filter all positive values of F3, and then filter the largest absolute value of M2. Enter this value, correspondence compressive force and shear to  $M1_{max}$ ,  $P_{M1}$  and  $V_{M1}$ .

Filter all negative values of F3, and then filter the largest absolute value of M2. Enter this value, correspondence tensile force and shear to  $M2_{max}$ ,  $P_{M2}$  and  $V_{M2}$ .

Filter all negative values of F3, and then filter the largest absolute value of F1. Enter this value, correspondence tensile force and shear to  $V_{max}$ ,  $T_v$  and  $M_v$ .

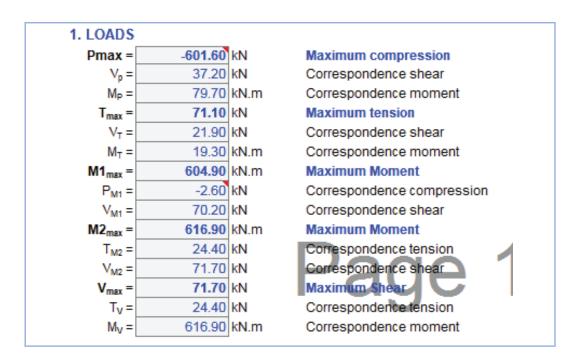


Figure 4 115 Internal force for calculation

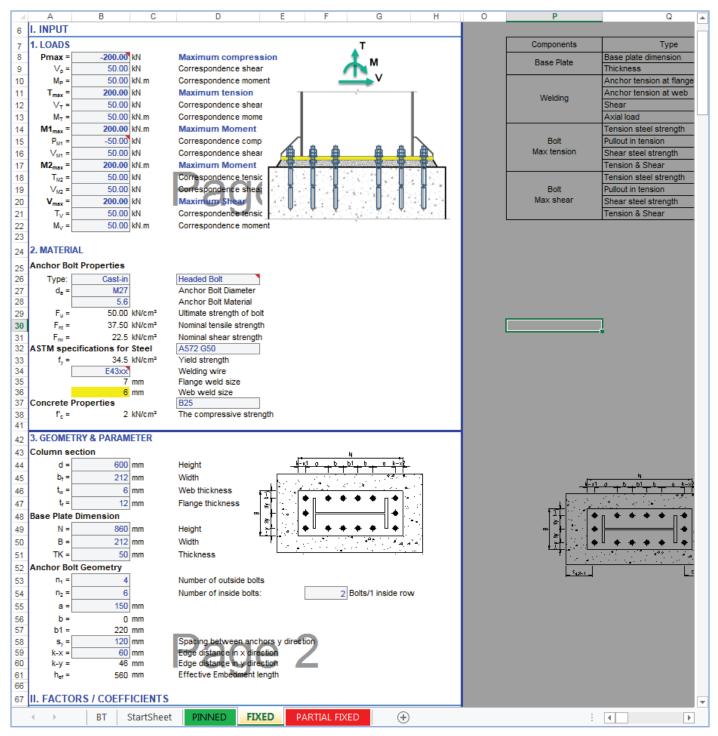


# NOTE

Compressive force will have negative sign "-", and tensile force will have positive sign "+" or no sign. If there is not tensile force at base, value for it will be taken as 1.

## 4.3.2. Material, geometry parameter

In order to illustrate a fixed base plate connection with full of dangerous cases, we assume information as figure below.



# 4.3.3. Checking base plate

Calculate base plate area	$A_1 = NB = 860 \times 212$	1823.2 cm²
Calculate concrete area:		
Assume spacing between outermost	anchor bolt and concrete edge is 200 mm	
Length of concrete foundation	$N_{c} = N + (c_{a,x-1} - k_{x1}) + (c_{a,x-2} - k_{x2})$	1140 mm
	$N_c = 860 + (200 - 60) + (200 - 60)$	

Width of concrete foundation	$B_{c} = B + (c_{a,y-1} - k_{y1}) + (c_{a,y-2} - k_{y2})$ $B_{c} = 212 + (200 - 46) + (200 - 46)$	520 mm
Calculate effective concrete area	$\mathcal{A}_{2} = \min\left(N_{c}\frac{B}{N}N_{c}, N_{c}B_{c}\right)$ $\mathcal{A}_{2} = \min\left(1140 \times \frac{212}{860} \times 1140, 1140 \times 520\right)$	320366.5 mm <sup>2</sup>
Maximum bearing stress between plate and concrete	$f_{p(\max)} = \frac{1}{\Omega} 0.85 f_c' \sqrt{\frac{A_2}{A_1}} = \frac{1}{2.31} 0.85 \times 20000 \times \sqrt{\frac{3203.7}{1823.2}}$	9755.36 kN/m²
Maximum resultant bearing force	$q_{\max} = f_{p(\max)}B = 9755.36 \times 0.212$	2068.14 kN/m
Critical base plate cantilever dimension is the larger of m, n:	$m = \frac{N - 0.95d}{2} = \frac{860 - 0.95\times600}{2}$	145 mm
	$n = \frac{B - 0.8b_f}{2} = \frac{212 - 0.8x212}{2}$	21.2 mm

# 4.3.3.1 Checking base plate with combo Mp + Pmax

The equivalent eccentricity of base plate	$e = \frac{M_P}{P_{\text{max}}} = \frac{50}{200} = 0.25 \mathrm{m}$	250 mm
The critical eccentricity	$e_{crit} = \frac{N}{2} - \frac{P}{2q_{\max}} = \frac{860}{2} - \frac{200}{2x2068.14}$	381.65 mm
Conclusion	$e < e_{crit} \Rightarrow$ Case of a base plate with small moment $\Rightarrow$ No anchor rod forces exists.	
The concrete bearing length	Y = N - 2e = 860 - 2x250	360 mm
Bearing stress between base plate and concrete	$f_p = \frac{P}{BY} = \frac{200}{0.212 \times 0.360}$	2620.55 kN/m <sup>2</sup>
Determine minimum plate thickness with m (Y>m)	$\frac{1}{\Omega}F_{y}\frac{Bt_{p}^{2}}{4} = f_{p}B\frac{m^{2}}{2}$	23.13 mm
	$ \rightarrow t_{p} = m \sqrt{\frac{2\Omega f_{p}}{F_{y}}} = 145 \sqrt{\frac{2x1.67x2620.55}{34.5x10^{4}}} $	

# 4.3.3.2 Checking base plate with combo $T_{max} + M_T$

It's always the case of base plate with large moment when considering combo momnent with tension.

$$f_p = f_{p,\max} = 9755.36 \frac{kN}{m^2}$$

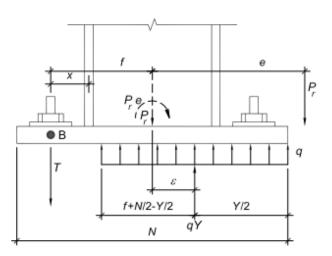


Figure 3.4.1. Base plate with large moment.

Determine bearing length Y:

- Equilibrium momentum with rotate center at the edge of compression concrete zone:

$$\begin{aligned} &\frac{P_1}{b_1 - Y} \Big[ (b_1 - Y)^2 + (b_2 - Y)^2 + (b_3 - Y)^2 + (b_4 - Y)^2 + (b_5 - Y)^2 \Big] + q_{\max} \frac{Y^2}{2} = M + P \bigg( \frac{N}{2} - Y \bigg) \\ \Leftrightarrow & R \bigg( \sum_{i=1}^5 b_i^2 - 2Y \sum_{i=1}^5 b_i + 5Y^2 \bigg) = M + P \bigg( \frac{N}{2} - Y \bigg) - q_{\max} \frac{Y^2}{2} \\ \Leftrightarrow & R = \frac{M + P \bigg( \frac{N}{2} - Y \bigg) - q_{\max} \frac{Y^2}{2}}{\bigg( \sum_{i=1}^5 b_i^2 - 2Y \sum_{i=1}^5 b_i + 5Y^2 \bigg)} = \frac{50 + 200 \bigg( \frac{0.86}{2} - Y \bigg) - 2068.14 \frac{Y^2}{2}}{12951 \times 10^{-4} - 2Y \times 2.15 + 5Y^2} = \frac{136 - 200Y - 1034.07Y^2}{12951 \times 10^{-4} - 4.3Y + 5Y^2} \end{aligned}$$

With:

h: 
$$\sum_{\substack{i=1\\5}}^{5} b_i^2 = 800^2 + 650^2 + 430^2 + 210^2 + 60^2 = 1295100 \text{ mm}^2$$
$$\sum_{\substack{i=1\\5}}^{5} b_i = 800 + 650 + 430 + 210 + 60 = 2150 \text{ mm}$$
$$R = \frac{P_1}{b_1 - Y}$$

$$\begin{array}{l} - & \text{Equilibrium force equation:} \quad \frac{P_1}{b_1 - Y} \Big[ (b_1 - Y) + (b_2 - Y) + (b_3 - Y) + (b_4 - Y) + (b_5 - Y) \Big] = P + q_{\max} Y \\ \Leftrightarrow R \bigg( \sum_{i=1}^5 b_i - 5Y \bigg) = P + q_{\max} Y \\ \Leftrightarrow (q_{\max} + 5R) Y = R \sum_{i=1}^5 b_i - P \\ \Leftrightarrow Y = \frac{R \sum_{i=1}^5 b_i - P}{(q_{\max} + 5R)} = \frac{2.15R - 200}{2068.14 + 5R} \end{array}$$

- Assume Y=0.001, set this value into Equation R:  $R = \frac{136 - 200 \times 0.001 - 1034.07 \times 0.001^{2}}{1295 \times 10^{-4} + 4.3 \times 0.001 + 5 \times 0.001^{2}} = 105.2$ 

- Set R into Equation Y: 
$$Y = \frac{2.15 \times 105.2 - 200}{2068.14 + 5 \times 105.2} = 0.01$$

$$\Delta Y = \frac{0.01 - 0.001}{0.001} = 9 > 0.05 => \text{Recalculate until this change not excess } 0.05$$

- Set Y=0.01 into Equation R: 
$$R = \frac{136 - 200 \times 0.01 - 1034.07 \times 0.01^2}{1295 \times 10^{-4} - 4.3 \times 0.01 + 5 \times 0.01^2} = 106.89$$

$$Y = \frac{2.15 \times 106.89 - 200}{2068.14 + 5 \times 106.89} = 0.01146$$

 $\Delta Y = \frac{0.01146 - 0.01}{0.01} = 0.146 > 0.05 => \text{Recalculate until this change not excess } 0.05$ 

- Set Y=0.01146 into Equation R: 
$$R = \frac{136 - 200 \times 0.01146 - 1034.07 \times 0.01146^2}{1295 \times 10^{-4} - 4.3 \times 0.01146 + 5 \times 0.01146^2} = 107.16$$

$$Y = \frac{2.15 \times 107.16 - 200}{2068.14 + 5 \times 107.16} = 0.01167$$

$$\Delta Y = \frac{0.01167 - 0.01146}{0.01146} = 0.018 < 0.05 => \text{OK}$$

Determine maximum anchor tension:  $R = \frac{P_1}{h_1 - Y} \Rightarrow P_1 = R(h_1 - Y) = 107.16(0.8 - 0.01167) = 84.5 \text{ kN}$ 

Tension in one outermost bolt:  $T_{\text{max}} = \frac{P_1}{n_1 / 2} = \frac{84.5}{4 / 2} = 42.25 \text{ kN}$ 

Determine minimum plate thickness with m (Y<m):

$$\frac{1}{\Omega}F_{y}\frac{Bt_{p}^{2}}{4} = f_{p}BY(m - \frac{Y}{2})$$

$$\rightarrow t_{p} = \sqrt{\frac{4\Omega f_{p}Y(m - \frac{Y}{2})}{F_{y}}} = \sqrt{\frac{4x1.67x9755.36x11.67x(145 - \frac{11.67}{2})}{34.5x10^{4}}} = 17.51 \text{ mm}$$

Determine minimum plate thickness follow tension in bolt:  $\frac{1}{\Omega} F_y \frac{Bt_p^2}{4} = P_1 x$ 

$$\rightarrow t_{p} = \sqrt{\frac{4\Omega P_{1x}}{BF_{y}}} = \sqrt{\frac{4x1.67x84.5x76}{212x34.5}} = 24.22 \text{ mm}$$

Calculate similarly for combos  $M1_{\text{max}} \& P_{M1}, M2_{\text{max}} \& T_{M2}, V_{\text{max}} \& M_V$  we find that required minimum base plate thickness is 44.62mm which is still smaller than currently thickness of 50mm.

Maximum tension in anchor bolts governed by combination  $M2_{max} \& T_{M2}$ 

Use these force to checking bolts:

Total axial force P = 50 kN

Total shear force  $V_g = 50 \text{ kN}$ 

Total moment M = 200 kNm

Maximum anchor tension Tmax = 82.41 kN

Shear for one bolt V = 5 kN

# 4.3.3.3 Design welding

#### Weld capacity due to anchor tension.

Load angle factor when force perpendicular with welding:

$$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5$$

Width of effective area resisting tension force of one bolt:

 $b_{eff} = 2a = 2x70 = 140 \text{ mm}$ 

Allowable bearing capacity of column flange weld:

$$\frac{R_{w}}{\Omega} = \frac{1}{\Omega} \chi 0.6F_{EXX} \frac{\sqrt{2}}{2} h_{f_{\text{flange}}} = \frac{1}{2} \times 1.5 \times 0.6 \times 43 \times \frac{\sqrt{2}}{2} \times 0.7 = 9.58 \frac{kN}{cm}$$

Force apply on column flange weld:

$$\frac{T_{\text{max}}}{b_{\text{eff}}} = \frac{82.41}{14} = 5.89 \frac{kN}{cm}$$

$$\longrightarrow \quad \text{Demand/capacity welding: } \frac{5.89}{9.58} = 0.615 < 1$$

$$\longrightarrow \quad \text{OK}$$

#### Elastic method weld shear capacity

Load angle factor when force parallel with welding:

$$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(0) = 1.0$$

Total web weld length:

$$L_{sbear} = 2d = 2x600 = 1200 \text{ mm}$$

Allowable bearing capacity of column web weld:

$$\frac{R_{w}}{\Omega} = \frac{1}{\Omega} \chi 0.6F_{EXX} \frac{\sqrt{2}}{2} h_{f_{-}web} = \frac{1}{2} \times 1.0 \times 0.6 \times 43 \times \frac{\sqrt{2}}{2} \times 0.6 = 5.47 \frac{kN}{cm}$$

Force apply on column web weld:

$$\frac{V_{\text{max}}}{L_{shear}} = \frac{200}{120} = 1.67 \frac{\&N}{cm}$$
Demand/capacity welding:  $\frac{1.67}{5.47} = 0.305 < 1$ 
OK

#### Elastic method weld axial capacity

Load angle factor when force perpendicular with welding:

$$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5$$
  
Total flange weld length

 $L = 4b_f = 4x212 = 848 \text{ mm}$ 

Inertia moment of weld:

$$I = 4b_f (d/2)^2 + 2(d-2t_f)^3 / 12 = 4x212x(600/2)^2 + 2x(600-2x12)^3 / 12 = 108170.5cm^3$$

Allowable bearing capacity of column web weld:

$$\frac{R_{w}}{\Omega} = \frac{1}{\Omega} \chi 0.6F_{EXX} \frac{\sqrt{2}}{2} b_{f_{\text{flange}}} = \frac{1}{2} \times 1.5 \times 0.6 \times 43 \times \frac{\sqrt{2}}{2} \times 0.7 = 9.58 \frac{kN}{cm}$$

Force apply on column flange weld:

$$\max\left(\frac{T_{\max}}{L} + \frac{M_T(d/2)}{I}, \frac{T_{M2}}{L} + \frac{M2_{\max}(d/2)}{I}\right) = \max\left(\frac{200}{84.8} + \frac{5000(60/2)}{108170.5}, \frac{50}{84.8} + \frac{20000(600/2)}{109170.5}\right) = 6.14$$

$$\implies \text{Demand/capacity welding: } \frac{6.14}{9.58} = 0.64 < 1$$

$$\implies \text{OK}$$

#### 4.3.4. Anchor bolt checking

#### Checking for combination maximum tension

Total axial force P = 50 kN Total shear force  $V_g = 50 \text{ kN}$ Total moment M = 200 kNm Maximum anchor tension  $T_{max} = 82.41 \text{ kN}$ Shear for one bolt V = 5 kN

# 4.3.4.1 Steel strength of a single anchor in tension

Allowable tensile capacity of each bolt:

$$\frac{R_{m}}{\Omega} = \frac{1}{\Omega} F_{m} \frac{\pi d_{b}^{2}}{4} = \frac{1}{2} 37.5 \times \frac{\pi 2.7^{2}}{4} = 107.35 \text{ kN}$$
  
Demand/Capacity:  $\frac{82.41}{107.35} = 0.77 < 1$   
OK

#### 4.3.4.2 Pullout of anchor in tension

The pullout strength in tension of a single headed bolt:

$$\frac{N_{pn}}{\Omega} = \psi_{c,P} \frac{A_{brg} 8 f'_{c}}{\Omega} = 1.4 \frac{8.83 \times 8 \times 2}{2.14} = 92.43 \text{ kN}$$
  
With:  $A_{brg} = \frac{\sqrt{3}}{2} F^2 - \frac{\pi d_b^2}{4} = \frac{\sqrt{3}}{2} 4.1^2 - \frac{\pi \times 2.7^2}{4} = 8.83 \text{ cm}^2$  (F is nominal dimension of bolt's head, take  $F \approx 1.5 d_b$ )

Assume there is no cracking in concrete so modification factor for pullout shall be taken as 1.4

Demand/Capacity = 
$$\frac{82.41}{92.43}$$
 = 0.89 < 1

# 4.3.4.3 Steel strength of anchors in shear

Allowable shear capacity of each bolt:

$$\frac{V_{sa}}{\Omega} = \frac{0.8F_{nv}\pi d_b^2 / 4}{\Omega} = \frac{0.8\times22.5\times\pi\times2.7^2 / 4}{2} = 51.53 \text{ kN}$$
  
Demand/Capacity =  $\frac{5}{51.53} = 0.097 < 1$   
OK

# 4.3.4.4 Interaction of tensile and shear forces of anchor:

 $\frac{0.89 + 0.097}{1.2} = 0.823 < 1$ 

#### Checking for combination maximum shear

Total axial force P = 50 kN Total shear force  $V_g = 200$  kN Total moment M = 50 kNm Maximum anchor tension  $T_{max} = 22.74$  kN Shear for one bolt V = 20 kN

#### 4.3.4.5 Steel strength of a single anchor in tension

Allowable tensile capacity of each bolt:

$$\frac{R_{m}}{\Omega} = \frac{1}{\Omega} F_{m} \frac{\pi d_{b}^{2}}{4} = \frac{1}{2} 37.5 \times \frac{\pi 2.7^{2}}{4} = 107.35 \text{ kN}$$
  
Demand/Capacity:  $\frac{22.74}{107.35} = 0.21 < 1$ 

#### 4.3.4.6 Pullout of anchor in tension

The pullout strength in tension of a single headed bolt:

$$\frac{N_{pn}}{\Omega} = \Psi_{c,P} \frac{A_{brg} 8f'_{c}}{\Omega} = 1.4 \frac{8.83 \times 8 \times 2}{2.14} = 92.43 \text{ kN}$$
  
With:  $A_{brg} = \frac{\sqrt{3}}{2} F^{2} - \frac{\pi d_{b}^{2}}{4} = \frac{\sqrt{3}}{2} 4.1^{2} - \frac{\pi \times 2.7^{2}}{4} = 8.83 \text{ cm}^{2}$  (F is nominal dimension of bolt's head, take  $F \approx 1.5 d_{b}$ )

Assume there is no cracking in concrete so modification factor for pullout shall be taken as 1.4

Demand/Capacity = 
$$\frac{22.74}{92.43} = 0.25 < 1$$
  
OK

# 4.3.4.7 Steel strength of anchors in shear

Allowable shear capacity of each bolt:

$$\frac{V_{sa}}{\Omega} = \frac{0.8F_{m}\pi d_b^2 / 4}{\Omega} = \frac{0.8 \times 22.5 \times \pi \times 2.7^2 / 4}{2} = 51.53 \text{ kN}$$

$$\implies \text{Demand/Capacity} = \frac{20}{51.53} = 0.39 < 1$$

$$\implies \text{OK}$$

# 4.3.4.8 Interaction of tensile and shear forces of anchor:

$$\frac{0.25 + 0.39}{1.2} = 0.53 < 1 : \text{OK}$$

# 4.4. Horizontal Knee Connection Design

# 4.4.1. Input 4.4.1.1 Loading

Run the model and specify all rafter having section 1K. Select the **Display > Show Tables** command to show **Choose Tables for Display** form. Make sure that the **Table: Element Forces - Frames** item and **all combinations** in the **Select Load Cases** form are selected before click the OK button.

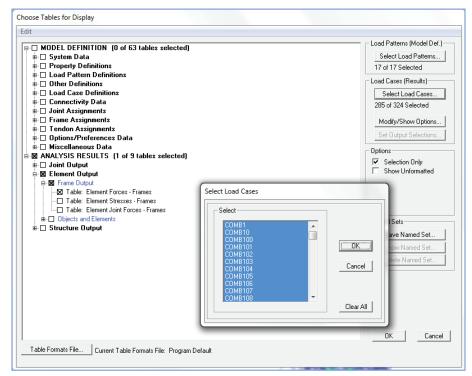


Figure 4 116 Choose Tables for Display form

In Element Forces tables, select **File > Export Current Table > To Excel**.

its: A	s Noted				Elemer	nt Forces - Fram	nes		
	Frame Text	Station m	OutputCase Text	CaseType Text	P KN	V2 KN	V3 KN	T KN-m	M2 KN-m
	41	0	COMB1	Combination	-9.237	-36.652	-0.001523	-0.0001683	0.0011
	41	3	COMB1	Combination	-8.64	-24.724	-0.001523	-0.0001683	0.0057
	41	6	COMB1	Combination	-8.057	-13.061	-0.001523	-0.0001683	0.0102
	41	0	COMB2	Combination	10.687	53.495	0.003232	-0.00005324	-0.0128
	41	3	COMB2	Combination	10.924	40.785	0.003232	-0.00005324	-0.0225
	41	6	COMB2	Combination	11.148	27.809	0.003232	-0.00005324	-0.0322
	41	0	COMB3	Combination	-4.345	39.083	-0.001714	-0.00002205	-0.0045
	41	3	COMB3	Combination	-4.108	31.261	-0.001714	-0.00002205	0.0006913
	41	6	COMB3	Combination	-3.885	23.173	-0.001714	-0.00002205	0.0058
	41	0	COMB4	Combination	4.005	16.883	0.001195	-0.00003876	-0.0118
	41	3	COMB4	Combination	4.242	17.099	0.001195	-0.00003876	-0.0154
	41	6	COMB4	Combination	4.465	17.049	0.001195	-0.00003876	-0.019
	41	0	COMB5	Combination	-10.615	4.607	-0.003614	-0.000006423	-0.0035
	41	3	COMB5	Combination	-10.378	8.978	-0.003614	-0.000006423	0.0073
	41	6	COMB5	Combination	-10.155	13.083	-0.003614	-0.000006423	0.0182
	41	0	COMB6	Combination	32.367	23.847	0.021	-0.0004379	0.016
	41	3	COMB6	Combination	32.604	14.789	0.021	-0.0004379	-0.0471
	41	6	COMB6	Combination	32.828	5.465	0.021	-0.0004379	-0.1102
	41	0	COMB7	Combination	5.329	6.621	-0.002999	-0.00001421	-0.0219
	41	3	COMB7	Combination	5.565	0.84	-0.002999	-0.00001421	-0.0129

Figure 4 117 Joint Reactions table

In excel file just exported, in Station column, select station that use K-type connection, usually "0". Filter the smallest value in "M3" column and enter this value, correspondence shear V2 and axial force P to "Bolt Connections" file.



### NOTE

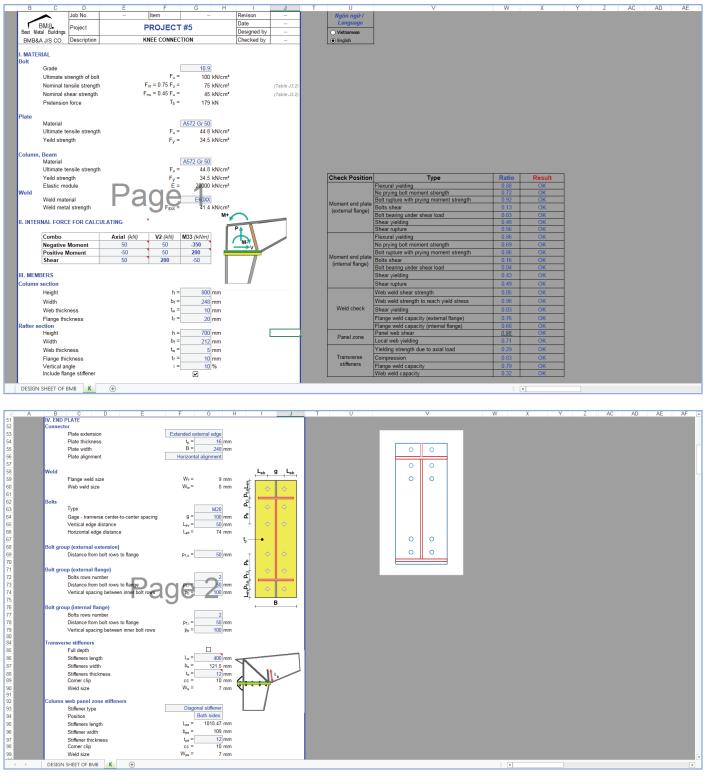
Sign of force in sap will be the same sign of force in calculation file.

a. Filter the largest value in "M3" column and enter this value, correspondence shear V2 and axial force P to "Bolt Connections" file.

b. Filter the largest value in "V2" column and enter this value, correspondence moment M3 and axial force P to "Bolt Connections" file.

## 4.4.1.2 Material and geometry

In order to illustrate a K-type connection with full of dangerous cases, we assume information as figure below.



# 4.4.2. Determind internal force for K connection

Because interal force are in the end of rafter so we have to transfer this load into internal force at top of column

#### Axial force 54.73 kN $P = P_1 \sin \alpha + V_1 \cos \alpha = 50 \sin 5.71^{\circ} + 50 \cos 5.71^{\circ}$ Shear force $V = P_1 \cos \alpha - V_1 \sin \alpha = 50 \cos 5.71^\circ - 50 \sin 5.71^\circ$ 44.78 kN Moment -350 kNm $M = M_1$ Tension force at $PufTop = -\frac{M}{b_{jk}} + \frac{P}{2} = -\frac{-350}{0.78} + \frac{54.73}{2}$ 476.08 kN external flange Tension force at $PufBot = \frac{M}{b_{ik}} + \frac{P}{2} = \frac{-350}{0.78} + \frac{54.73}{2}$ -421.35 kN internal flange

#### Internal force for Combo of negative moment

#### Internal force for Combo of positive moment

Axial force	$P = P_1 \sin \alpha + V_1 \cos \alpha = -50 \sin 5.71^\circ + 50 \cos 5.71^\circ$	44.78 kN
Shear force	$V = P_1 \cos \alpha - V_1 \sin \alpha = -50 \cos 5.71^\circ - 50 \sin 5.71^\circ$	-54.73 kN
Moment	$M = M_1$	200 kN
Tension force at external flange	$PufTop = -\frac{M}{h_{jk}} + \frac{P}{2} = -\frac{200}{0.78} + \frac{44.78}{2}$	-234.02 kN
Tension force at internal flange	$PufBot = \frac{M}{h_{jk}} + \frac{P}{2} = \frac{200}{0.78} + \frac{44.78}{2}$	278.80 kN

#### Internal force for Combo of shear max

Axial force	$P = P_1 \sin \alpha + V$	$a_1 \cos \alpha = 50 \sin 50$	$0.71^{\circ} + 200\cos 5.71^{\circ}$		203.98 kN
Shear force	$V = P_1 \cos \alpha - U$	$V_1 \sin \alpha = 50 \cos \alpha$	5.71° – 200 sin 5.71°		29.85 kN
Moment	$M = M_1$				-50 kN
Tension force at external flange	$PufBot = \frac{M}{h_{jk}} +$	$\frac{P}{2} = \frac{-50}{0.78} + \frac{20}{2}$	$\frac{3.98}{2}$		166.09 kN
Tension force at internal flange	$PufBot = \frac{M}{h_{jk}} + \frac{1}{2}$	$\frac{1}{2} = \frac{-50}{0.78} + \frac{203.9}{2}$	8		37.89 kN
Combo	Axial <b>(kN)</b>	V2 <b>(kN)</b>	M33 (kNm)	PufTop <b>(kN)</b>	PufBot <b>(kN)</b>
Negative Moment	54.73	44.78	-350.00	476.08	-421.35
Positive Moment	44.78	54.73	200.00	-234.02	278.80

-50.00

166.09

37.89

29.85

Shear

203.98

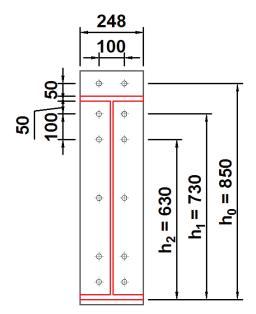
# 4.4.3. Checking moment end plate (external flange)

## Flexural yielding

Calculate total length of yielding line of external flange from Table 4-4 DG16:

$$Y = \frac{b_p}{2} \left( \frac{b_1}{p_{f,i}} + \frac{b_2}{s} + \frac{b_0}{p_{f,o}} - \frac{1}{2} \right) + \frac{2}{g} \left[ b_1 \left( p_{f,i} + 0.75 p_b \right) + b_2 \left( s + 0.25 p_b \right) \right] + \frac{g}{2} \\ = \frac{248}{2} \left( \frac{730}{50} + \frac{630}{78.74} + \frac{850}{50} - \frac{1}{2} \right) + \frac{2}{100} \left[ 730 \left( 50 + 0.75 \times 100 \right) + 630 \left( 78.74 + 0.25 \times 100 \right) \right] + \frac{100}{2} \\ = 8030.65 mm$$

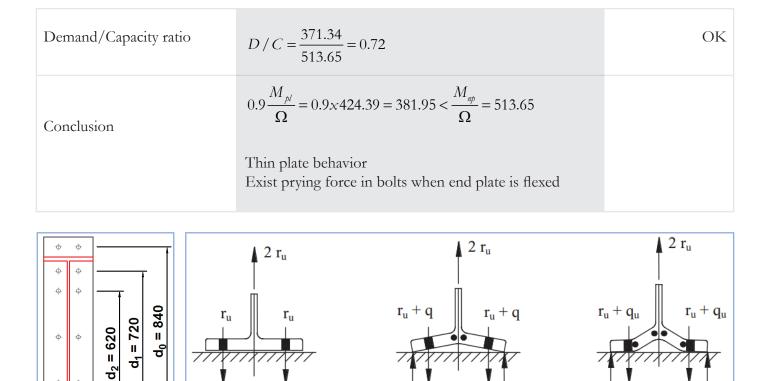
With:  $s = \frac{1}{2}\sqrt{b_p g} = \frac{1}{2}\sqrt{248 \times 100} = 78.74 \text{ mm}$ 



Flexural moment capacity of external end plate	$\frac{M_{pl}}{\gamma_r \Omega} = \frac{1}{\gamma_r \Omega} F_y t_p^2 Y = \frac{1}{1 \times 1.67} 34.5 \times 1.6^2 \times 803.065$ ( $\gamma_r = 1$ for extended connections)	424.71 kNm
Demand flexural moment	$M_{an} =  M  + P\frac{h_{jk}}{2} =  -350  + 54.73\frac{0.78}{2}$	371.34 kNm
Demand/Capacity ratio	$D/C = \frac{371.34}{424.71} = 0.88$	OK

## No prying bolt moment strength

Tension capacity of one bolt	$P_{t} = F_{nt} \frac{\pi d_{b}^{2}}{4} = 75 \frac{\pi \times 2^{2}}{4}$	235.62 kN	
Flexural moment capacity of bolt with no prying force	$\frac{M_{np}}{\Omega} = \frac{1}{\Omega} 2P_t \sum d_n = \frac{1}{2} 2 \times 235.62(0.840 + 0.720 + 0.620)$	513.65 kNm	



(b) intermediate

 $\mathbf{q}_{\mathrm{u}}$ 

q<sub>u</sub>

(c) thin

#### Bolt rupture with prying moment strength

Maximum prying force of inner bolts:

ф (ф

φ

$$\mathcal{Q}_{\max,i} = \frac{w't_p^2}{4a_i} \sqrt{F_{py}^2 - 3\left(\frac{F_i'}{w't_p}\right)^2} = \frac{101\times16^2}{4\times45.72} \sqrt{3450^2 - 3\left(\frac{93.94}{101\times16}\right)^2} = 47.10 \text{ kN}$$

With:  $w' = b_p / 2 - d_b = 248 / 2 - 22 = 102 \text{ mm}$ 

(a) thick

$$a_{i} = 25.4 \left( 3.682 \left( \frac{t_{p}}{d_{b}} \right)^{3} - 0.085 \right) = 25.4 \left( 3.682 \left( \frac{16}{20} \right)^{3} - 0.085 \right) = 45.72 \text{ mm}$$

$$F_{i}' = \frac{t_{p}^{2} F_{y} \left( 0.85 \frac{b_{p}}{2} + 0.80 w' \right) + \frac{\pi d_{b}^{3} F_{nt}}{8}}{4 p_{f,i}} = \frac{16^{2} \times 0.345 \left( 0.85 \frac{248}{2} + 0.8 \times 102 \right) + \frac{\pi \times 20^{3} \times 0.75}{8}}{4 \times 50} = 94.30 \text{ km}$$

Maximum prying force of outer bolts:

$$\mathcal{Q}_{\max,o} = \frac{w't_{p}^{2}}{4a_{o}}\sqrt{F_{py}^{2} - 3\left(\frac{F_{o}}{w't_{p}}\right)^{2}} = \frac{101\times16^{2}}{4\times45.72}\sqrt{3450^{2} - 3\left(\frac{93.94}{101\times16}\right)^{2}} = 47.10 \text{ kN}$$

With:  $w' = b_p / 2 - d_b = 248 / 2 - 22 = 102 \text{ mm}$ 

$$a_{a} = \min\left(L_{ev}; 25.4\left(3.682\left(\frac{t_{p}}{d_{b}}\right)^{3} - 0.085\right)\right) = \min\left(50; 25.4\left(3.682\left(\frac{16}{20}\right)^{3} - 0.085\right)\right) = 45.72 \text{ mm}$$

$$F_{a}' = \frac{t_{p}^{2}F_{y}\left(0.85\frac{b_{p}}{2} + 0.80w'\right) + \frac{\pi d_{b}^{3}F_{m}}{8}}{4p_{f,o}} = \frac{16^{2}\times0.345\left(0.85\frac{248}{2} + 0.8\times102\right) + \frac{\pi\times20^{3}\times0.75}{8}}{4\times50} = 94.30 \text{ kN}$$

Moment strength	$\begin{split} \frac{M_{q}}{\Omega} &= \frac{1}{\Omega} \max \begin{cases} \begin{bmatrix} 2(P_{t} - Q_{\max,o})d_{0} + 2(P_{t} - Q_{\max,i})d_{1} + 2T_{b} d_{2} \end{bmatrix} \\ 2(P_{t} - Q_{\max,o})d_{0} + 2T_{b} (d_{1} + d_{2}) \end{bmatrix} \\ 2(P_{t} - Q_{\max,i})d_{1} + 2T_{b} (d_{0} + d_{2}) \end{bmatrix} \\ &= \frac{1}{2} \max \begin{cases} 2(235.62 - 47.10) 840 + 2(235.62 - 47.10) 720 + 2x179x620 \\ 2(235.62 - 47.10) 840 + 2x179x(720 + 620) \\ 2(235.62 - 47.10) 720 + 2x179x(840 + 620) \\ 2x179x(840 + 720 + 620) \end{cases} \end{cases} $ With: $P_{t} = F_{nt} \frac{\pi d_{b}^{2}}{4} = 75 \frac{\pi \times 2^{2}}{4} = 235.62 \text{ kN}$	405.07 kNm	
Demand/ Capacity ratio	$D / C = \frac{371.34}{405.07} = 0.92$	OK	

#### **Bolts shear**

Shear strength of six bolts in external flange	$\frac{R_n}{\Omega} = \frac{1}{\Omega} F_{nv} \frac{\pi d_b^2}{4} n = \frac{1}{2} 45 \frac{\pi x 2^2}{4} 6$	424.12 kN
Demand shear strength is the maximum shear force correspondent with negative axial force of top flange		54.73 kN
Demand/Capacity ratio	$D/C = \frac{54.73}{424.12} = 0.13$	OK

#### Bolts bearing under shear load

Distance from edge of outermost hole to edge of plate:

$$l_{lc1} = L_{ev} - d_b l 2 = 50 - 22 / 2 = 39 \text{ mm}$$

Distance from edge of hole to hole:

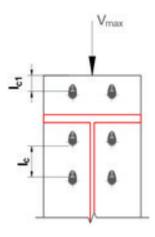
$$l_c = p_b - d_b = 100 - 22 = 78 \,\mathrm{mm}$$

Bolts bearing capacity at outermost bolts:

$$r_{n2} = 2(0.6F_n l_c t) = 1.2 l_c t F_n = 1.2 \times 7.8 \times 1.6 \times 44.8 = 670.92 \text{ kN}$$
  
Bolts bearing capcity between bolts:

 $r_{n2} = 2(0.6F_nl_ct) = 1.2l_ctF_n = 1.2x7.8x1.6x44.8 = 670.92$  kN Maximum bearing capacity:

 $r_{n(\max)} = 2(0.6F_n 2dt) = 2.4dtF_n = 2.4 \times 2 \times 1.6 \times 44.8 = 344.06 \text{ kN}$ 



Total bearing capacity	
	$\frac{R_n}{\Omega} = \frac{1}{\Omega} \left[ 2\min(r_{n1}, r_{n(\max)}) + 2(\frac{n}{2} - 1)\min(r_{n2}, r_{n(\max)}) \right]$ $= \frac{1}{2} \left[ 2\min(335.46, 344.06) + 2\left(\frac{6}{2} - 1\right)\min(670.92, 344.06) \right]$ $= 1023.58 \text{ kN}$
Demand shear strength	$V_a = \max(44.78, 54.73, 29.85) = 54.73 \text{ kN}$
Demand/Capacity ratio	$D/C = \frac{54.73}{1023.58} = 0.05$ : OK

## Shear yielding of plate

Shear yielding strength of plate:

$$V_a = \frac{Puftop}{2} = \frac{\max(476.08, 234.02, 166.09)}{2} = 238.04 \text{ kN}$$
  
Demand shear strength:

 $V_a = \frac{Puftop}{2} = \frac{\max(476.08, 234.02, 166.09)}{2} = 238.04 \text{ kN}$ Demand/Capacity ratio:

$$D/C = \frac{238.04}{491.84} = 0.48$$
 : OK

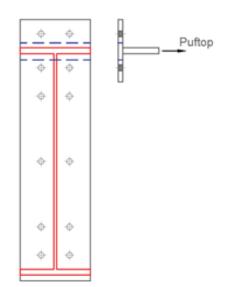
#### Shear rupture of plate

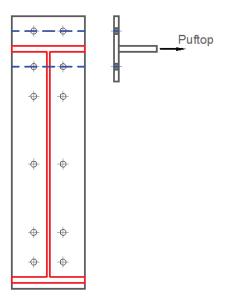
Shear rupture strength of plate:

 $\frac{R_n}{\Omega} = \frac{1}{\Omega} 0.6F_n [b_p - 2(d_b + 1.6)] t_p$  $\frac{R_n}{\Omega} = \frac{1}{2} \times 0.6 \times 0.45 \times [248 - 2(22 + 1.6)] 16 = 431.99 \text{ kN}$ Demand shear strength:

$$V_a = \frac{Puftop}{2} = \frac{\max(476.08, 234.02, 166.09)}{2} = 238.04 \text{ kN}$$

Demand/Capacity ratio:  $D/C = \frac{238.04}{431.99} = 0.55$  : OK





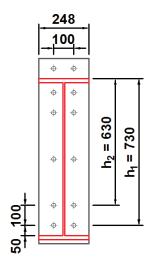
# 4.4.4. Checking moment end plate (internal flange)

## Flexural yielding

Calculate total length of yielding line of internal flange from Table 3-3 DG16:

$$Y = \frac{b_p}{2} \left( \frac{b_1}{p_f} + \frac{b_2}{s} \right) + \frac{2}{g} \left[ b_1 \left( p_f + 0.75 p_b \right) + b_2 \left( s + 0.25 p_b \right) \right] + \frac{g}{2}$$
  
=  $\frac{248}{2} \left( \frac{730}{50} + \frac{630}{78.74} \right) + \frac{2}{100} \left[ 730 \left( 50 + 0.75 \times 100 \right) + 630 \left( 78.74 + 0.25 \times 100 \right) \right] + \frac{100}{2}$   
= 5984.65mm

With: 
$$s = \frac{1}{2}\sqrt{b_p g} = \frac{1}{2}\sqrt{248 \times 100} = 78.74 \text{ mm}$$



Flexural moment capacity of internal end plate	$\frac{M_{pl}}{\gamma_r \Omega} = \frac{1}{\gamma_r \Omega} F_y t_p^2 Y = \frac{1}{1.25 \times 1.67} 34.5 \times 1.6^2 \times 598.465 = 253.20 \text{ kNm}$ ( $\gamma_r = 1.25$ for flush connections)
Demand flexural moment	$M_{ap} = M + P \frac{b_{fk}}{2} = 200 + 44.78 \frac{0.78}{2} = 217.46 \text{ kNm}$
Demand/Capacity ratio	$D/C = \frac{217.46}{253.20} = 0.86$ : OK

# No prying bolt moment strength

$ \begin{array}{c ccccccccccccccccccccccccccccccccccc$	$r_u$ $r_u$ (a) thick	$r_u + q$ $r_u + q$ q $q$ $q$ $q$ $(b) intermediate$	$r_{u} + q_{u}$ $r_{u}$ $r_{u} + q_{u}$ $r_{u}$ $r_{u}$ $r_{u}$ $r_{u}$ $r_{u}$ $r_{u}$ $r_{u}$ $r_{u}$ $r_{u}$	$q_u$
Tension capacity of one bolt	$P_t = F_m \frac{\pi d_b^2}{4} = 75 \frac{\pi \times 2^2}{4}$			235.62 kN
Flexural moment capacity of bolt with no prying force	$\frac{M_{np}}{\Omega} = \frac{1}{\Omega} 2P_t(\sum d_n) = \frac{1}{2} 2 \times 235$	5.62(0.720+0.620)		315.73 kNm
Demand/Capacity ratio	$D/C = \frac{217.46}{315.73} = 0.69$			OK

Conclusion  $0.9 \frac{M_{pl}}{\Omega} = 0.9 \times 253.20 = 227.88 < \frac{M_{np}}{\Omega} = 315.73$ Thin plate behavior Exist prying force in bolts when end plate is flexed

## Bolt rupture with prying moment strength

## **Bolts shear**

Shear strength of four bolts in internal flange	$\frac{R_n}{\Omega} = \frac{1}{\Omega} F_{nv} \frac{\pi d_b^2}{4} n = \frac{1}{2} 45 \frac{\pi 2^2}{4} 4$	282.74 kN
Demand shear strength is the maximum shear force correspondent with negative axial force of bottom flange	$V_{a}$	44.78 kN
Demand/Capacity ratio	$D/C = \frac{44.78}{282.74} = 0.16$	OK

# Bolts bearing under shear load

Distance from edge of outermost hole to edge of plate	$l_{k1} = L_{ev} - d_b l^2 = 50 - 22 / 2$	39 mm	V <sub>max</sub>
Distance from edge of hole to hole	$l_{c} = p_{b} - d_{b} = 100 - 22$	78 mm	
Bolts bearing capacity at outermost bolts	$r_{n1} = 2(0.6F_n l_{c1}t) = 1.2l_{c1}tF_n$ = 1.2 × 3.9 × 1.6 × 44.8	335.46 kN	
Bolts bearing capcity between bolts	$r_{n2} = 2(0.6F_n l_c t) = 1.2 l_c t F_n$ = 1.2 × 7.8 × 1.6 × 44.8	670.92 kN	•
Maximum bearing capacity	$r_{n(\max)} = 2(0.6F_u 2dt) = 2.4 dt F_u$ = 2.4 × 2 × 1.6 × 44.	344.06 kN	

Total bearing capacity	$\frac{R_n}{\Omega} = \frac{1}{\Omega} \left[ 2\min(r_{n1}, r_{n(\max)}) + 2(\frac{n}{2} - 1)\min(r_{n2}, r_{n(\max)}) \right]$ $\frac{R_n}{\Omega} = \frac{1}{2} \left[ 2\min(335.46, 344.06) + 2\left(\frac{4}{2} - 1\right)\min(670.92, 344.06) \right]$	679.52 kN	
Demand shear strength	$V_a = \max(44.78; 54.73; 29.85)$	54.73 kN	
Demand/ Capacity ratio	$D / C = \frac{54.73}{679.52} = 0.08$	ОК	

# Shear yielding of plate

Shear yielding strength of plate	$\frac{R_n}{\Omega} = \frac{1}{\Omega} 0.6 F_y b_p t_p$ = $\frac{1}{1.67} 0.6 \times 34.5 \times 24.8 \times 1.6$	491.84 kN	\$\phi\$         \$\phi\$           \$\phi\$         \$\phi\$           \$\phi\$         \$\phi\$           \$\phi\$         \$\phi\$
Demand shear strength	$V_a = \frac{Pufbot}{2} = \frac{\max(421.35; 278.8; 37.89)}{2}$	210.68 kN	\$ \$
Demand/ Capacity ratio	$D/C = \frac{210.68}{491.84} = 0.43$	OK	

# Shear rupture of plate

Shear rupture strength of plate	$\frac{R_n}{\Omega} = \frac{1}{\Omega} 0.6F_n [b_p - 2(d_b + 1.6)] t_p$ $= \frac{1}{2} \times 0.6 \times 0.45 [248 - 2(22 + 1.6)] 16$	431.99 kN	•     •       •     •       •     •       •     •
Demand shear strength	$V_{a} = \frac{Pufbot}{2} = \frac{\max(421.35; 278.8; 37.89)}{2}$	310.68 kN	φ φ
Demand/ Capacity ratio	$D/C = \frac{210.68}{431.99} = 0.49$	ОК	

# 4.4.5. Weld check

# Web weld shear strength

Area of web weld (assume only weld at compresion area assist shear force)	$\mathcal{A}_{w} = \frac{\sqrt{2}}{2} W_{w} L = \frac{\sqrt{2}}{2} W_{w} \frac{h_{jk}}{2} = \frac{\sqrt{2}}{2} 8 \times \frac{780}{2}$	2206.17 mm
Shear strength of web weld	$\frac{R_n}{\Omega} = 2\left(\frac{1}{\Omega}0.6F_{EXX}A_w\right) = 2\left(\frac{1}{2}0.6\times41.4\times22.06\right)$	547.97 kN
Demand shear strength	$V_a = \max(44.78; 54.73; 29.85)$	54.73 kN
Demand/Capacity ratio	$D/C = \frac{54.73}{547.97} = 0.10$	OK

# Web weld strength to reach yield stress

Load angle factor when force perpendicular with weld	$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90)$	1.5
Weld strength of web weld	$\frac{R_{w}}{\Omega} = 2\left(\frac{1}{\Omega}0.6F_{EXX}\chi\frac{\sqrt{2}}{2}W_{w}\right)$ $= 2\left(\frac{1}{2}0.6\times41.4\times1.5\times\frac{\sqrt{2}}{2}0.8\right)$	21.07 kN/cm
Demand yield stress	$\frac{R_n}{\Omega} = \frac{1}{\Omega} F_y t_w = \frac{1}{1.67} 34.5 \times 1$	20.64 kN/cm
Demand/Capacity ratio	D/C = 20.64/21.08 = 0.98	ОК

# Shear yielding of web

Shear yielding strength of web	$\frac{R_n}{\Omega} = \frac{1}{\Omega} 0.6F_y L_p t_w = \frac{1}{1.5} 0.6 \times 34.5 \times 80 \times 1$	1104 kN
Demand shear strength	D/C = 54.73/1104 = 0.05	54.73 kN
Demand/Capacity ratio	D/C = 54.73/1104 = 0.05	OK

# Flange weld capacity (external flange)

Area of outsite flange weld	$A_{w} = \frac{\sqrt{2}}{2} W_{f} L = \frac{\sqrt{2}}{2} W_{f} (2b_{f} - t_{w} + 2t_{f})$ $= \frac{\sqrt{2}}{2} 9 (2 \times 248 - 10 + 2 \times 20)$	3347.44 mm <sup>2</sup>
Load angle factor when force perpendicular with weld	$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90)$	1.5
Weld strength of flange weld	$\frac{R_n}{\Omega} = \frac{1}{\Omega} 0.6F_{EXX} \chi A_w$ $= \frac{1}{2} 0.6 \times 41.4 \times 1.5 \times 33.47$	623.55 kN

Demand shear strength	$V_a = Puftop \ kN$	476.08 kN
Demand/Capacity ratio	$D/C = \frac{476.08}{623.55} = 0.76$ : OK	

# Flange weld capacity (internal flange)

Area of inside flange weld	$A_{w} = \frac{\sqrt{2}}{2} W_{f} L = \frac{\sqrt{2}}{2} W_{f} (2b_{f} - t_{w} + 2t_{f})$
Load angle factor when force perpendicular with weld Weld strength of flange	$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5$ $\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5$
weld Demand shear strength	$\frac{R_n}{\Omega} = \frac{1}{\Omega} 0.6F_{EXX} \chi A_w = \frac{1}{2} 0.6x41.4x1.5x33.47 = 623.55 \text{ kN}$ $V_a = Pufbot = 421.35 \text{ kN}$
Demand/Capacity ratio	$D/C = \frac{421.35}{623.55} = 0.68$ : OK

# 4.4.6. Checking panel zone

## Panel web shear

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- Shear yield strength of the rafter:

$$P_{c} = 0.6F_{y}A_{g} = 0.6F_{y}(2b_{f}t_{f} + b_{w}t_{w}) = 0.6x34.5x(2x212x10 + 680x5)/100 = 1581.48 \text{ kN}$$
  
For: 
$$P_{r} = \max(Puftop, Pufbot) = \max(476.08, 421.35) = 476.08$$
$$P_{r} = 476.08 < 0.4P_{c} = 0.4x158148 = 632.59$$

$$\frac{R_n}{\Omega} = \frac{0.6F_y bt_w}{1.67} = \frac{0.6x34.5x70x0.5}{1.67} = 433.83 \text{ kN}$$

- Shear yield strength of the diagonal stiffeners (consider diagonal stiffeners as compresion elements): Inertia radius of stiffeners:  $r = t_{ps} / \sqrt{12} = 12 / \sqrt{12} = 3.46$  mm

Slenderness of stiffeners:  $KL / r = 0.65 \times 1018.47 / 3.46 = 191.10$ 

$$\frac{KL}{r} = 191.10 > 4.71 \sqrt{\frac{E}{F_y}} = 4.71 \sqrt{\frac{20000}{34.5}} = 113.4$$

$$F_{cr} = 0.877 F_e = 0.877 \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = 0.877 \frac{\pi^2 \times 20000}{191.10^2} = 4.74 \frac{kN}{cm^2}$$

$$\frac{R_{nps}}{\Omega} = 2\frac{1}{\Omega}F_{rr}A_{g}\cos\theta = 2\frac{1}{\Omega}F_{rr}(b_{ps}-\omega)t_{ps}\frac{b_{c}-2t_{fr}}{L_{ps}}$$
  
Strength of diagonal stiffeners:  
$$= 2\frac{1}{1.67}4.74x(109-10)x12x10^{-2}\frac{800-2x20}{1018.47}$$
$$= 50.32kN$$

Total panel web shear resitance:  $\frac{R_n}{\Omega} + \frac{R_{nps}}{\Omega} = 433.83 + 50.32 = 484.15$  kN

Demand shear strength:  $V_a = \max(Puftop, Pufbot) = \max(476.08, 421.35) = 476.08 \text{ kN}$ 

Demand/Capacity ratio: 
$$D/C = \frac{476.08}{484.15} = 0.98$$

OK OK

### 4.4.7. Local web yielding

Web resistance in local web yielding:

$$\frac{1}{\Omega} = \frac{1}{\Omega} \left[ 0.5(6k + 2t_p) + N \right] F_{yw} t_w = \frac{1}{1.5} \left[ 0.5(6x16 + 2x16) + 38 \right] 34.5x10 \quad x5 = 117.3 \text{ kN}$$

With:  $k = t_p = 16 \text{ mm}$ 

 $N = t_f + 2W_f = 10 + 2x9 = 38 \text{ mm}$ Stiffeners resistance in local web yielding:

$$\frac{R_{ns}}{\Omega} = 2\frac{1}{\Omega}F_{y}A_{s} = 2\frac{1}{1.67}34.5\times10^{-2}\times1338 = 552.83\,\text{kN}$$

With:  $A_s = t_s(b_s - clip) = 12(121.5 - 10) = 1338 mm^2$ Total resistance of web:

$$\frac{R_n}{\Omega} + \frac{R_{ns}}{\Omega} = 117.3 + 552.83 = 670.13 \text{ kN}$$

Demand shear strength:

 $V_a = \max(Puftop, Pufbot) = \max(476.08, 421.35) = 476.08 \text{ kN}$ Demand/Capacity ratio:

$$D/C = \frac{476.08}{670.13} = 0.71$$

# 4.4.8. Transverse stiffeners Yielding strength due to axial load

Gross section area of transverse stiffeners:  $A_g = 2(b_s - clip)t_s = 2x(121.5 - 10)x12 = 2676mm^2$ 

Tensile yielding strength of stiffeners:  $\frac{V}{\Omega} = \frac{1}{\Omega} F_y A_g = \frac{1}{1.67} 34.5 \times 10^{-2} \times 2676 = 552.83 \text{ kN}$ Demand axial strength is calculated by subtracting web resistance in local web yielding from maximum tension force at bottom flange:  $P_{tension} = \max(-421.35, 278.80, 37.89) - 117.3 = 161.5 \text{ kN}$ 

Demand/Capacity ratio: 
$$D/C = \frac{161.5}{552.83} = 0.29$$

#### Compression

Inertia radius of stiffeners:

 $r = t_s / \sqrt{12} = 12 / \sqrt{12} = 3.46 \text{ mm}$ Slenderness of stiffeners:

$$KL / r = 0.65 \times 400 / 3.46 = 75.14$$

$$\frac{KL}{r} = 75.14 < 4.71 \sqrt{\frac{E}{F_y}} = 4.71 \sqrt{\frac{20000}{34.5}} = 113.4$$

$$F_{\sigma} = \left[0.658^{-1}\right] F_{y} = \left[0.658^{\frac{34.5}{34.96}}\right] 34.5 = 22.83 \frac{kN}{cm}$$
$$F_{\sigma} = \left[0.658^{\frac{F_{y}}{F_{e}}}\right] F_{y} = \left[0.658^{\frac{34.5}{34.96}}\right] 34.5 = 22.83 \frac{kN}{cm^{2}}$$

Compressive strength shall be determined based on the limit state of flexural buckling:

$$\frac{P_n}{\Omega} = 2\frac{1}{\Omega}F_{\sigma}A_g = 2\frac{1}{1.67}22.83x13.38 = 365.83\,\text{kN}$$

Demand axial strength is calculated by subtracting web resistance in local web yielding from maximum compressive force at bottom flange:

 $P_{compression} = |min(-421.35, 278.80, 37.89)| - 117.3 = 304.05 \text{ kN}$ Demand/Capacity ratio:

$$D/C = \frac{304.05}{365.83} = 0.83$$

#### Welding stiffeners with rafter flange

Area of flange weld:

$$A_{w} = \frac{\sqrt{2}}{2} W_{s} L = \frac{\sqrt{2}}{2} W_{s} (b_{st} - clip - W_{s}) = \frac{\sqrt{2}}{2} 7 (121.5 - 10 - 7) = 517.25 mm^{2}$$
  
Load angle factor when force perpendicular with weld

Load angle factor when force perpendicular with weld

$$\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5$$

Weld strength of total flange weld:

$$V_a = \max(P_{tension}, P_{compression}) = \max(161.5, 304.05) = 304.05 \text{ kN}$$
  
Demand shear strength:

 $V_a = \max(P_{tension}, P_{compression}) = \max(161.5, 304.05) = 304.05 \text{ kN}$ Demand/Capacity ratio:

$$D/C = \frac{304.05}{385.27} = 0.79$$
  
OK

#### Welding stiffeners with rafter web

Area of web weld:

$$\frac{K_n}{\Omega} = 4 \left( \frac{1}{\Omega} 0.6F_{EXX} \chi A_{w} \right) = 4 \left( \frac{1}{2} 0.6 \times 41.4 \times 1.0 \times 18.96 \right) = 941.93$$

Load angle factor when force parallel with weld  $\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(0) = 1.0$ Weld strength of total web weld:

$$\frac{R_n}{\Omega} = 4 \left( \frac{1}{\Omega} 0.6 F_{EXX} \chi A_{\mu} \right) = 4 \left( \frac{1}{2} 0.6 \times 41.4 \times 1.0 \times 18.96 \right) = 941.93 \,\mathrm{kN}$$

Demand shear strength:  $V_a = \max(P_{tension}, P_{compression}) = \max(161.5, 304.05) = 304.05 \text{ kN}$ 

Demand/Capacity ratio:  $D/C = \frac{304.05}{941.93} = 0.32$ 

OK 

#### 4.4.9. Checking diagonal stiffeners

Yielding strength due to axial load

Section area of diagonal stiffeners:  $A_g = 2(b_{ps} - clip)t_{ps} = 2(109 - 10)12 = 2376 \text{ mm}^2$ 

Tensile yielding strength of stiffeners:  $\frac{R_n}{\Omega} = \frac{1}{\Omega} F_y A_g = \frac{1}{1.67} 34.5 \times 23.76 = 490.85 \text{ kN}$ 

Demand axial strength is calculated by subtracting web resistance in panel web shear from maximum of tensile force at bottom flange and compressive force at top flange:

$$P_{tension} = \frac{\max\left[\left|\min\left(476.08, -234.02, 166.09\right)\right|, \max(-421.35, 278.80, 37.89)\right] - 433.83}{\cos\theta} = \frac{-155.03}{\cos\theta} < 0 \text{ kN}$$

Because required strength is negative so it's not necessary to check tensile yielding strength of diagonal stiffeners.

Checking compressive strength of diagonal stiffeners

Inertia radius of stiffeners:  $r = t_{ps} / \sqrt{12} = 12 / \sqrt{12} = 3.46 \text{ mm}$ 

Slenderness of stiffeners:  $\frac{KL}{r} = 191.33 > 4.71 \sqrt{\frac{E}{F_y}} = 4.71 \sqrt{\frac{20000}{34.5}} = 113.4$ 

$$\implies \frac{KL}{r} = 191.33 > 4.71 \sqrt{\frac{E}{F_y}} = 4.71 \sqrt{\frac{20000}{34.5}} = 113.4$$

$$F_{e} = \frac{\pi^{2}E}{\left(\frac{KL}{r}\right)^{2}} = \frac{\pi^{2}x20000}{191.33^{2}} = 5.39\frac{kN}{cm^{2}}$$
$$F_{r} = 0.877F_{e} = 0.877x5.39 = 4.73\frac{kN}{cm^{2}}$$

Compressive strength shall be determined based on the limit state of flexural buckling:

$$\frac{P_n}{\Omega} = 2\frac{1}{\Omega}F_{cr}A_g = 2\frac{1}{1.67}4.74 \times 11.88 = 67.44 \text{ kN}$$

Demand axial strength is calculated by subtracting web resistance in panel web shear from maximum of compressive force at bottom flange and tensile force at top flange:

$$P_{compression} = \frac{\max\left[\max\left(476.08, -234.02, 166.09\right), \left|\min\left(-421.35, 278.80, 37.89\right)\right|\right] - 433.83}{42.25}$$
$$= \frac{42.25}{\left(b_c - 2t_{fc}\right)/L_{ps}} = \frac{42.25}{\left(800 - 2x20\right)/1018.47} = 56.62kN$$

Demand/Capacity ratio:  $D/C = \frac{56.62}{67.44} = 0.84$ OK

# Welding stiffeners with rafter flange

Area of flange weld:  $A_{w} = \frac{\sqrt{2}}{2} W_{ps} L = \frac{\sqrt{2}}{2} W_{ps} (b_{ps} - clip - W_{ps}) = \frac{\sqrt{2}}{2} 7(109 - 10 - 7) = 455.38 mm^{2}$ Load angle factor when force perpendicular with weld:  $\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(90) = 1.5$ 

Weld strength of total flange weld:  $\frac{R_n}{\Omega} = 4\left(\frac{1}{\Omega}0.6F_{EXX}\chi A_w\right) = 4\left(\frac{1}{2}0.6x41.4x1.5x4.55\right) = 339.07 \text{ kN}$ Demand shear strength:  $V_a = \max\left(P_{tension}, P_{compression}\right) = \max\left(0, 56.62\right) = 56.62 \text{ kN}$ 

Demand/Capacity ratio:  $D/C = \frac{56.62}{339.07} = 0.17$ 

OK OK

#### Welding stiffeners with rafter web

Area of web weld:  $A_{\mu\nu} = \frac{\sqrt{2}}{2} W_{\rho s} L = \frac{\sqrt{2}}{2} W_{\rho s} \left( L_{\rho s} - clip - W_s \right) = \frac{\sqrt{2}}{2} 7 (1018.47 - 10 - 7) = 4957.02 mm^2$ Load angle factor when force parallel with weld  $\chi = 1 + 0.5 \sin^{1.5}(\theta) = 1 + 0.5 \sin^{1.5}(0) = 1.0$ 

Weld strength of total web weld:  $V_a = \max(P_{tension}, P_{compression}) = \max(0, 56.62) = 56.62 \text{ kN}$ Demand shear strength:  $V_a = \max(P_{tension}, P_{compression}) = \max(0, 56.62) = 56.62 \text{ kN}$ 

Demand/Capacity ratio:  $D/C = \frac{56.62}{2462.64} = 0.02$ 

OK OK

# 4.5. Single Plate Connection Design

## 4.5.1. Input

## 4.5.1.1 Loading

Run the model and specify all beam having section B1. Select the **Display > Show Tables** command to show **Choose Tables for Display** form. Make sure that **the Table: Element Forces - Frames i**tem and **ENVE** combination in the **Select Load Cases** form are selected before click the OK button.

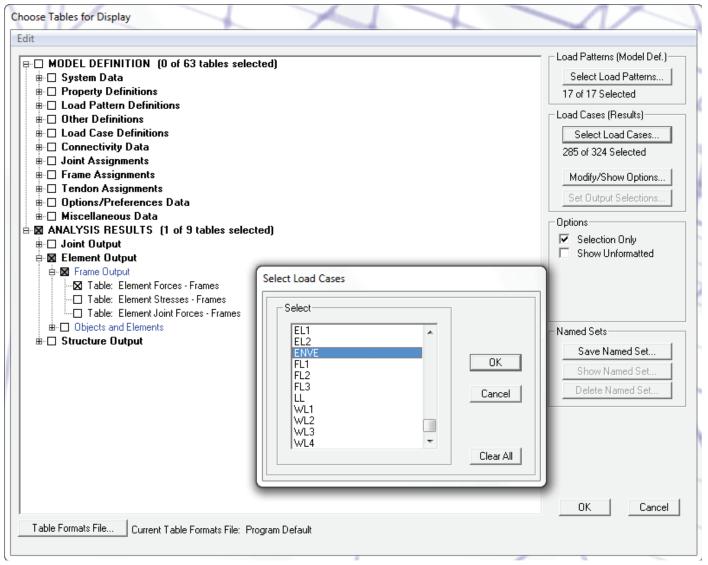


Figure 4 118 Choose Tables for Display form

In Element Forces tables, select File > Export Current Table > To Excel.

Jnits: As Noted Element Forces - Frames						S			
	Frame Text	Station m	OutputCase Text	CaseType Text	StepType Text	P KN	V2 KN	V3 KN	T KN-m
	261	0	ENVE	Combination	Max	3.01	-0.603	0	0.0002593
	261	0.5	ENVE	Combination	Max	3.01	-0.528	0	0.0002593
	261	1	ENVE	Combination	Max	3.01	-0.453	0	0.0002593
	261	1.5	ENVE	Combination	Max	3.01	-0.377	0	0.0002593
	261	2	ENVE	Combination	Max	3.01	-0.302	0	0.0002593
	261	2.5	ENVE	Combination	Max	3.01	-0.226	0	0.0002593
	261	3	ENVE	Combination	Max	3.01	-0.151	0	0.0002593
	261	3.5	ENVE	Combination	Max	3.01	-0.075	0	0.0002593
	261	4	ENVE	Combination	Max	3.01	0	0	0.0002593
	261	4.5	ENVE	Combination	Max	3.01	0.126	0	0.0002593
	261	5	ENVE	Combination	Max	3.01	0.251	0	0.0002593
	261	5.5	ENVE	Combination	Max	3.01	0.377	0	0.0002593
	261	6	ENVE	Combination	Max	3.01	0.503	0	0.0002593
	261	6.5	ENVE	Combination	Max	3.01	0.628	0	0.0002593
	261	7	ENVE	Combination	Max	3.01	0.754	0	0.0002593
	261	7.5	ENVE	Combination	Max	3.01	0.88	0	0.0002593
	261	8	ENVE	Combination	Max	3.01	1.006	0	0.0002593
	261	0	ENVE	Combination	Min	-0.141	-1.006	0	-0.0004561
	261	0.5	ENVE	Combination	Min	-0.141	-0.88	0	-0.0004561
	261	1	ENVE	Combination	Min	-0.141	-0.754	0	-0.0004561

Figure 4 118 Choose Tables for Display form



## NOTE

Take absolute value for both shear and axial force to enter into "Bolt Connections" file. In excel file just exported, in Station column, select stations at two ends of beam. Filter the largest value in "V2" column and largest value in "P", enter these value to "Bolt Connections" file.

## 4.5.1.2 Material and geometry

In order to illustrate a single plate connection with full of dangerous cases, we assume information as figure below.

			0		F	-	0					_
	A	В	C	D	E	F GLE PLATE CO		H I	J	М		*
5		BMB&/	A J/S CO.	Description	SING	SLE PLATE CU	NNECTION	Checked by		-	<ul> <li>English</li> </ul>	
6		I. MATER										
8		Bolt	IAL									
9		Don	Grade				10.9					
10				trength of bolt		Fu =	100 kN/c	m²				
11				ensile strength		F <sub>nt</sub> =	75 kN/c	m²	(Table J3.2)			
12			Nominal s	hear strength		F <sub>nv</sub> =	45 kN/c	m²	(Table J3.2)			
13												
14		Plate					70 0 50					
15 16			Material	o n a th			72 Gr 50					
10			Tensile str Yeild strer			F <sub>u</sub> = F <sub>v</sub> =	44.8 kN/ci 34.5 kN/ci					
18			Tellu Strei	igui		1 v -	54.5 KN/C					
19		Beam, G	irder									
20		l í	Material			A5	72 Gr 50					
21			Ultimate te	ensile strength	1	Fu =	44.8 kN/c	m²				
22			Yeild strer	_		F <sub>y</sub> =	34.5 kN/c	m²				
23			Elastic mo			É =	20000 kN/c	m²				
24												
25		II. INTER	NAL FORC	E FOR CALC	ULATING ষ				ф-			
26			Shear			V =	200 kN	1.0	•P			
27			Axial				50 kN					
28 29		III. DIME				いて		1.6	• •			
30		Beam se										
31		Dealli Se	Height			h =	500 mm				Structural	
32			Flange wid	dth		b <sub>f</sub> =	200 mm					
33			Web thick			t <sub>w</sub> =	10 mm					
34			Flange thi	ckness		t <sub>f</sub> =	20 mm					
35			Beam set	back		sb =	10 mm					
36											Dista	
37 38		Girder se		4+1-		h =	212 mm				Plate	
39			Flange wid Web thick			b <sub>f</sub> = t <sub>w</sub> =	8 mm					
40			Web then	11633			0	sbL <sub>eh</sub> g	Leh			
41		Plate							at a			
42			Thickness			t <sub>p</sub> =	14 mm		⇒ <mark>–</mark>			
43			Height			L <sub>p</sub> =	305 mm	<u>ф</u>	° v			
44		L .							• -			
45		Bolt	т				1120	·⊕ · ·	° ≱_		Beam	
46 47			Type Bolt rows			n -	M20 4		┛┛┼			
47			Bolt colum	ne		n = n <sub>c</sub> =	2					
40				nsverse spaci	na	g =	100 mm					
50				gitudinal spac		9 s =	75 mm		]			
51			Vertical ed	dge distance	-	Lev =	40 mm		-			
52			Horizontal	edge distance	9	Leh =	40 mm					
53				between weld	and bolts	a =	152 mm					
54			Eccentric	distance		e =	202 mm					
55 56		IV CALC	ULATING							-		
50			boam eide									Ŧ
	•	DESIGN	SHEET OF E	SMB K	SP TAP B	s 🕂		:	4		Þ	

**DESIGN GUIDELINES** 

## 4.5.2. Plate (beam side) 4.5.2.1 Bolt shear

Shear strength of bolt group:

$$\frac{R_{n}}{\Omega} = \frac{CF_{nv}A_{b}n_{v}}{\Omega} = \frac{3.08\times45\times\frac{\pi\times2^{2}}{4}\times1}{2} = 217.71 \,\mathrm{kN}$$

Where:

C: coefficient for eccentrically loaded bolt groups (represents the number of bolts that are effective in resisting the eccentric shear force) is selected from table 7-8 AISC Steel Construction Manual 13<sup>th</sup>, with n=4, s=75mm (3in), e<sub>x</sub>=202mm (7.95in), using interpolation method we find out C value is approximately 3.08.

7-38

#### DESIGN CONSIDERATIONS FOR BOLTS

Co	Table 7–8 Coefficients C for Eccentrically Loaded Bolt Groups Angle = $0^{\circ}$												
Available Strength of a bolt group, $\phi R_n$ or $R_n/\Omega$ , is determined with $R_n = C \times r_n$ $\phi = 0.75$ $\Omega = 2.00$ orwhere $P$ = required force, $P_u$ or $P_a$ , kips $r_n$ = nominal strength per bolt, kips 													
<i>s</i> , in.	e <sub>x</sub> , in.	Number of Bolts in One Vertical Row, n											
0,	<i>x</i> ,	1	2	3	4	, 5	6	7	8	9	10	11	12
	2 3 4 5 6	0.84 0.65 0.54 0.45 0.39	2.54 2.03 1.67 1.42 1.22	4.48 3.68 3.06 2.59 2.25	6.59 5.67 4.86 4.21 3.69	8.72 7.77 6.84 6.01 5.32	10.8 9.91 8.93 8.00 7.17	12.9 12.1 11.1 10.1 9.16	15.0 14.2 13.2 12.2 11.2	17.0 16.3 15.4 14.4 13.4	19.0 18.3 17.5 16.5 15.5	21.0 20.4 19.6 18.7 17.7	23.0 22.5 21.7 20.8 19.8
	7 8	0.35 0.31	1.08 0.96	1.99 1.78	3.27 2.93	4.74 4.27	6.46 5.86	8.33 7.60	10.3 9.50	12.4 11.5	14.5 13.6	16.7 15.7	18.8 17.8
3	9 10 12 14 16 18 20 24 28 32 36 <i>C</i>	0.31 0.28 0.26 0.22 0.19 0.17 0.15 0.14 0.12 0.10 0.09 0.08 2.94	0.96 0.86 0.78 0.66 0.57 0.51 0.45 0.41 0.34 0.29 0.26 0.23 8.33	1.76 1.60 1.46 1.24 1.08 0.95 0.85 0.77 0.65 0.56 0.49 0.43 15.8	2.93 2.65 2.42 2.06 1.78 1.57 1.41 1.27 1.07 0.92 0.80 0.72 26.0	4.27 3.87 3.53 3.01 2.62 2.32 2.07 1.88 1.58 1.36 1.19 1.06 38.7	5.36 5.34 4.90 4.19 3.66 3.24 2.63 2.21 1.90 1.67 1.49 54.2	6.97 6.42 5.51 4.82 4.27 3.83 3.48 2.93 2.53 2.22 1.98 72.2	8.75 8.10 7.01 6.15 5.47 4.92 4.47 3.77 3.25 2.86 2.55 93.1	10.7 9.91 8.63 7.61 6.79 6.11 5.55 4.69 4.05 3.57 3.18 117	12.7 11.8 10.4 9.19 8.23 7.43 6.76 5.72 4.95 4.36 3.90 143	14.7 13.8 12.2 10.9 9.78 8.85 8.07 6.85 5.93 5.23 4.67 172	16.8 15.9 14.2 12.7 11.4 10.4 9.48 8.06 7.00 6.18 5.52 204

-  $n_v$ : number of shear plane through bolt. In this case,  $n_v = 1$ .

Demand shear strength for bolt group:

$$V_a = \sqrt{P^2 + V^2} = \sqrt{50^2 + 200^2} = 206.16 \,\mathrm{kN}$$

Demand/Capacity ratio

$$D/C = \frac{206.16}{217.71} = 0.95 : OK$$

## 4.5.2.2 Shear yielding/buckling and flexure yielding

Shear yielding strength of plate	$V_{c} = \frac{1}{\Omega} 0.6F_{y}A_{g} = \frac{1}{1.5} 0.6 \times 34.5 \times 30.5 \times 1.4$	589.26 kN
Plastic section modul of plate	$Z_{pl} = \frac{t_p L_p^2}{4} = \frac{1.4 \times 30.5^2}{4}$	325.59 cm <sup>2</sup>
Flexural yielding strength of plate	$M_{c} = \frac{1}{\Omega} F_{y} Z_{pl} = \frac{1}{1.67} 34.5 \times 325.59 \times 10^{-3}$	67.26 kNm
Demand moment strength	$M_a = Ve = 200 \times 0.202$	40.4 kNm
Demand shear strength	$V_{a}$	200 kN
Demand/Capacity ratio	$D/C = \left(\frac{200}{589.26}\right)^2 + \left(\frac{40.4}{67.26}\right)^2 = 0.48$	ОК

## 4.5.2.3 Bolt bearing under shear load

Distance from edge of outermost hole to edge of plate	$l_{lc1} = L_{ev} - d_b l^2 = 40 - 22 / 2$	29 mm
Distance from edge of hole to hole	$l_c = p_b - d_b = 75 - 22$	53 mm
Bolts bearing capacity at outermost bolts	$r_{n1} = 2(0.6F_n l_{c1}t) = 1.2l_{c1}tF_n = 1.2 \times 2.9 \times 1.4 \times 44.8$	218.27 kN
Bolts bearing capcity between bolts	$r_{n2} = 2(0.6F_{u}l_{c}t) = 1.2l_{c}tF_{u} = 1.2 \times 5.3 \times 1.4 \times 44.8$	398.90 kN
Maximum bearing capacity	$r_{n(\max)} = 2(0.6F_u 2dt) = 2.4dtF_u = 2.4 \times 2 \times 1.4 \times 44.8$	301.06 kN
Total bearing capacity	$\frac{R_n}{\Omega} = \frac{1}{\Omega} C \min(r_{n1}, r_{n2}, r_{n(\max)})$ $= \frac{1}{2} 3.08 \times \min(218.27; 398.90; 301.06)$	336.14 kN
Demand shear strength	$V_{a}$	200 kN
Demand/Capacity ratio	$D/C = \frac{200}{336.14} = 0.59$	OK

**DESIGN GUIDELINES** 

## 4.5.2.4 Shear yielding

Shear yielding strength of plate	$\frac{R_{y}}{\Omega} = \frac{1}{\Omega} 0.6F_{y}A_{g} = \frac{1}{1.5} 0.6 \times 34.5 \times 30.5 \times 1.4$	589.26 kN
Demand shear strength	$V_{a}$	200 kN
Demand/Capacity ratio	$D/C = \frac{200}{589.26} = 0.34$	ОК

## 4.5.2.5 Shear rupture

Net area of plate	$\mathcal{A}_{nv} = [L_{p} - n(d_{b} + 1.6)]t = [305 - 4(22 + 1.6)]14$	2948.4 mm <sup>2</sup>
Shear rupture strength of plate	$\frac{R_n}{\Omega} = \frac{0.6F_n A_{nv}}{\Omega} = \frac{0.6 \times 44.8 \times 29.48}{2}$	396.21 kN
Demand shear strength	V <sub>a</sub>	200 kN
Demand/Capacity ratio	$D/C = \frac{200}{396.21} = 0.50$	ОК

## 4.5.2.6 Block shear rupture

Net area subject to shear	$A_{\mu\nu} = [(n-1)s + L_{e\nu} - (n-0.5)(d_b + 1.6)]t$ = $[(4-1)75 + 40 - (4-0.5)(22+1.6)]14$	2553.6 mm <sup>2</sup>
Net area subject to tension	$\mathcal{A}_{\mu t} = [L_{eb} + (n_c - 1)g - (n_c - 0.5)(d_b + 1.6)]t$ = $\begin{bmatrix} 40 + (2 - 1)100 - (2 - 0.5)(22 + 1.6) \end{bmatrix} 14$	1464.4 mm <sup>2</sup>
Gross area subject to shear	$A_{gv} = [(n-1)s + L_{ev}]t = [(4-1)75 + 40]14$	3710 mm <sup>2</sup>
Block shear rupture strength of plate	$\frac{R_n}{\Omega} = \frac{1}{\Omega} [U_{bs} F_u A_{ut} + \min(0.6F_y A_{gr}; 0.6F_u A_{ur})] \\ = \frac{1}{2} [0.5 \times 44.8 \times 14.64 + \min(0.6 \times 34.5 \times 37.1; 0.6 \times 44.8 \times 25.54)]$	507.4 kN
	$(U_{bs}$ is taken equal to 0.5 due to the tension stress is nonuniform for two of column of bolts)	
Demand shear strength	$V_{a}$	200 kN
Demand/Capacity ratio	$D/C = \frac{200}{507.4} = 0.39$	OK

## 4.5.2.7 Flexure rupture

Plastic section modul of net section will be calculated as below	$Z_{net} = \frac{t}{4} \left[ L_p^2 - \frac{s^2 n (n^2 - 1) (d_b + 1.6)}{L_p} \right]$ $= \frac{14}{4} \left[ 305^2 - \frac{75^2 x 4 x (4^2 - 1) (22 + 1.6)}{305} \right]$	234186 mm³
Flexure rupture strength of plate	$\frac{M_n}{\Omega_b} = \frac{F_n Z_{net}}{\Omega_b} = \frac{44.8 \times 234.19}{2}$	52.46 kNm

Demand moment strength	$M_a = V_a a = 200 \times 0.152$	30.4 kNm
Demand/Capacity ratio	$D/C = \frac{30.4}{52.46} = 0.58$	OK

## 4.5.2.8 Bolt bearing under axial load

Distance from edge of outermost hole to edge of plate	$l_{k1} = L_{ab} - d_b l^2 = 40 - 22 / 2$	29 mm
Distance from edge of hole to hole	$l_c = g - d_b = 100 - 22$	78 mm
Bolts bearing capacity at outermost bolts	$r_{n1} = 2(0.6F_n l_{c1}t) = 1.2l_{c1}tF_n = 1.2 \times 2.9 \times 1.4 \times 44.8$	218.27 mm
Bolts bearing capcity between bolts	$r_{n2} = 2(0.6F_{n}l_{c}t) = 1.2l_{c}tF_{n} = 1.2 \times 7.8 \times 1.4 \times 44.8$	587.06 kN
Maximum bearing capacity	$r_{n(\max)} = 2(0.6F_{u}2dt) = 2.4dtF_{u} = 2.4 \times 2 \times 1.4 \times 44.8$	301.06 kN
Total bearing capacity	$\frac{R_n}{\Omega} = \frac{1}{\Omega} C \min(r_{n1}, r_{n2}, r_{n(\max)})$ $= \frac{1}{2} 3.08 \times \min(218.27; 587.06; 301.06)$	336.14 kN
Demand axial strength	$P_a$	50 kN
Demand/Capacity ratio	$D/C = \frac{50}{336.14} = 0.15$	OK

## 4.5.2.9 Tension yielding

Tension yielding strength of plate	$\frac{R_{\pi}}{\Omega} = \frac{1}{\Omega} F_{y} A_{g} = \frac{1}{1.67} 34.5 \times 30.5 \times 1.4$	882.13 kN
Demand axial strength	$P_a$	50 kN
Demand/Capacity ratio	$D/C = \frac{50}{882.13} = 0.06$	OK

## 4.5.2.10 Tension rupture

Net area subject to tension	$\mathcal{A}_{np} = [L_{p} - n(d_{b} + 1.6)]t = \lfloor 305 - 4(22 + 1.6) \rfloor 14$	2948.4 mm <sup>2</sup>
Axial rupture strength of plate	$\frac{R_n}{\Omega} = \frac{F_u A_{np}}{\Omega} = \frac{44.8 \times 29.48}{2}$	660.35 kN
Demand axial strength	$P_a$	50 kN
Demand/Capacity ratio	$D/C = \frac{50}{660.35} = 0.08$	OK

## 4.5.2.11 Tear out under axial load

Net area subject to shear	$A_{nv} = 2[(n_c - 1)g + L_{eb} - (n_c - 0.5)(d_b + 1.6)]t$ = 2[(2-1)100 + 40 - (2-0.5)(22+1.6)]14	2928.8 mm <sup>2</sup>
Net area subject to tension	$A_{nt} = (n-1)[s - (d_b + 1.6)]t$ = $(4-1)[75 - (22+1.6)]14$	2158.8 mm <sup>2</sup>
Gross area subject to shear	$A_{g^{p}} = 2[(n_{c} - 1)g + L_{eb}]t = 2[(2 - 1)100 + 40]14$	3920 mm <sup>2</sup>
Block axial rupture strength of plate	$\frac{R_n}{\Omega} = \frac{1}{\Omega} [U_{bs} F_n A_{mt} + \min(0.6F_y A_{gy}; 0.6F_n A_{mt})]$ = $\frac{1}{2} [1 \times 44.8 \times 21.59 + \min(0.6 \times 34.5 \times 39.2; 0.6 \times 44.8 \times 29.29)]$ (U <sub>bs</sub> is taken equal to 1 because the tension stress is uniform for all of rows of bolts)	877.27 kN
Demand axial strength	$P_a$	50 kN
Demand/Capacity ratio	$D / C = \frac{50}{877.27} = 0.06$	OK

## 4.5.3. Beam checking

Checking bolt bearing under shear load and tension load, shear yielding, tension rupture and tear out under axial load similarly for beam. The other limit states are not necessary.

# **CHAPTER 5. CRANE SYSTEMS DESIGN** 5.1. Crane Types

The most common crane systems used in steel buildings are given in Table 5 1.

These categories of crane service classification have been established in Table 5 2.

This document refers only to Classes A through D.

Table 5 1	General	Ranae	of Crane	Types
	Generat	nunge	of cranc	iypes

Crane Type	Power Source	Description	Span or Reach (m)	Capacity (Tons)
Under hung or Monorails	Hand Geared or Electric	Single Girder	3 ÷ 15.24	1/2÷10
	Hand Geared or Electric	Single Girder	3 ÷ 15.24	1/2÷10
	Electric	Double Girder	6 ÷ 18.29	5 ÷ 25
Top Running	Electric	Box Girder Pendant-Operated 4-Wheel End Truck	6 ÷ 27.43	5 ÷ 25
	Electric	Box Girder Cab Operated 4-Wheel End Truck	15.24 ÷ 30.48	Up to 60
	Electric	Box Girder Cab Operated 8-Wheel End Trucks	15.24 ÷ 30.48	Up to 250
JIB Cranes	Hand Geared or Electric	Floor Mounted 280° ÷ 360° Column Mounted 180°	2.43 ÷ 6	1/4 ÷ 5

#### Table 5 2 CMAA Crane Service Classes<sup>5</sup>

Class	Usage range	Description
<b>CLASS A</b> (Standby or Infrequent Service)	Power houses, public utilities, turbine rooms, motor rooms, and transformer stations, etc.	slow speeds and with long periods of idling between each lift.
<b>CLASS B</b> (Light Service)	light assembly facilities, repair shops, service buildings, and warehousing, etc.	<ul> <li>slow speeds and light service.</li> <li>2 to 5 lifts per hour,</li> <li>average lift distance of 10 feet</li> </ul>
<b>CLASS C</b> (Moderate Service)	manufacturing, machine shops, or paper mill machine rooms, etc.	<ul> <li>average loads is 50% of the rated capacity,</li> <li>5 to 10 lifts per hour</li> <li>average lift distance of 15 feet</li> </ul>
<b>CLASS D</b> (Heavy Service)	heavy machine shops, foundries, fabricating plants, steel warehouses, container yards, lumber mills, etc.	<ul> <li>quick speeds and moved constantly</li> <li>loads approaching 50% of the capacity throughout the workday but not over 65%,</li> <li>10 to 20 lifts per hour</li> <li>average heights of 15 feet</li> </ul>
<b>CLASS E</b> (Severe Service)	Magnet/bucket combination cranes for heavy items like crap yards or production mills, cement mills, lumber mills, fertilizer plants, and container handling, etc.	
<b>CLASS F</b> (Continuous Severe Service)	industrial settings, etc.	<ul> <li>high capacity loads with constant frequency</li> <li>require the ability to handle the most extreme working conditions.</li> </ul>

<sup>5</sup>Crane Manufacturers Association of America CMAA (2008), Specification No. 70, Charlotte, NC.

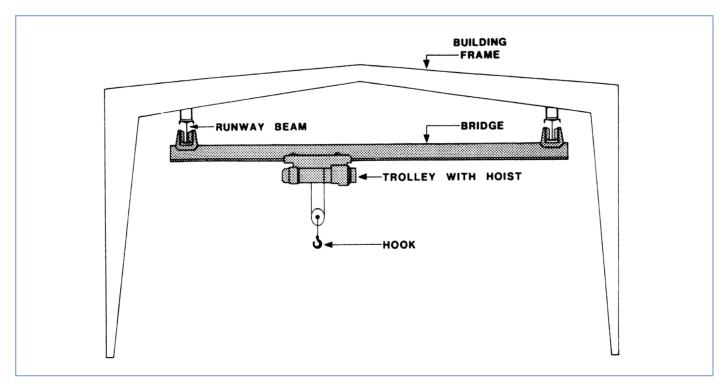


Figure 5 1 Underhung Bridge Crane

The monorail cranes are characterized by the hoist being suspended from the lower flange of a single supporting runway beam (See Figure 5 2). Monorails are used where the need to lift and move items can be confined to one direction.

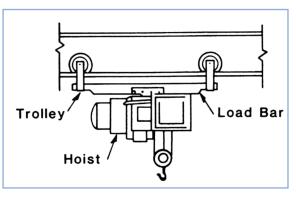


Figure 5 2 Monorail Crane

Top Running Cranes as show in Figure 5 3 are generally used in workshops and warehouses where lifting capacity is required over a large span of the floor area.

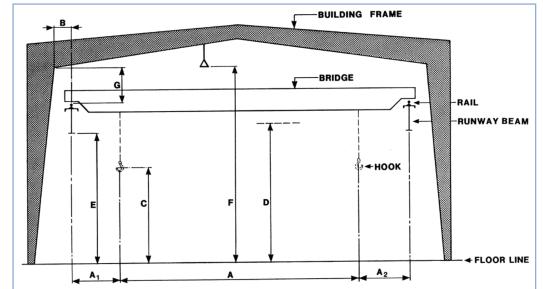
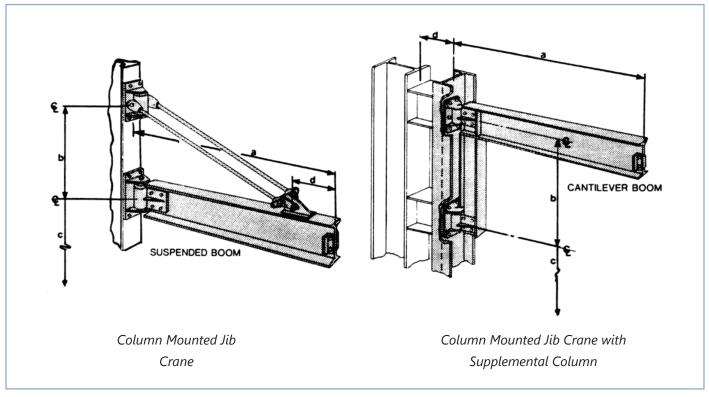


Figure 5 3 Overhead Crane in Design Example

Top Running Cranes usually provide greater hook height and clearance below the crane girder than underhung cranes.



The JIB crane is a crane that has a rotating horizontal boom attached to a fixed.

Bridge cranes can be designed having either single girder, double girder or box girder.

Single girder is generally used on shorter spans and lower capacities or service classifications. Double girder cranes provide greater hook height, but are no more durable than single girder cranes. However, crane and accessories such as bridge cranes, hoist, and trolley, etc. are provided by the manufacturer, not by BMB&A.

Figure 5 4 Jib Crane

## 5.2. Design procedure

The clients will usually show their basic requirements in the design brief. Then the designers need to establish various parameters that will influence the structural design of the building.

These may include like below:

- (1) Crane type (top running, underhung, etc.);
- (2) Capacity (rated in tons) and service classification (CMAA);
- (3) Power source: most use electric instead of hand geared.

For electric powered cranes, method of operation (pendant, cab, or radio);

- (4) Crane load:
- a. The self-weight: Bridge weight (CW) and weight of trolley with hoist (HT);
- b. Maximum wheel load without impact;

c. Special allowances for vertical impact, lateral force, longitudinal force, or the loads factored for dynamic effects and lateral loads, if required.

- (5) Dimensions:
- a. The building layouts

b. Number wheelbase (NWb), distance between cranes (LCr), end-truck length (N), distance wheelbase(W);

c. Level of the top of rail (TOR), the clearance above the top of the rail, Horizontal clearance, vertical clearance, and clearance beneath the runway beam or hook height.

(6) Deflection limits for the crane runway beam and portal frame.

Utilization and state of loading for fatigue assessment.



## NOTE

There can be a significant difference in wheel loads and geometry between single and double girder cranes. If the designer cannot establish the make of the crane, then a contingency of 10% load could be added to the load provided by one manufacturer to allow for other make which might be adopted.

The speed of hand-geared cranes is low, and the impact forces which supporting structures may resist are low compared to the faster electric powered cranes.

Once the crane wheel loads and overall geometry have been established, the general design procedure is as given below:

- (1) Design the runway beams
- (2) Determine the crane load reactions on the bracket and load combinations
- (3) Design of main frames.

# 5.3. Design the runway beam

It is assumed the crane runway beams are simple beams supported at the brackets that cantilever from the main portal columns.

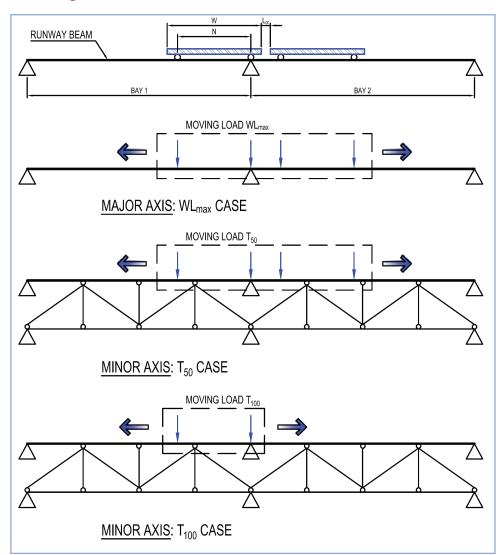


Figure 5 5 Schematics of Crame Runway Beam



## NOTE

## Bracing

When runway beam depth is less than **900mm**, top flange beam need to be braced by I-260x164x5x6, and V50x50x5 or larger if necessary.

When runway beam depth is not less than 900mm, need to use both top and bottom flange bracing, by I-260x164x5x6, and V50x50x5 or larger if necessary.

## Width of runway beam's top flange

When length of building is greater than **50m**, the width of runway beam's top flange need to be greater than **200mm** to avoid errors in rail installation.

## 5.3.1. Crane Wheel Load

#### Wheel Load

The maximum wheel load ( $^{WL}_{max}$ ) and the minimum wheel load ( $^{WL}_{min}$ ) can be provided by the crane manufacturer or may be conservatively approximated from the crane loads as follows:

WL <sub>max</sub>	$x = \frac{RC + HT + 0.5CW}{NWb}$	(2012MBSM, Eq. 2.4.1-1)
WL <sub>mir</sub>	$h_{n} = \frac{0.5CW}{NWb}$	
where	,	
WL	= Maximum wheel load	
RC	= Rated capacity of the crane	
ΗT	= Weight of hoist with trolley	
CW	= Weight of the crane excluding the hoist with	trolley

%

25

25

10

0

NWb = Number of end truck wheels at one end of the bridge

#### Vertical Impact Force

Crane Type The maximum wheel load (WLmax) used for the design of runway beams, including monorails, Monorail cranes (powered) their connections and support brackets, shall Cab-operated or radio operated bridge be increased by the percentage given below to cranes (powered) determine the induced vertical impact or vibration force. Pendant-operated bridge cranes Vertical impact shall not be required for the design (powered) of frames, support columns, or the building Bridge cranes or monorail cranes with foundation. hand-geared bridge, trolley and hoist

#### Lateral Force

The lateral force shall be assumed to act horizontally at the traction surface of a runway beam, in either direction perpendicular to the beam, and shall be distributed with due regard to the lateral stiffness of the runway beam and supporting structure.

The lateral force in each end-truck wheel on crane runway beams with electrically powered trolleys shall be calculated as below:

$$T = 20\% \frac{RC + HT}{NWb}$$

## Longitudinal Force

The longitudinal force, acting horizontally at the top of the rails and in each direction parallel to each runway beam on crane runway beams, shall be calculated as 10 percent of the maximum wheel loads excluding vertical impact:

$$P = 10\% WL_{max}$$

## 5.3.2. Design runway beam using SAP 2000

Step-by-step moving-load analysis is initiated through the following process:

- Define Vehicles type through *Define > Moving Loads > Vehicles*: VEH-X: defines the maximum wheel load.
   VEH-Y-T50: defines 100% Lateral Forces.
   VEH-Y-T100: defines 50% Lateral Forces.
- Define The Paths through *Define > Moving Loads > Paths*.
   PATH-X: defines the lane in X-axis.
   PATH-Y: defines the lane in Y-axis.
- Define Moving-load Patterns and Cases through *Define > Load Patterns* and *Load Cases*:
   DEAD: defines self-weight of model.
   WLmax: defines moving-load VEH-X on PATH-X lane.

VIIIIdx. defines moving-load VER-A on FAIR-A lane.

**T100**: defines moving-load **VEH-Y-T100** on **PATH-Y** lane. **T50**: defines moving-load **VEH-Y-T50** on **PATH-Y** lane.

Set the load-case type to Moving Load, and then specify the vehicles and paths assigned to this moving load, as shown below:

Load Case Data - Mo∨ing Load				
Load Case Name Notes ACASE1 Set Def Name Modify/Show	Load Case Type Moving Load			
<ul> <li>Stiffness to Use</li> <li>C Zero Initial Conditions - Unstressed State</li> <li>C Stiffness at End of Nonlinear Case</li> <li>Important Note: Loads from the Nonlinear Case are NOT included in the current case</li> </ul>				
Loads Applied Vehicle Scale Factor Loaded Loaded Assign Class Lanes Lanes Lanes Loaded VECL1  1 0 0 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	Lane Definitions Loaded for Assignment List of Lane Definitions Definitions			
Add Modify Delete	Add -> <- Remove			

Figure 5 6 Load-case Data

## Combinations

Checklist	Combinations	Load case	Output
Strength	M33	DL + WLmax	M33 & M22
	M22-T100	T-100	M22 & V33
	M22-T50	T-50	M22 & V33
	CV-X	DL + WLmax/ $\alpha$ One crane on single aisle: $\alpha$ = Vertical Impact I Other case: $\alpha$ = 1	$\Delta x$
Deflection	CV-Y-T100	T-100	Δy
	CV-Y-T50	T-50	Δy

## Analysis

- Set Analysis Options: *Plane Frame*.
- Sum ratios of runway beam both axis should be less than 1.0.
- Check Deflection as table below:

## Table 5 3 Types Deflection Limitations for Top Running Cranes<sup>6</sup>

	DEFORMATION	REMARK
Top Running Cranes		
	L/600 (Classes A/B/C)	
1/ Runway Beam (vertical deflection)	L/800 (Class D)	CV-X
	L/1000 (Class E/F)	
2/ Runway Beam (horizontal deflection)	L/400	CV-Y-T100 & CV-Y-T50
Underhung and Monorail Crane		
Runway Beam (vertical deflection)	L/450 (Classes A/B/C)	CV-X
Crane Bracket (horizontal deflection)	DL + CR or WL 10yr.	
1/ Cab or Radio - Operator cranes	$H/240 \text{ or} \le 5.08 \text{cm}$	Crane lateral or WL 10yr.
2/ Pendant - Operator cranes	H/100	Grane fateral of w12 foyl.

## Output

Using BMB's calculation sheet presents the images and data of the runway-beam design.

# 5.4. Design Steel Frame

## 5.4.1. General

The general design main frame procedure is as given below:

- 1. Determine the load reactions on the bracket.
- 2. Add the crane runway beam dead load to the dead load and add the following new load case:
- a. Crane loads with the maximum wheel load at left column.
- Lateral crane loads with maximum at left column and acting from right to left.
- b. Crane loads with maximum load at right column.
- Lateral crane loads with maximum at right column and acting from left to right

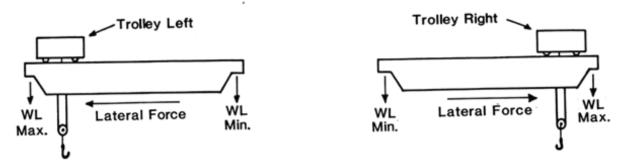


Figure 5 7 Crane Loading Conditions

Crane buildings have single or multiple cranes acting in one or more aisles shall be designed with the crane or cranes located longitudinally in the aisle or aisles in the positions that produce the most unfavorable effect. Unbalanced loads shall be applied as induced by a single crane operating in a crane aisle, and by a crane or cranes operating in one crane aisle of a building with multiple crane aisles. See Table 5 4 for a summary of these provisions.

		Schematic	Vertical Impact	Lateral Force
aisles with multiple cranes An any An c a cran	Any one crane in any aisle		0%	100%
	Any two adjacent cranes in any aisle		0% Both cranes	50% Both cranes, or 100% Either crane
	Any one crane in any two adjacent aisles	RL CAME RL RB	0% Both cranes	50% Both cranes, or 100% Either crane
	Any two adjacent cranes in any aisle and one crane in any other nonadjacent aisle	FL	0% All cranes	50% All three cranes, or 100% Any one crane

 Table 5 4 Loading for Building Frames and Support Columns

## 5.4.2. Bracket System

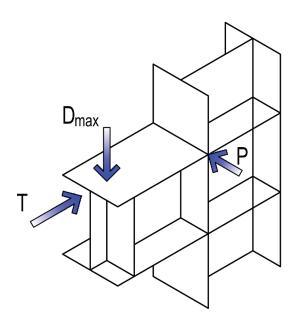


Figure 5 8 Loading on bracket

## 5.4.3. Bracing

The lateral load on a crane runway beam is transmitted to the main frame column by the bracing systems in the minor axis.

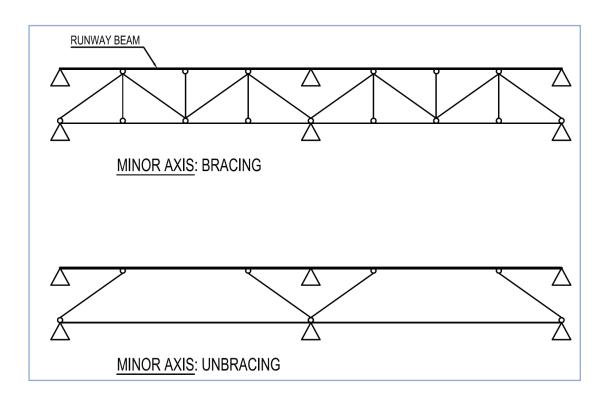


Figure 5 9 Bracing of minor axis

# 5.5. Design example - One Crane per Aisle

## 5.5.1. Input

Bay spacing	B = 7.50 m
Crane Data	5 Tons Top Running Crane, 13.72 m span, Class C, top of rail 5.75 m
Bridge weight	CW = 2.79 Tons
Hoist and trolley weight	HT = 0.5 Tons
Service classification	Class C, Radio Operation
2 End-truck wheel	End-truck length N = 2.184 m, Distance Wheelbase W = $1.829$ m

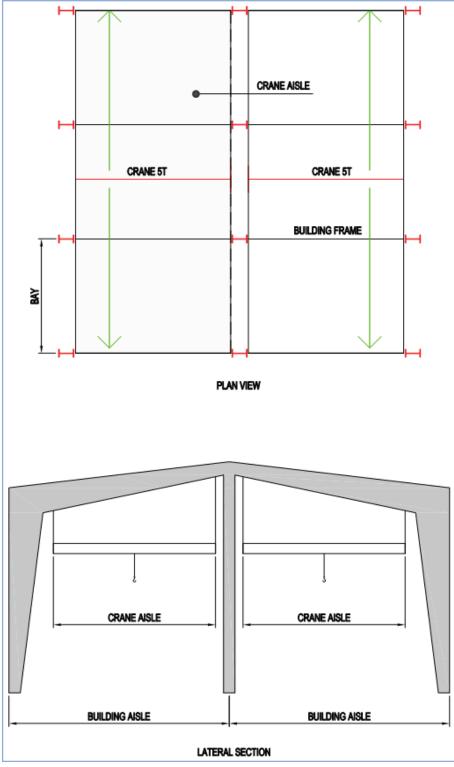


Figure 5 10 Building Layout

## 5.5.2. Crane Load

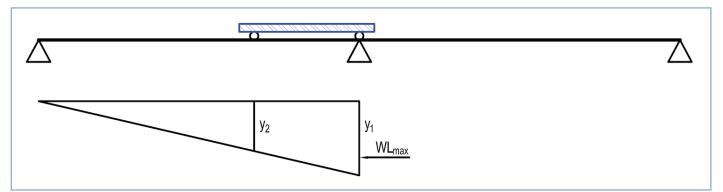
## Crane Wheel Load

Crane Wheel Load	$WL \max = (5 + 0.5 + 0.5 \times 2.79) / 2 \times 10$	34.5 (kN)
	$WL\min = 2.79 / (2 \times 2) \times 10$	7.0 (kN)
Lateral Force	$T = 20\%(5 + 0.5) / 2 \times 10$	5.5 (kN)
Longitudinal Force	$P = 10\% \times 34.5$	3.45 (kN)

## Loading for Runway Beam

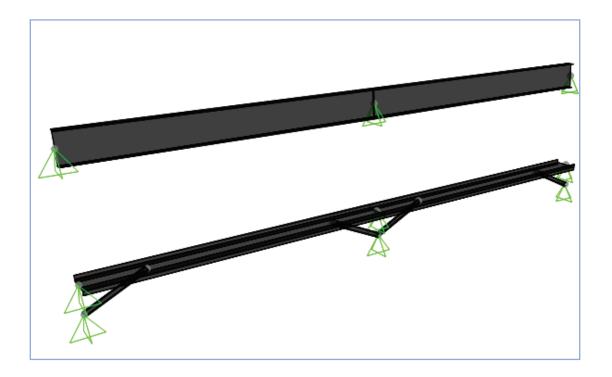
Vertical impact	I (Radio Operation)	1.25
Maximum vertical wheel load	$WL \max I = 34.5 \times 1.25$	43.1 (kN)
100% Lateral Force	T100 = T = 5.5	5.5 (kN)
Longitudinal Force	$P = 0.1 \times WL \max \times NWb = 0.1 \times 34.5 \times 2$	6.9 (kN)

## Loading for Building Columns



Vertical load on column near trolley	$D \max = WL\max \times (1 + (B - W) / B)$ = 34.5 \times (1 + (7.5 - 1.829) / 7.5)	60.6 (kN)
Vertical load on column away from trolley	$D\min = WL\min \times (1 + (B - W) / B)$ = 7.0 × (1 + (7.5 - 1.829) / 7.5)	12.3 (kN)
Lateral load on building columns	$T \max = T \times (1 + (B - W) / B)$ = 5.5 \times (1 + (7.5 - 1.829) / 7.5)	9.66 (kN)
Longitudinal load on building columns	$P = 0.1 \times WL \max \times NWb$	6.9 (kN)

## 5.5.3. Runway Beam Modeling



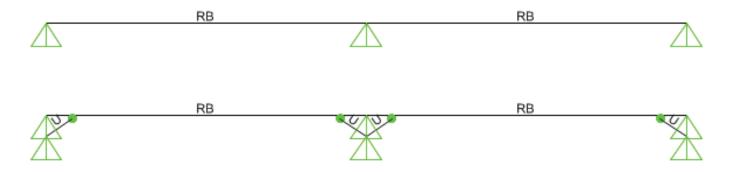
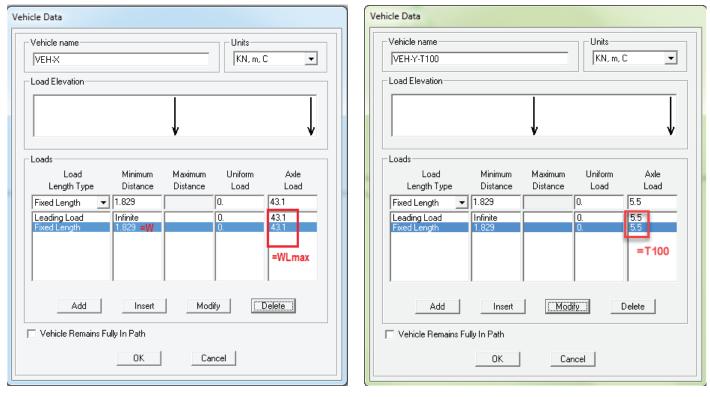


Figure 5 11 Runway Beam in SAP2000 Modeling Section: RB (I – 766x184x8x10), U (U 120x52x4.8x7.8)

## Define Vehicles type

Define Vehicles			
Vehicles VEH-X VEH-Y-T50 VEH-Y-T100	Click to: Add Vehicle Add Copy of Vehicle		
	Modify/Show Vehicle Delete Vehicle		

(a) Define Vehicles



(b) Vehicles "VEH-X" defined for WLmax

(c) Vehicles "VEH- Y-T100" defined for T100

Figure 5 12 Define Vehicles





Figure 5 13 Define Paths of Moving Loads \*

PATH-X defined the path on the Major-axis (beams labeled 1 and 2) PATH-Y defined the path on the Minor-axis (beams labeled 3 and 4)



## NOTE

The **discretization** is as same as output-station spacing of frame. The effect of refining path discretization is apparent in the influence lines which follow (Figure 5 14):

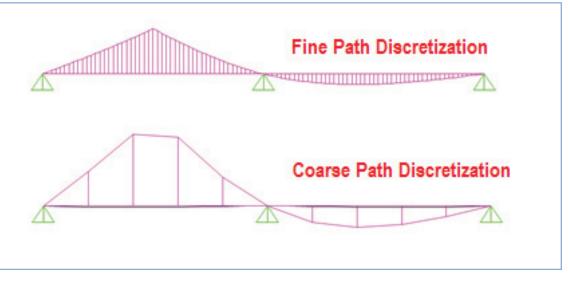
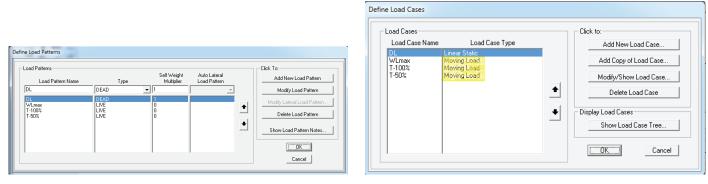


Figure 5 14 Influence line



#### Define Load-Patterns

#### Define Load-Case

Load Case Data - Moving Load		Load Case Data - Moving Load	
Load Case Name UMLmax Set Def Name Modify/Show	Load Case Type Moving Load  Design	Load Case Name    Load Case Name	Load Case Type Moving Load   MultiPath Scale Factors
Stiffness to Use C Zero Initial Conditions - Unstressed State C Stiffness at End of Nonlinear Case Important Note: Loads from the Nonlinear Case are NOT included in the current case	MultiPath Scale Factors Number of Reduction Paths Scale Factor Loaded 1 1 1 1 1 Modify	Summess to Use  C Zero Initial Conditions - Unstressed State  C Stiffness at End of Nonlinear Case Important Note: Loads from the Nonlinear Case are NOT included in the current case	Mumber of Reduction       Paths     Scale Factor       Loaded     1       2     1.
Loads Applied Vehicle Scale Factor Loaded Loaded Number VEHX V 1. 0 0 Loaded 1 VEHX 1. 0 0 Some	Paths Loaded for Assignment 1 List of Path Selected Path Definitions Definitions PATHX PATHX	Loads Applied Vehicle Scale Factor Loaded Loaded Assign Class Paths Paths Paths Number VEH-Y-T1 v 1. 0 0 Loaded 1 VEH-Y-T100 1. 0 0 Some	Paths Loaded for Assignment 1 List of Path Selected Path Definitions Definitions PATHX
Add Modify Delete	Add -> <- Remove	Add Modify Delete	Add -> <- Remove
Cancel		Cancel	MSSSRC1

Load-case WLmax Data

Load-case T100 Data

#### Output

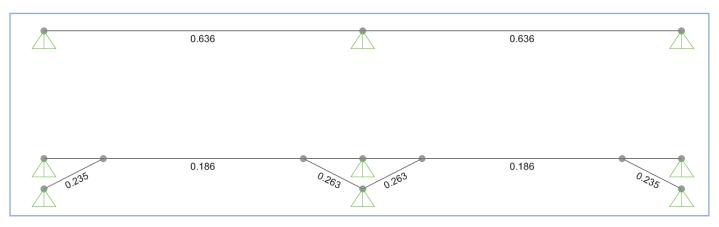


Figure 5 15 Steel P-M Interaction Ratios of Runway Beam\*



## NOTE

This ratio is only used for reference, since the SAP model does not include the longitudinal force. The total ratio of the major and minor axis must be less than **1.00**.

#### Check Runway Beam by Calculation Sheet

#### I. INPUT DATA

1. Material

Steel		A572 Grade 50
Yield strength of steel	F <sub>y</sub> =	345 MPa
Modulus of elasticity of steel	E =	200000 MPa

2. Section

Section of runway beam

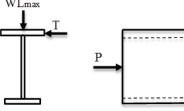
Section	Height	Web thickness         Compression Flange         Tension Fla		Compression Flange		Flange
Built-up member	d (mm)	t <sub>w</sub> (mm)	b <sub>1</sub> (mm)	t <sub>f1</sub> (mm)	b <sub>2</sub> (mm)	t <sub>f2</sub> (mm)
l - 766x184x8x10	766	8	184	10	184	10

Factor

Length	Length	Factor	Stiffeners	Factor
L <sub>b</sub> (mm)	L(mm)	К	@a (mm)	Cb
4700	7500	1	2000	1.088

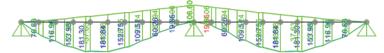
#### II. FORCE

Loading for runway beam			Crane 1
Maximum vertical wheel load	WL <sub>max</sub> =	kN	43.11
100% side thrust in each end-truck wheel max	T=	kN	6.88
Longitudinal force	P =	kN	6.90
Vertical impact	I.		1.25
WI may			



Force of runway beam					
Axial force	Axial force Moment Shear force				
P (kN)	M <sub>33</sub> (kNm )	M <sub>22</sub> (kNm)	V <sub>22</sub> (kN)	V <sub>33</sub> (kN)	
-6.90	181.84	5.76	140.6	10.95	

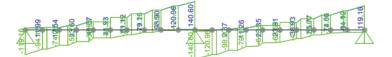
Resultant Moment M33, kNm



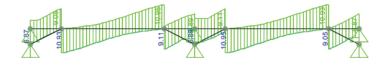
Resultant Moment M22, kNm



Resultant Shear V22, kN



Resultant Shear V33, kN





#### NOTE

\*Length Lb: the largest un-bracing length of runway beam \*Factor Cb: get from Detail of Steel Stress Check Data

#### **III. CHECK OF MENBERS FOR STRENGTH CAPACITY**

1. CHECK OF RUNWAY BEAM FOR STRENGTH CAPACITY

#### a. Design of runway beam for Flexure

Design of runway beam for flexure - major axis

	M <sub>cx</sub> =	228.48 kNm	
	M <sub>cx/No LTB</sub> =	432.92 kNm	[Chapter F5]
Design of runway beam for flexure - minor axis			
	M <sub>cy</sub> =	37.29 kNm	[Chapter F6]
b. Design of runway beam for Compression			
	P <sub>c</sub> =	488.71 kN	[Chapter E]
c. Design of runway beam for Shear			
Design of runway beam for shear - major ax is			
	V <sub>cx</sub> =	435.52 kN	[Chapter G2]
Design of runway beam for shear - minor axis			
	V <sub>cy</sub> =	456.14 kN	[Chapter G7]

d. Design of runway beam for Combined Forces and Torsion

Design of runway beam for combined flexure and axial force

AXL	B-MAJ	B-MIN	D/C ratio
Pr/Pc	(M 33/M cx)^2	M22/Mcy	H1.3b,H1-2
0.014	0.633	0.154	0.802

Design of runway beam for Shear

$$\frac{V_r}{V_{cy}} = \frac{V_{22}}{V_{cy}} = 0.323 \quad \text{[Satisfactory]}$$

$$\frac{V_r}{V_{cx}} = \frac{V_{33}}{V_{cx}} =$$
 **0.024** [Satisfactory]

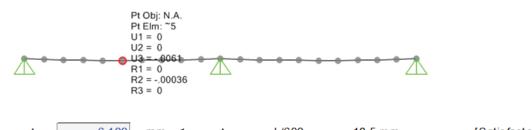
## 2. CHECK OF BRACING FOR STRENGTH CAPACITY

Design of bracing for Compression

Pc = 41.2 kN [Satisfactory] [Chapter E5]

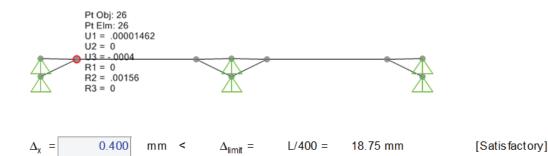
## **III. CHECK DEFORMATION**

1. Vertical Defection





2. Horizontal Defection



## I. INPUT DATA

#### 1. Bolt

T. DOR	Grade		10.9	
	Туре		M24	
	Ultimate strength of bolt	F <sub>u</sub> =	100 kN/cm <sup>2</sup>	
	Ultimate tensile strength	$F_{nt} = 0.75 F_{u} =$	75 kN/cm <sup>2</sup>	
	Shear strength	$F_{nv} = 0.45 F_{u} =$	45 kN/cm <sup>2</sup>	ſF
2. Force	No. of bolt	n =	4	
	Reaction at support	V <sub>max</sub> =	75.7 kN	

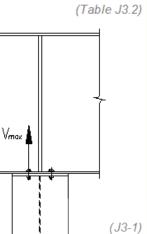
## **II. CALCULATING**

Bolt resisting tension

Hence:

Tensile resistance of bolt

$$\frac{R_{nt}}{\Omega} = \frac{F_{nt} \times A_b}{\Omega} = 169.65 \text{ kN}$$
$$\Omega = 2$$



Where: Max. Tension in outermost bolt

$$N_{bM} = \frac{V_{max}}{m_r} = 37.9 \text{ kN}$$

Where: mr is no. of bolt on 1 row Total of tension on 1 bolt

$$T_{b} = 37.9 \text{ kN}$$
  
 $\frac{R_{nt}}{\Omega} \ge T_{b}$ 

0K

## Check Welding at Bracket

## I. MATERIAL

#### 1. Bracket

2. Weld

ASTM specifications for Steel	A572 Grade 50
Yeild strength	F <sub>y</sub> = 345 MPa
	E60XX
Strength of the weld	F <sub>EXX</sub> = 414 MPa
Height of the fillet vertical weld	h <sub>v</sub> = 5 mm
Height of the fillet horizontal weld	h <sub>h</sub> = 7 mm

h<sub>h</sub> = 7 mm  $I_v =$ 900 mm l<sub>h</sub> = 800 mm

## **II. INTERNAL FORCE FOR CALCULATING**

Total length of horizontal weld

Total length of vertical weld

Shear	V = 75.70 kN
III. DIMENSION	
Web height of bracket	d = 500 mm
Width flange	b <sub>f</sub> = <u>212</u> mm
Distance	e = <u>300</u> mm

#### **IV. CALCULATING**

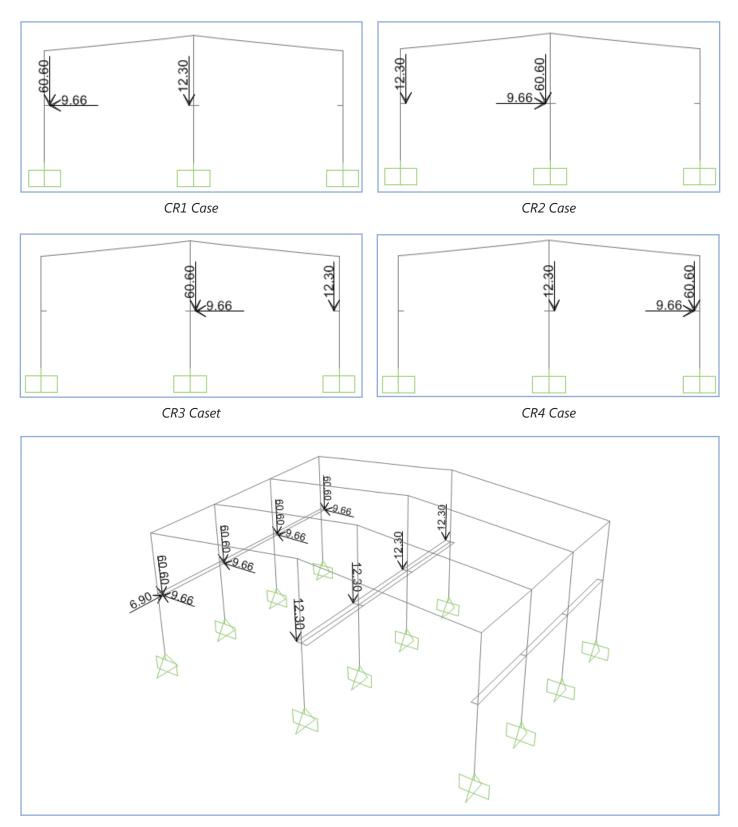
Eccentric moment		22.7101 kNm		
Allowable shear stress of the				
	$\frac{R_n}{\Omega} = \frac{F_{nw} \times 1m^2}{\Omega} =$	103,500 kN/n	n <sup>2</sup>	
Where:	$F_{nw} = 0.6F_{EXX} =$	207000 kN/n	1 <sup>2</sup>	(J2-5)
	Ω =	2		
Weld area				
		0.006 m <sup>2</sup>		
Moment of inertia of the weld				
	I <sub>w</sub> =	0.000308 m <sup>4</sup>		
Section modulus				
	$W_w = \frac{2I_w}{(d+h)} =$	0.0012 m <sup>3</sup>		
Checking stress of the weld	(u+n)			
$\tau = \sqrt{\tau_M^2 + \tau_V^2}$	$=\sqrt{\left(\frac{M}{W_w}\right)^2 + \left(\frac{V}{A_{we}}\right)^2} =$	22469 kN/n	n <sup>2</sup>	
Hence:	τ	$\leq \frac{R_n}{R_n}$		OK
Check the weld length to we		Ω		
	l <sub>w</sub> /h =	90.00	< 100	
	0 -1 0 0 000/L /b) -	1.02		
	β =1.2-0.002(l <sub>w</sub> /h) =		~ 1	(J2-1)
	$\rightarrow \beta =$	1.00		
The capacity of the weld	D RD			
	$\frac{R_{n1}}{\Omega} = \frac{\beta R_n}{\Omega} =$	103,500 kN/n	n <sup>2</sup>	
Hence:	R <sub>n1</sub> / Ω	≥ τ		OK

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## 5.5.4. Main Frame

## The load frame

Columns supporting crane fixed about the major axes and pinned about minor axes.



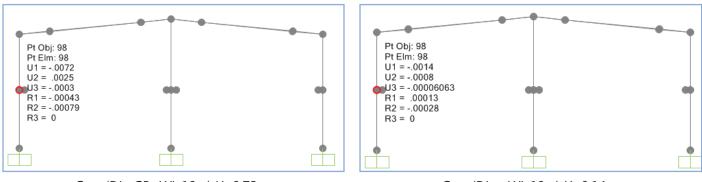
Typical Longitudinal Force for "CR1 Case"

The crane runway beam and rail are to be included as dead loads, applied at the bracket. Note that for clarity, dead loads (DL), live roof loads (LL), wind loads (WL) are not shown in this example.

## The combinations

COMBO 1	DL + LL	COMBO 16	DL + 0.75CR3 + 0.45WL1
COMBO 2	DL + 0.6WL1	COMBO 17	DL + 0.75CR4 + 0.45WL1
COMBO 3	DL + 0.6WL2	COMBO 18	DL + 0.75CR1 + 0.75CR3 + 0.45WL1
COMBO 4	DL + 0.75LL + 0.45WL1	COMBO 19	DL + 0.75CR1 + 0.75CR4 + 0.45WL1
COMBO 5	DL + 0.75LL + 0.45WL2	COMBO 20	DL + 0.75CR2 + 0.75CR3 + 0.45WL1
COMBO 6	DL + CR1	COMBO 21	DL + 0.75CR2 + 0.75CR4 + 0.45WL1
COMBO 7	DL + CR2	COMBO 22	DL + 0.75CR1 + 0.45WL2
COMBO 8	DL + CR3	COMBO 23	DL + 0.75CR2 + 0.45WL2
COMBO 9	DL + CR4	COMBO 24	DL + 0.75CR3 + 0.45WL2
COMBO 10	DL + CR1 + CR3	COMBO 25	DL + 0.75CR4 + 0.45WL2
COMBO 11	DL + CR1 + CR4	COMBO 26	DL + 0.75CR1 + 0.75CR3 + 0.45WL2
COMBO 12	DL + CR2 + CR3	COMBO 27	DL + 0.75CR1 + 0.75CR4 + 0.45WL2
COMBO 13	DL + CR2 + CR4	COMBO 28	DL + 0.75CR2 + 0.75CR3 + 0.45WL2
COMBO 14	DL + 0.75CR1 + 0.45WL1	COMBO 29	DL + 0.75CR2 + 0.75CR4 + 0.45WL2
COMBO 15	DL + 0.75CR2 + 0.45WL1	COMBO 30	0.6DL + 0.6WL1
		COMBO 31	0.6DL + 0.6WL2

## **Deflection Checking**



Case 'DL+CR+WL 10yr.': Y=0.72cm

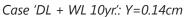


Figure 5 16 Crane Bracket Horizontal Deflection

 $[Y] = \min[5.08cm; (575/240)] = 5.08cm$ 

## Steel stress check



## NOTE

When check steel column need to overwrite Factor "Unbraced Length Ratio (Major)" to 1.00 as Figure 5 17.

			Item Description
	ltem	Value 🔺	
1	Current Design Section	Program Determined	
2	Framing Type	Program Determined	
3	Omega0	Program Determined	
4	Consider Deflection?	No	
5	Deflection Check Type	Program Determined	
6	DL Limit, L /	Program Determined	
7	Super DL+LL Limit, L /	Program Determined	
8	Live Load Limit, L /	Program Determined	
9	Total Limit, L/	Program Determined	
10	TotalCamber Limit, L/	Program Determined	
11	DL Limit, abs	Program Determined	
12	Super DL+LL Limit, abs	Program Determined	
13	Live Load Limit, abs	Program Determined	
14	Total Limit, abs	Program Determined	
15	Total-Camber Limit, abs	Program Determined	
16	Specified Camber	Program Determined	
17	Net Area to Total Area Ratio	Program Determined	
18	Live Load Reduction Factor	Program Determined	
19	Unbraced Length Ratio (Major)	1.	
20	Unbraced Length Ratio (Minor)	Program Determined	
21	Unbraced Length Ratio (LTB)	Program Determined	
22	Effective Length Factor (K1 Major)	Program Determined	Explanation of Color Coding for Values
23	Effective Length Factor (K1 Minor)	Program Determined	
24	Effective Length Factor (K2 Major)	Program Determined 🗨	Blue: All selected items are program determined
et To	Prog Determined (Default) Values	Reset To Previous Values	Black: Some selected items are user defined
	All Items Selected Items	All Items Selected Items	Red: Value that has changed during the current session

Figure 5 17 Define ratio for steel stress check

DESIGN GUIDELINES

# **CHAPTER 6. MEZZANINE FLOOR DESIGN 6.1. General**

Mezzanines and platforms are often required in industrial buildings. The type of usage dictates design considerations. For proper design the designer needs to consider the following design parameters:

- 1. Occupancy or Use.
- 2. Design Loads (Uniform and Concentrated).

The dead load includes the weight of panel, concrete slab, finish floor and, wall on floor, and self-weight of joist.

The live load depends on the purpose of the floor. Refer Chapter 2 for more details.

3. **Type of slab**: composite slab, checkered plate, gratings, expanded metals, etc. Each type of floor requires different design of joists.

Mezzanine joists are analyzed and designed as simple span members.

Joist beam spacing needs being less than **2000mm** and depend on bay spacing for equal distances.

- 4. Stair, Opening, Guard rail requirements.
- 5. Design Criteria (if required): Deflection limitation, Vibration Control, Lateral Stability Requirements.
- 6. Future Expansion.

## **Design Procedure**

- 1. Design of Joists:
- 2. Design of Flooring
- 3. Design of Main Frames

# 6.2. Composite Slab

Composite Metal Deck Slabs - most commonly used today. Advantages:

- Stay in place form.
- Slab shoring typically not required.
- Metal deck serves as positive reinforcement.
- Metal deck serves as construction platform.

Shear connector can use Steel Headed Stud or Steel Channel Anchors.

Steel headed stud anchors shall be welded through the deck to the steel cross section. Such anchorage shall be provided by steel headed stud anchors, a combination of steel headed stud anchors and arc spot (puddle) welds, or other devices specified by the contract documents.

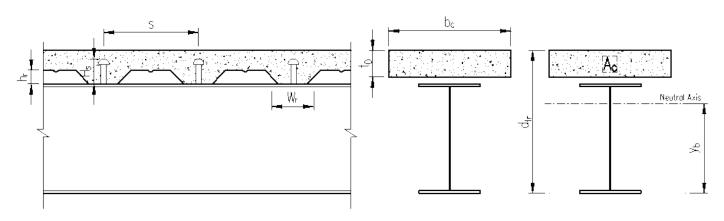


Figure 6 1 Composite Beam Dimensions with Ribs Perpendicular to Beam Span

## 6.2.1. Section Properties

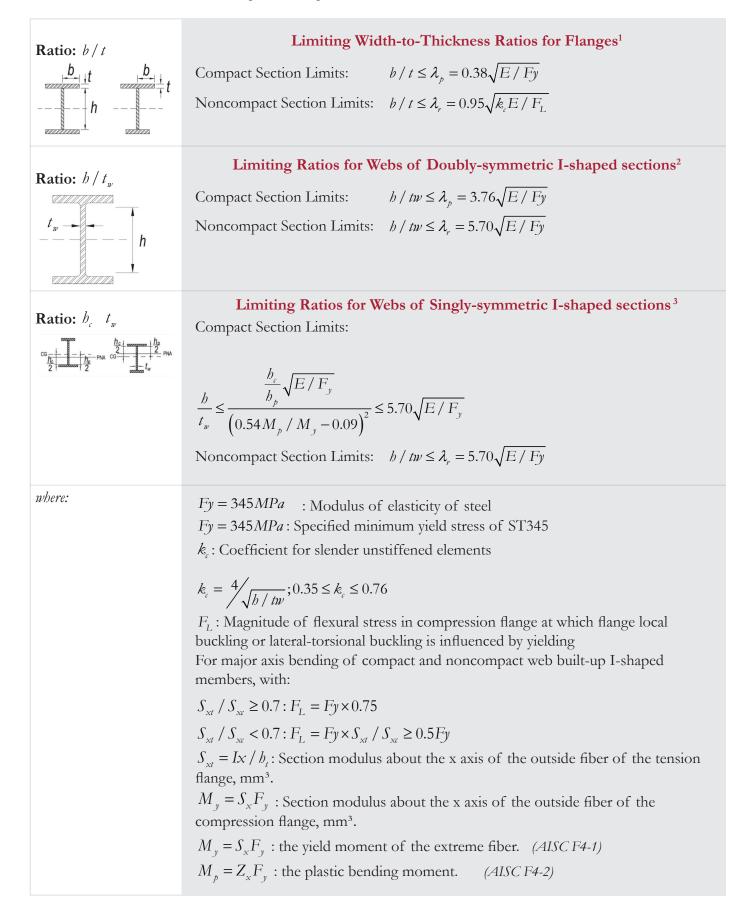
The section properties are reported with respect to the section local axes (2-3). Furthermore the section properties are reported assuming that the entire section is transformed into an equivalent area of the specified base material.

Cross-section area	$A = 2 \times bf \times tf + tw \times hw$
Moment of Inertia	$Ix = \sum \left(\frac{bb^{3}}{12}\right) + \sum Ay_{i}^{2} = \frac{t_{w} \times b_{w}^{3}}{12} + bf \times \frac{b^{3} - b_{w}^{3}}{12}$
Plastic Section Modulus	$Zx = \sum A_i y_i = 2 \times bf \times tf \times \frac{H - tf}{2} + tw \times \frac{hw^2}{4}$
Elastic Section modulus	$Sx = \frac{Ix}{H/2}$
Radius of Gyration	$r_X = \sqrt{I_X / A}$
Center of gravity	$x_{C} = \frac{\sum F_{i} x_{i}}{\sum F_{i}}; y_{C} = \frac{\sum F_{i} y_{i}}{\sum F_{i}}$

#### **Compact and Non-compact Requirements**

For flexure, sections are classified as compact, non-compact or slender-element sections.

Width-to-Thickness Ratios of I-shaped built-up sections are shown as below:



<sup>8</sup>AISC 360-10, Table B4.1 Case 15.

<sup>9</sup>AISC 360-10, Table B4.1 Case 16.

## **Checking Dimension Requirements**

Concrete slabs on formed steel deck connected to steel beams need to be satisfied the following requirements<sup>10</sup>: (1) The nominal rib height (hr) shall not be greater than 3 in. (75 mm).

(2) The average width of concrete rib or haunch (wr) shall be not less than 2 in. (50 mm), but shall not be taken in calculations as more than the minimum clear width near the top of the steel deck.

(3) The slab thickness above the steel deck shall be not less than 2 in. (50 mm).

(4) a. The concrete slab shall be connected to the steel beam with welded steel headed stud anchors, not larger than 19 mm in diameter.

Steel headed stud anchors, after installation, shall be extended not less than 38mm above the top of the steel deck and there shall be at least 13 mm of specified concrete cover above the top of the steel headed stud anchors.

b. The diameter of a steel headed stud anchor shall not be greater than 2.5 times the thickness of the base metal to which it is welded, unless it is welded to a flange directly over a web.

c. Steel anchors shall have at least 1 in. (25 mm) of lateral concrete cover in the direction perpendicular to the shear force, except for anchors installed in the ribs of formed steel decks. The minimum distance from the center of an anchor to a free edge in the direction of the shear force shall be 8 in. (203 mm) if normal weight concrete is used and 10 in. (250 mm) if lightweight concrete is used. (5) The Stud spacing

a. Steel deck shall be anchored to all supporting members at spacing not to exceed 460 mm.

b. The minimum center-to-center spacing of steel headed stud anchors shall be six diameters along the longitudinal axis of the supporting composite beam and four diameters transverse to the longitudinal axis of the supporting composite beam, except that within the ribs of formed steel decks oriented perpendicular to the steel beam the minimum center-to-center spacing shall be four diameters in any direction.

c. The maximum center-to-center spacing of steel anchors shall not exceed eight times the total slab thickness or 36 in. (900 mm).

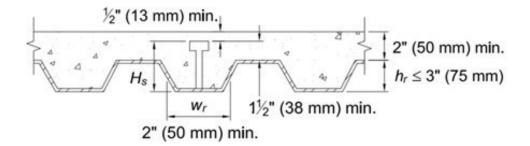


Figure 6 2 Dimension Requirements of Steel headed stud anchors

## 6.2.2. Beam Bending Capacities for Fully Composite Beam

#### Properties of Transformed Section

To calculate the moment capacity with an elastic stress distribution, first need to calculate the location of the elastic neutral axis (ENA) and the transformed section moment of inertia as below:

Effective Width	$b_{eff} = \frac{\min(L/4;B)}{n} $ (AISC)	360-10 I3.1a)
Effective Thickness of Slab	$t_o = t_s - b_r / 2$ where: $n = E_s / E_c$ : Modular ratio $E_c$ : Modulus of elasticity of steel $E_c$ : Modulus of elasticity of concrete	
Height of Composite Section	$d_{ir} = H + b_s$	
Distance from <b>Elastic Neutral</b> <b>Axis(ENA)</b> to steel bottom of composite member	$y_b = \frac{\sum A_i y_i}{\sum A_i}$	
The transformed section moment of inertia, about its ENA	$I_{rr}$	
Section modulus of bottom edge	$y_b \ge H : S_{tr} = I_X / (0.5b)$ $y_b \ge H : S_{tr} = I_X / (0.5b)$	
Section modulus of top edge	$S_t = \frac{I_{tr}}{d_{tr} - y_b}$	

#### **Check Bending Capacities**

Moment Capacity for Positive Bending

$$f_b = \frac{M_{\max}}{S_{tr}} \le \frac{F_y}{\Omega_b}; \Omega_b = 1.67$$

Moment Capacity for Negative Bending

$$f_c = \frac{M_{\text{max}}}{nS_t} \le 0.45 f_c$$

# 6.2.3. Beam Shear Capacity

## Shear Capacity

The nominal shear strength of unstiffened or stiffened webs of all other doubly symmetric shapes and singly symmetric shapes and channels, is determined as follows:

	$V_n = 0.6F_y \mathcal{A}_w C_v$		(AISC 14.2, G2-1)
where:			
$C_{v}$	The web shear coefficient, is given by:		(AISC G2.1b, G2-3)
	Conditions	$C_{\nu}$	(AISC G2.1b, G2-4)
			(AISC G2.1b, G2-5)
	$hw / tw \le 1.10 \sqrt{\frac{k_v E}{F_{\gamma}}}$	1.00	
	kE $kE$	$1.10\sqrt{kE/Fy}$	
	$1.10\sqrt{\frac{k_vE}{Fy}} < bw / tw \le 1.37\sqrt{\frac{k_vE}{Fy}}$	$\frac{h}{h} \frac{h}{tw}$	
	$k_{\mu}E$	1.51k,E	
	$hw / tw > 1.37 \sqrt{\frac{k_r E}{Fy}}$	$\frac{1.51k_{v}E}{\left(b/tw\right)^{2}Fy}$	
$k_{v}$	The web plate buckling coefficient, is give	n by:	(AISC G2.1b
	For unstiffened webs with $hw / tw < 260$ :	$k_{v} = 5$	
	For stiffened webs:		
	$+ k_v = 5 + \frac{5}{(a/b)^2}$		
	+ $k_{p} = 5$ for $a / b > 3.0$ or $a / b > \left(\frac{260}{b / tw}\right)$	$\Big)^2$	
$A_{w}$	dt : area of web		
b	for built-up welded sections, the clear dist	ance between flanges	

#### Checking the Beam Shear

The beam shear at the ends of the beam is checked using the following equation.

 $V_{\max} \leq \frac{V_n}{\Omega_p} \qquad (AISC 360-10 G2-1)$ where: (AISC G2.1b, 1)  $\Omega_p = 1.67 \quad \text{Safety factor for shear}$   $V_{\max} \quad \text{The required shear strength}$   $V_n \quad \text{Shear capacity}$ 

## 6.2.4. Steel Anchors

Shear connector capacities are defined for both shear studs and channel shear connectors. Next the equations used for determining the number of shear connectors on the beam are provided.

The nominal shear force between the steel beam and the concrete slab transferred by steel anchors for Positive Flexural Strength:

$$V' = 0.5 \min \begin{cases} 0.85 f_c' A_c \\ F_y A_s \end{cases}$$
(AISC 360-10 G2.1b)

where:

 $A_c \quad b_{eff} \times t_o \times n$ : area of concrete slab within effective width

 $A_s$  area of steel cross section

## Steel Headed Stud Anchor

The nominal shear strength of one steel headed stud anchor embedded in a solid concrete slab

	$Q_n = \min \begin{cases} 0.5 \mathcal{A}_{sa} \sqrt{f_c' E_c} \\ R_g R_p \mathcal{A}_{sa} F_u \end{cases}$	(AI.	SC 360-1	0 I8.2a)
where:				
$A_{sa}$	cross-sectional area of steel headed stud anchor, mm <sup>2</sup> .			
$E_{c}$	$0.043 \mathrm{w}_{c}^{1.5} \sqrt{f'_{c}} (MPa)$ : Modulus of elasticity of concrete			
$\mathbf{W}_{c}$	Weight of concrete per unit volume, assumed 2300 kg/m <sup>3</sup>			
$f_u$	specified compressive strength of concrete, MPa			
$f_u$	specified minimum tensile strength of a steel headed stud anchor			
$R_g; R_p$	The table below presents values for $R_g$ and $R_g$ for several cases. Capacities for steel headed stud anchors can be found in the Manu	ual.		
	Condition		R <sub>e</sub>	R <sub>p</sub>
	No decking		1.0	0.75
	Decking oriented parallel to the steel shape $w_r / b_r < 1.$	5	1.0	0.75
	$w_r / b_r < 1.$	5	0.85 <sup>(a)</sup>	0.75
	Decking oriented perpendicular to the steel shape	1	1.0	0.6 <sup>(b)</sup>
	Number of steel headed stud anchors occupying	2	0.85	0.6 <sup>(b)</sup>
	the same decking rib	3 or more	0.7	$0.6^{(b)}$
	where:			
	$w_r$ : nominal rib height			
	$w_r$ : average width of concrete rib or haunch (a) for a single steel headed stud anchor			
	(b) this value may be increased to 0.75 when $e_{mid-bt} \ge 2in = 51mm$	2		
ℓ <sub>mid−ht</sub>	distance from the edge of steel headed stud anchor shank to the s mid-height of the deck rib, and in the load bearing direction of the other words, in the direction of maximum moment for a simply supported beam).	steel deck web, m		

#### **Steel Channel Anchors**

The nominal shear strength of one hot-rolled channel anchor embedded in a solid concrete slab shall be determined as follows:

$$Q_n = 0.3(tf + 0.5tw)l_a\sqrt{f_c E_c}$$

where:

 $l_a$  length of channel anchor

tf thickness of flange of channel anchor

tw thickness of channel anchor web

#### **Check Shear Connector Capacity**

The number of shear connectors is given as follows:

$$\sum Q_n = Q_n N_r \frac{L}{2s} \ge V$$

(AISC 360-10 I3.1c)

(AISC 360-10 I8.2a)

where:

- $N_r$  Number of Stud each position
  - s Shear Connectors Spacing
- *L* Length of joist beam

#### 6.2.5. Beam Deflection and Camber

#### Checking Stress when made camber

Camber	$\Delta_{DL} = \frac{5}{384} \times \frac{q_{DL}L^4}{EI_s}$
Checking Stress	$\frac{M_{DL}}{S_s} < \frac{Fy}{\Omega_b}$

#### **Checking Deflection of Joist Beam**

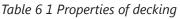
Effective moment of inertia	$I_{eff} = 0.75 I_{tr}$	(AISC 360-10 I3.2)
Deflection by Floor Load	$\Delta_{FL} = \frac{5}{384} \times \frac{q_{FL}L^4}{EI_{eff}} \leq \left[\Delta\right] = \frac{L}{360}$	
Deflection by Dead load & Floor load	$\Delta = \frac{5}{384} \times \frac{qL^4}{EI_{eff}} \leq \left[\Delta\right] = \frac{L}{240}$	

## 6.2.6. Check Decking

### **Decking Properties**

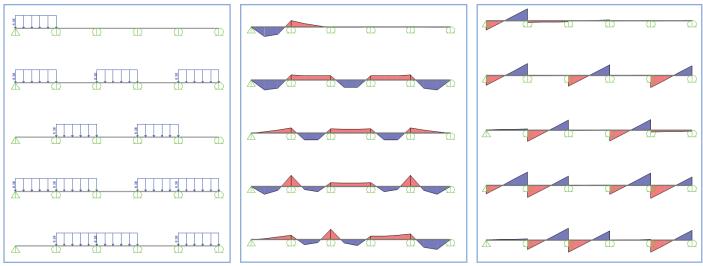
It is necessary to determine decking's material each project information. Normally, yielding strength of decking is 235 MPa.

[width ]-[rid depth ]-[rib spacing ]-[ average width rib ]		Thickness	
[width ]-[nd deput ]-[nd spacing ]-[ average width nd ]	0.75	0.95	1.15
[1000]-[50]-[334]-[166]	455282	574099	691840
	17842	22441	26975
[0870]-[75]-[ 287]-[ 148]	922823	1168944	1415089
	23643	29917	36178



#### Checking the decking

Loading impact on decking:  $q_c = B_r \times (q_{CL} \times n + G_{concrete} \times t_o)$ 



Load assign decking (5 cases)

Moment 3-3 Diagram

Figure 6 3 Schematics of Decking

Shear 2-2 Diagram

Checking capacity:

$$\frac{M_{\max}}{S_x} \le \frac{Fy}{\Omega_b}; \Omega_b = 1.67$$

Checking deflection :

$$f_{\max} \leq \left[ f \right] = \frac{B}{180}$$

# 6.2.7. Design Example 8a - Composite Slab

#### Input

Loading	Floor load (FL)	5.0	$kN/m^2$
	Brick wall 1m height 0.2m thickness Wall load $WL = 18 \times 1 \times 0.2$	3.6	kN/m
Slab thickness	Included: Surface sika, thickness 10 <i>mm</i> , density 22 <i>kN/mm</i> <sup>3</sup> Mortar, thickness 20 <i>mm</i> , density 18 <i>kN/mm</i> <sup>3</sup>	120.0	mm
Joist beam	Section Length Spacing	I — 508x148x5x6 8000.0 1500.0	mm mm
Decking	Type / Thickness (mm) Construction method Yielding strength, Fy Safety factor, n	[B1000 – hr50]/[0.7 Don't Use Support ( 235 1.5	
Shear Connector	Diameter/Height/Qty. each position/spacing	[16]/[90]/[02]/[334]	
Concrete strength: B20	Compression strength, fc' Elastic modulus, Ec	16 18800	MPa MPa
Steel	<b>ASTM specifications: A572 Grade 50</b> Yield strength, Fy Ultimate strength, Fu Modulus of elasticity, E	345 450 200000	MPa MPa MPa

# 6.2.7.1 Design Composite Joist Beam using Calculation Sheet

# Slab Loading

Load Case	Include	Thickness (mm)	$G (kN/m^3)$	Design Load
Dead Load (DL)	Mass of joist Wall load Surface sika	10	22	0.334 kN/m 3.60 kN/m
	Concrete slab Mortar	$t_o = 95$ 20	25 18	$2.955 \text{ kN/m}^2$

Dead load applied on each joist:  $q_{DL} = 0.334 + 3.6 + 1.5 \times 2.955 = 8.37(kN / m)$ Hence,

Moment by Dead Load	$M_{DL} = qL^2 / 8 = 8.37 \times 8^2 / 8$	66.99 kN/m
Moment by Dead Load & Floor Load	$M_{DL+FL} = qL^2 / 8 = (8.37 + 1.5 \times 5) \times 8^2 / 8$	126.93 kN/m
Maximum Shear by (DL+FL)	$V_{DL+FL} = qL/2 = (8.37 + 1.5 \times 5) \times 8/2$	63.48 kN

## **Section Properties**

Cross-section area	$A = 2 \times 148 \times 6 + 496 \times 5$	4256	mm <sup>2</sup>
Mass	A×78.5	0.334	kN/m
Moment of Inertia	$I_X = \frac{5 \times 496^3}{12} + 148 \times \frac{508^3 - 496^3}{12}$	1.63E8	mm4
Plastic Section Modulus	$Z_{\mathcal{X}} = 2 \times \left( 148 \times 6 \times \frac{502}{2} + 5 \times \frac{496}{2} \times \frac{496}{4} \right)$	753296	mm <sup>3</sup>
Elastic Section modulus	$Sx = 2 \times 1.63E + 08 / 508$	6.41E5	mm <sup>3</sup>
Radius of Gyration	$rx = \sqrt{1.63E + 08/4256}$	195.5	mm

Classification of Filled Composite Sections for Local Buckling

Flanges:	$\lambda p = 0.38\sqrt{200000 / 345}$	9.15
Noncompact	$\lambda r = 0.95 \sqrt{\frac{0.45 \times 200000}{0.75 \times 345}}$	18.65
	$k_C : 0.35 \le \left[ k_C = \frac{4}{\sqrt{396/5}} = 0.45 \right] \le 0.76$	0.45
	$S_{xt} / S_{xx} = 1.00 \Longrightarrow F_L = 0.75 Fy$	
	$0.5B/t = 0.5 \times 148/6 = 12.33 \Longrightarrow \lambda_p < (0.5B/t) < \lambda_r$	
Web:	$\lambda p = 3.76\sqrt{E/Fy} = 3.76\sqrt{200000/345}$	90.53
Noncompact	$\lambda r = 5.70\sqrt{200000 / 345}$	137.23
	$h / tw = 496 / 5 = 99.2 \Longrightarrow \lambda_p < (h / tw) < \lambda_r$	
Class: Noncompact		

# **Checking Dimension Requirements**

<ol> <li>(1) The nominal rib height, hr</li> <li>(2) The average width of rib, wr</li> <li>(3) The slab thickness</li> <li>(4) Steel headed stud anchors</li> </ol>	50 mm 166 mm 120 mm	$\leq 75 mm$ > 50 mm $\geq hr + 50 = 100 mm$	OK OK OK
Diameter Height Stud spacing	16 <i>mm</i> 90 <i>mm</i>	$\leq 19 mm$ $\leq 2.5 \times tf 1 = 2.5 \times 6 = 15$ $\geq hr + 38 = 88mm$ $\geq 5 \times 16 = 80mm$ < 120 - 13 = 107mm  (slab cover above) $\leq 460 mm$	OK
	334 <i>mm</i>	> $Min. = 6 \times 16 = 96mm$ < $Max. = max(8 \times 120; 900) = 960mm$	

# Determining Composite Properties for Plastic Design

Effective Width	$b_{eff} = \min(8000 / 4; 1500) / 10.64$	141	mm
Modular ratio	n = 200000 / 18800	10.64	
Effective Thickness of Slab	$t_o = 120 - 50 / 2$	95	mm
Height of Composite Section	$d_{ir} = 120 + 508 = 628mm$		
Distance from ENA to steel bottom of composite member	$y_b = \frac{141 \times 95 \times (508 + 120 - 95 / 2) + 4256 \times 508 / 2}{141 \times 95 + 4256}$	501.77	mm
The transformed section moment	$I_{tr} = I_s + I_c = 4.24 E8 + 9.31 E7$	5.17E8	mm4
of inertia, about its ENA	$I_{s} = Ix + A(y_{c} - y_{b})^{2} = 1.63E8 + 4256(254 - 501.77)^{2}$	4.24E8	mm4
	$I_c = 141 \times 95^3 / 12 + 141 \times 95 \times (628 - 95 / 2 - 501.77)^2$	9.31E7	mm4
Section modulus of bottom edge	$y_b = 501.77 < H = 508 : S_{tr} = \frac{5.17E8}{501.77}$	1.03E6	mm <sup>3</sup>
Section modulus of top edge	$S_{t} = \frac{5.17E8}{628 - 501.77}$	4.1E6	mm <sup>3</sup>

## **Check Bending Capacities**

Positive Bending	$f_b = \frac{126.93}{1.03E6 / 1000^3}$	1.23E5	$kN/m^2$
$f_b < \frac{F_y}{\Omega_b}$ : OK	$\frac{F_{y}}{\Omega_{b}} = \frac{345E3}{1.67}$	2.10E5	kN/m <sup>2</sup>
Negative Bending $f_c < 0.45 f'_c$ : OK	$f_c = \frac{126.93}{10.64 \times 4.1E6 / 1000^3}$	2.9E3	kN/m²
	$0.45 f_c' = 0.45 \times 16E3$	7.2E3	kN/m <sup>2</sup>

# **Check Shear Capacities**

The web shear coefficient	$\frac{h_w}{t_w} = \frac{496}{5} = 99.2 > 1.37 \sqrt{\frac{5 \times 200000}{345}} = 73.76$	
	$k_{p} = 5$ for $a / b = 5.4 > 3.0$ where: $a / b = 2500 / 496 = 5.4 > 3.0$	
	$\Rightarrow C_{v} = \frac{1.51 \times 5 \times 200000}{\left(496 / 5\right)^{2} \times 345}$	0.44
The available shear strength	$\frac{V_n}{\Omega_v} = \frac{0.6 \times 345E3 \times 0.496 \times 0.005 \times 0.44}{1.67}$	135.26 kN
Check Shear Capacity	$\frac{V_n}{\Omega_p} > V_{\max} = 63.48 \text{kN} : \text{OK}$	

# **Check Shear Connector Capacity**

The required Shear for Shear Connector	$V' = 0.5 \min \begin{cases} 0.85 \times 16E3 \times 0.141 \times 0.095 \times 10.64 = 1938\\ 345E3 \times 4256 / 1000^2 = 1468 \end{cases}$	734.16 kN
The nominal shear strength of each stud	$Q_n = \min \begin{cases} 0.5 \times \frac{201.06}{1000^2} \sqrt{16 \times 18800} \times 1000 = 55.13 \text{ kN} \\ 1.0 \times 0.75 \times \frac{201.06}{1000^2} \times 450 \times 1000 = 67.86 \text{ kN} \end{cases}$	55.13 kN
The number of shear connectors	$\sum Q_n = 55.13 \times 2 \times \frac{8000}{2 \times 334} = 1321 \text{ kN} \ge V' = 734.16 \text{ kN}$	: OK
where:	$A_{sa} = \pi \times 16^{2} / 4 = 201.06 (mm^{2})$ $f_{c}' = 16MPa$ $E_{c} = 0.043 \times 2300^{1.5} \sqrt{16} = 18800MPa$ $w_{r} / b_{r} = 166 / 50 = 3.32 \ge 1.5 \Longrightarrow \begin{cases} R_{g} = 1.00 \\ R_{p} = 0.75 \end{cases}$ $f_{u} = 450MPa$	

#### **Beam Deflection and Camber**

Camber	$\Delta_{DL} = \frac{5}{384} \times \frac{8.37 \times 8^4}{200000 \times \frac{1.63E8}{1000^4}}$	13.7 mm
Checking Stress	$\frac{66.99}{6.41E5/1000^3} = 1.05E5 < \frac{345E3}{1.67} = 2.07E6(kN/m^2)$	: OK

# Checking Deflection of Joist Beam

Effective moment of inertia	$I_{gf} = 0.75 \times 5.17 \text{E8}$ 3.88E8 mm <sup>4</sup>
Deflection by Floor Load	$\Delta_{FL} = \frac{5}{384} \times \frac{5 \times 1.5 \times 8^4}{200000 \times 3.88 \text{E} - 4} = 5.26 \text{mm} < [\Delta] = \frac{8000}{360} = 22.2 \text{mm}$
Deflection by Dead load & Floor load	$\Delta = \frac{5}{384} \times \frac{(5 \times 1.5 + 8.37) \times 8^4}{EI_{\text{eff}}} 10.91 \text{ mm} \le \left[\Delta\right] = \frac{8000}{240} 33.3 \text{ mm}$

# Design Data

Components	Types	Ratio	Conclusion
Section	Preliminary dimension	0.842	OK
	Top flange	0.405	
	Bottom flange	0.596	
	Shear strength	0.464	
Studs	Spacing	0.556	OK
Camber	Camber	14	mm
	Capacity	0.506	OK
Deflection	Live load	0.232	OK
	Dead load & Live load	0.327	

# 6.2.7.2 Design Composite Mezzanine Decking using Calculation Sheet

Decking [1000]-[50]-[334]-[166] thickness 0.75mm:  $S_x = 1.78E4(mm^3)$ ;  $S_x = 1.78E4(mm^3)$ 

# Checking the decking

Loading impact on decking	$q_{c} = B_{r} \times (q_{CL} \times n + G_{concrete} \times t_{o})$ $q_{c} = 1 \times (4 \times 1.5 + 25 \times 0.095)$	8.375 kN/m
Maximum Moment	$M_{\rm max} = 1.86(kNm)$	
Checking capacity	$\frac{1.86}{1.78E - 5} 1.02E5 \le \frac{Fy}{\Omega_b} = \frac{235000}{1.67} = 1.4E5(kN / m^2)$	: OK
Checking deflection	$f_{\text{max}} = 4.52  mm \le \left[ f \right] = \frac{B}{180} = \frac{1500}{180} = 8.33  mm$	: OK

# 6.3. Checkered Plate Slab

NOTE

# 6.3.1. General

Allowable thicknesses of Checkered Plate are 4mm, 5mm, 6mm or 8mm (included ribbed plate). Checkered Plate Slab can be designed with or without Flat Bar.



Incase floor load is not less than 4  $kN/m^2$ ; designers should use Checkered Plate Slab with Flat Bar under.

If floor load is less than 4 kN/m<sup>2</sup>, Checkered Plate Slab without Flat Bar can be used with the joists having maximum spacing  $1.2\div1.5m$ .

## 6.3.2. Detailing Requirement

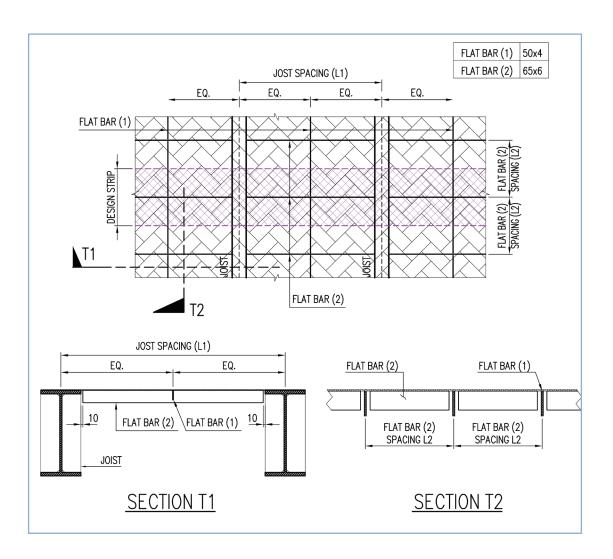


Figure 6 4 Checkered Plate with Flat Bar

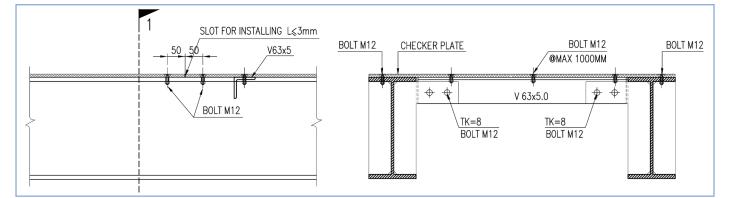


Figure 6 5 Bolt Connection of Checkered Plate and Joists

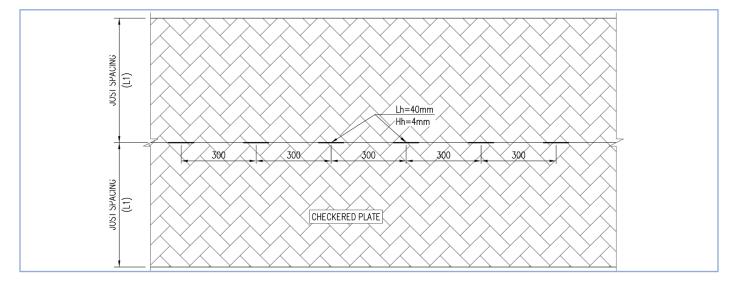


Figure 6 6 Weld connection at the intersection of Checkered Plates

## 6.3.3. Checkered Plate with Flat Bar

#### **Design Strip Properties**

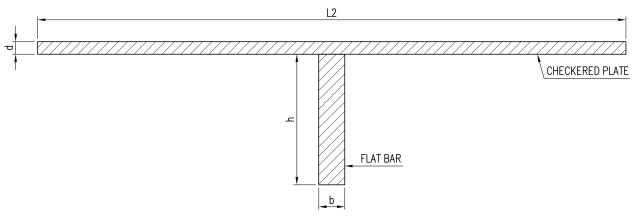


Figure 6 7 Design Strip of Checkered Plate with Flat Bar

Section area	$\mathcal{A} = d \times L_2 + b \times b$
Distance from Neutral Axis to bottom	$y_{base} = \sum A_i y_i / \sum A_i$
Inertia moment of strip	$Ix = \sum \left( \frac{bb^3}{12} \right) + \sum Ay_i^2$
Cylindrical rigidity of plate	$D = E h_e^3 / 12 \times L_2$
where,	E, Young's Modulus
	$b_e$ , elastic thickness v = 0.3, Poisson's Ratio for steel

#### **Check Deflection**

Checkered Plate is considered as a uniform loaded long rectangular plate with longitudinal edges which are free to rotate but cannot move toward each other during bending. The design strip cut out this plate is in the condition of a uniformly loaded bar submitted to the action of an tensile force H, as show in Figure 6-8. The magnitude of H is such as to prevent the ends of bar from moving along the x-axis.

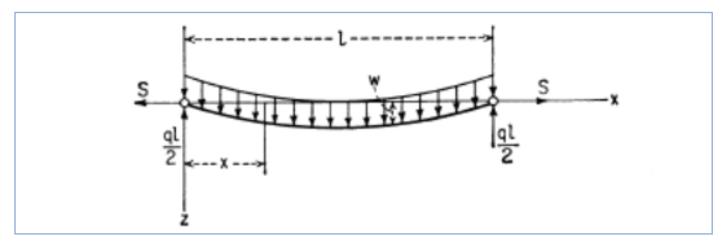


Figure 6 8 Uniformly Loaded Rectangular Plates with Simply Supported Edges

The maximum deflection	$\Delta = \Delta_{a} \frac{1}{1+\alpha} \leq \left[\Delta\right] = \frac{L}{180}$
The maximum deflection caused by uniform load $q$	$\Delta_{o} = \frac{5}{384} \frac{ql^{4}}{E_{1}I_{x}}; \text{ where } E_{1} = \frac{E}{1 - \upsilon^{2}}$
The factor for the effect of the tensile reactions at the ends of the strip	$\frac{1}{1+\alpha}$
The ratio of the tensile force $\alpha$ to the Euler critical load of the elemental strip, $\alpha$ , is determined from the equation:	$\alpha \left(1+\alpha\right)^2 = 3 \left(\frac{\Delta_{e}}{t}\right)^2$

#### **Check Bending**

The tensile force subject on joint of strip	$H = \frac{\pi^2 D\alpha}{l^2}$
The maximum bending moment	$M_{\rm max} = \frac{ql^2}{8} - H\Delta$
Check Stress	$\sigma = \frac{H}{A} + \frac{M_{\text{max}}}{S_x} \le \frac{F_y}{\Omega}; \Omega = 1.67$

#### **Check Weld Connection**

Tensile force subject to weld	$T \leq \frac{R_n}{\Omega} = \frac{F_{nv} \times A_{nv}}{\Omega}; \Omega = 2.00$
Allowable bearing capacity of weld	$T \leq \frac{R_n}{\Omega} = \frac{F_{nv} \times A_{nv}}{\Omega}; \Omega = 2.00$
Nominal stress of the weld	$F_{m\nu} = 0.6F_{EXX} \left( 1 + 0.5\sin^{1.5}\theta \right)$
Area of the weld	$\mathcal{A}_{w} = \frac{\sqrt{2}}{2} \times b_{f}$

#### 6.3.4. Checkered Plate without Flat Bar

Formulae show ratio of design length and slab thickness(According to formulae of A.L.Teloian):

$$\frac{l}{t} = \frac{4n_o}{15} \left( 1 + 75 \frac{E_1}{n_o^4 q} \right) \Longrightarrow l = t \times \frac{4n_o}{15} \left( 1 + 75 \frac{E_1}{n_o^4 q} \right)$$

where  $\frac{1}{n_o} = \left[\frac{f}{L}\right] = \frac{1}{180}$ : Ratio of allowable deflection of slab. Hence, available joist spacing (distance edge flange to edge flange of joists)  $\left[l_o\right] \le l$ .

# 6.3.5. Design Example 8b - Checkered Plate with Flat Bar Input

Spacing of Joist	$L_1$	1500	mm
Spacing of Flat bar	$L_2$	600	mm
Checkered Plate Thickness	d	4	mm
Flat bar	Width, b	4	mm
	Height, h	50	mm
Material	ASTM specifications for Steel	JIS 3101 /SS400	
	Yield strength, $F_{\mu}$	235	MPa
	Ultimate strength, $F_{\mu}$	400	MPa
	Modulus of elasticity of steel, E	200000	MPa
Welding	Welding wire	E60xx	
	Height of fillet weld	3	mm
Floor load	FL	5	$kN/m^2$

# Properties Of Design Strip

Section area	$A = 4 \times 600 + 4 \times 50 = 2600  mm^2$	2	600	mm <sup>2</sup>
Distance from Neutral Axis to bottom	$y_{base} = \frac{600 \times 4 \times 52 + 4 \times 50 \times 25}{2600}$	4	9.9323	mm
Inertia moment of strip	$I = \frac{600 \times 4^{3}}{12} + 600 \times 4 \times (52 - y_{base})^{2}$ $-\frac{4 \times 50^{3}}{12} + 4 \times 50 \times (25 - y_{base})^{2}$	1	79451	mm <sup>4</sup>
Section modulus	$S_x = \frac{I_x}{0.5 y_x} = \frac{179451}{0.5 \times 49.9323}$	7	187.8	mm <sup>3</sup>
Cylindrical rigidity of plate	$D = \frac{2E8 \times 0.004^3}{12} \times 0.6$	0	.64	kNm

## **Check Deflection**

Modulus $E_1$	$E_1 = \frac{200000}{1 - 0.3^2} = 2.2E5(MPa)$	2.2E5	MPa
Dead load	$q_{DL} = \frac{2600}{1000^2} \times 78.5 \times 1.05$	0.214	$kN/m^2$
Uniform load q	$q = q_{DL} + q_{FL} = (0.214 + 5) \times 0.6$	3.13	$kN/m^2$
The maximum deflection caused by uniform load $q$	$\Delta_{o} = \frac{5}{384} \frac{3.13 \times 1.5^{4}}{2.2E8 \times 1.79E - 7}$	5.24E-3	m
The ratio $\alpha$	$\alpha (1+\alpha)^2 = 3 \left(\frac{5.24E-3}{4E-3}\right)^2 = 5.15 \Longrightarrow \begin{cases} \alpha = 1.535\\ \alpha = -0.77 \end{cases}$ Choose $\alpha = 1.535$		
The maximum deflection	$\Delta = (5.4E - 3) \times \frac{1}{1 + 1.535}$	2.1E-3	m
Check deflection	$\Delta \le \left[\Delta\right] = \frac{1.5}{180} = 0.0833m$	ОК	

# **Check Bending**

The tensile force	$H = \frac{3.14^2 \times 0.64 \times 1.535}{1.5^2}$	4.3	kN
The maximum bending moment	$M_{\rm max} = \frac{3.13 \times 1.5^2}{8} - 4.4 \times 2.1E - 3$	0.87	kNm
Tensile stress	$\sigma = \frac{4.4}{2.6E - 3} + \frac{0.87}{7.19E - 6}$	1.23E8	$kN/m^2$
Check Stress	$\sigma \le \frac{F_y}{\Omega} = \frac{245E3}{1.67} = 1.47E6 (kN / m^2)$	OK	

#### **Check Weld Connection**

Factor $\beta$	$\beta = 1.2 - \frac{0.002}{3  /  600} = 0.8 < 1$	0.8
Tensile force subject to weld	$T = \frac{H}{n\beta b_{\text{eff}}} = \frac{4.3}{0.6 \times 0.8}$	8.95 kN/m
Bearing capacity of weld	$\frac{R_{\pi}}{\Omega} = \frac{3.73E5 \times 2.12E - 3}{2}$	394.96 kN/m
	$A_w = \sqrt{2} / 2 \times 0.003 \times 1$	2.12E-3 m <sup>2</sup>
	$F_{nw} = 0.6 \times 4.14 E5 \times \left(1 + 0.5 \sin^{1.5} 90^{\circ}\right)$	3.73E5 kN/m <sup>2</sup>
	$F_{EXX} = 60 (ksi)$	4.14E5 kN/m <sup>2</sup>
Check Welding	$T = 8.95 < \frac{R_n}{\Omega} = 394.969 (kN / m^2)$	OK

# 6.3.6. Design Example 8c - Checkered Plate without Flat Bar

Input: as same as Design Example 8b, Checkered Plate without Flat Bar.

Specify Thickness: 4.00 mm.

Dead load	$q_{DL} = \frac{4}{1000} \times 78.5 \times 1.05$	0.33 kN/m <sup>2</sup>
Uniform load q	$q = q_{DL} + q_{FL} = 0.33 + 5.00$	5.33 kN/m <sup>2</sup>
Joist spacing	$[l_{o}] \le 0.004 \times \frac{4 \times 180}{15} \left(1 + 75 \frac{2.2E8}{180^{4} \times 5.33}\right) = 0.758m$	0.8 m

Then, check Deflection and Bending as same as Example 8a.

#### Conclusion

The type of checkered plate (with or without flat bar) depends on the load and thickness (if required).

# 6.4. Grating Slab

# 6.4.1. General

Open steel rectangular pattern floor is constructed from mild steel and is constructed by a forge-welding process in which the load bearing and transverse bars are heated and joined under pressure.

Gratings is suitable for most floor walkways, gantries, platforms, etc. and could even be used upright as fencing.

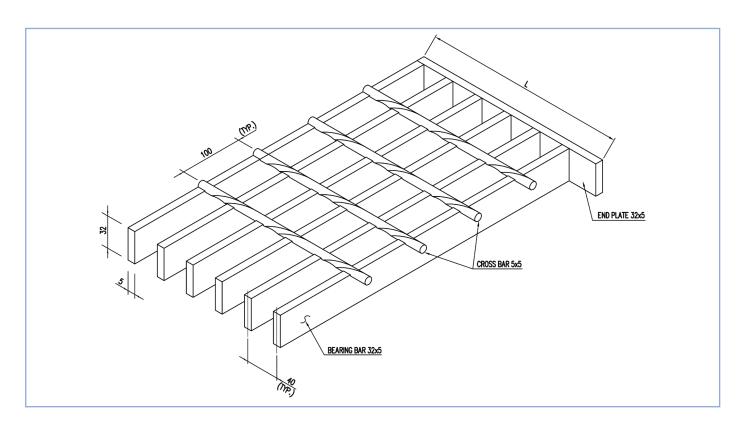


Figure 6 9 Standard Grating

Standard Range – Readily available with either plain or serrated load bearings bars (5mm thick) at 40 pitch and twisted transverse bars at 50 or 100 centers.

## 6.4.2. Design Example 8c - Grating Slab

#### Input

Floor load	$q_{FL}$	3.0	kN
Span/Trench width	b	1300	mm
Bearing bar x pitch		(32 x 5) @max 40	mm
Allowable stress	$F_{\mathcal{Y}}$	235	MPa
Modulus of elasticity of steel	E	200000	MPa
Allowable deflection	$\left[\Delta\right] = L / 240$	5.41	mm

Consider 1m width of grating slab.

Moment of inertia	$I_{\mathcal{X}} = \frac{5 \times 32^3}{12}$	13653	$\mathrm{mm}^4$
Elastic section modulus	$S_X = \frac{13653}{32/2}$	853	mm <sup>3</sup>
Load per each Bearing bar	$P = \frac{q_{FL}}{n \times b} = \frac{3}{1000 / 40 \times 1.3 \times 1.0}$	0.0923	kN/m
Maximum Moment	$M_{\rm max} = \frac{ql^2}{8} = \frac{0.0923 \times 1.3^2}{8}$	0.0195	kNm
Maximum Stress	$\sigma = \frac{M_{\text{max}}}{S_x} = \frac{0.0195}{8.53E - 7}$	1.83E5	$kN/m^2$
Check stress	$\sigma \le \frac{F_y}{\Omega} = \frac{245E3}{1.67} = 1.47E6 (kN / m^2)$	OK	
Check deflection	$\Delta = \frac{5}{384} \frac{ql^4}{EI} = \frac{5}{384} \frac{0.0923 \times 1.3^4}{2E8 \times 1.36E - 8} = 0.00126m < [\Delta]$	OK	

DESIGN GUIDELINES

# CHAPTER 7. ACCESSORIES 7.1. Purlins and Girts Design

# 7.1.1. General

Purlins and girts are the immediate supporting member for roof and wall sheeting respectively. They act principally as beams, but also perform as struts and as compression braces in restraining rafters and columns laterally against buckling.

Purlins and girts are almost universally zed (Z), channel (C), or double channel section members.

**C-sections** have equal flanges and may be used in single spans and un-lapped continuous spans in multi-bay buildings. Their freestanding stable shape allows easy handling and storage and is easily adapted for use in small and medium sized buildings as structural framework.

**Z-sections** feature one broad and one narrow flange allowing the two sections to fit together snugly, making them suitable for lapping. Z sections of the same depth and different thickness' can be lapped in any combination. Purlins and Girts that are lapped form a structurally continuous line along the length of the building, a factor that contributes significantly to the reduction in building costs.

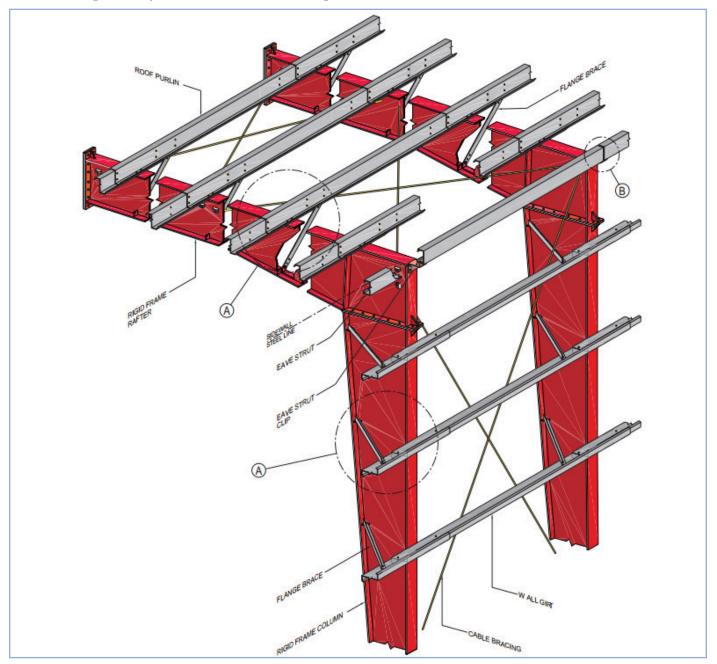


Figure 7 1 Typical Purlins and Girts Details

## 7.1.2. Design procedure

Purlins and girts can be calculated as simply supported spans, lapped continuous spans or increased thickness end spans in lapped continuous systems.

## 7.1.2.1 Purlin/Girt spacing

Purlin spacing must be chosen to suit the type of roof sheeting and ceiling system if any. The use of translucent fiberglass roof sheeting will also restrict the purlin spacing. Some suspended ceiling systems require a maximum purlin spacing of 1200mm (DONGIL project). Purlin deflections must also be controlled.

Roof purlins are to be arranged according to the following guide lines as applicable:

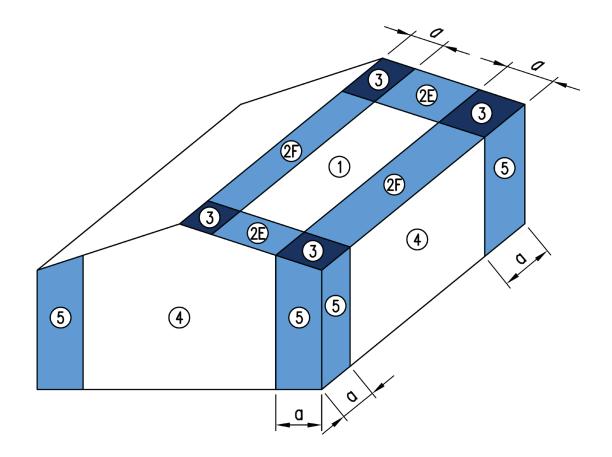
- 700 mm between first roof purlin and the eave strut
- Intermediate spacing not exceeding 1600 mm.

## 7.1.2.2 Loading

The maximum spans are determined not only from wind load considerations, but also from live load requirements. Refer Chapter 2 for more detail.

Collateral load applied on purlin such as Fire Sprinkler, MEP System, Plaster Ceiling System, etc. should be clearly specified as concentrated or uniformed load to identify the most dangerous case.

Components	Load Case	Load Combination
Purlins	Gravity Loads: $DL$ , $AL$ , $WL_1$ Windload: $WL_1$ , $WL_{2E}$ , $WL_{2E}$ , $WL_3$	Case 1: $DL + AL + LLr$ Case 2: $\begin{cases} 0.6DL + 0.6AL + 0.6WL_{1} \\ 0.6DL + 0.6AL + 0.6WL_{2E} \end{cases}$ Case 3: $\begin{cases} 0.6DL + 0.6AL + 0.6WL_{2F} \\ 0.6DL + 0.6AL + 0.6WL_{3} \end{cases}$
Eave struts	Gravity Loads: <i>LLr</i> , <i>LLr</i> Rainload: $WI$ Windload: $WL_1$ , $WL_{2E}$ , $WL_{2E}$ , $WL_3$	Case 1: $DL + LLr$ Case 2: $DL + P_r$ Case 2: $0.6DL + 0.6WL_3$
Girts	Gravity Loads: $AL$ , $AL$ , $WL_4$ Windload: $WL_4$ , $WL_5$	Case 1: $\begin{cases} 0.6DL + 0.6AL + 0.6WL_4 \\ 0.6DL + 0.6AL + 0.6WL_5 \end{cases}$



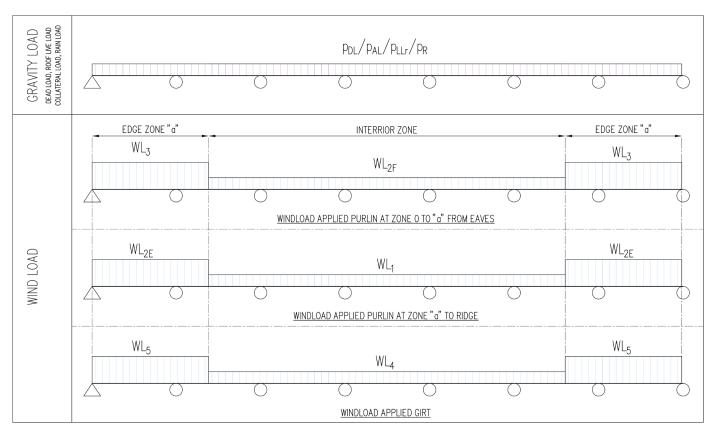


Figure 7 2 Schematics of Purlin/Girt/Eave strut

The peak local pressures zones around the perimeter of the roof govern the purlin spacing these areas, and the purlins in end bays is usually adopted for the rest of the roof because of the difference in loads and bending moments between end bays and internal span purlins. It is therefore advantageous for economical design to consider:

- Increased wall thicknesses in end span purlins, or
- Reduced end bay spacing, or

- Extra purlin spacing, extra purlins, or increase lapped length in end spans, provided this increases the design strength of the purlins.

It is necessary to provide bridging between purlins to reduce the effective lengths to control flexural-torsion buckling. BMB recommends at least one row or purlin bracing in every span, and that un-braced length be restricted to less than 20 times the section depth, or 4000 mm, whichever is less.

Bay Spacing	Number of purlin bracing
< 8m	01
8 ÷ 9m	02
> 9m	03

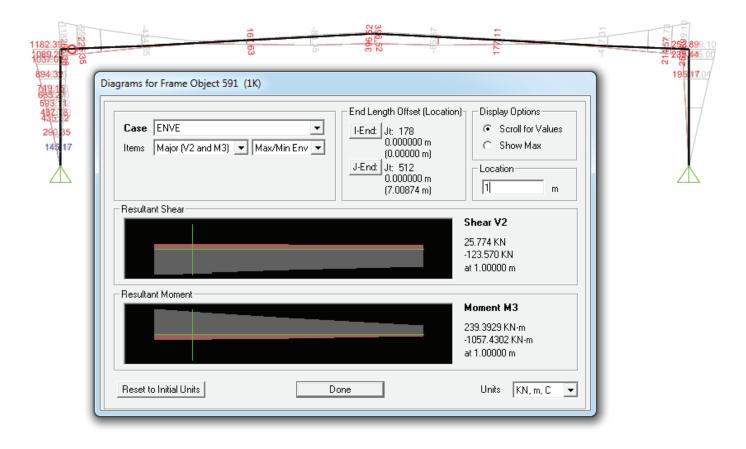
# 7.1.2.3 Lapped Length

The lapped length for Z-section purlins is a minimum of 10 percent of the span.

## 7.1.2.4 Purlin bolts

Standard purlins and girts cleats are generally used without analysis or design. Purlin cleats are subjected not only to axial load but also to bending moments. Angel cleats also provide greater robustness during transport and erection. One yardstick for robustness is that girt cleats should not yield when stood on by a heavy worker. This would equate to a 1.1 kN load applied to the tip of the cleat with a 1.5 load factor to allow for dynamic effects as the worker climbs the steel work.

The standard bolt is an M12 Type 4.6. Refer Figure 7 6 for more details.



## Input

Flange Brace section		
Yield strength: $F_y = 235MPa$ Elastic modulus: $E = 200000MPa$	V60x60x2.5	
Bending moment h	1057	kN
Height of rafter <i>b</i>	1.65	m
Length of Flange Brace $l = b\sqrt{2}$	2.33	m
Length coefficient k	1.00	
Section properties		
Area A	2.938E-4	m <sup>2</sup>

Checking ratio of slenderness: (compressive member with pin connection double end)

r

$$\frac{kL}{r} = \frac{1 \times 2.3}{0.019} = 121 < 200$$

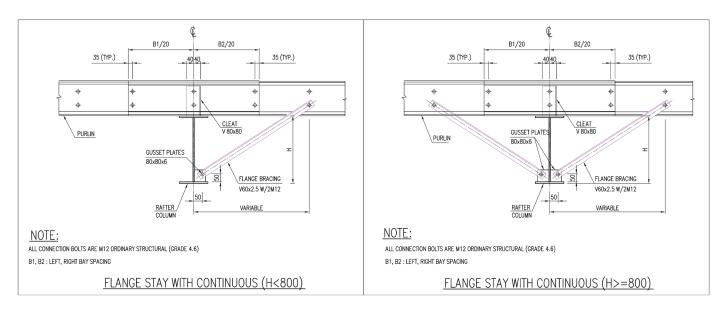
Radius of Gyration

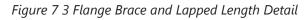
Checking capacity the compression

Elastic buckling stress	$F_{e} = \frac{\pi^{2}E}{\left(kL/r\right)^{2}} = \frac{3.14^{2} \times 2E8}{121^{2}} = 1.3E5(kN/m^{2})$	(AISC E <b>3-4</b> )
Critical stress	$\frac{kL}{r} < 4.71 \sqrt{\frac{E}{Fy}} = 4.71 \sqrt{\frac{200000}{345}} = 113$ $F_{cr} = 0.658^{F_y/F_e} \times F_y = 1.14E5(kN / m^2)$	(AISC E3-2, E3-3)
Capacity of the compression member	$[P] = \frac{P_{u}}{\Omega} = \frac{F_{cr} \times A_{g}}{1.67} = \frac{1.14E5 \times 2.938E - 4}{1.67} = 20 \text{ kN}$	(AISC E3-1)
The Transverse Compression load	$P = 2\% \frac{M}{b} = 0.02 \times 1057 / 1.65 = 12.81 \text{kN} < [P]$	(AISC Section J1.4)

0.019 m

# 7.1.3. Detailing Requirement





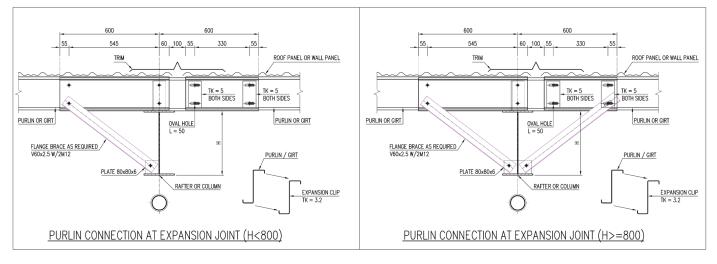


Figure 7 4 Alternative Flange Brace at Expansion Joint Detail

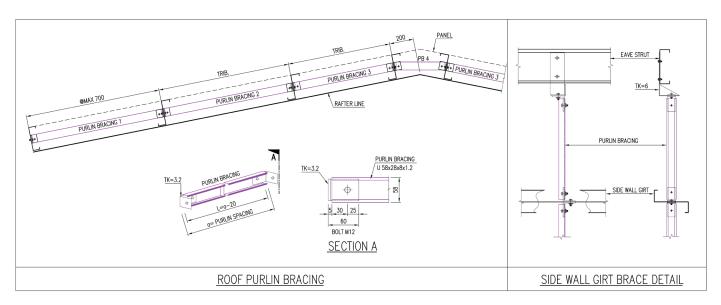


Figure 7 5 Typical Purlin Bracing Details

PURLIN CLEAT SCHEDULE			
PURLIN Z		PURLIN C	
،γ،	CLEAT SIZE	،γ،	CLEAT SIZE
< 500	V80x6	100 - 200	5 MM
> 500	V80x8	201 – 350	6 MM
		351 - 550	8 MM
		551 – 750	10 MM

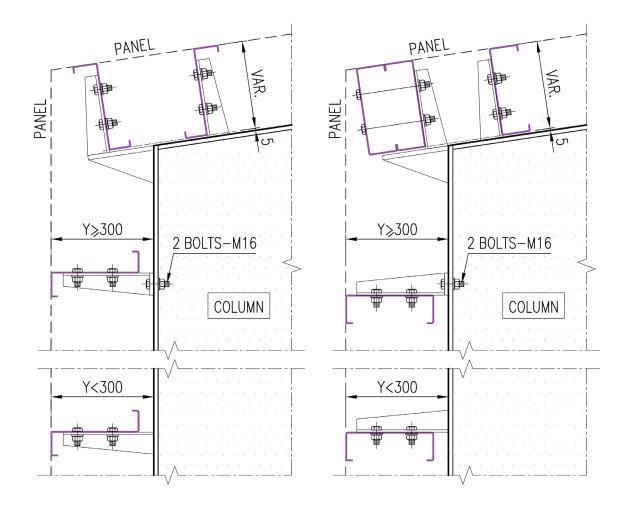


Figure 7 6 Typical Purlin Cleat Details



# NOTE

For purlin Z300, C300, use X sag rod bridging.

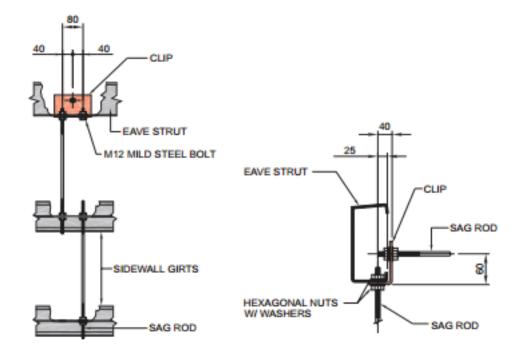


Figure 7.7 Detail A – Sag rod at wall

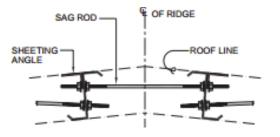


Figure 7.8 Detail B – Sag rod at eave



Figure 7.9 Detail C – Sag rod at ridge

Figure 7.10 Detail D – Sag rod at roof

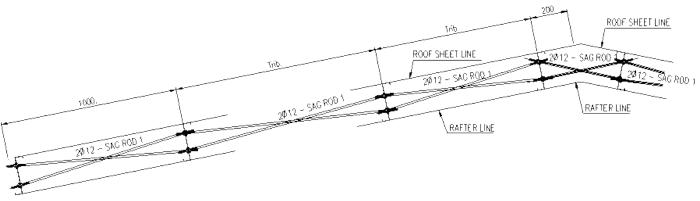


Figure 7.11 X - Sag rod

# 7.2. Drainage

The purpose of this section is to provide information for gutter capacity, the spacing of drainage downspouts (conductors), and secondary emergency overflow design for the roofs of metal buildings.

Basic eq	uation for rectangular gutter capacit	y:	
	$[L] = \left[\frac{w^{\frac{3}{7}} \times d^{\frac{3}{7}} \times \left(\frac{4320}{R \times R}\right)}{0.481}\right]$	$\frac{0}{1} \int_{-\frac{1}{2}}^{\frac{5}{14}} \left[ \frac{28}{13} \right]_{-\frac{1}{2}}^{\frac{28}{13}}$	(2012 MBSM, Eq. A4.3)
Basic eq	juation for downspout capacity:		
	$\left[\mathcal{A}\right] = \frac{I \times R \times L}{1200}$		(2012 MBSM, Eq. A4.4)
Roof D	rainage Area: $L \times W \times B$		
where: R L w d I	Width of roof to be drained, ft. Length of gutter to be drained, ft. Gutter width, ft. Gutter depth, ft. Rainfall intensity, in./hour. As per BMB&A standard I=250 mm/h.		
	Multiplier to modify drainage area	based on roof pitch	
В	Pitch	Factor "B"	
	Level to 25%	1.00	
	> 25% to $40%$	1.05	
	> 40% to $65%$	1.10	
	> 65% to 90%	1.20	
	> 90% to 100% 1.30		

#### **Emergency overflow**

Where roof drains are required, secondary (emergency overflow) roof drains or scuppers shall be provided where the roof perimeter construction extends above the roof in such a manner that water will be entrapped if the primary drains allow buildup for any reason. **DESIGN GUIDELINES** 

# 7.2.2. Example

Building Dimensions: 80m wide x 80m long (10 @ 8m bays).

Symmetrical gable roof configuration. Roof slope: 5%.

Rainfall intensity: I=250 mm/h= 9.84 in./h

#### Solve

Width of roof to be drained: L = 8m = 26.25 ftLength of gutter to be drained: L = 8m = 26.25 ftFactor: B = 1.00

Gutter Size 720(width x height):  $0.18m \times 0.202m = 0.59 \text{ ft} \times 0.66 \text{ ft}$ 

Gutter capacity:

$$[L] = \left[\frac{0.59^{\frac{3}{7}} \times 0.66^{\frac{3}{7}} \times \left(\frac{43200}{131.23 \times 9.84 \times B}\right)^{\frac{5}{14}}}{0.481}\right]^{\frac{28}{13}} \times 0.3048 = 9m \ge L = 8m : OK$$

Downspout capacity: Diameter 220mm thickness 6.6mm

$$[A] = \frac{9.84 \times 131.23 \times 26.25 \times 1}{1200} \times 0.00064516 = 0.018m^2 > A = \pi \frac{0.214^2}{4} = 0.036m^2 : OK$$

# **APPENDIX A: HOT ROLLED SECTION**

Section	Weight (kg/m)	Section	Weight (kg/m)
I 150 $\times$ 75 $\times$ 5 $\times$ 7	14.0	$H100 \times 100 \times 6 \times 8$	17.2
I 198 $\times$ 99 $\times$ 4.5 $\times$ 7	18.2	$H125 \times 125 \times 6.5 \times 9$	23.8
I $200 \times 100 \times 5.5 \times 8$	21.3	$H\ 150 \times 150 \times 7 \times 10$	31.5
I $248 \times 124 \times 5 \times 8$	25.7	H 194 $\times$ 150 $\times$ 6 $\times$ 8	30.6
$I 250 \times 125 \times 6 \times 9$	29.6	$H\ 175 \times 175 \times 7.5 \times 11$	40.4
$I 298 \times 149 \times 5.5 \times 8$	32.0	$H\ 200\times 200\times 8\times 12$	49.9
I $300 \times 150 \times 6.5 \times 9$	36.7	$H 244 \times 175 \times 7 \times 11$	44.1
I $346 \times 174 \times 6 \times 9$	41.4	$H\ 250\times 250\times 9\times 14$	72.4
I $350 \times 175 \times 7 \times 11$	49.6	H 294 × 200 × 8 × 12	56.8
$1400 \times 200 \times 8 \times 13$	66.0	$H 300 \times 300 \times 10 \times 15$	94.0
I $450 \times 200 \times 9 \times 14$	76.0	H $350 \times 350 \times 12 \times 19$	137.0
I 496 $\times$ 199 $\times$ 9 $\times$ 14	79.5	$H 390 \times 300 \times 10 \times 16$	107.0
$I 500 \times 200 \times 10 \times 16$	89.6	$H 400 \times 400 \times 13 \times 21$	172.0
$I 600 \times 200 \times 11 \times 17$	106.0	$H 482 \times 300 \times 11 \times 15$	114.0
$I 700 \times 300 \times 13 \times 24$	185.0	H 582 $\times$ 300 $\times$ 12 $\times$ 17	137.0
		$H 588 \times 300 \times 12 \times 20$	151.0

Table A.1 I Section

Table A.2 Channel Section

Section	Weight (kg/m)
U 200 x 80 x 7.5 x 11 x 12m	24.60
U 250 x 90 x9 x 13 x 12m	34.60
U 300 x 90 x 9 x 13 x 12m	38.10
U 100 x 50 x5 x 6m	9.36
U 120 x 53 x 5.5 x 12m	12.06
U 140 x 58 x 6 x 12m	14.50
U 150 x 75 x 6.5 x 12m	18.60
U 160 x 56 x 5.2 x12m	12.50
U 160 x 65 x 8,5 x 12m	19.80
U 180 x 67 x 5.4 x 12m	15.00
U 200 x 73 x 7 x 12m	22.60
U 200 x 75 x 9 x 12m	23.60
U 250 x 78 x 7 x 12m	24.60
U 300 x 87 x 9.5 x12m	39.10

Table A.3 Angle	Section
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Section	Weight (kg/piece)	Section	Weight (kg/piece)
V 25 x 3 x 6m	6.72	V 90 x 6 x 6m	51.0
V 30 x 3 x 6m	8.16	V90 x 7 x 6m	57.7
V 40 x 3 x 6m	11.10	V 90 x 8 x 6m	65.4
V 40 x 4 x 6m	14.52	V 100 x 7 x 6m	64.8
V 50 x 3 x 6m	13.92	V 100 x 8 x 6m	73.2
V 50 x 4 x 6m	18.30	V 100 x 10 x 6m	87.0
V 50 x 5 x 6m	22.62	V 120 x 8 x 12m	14.2
V 63 x 4 x 6m	23.40	V 120 x 10 x 12m	18.3
V 63 x 5 x 6m	28.86	V 120 x 12 x 12m	21.6
V 63 x 6 x 6m	34.32	V 130 x 10 x 12m	19.75
V 70 x 6 x 6m	38.34	V 130 x 12 x12m	23.4
V 70 x 7 x 6 m	44.34	V 150 x 10 x 12m	22.9
V 75 x 6 x 6m	41.34	V 150 x 12 x 12m	27.3
V 75 x 8 x 6m	54.12	V 150 x 15 x 12m	33.6
V 80 x 6 x 6m	44.16		
V 80 x 8 x 6m	51.06		

# Table A.4 Pipe Section

Section	Weight (kg/piece)	Section	Weight (kg/piece)
Ø60 x 2.3	Ø76 x 5	Ø114 x 3.2	Ø168.3 x 5.4
Ø60 x 2.6	Ø90 x 2.3	Ø114 x 3.6	Ø168.3 x 5.56
Ø60 x 2.9	Ø90 x 2.5	Ø114 x 4	Ø168.3 x 6.35
Ø60 x 3	Ø90 x 2.7	Ø114 x 4.5	Ø168.3 x 6.55
Ø60 x 3.2	Ø90 x 2.9	Ø114 x 5	Ø168.3 x 7.11
Ø60 x 3.6	Ø90 x 3	Ø114 x 5.4	Ø219.1 x 3.96
Ø60 x 4	Ø90 x 3.2	Ø141.3 x 3.96	Ø219.1 x 4.78
Ø60 x 4.5	Ø90 x 3.6	Ø141.3 x 4.78	Ø219.1 x 5.16
Ø76 x 2.5	Ø90 x 4	Ø141.3 x 5.16	Ø219.1 x 5.56
Ø76 x 2.7	Ø90 x 4.5	Ø141.3 x 5.56	Ø219.1 x 6.35
Ø76 x 2.9	Ø90 x 5	Ø141.3 x 6.35	Ø219.1 x 6.55
Ø76 x 3.2	Ø114 x 2.5	Ø141.3 x 6.55	Ø219.1 x 7.11
Ø76 x 3.6	Ø114 x 2.7	Ø168.3 x 3.96	Ø219.1 x 8.18
Ø76 x 4	Ø114 x 2.9	Ø168.3 x 4.78	
Ø76 x 4.5	Ø114 x 3	Ø168.3 x 5.16	

Table A.5 Tube in Square Section

Section	$\mathbf{W}(\mathrm{kg}/\mathrm{m})$	Section	<b>W</b> (kg/m)	Section	<b>W</b> (kg/m)
□ - 20x20x1.2	0.7	□ - 80x80x4.5	10.3	□ - 250x250x9	66.5
□ - 20x20x1.6	0.87	□ - 90x90x2.3	6.23	□ - 250x250x12	86.8
□ - 25x25x1.2	0.87	□ - 90x90x3.2	8.51	□ - 250x250x16	112.4
□ - 25x25x1.6	1.12	□ - 90x90x4.5	11.7	□ - 300x300x6	54.7
□ - 25x25x2.3	1.53	□ - 90x90x6	15.1	□ - 300x300x9	80.6
□ - 25x25x3.2	1.98	□ - 100x100x2.3	6.95	□ - 300x300x12	106
□ - 30x30x1.2	1.06	□ - 100x100x3.2	9.52	□ - 300x300x16	138
□ - 30x30x1.6	1.38	□ - 100x100x4	11.7	□ - 300x300x19	160
□ - 30x30x2.3	1.73	□ - 100x100x4.5	13.1	□ - 350x350x6	64.1
□ - 30x30x3.2	2.48	□ - 100x100x6	17	□ - 350x350x9	94.7
□ - 40x40x1.6	1.88	□ - 100x100x9	24.1	□ - 350x350x12	124
□ - 40x40x2.3	2.62	□ - 100x100x12	30.2	□ - 350x350x16	163
□ - 40x40x3.2	3.49	□ - 125x125x3.2	12	□ - 350x350x19	190
□ - 50x50x1.6	2.38	□ - 125x125x4.5	16.6	□ - 400x400x9	109
□ - 50x50x2.3	3.34	□ - 125x125x5	18.3	□ - 400x400x12	143
□ - 50x50x3.2	4.5	□ - 125x125x6	21.7	□ - 400x400x14	166
🗆 - 50x50x4.5	60.2	□ - 125x125x9	31.1	□ - 400x400x16	188
🗆 - 50x50x6	7.56	□ - 125x125x12	39.7	🗌 - 400x400x19	220
□ - 60x60x2.3	4.06	□ - 150x150x4.5	20.1	□ - 400x400x22	251
□ - 60x60x3.2	5.5	□ - 150x150x6	26.4	□ - 450x450x9	122
🗆 - 60x60x4.5	7.43	□ - 150x150x9	38.2	□ - 450x450x12	160
🗆 - 60x60x6	9.45	□ - 175x175x4.5	23.7	□ - 450x450x16	209
□ - 75x75x2.3	5.14	□ - 175x175x6	31.1	□ - 450x450x19	250
□ - 75x75x3.2	7.01	□ - 200x200x4.5	27.2	□ - 450x450x22	286
🗌 - 75x75x4.5	9.55	□ - 200x200x6	46.9	□ - 500x500x12	181
🗆 - 75x75x6	12.3	□ - 200x200x9	52.3	□ - 500x500x16	238
□ - 80x80x2.3	5.5	□ - 200x200x12	67.9	□ - 500x500x19	280
□ - 80x80x3.2	7.51	□ - 250x250x6	45.2	□ - 500x500x22	320

## Table A.6 Tube in Rectangular Section

Section	W(kg/m)	Section	<b>W</b> (kg/m)	Section	W(kg/m)
□ - 30x20x1.6	1.38	□ - 75x45x2.3	4.06	□ - 150x75x6	19.34
□ - 30x20x2.3	1.53	□ - 75x45x3.2	5.50	□ - 150x100x3.2	12.00
□ - 30x20x3.2	1.98	□ - 75x45x4.5	7.43	□ - 150x100x4.5	16.60
□ - 40x20x1.6	1.38	□ - 75x50x2.3	4.24	□ - 150x100x6	21.70
□ - 40x20x2.3	1.73	□ - 75x50x3.2	5.75	□ - 150x100x9	31.10
□ - 40x20x3.2	2.47	□ - 75x50x4.5	7.79	□ - 200x100x4.5	20.10
□ - 50x20x1.6	1.63	□ - 75x50x6	9.92	□ - 200x100x6	26.40
□ - 50x20x2.3	2.25	□ - 100x50x2.3	5.14	□ - 200x100x9	38.20
□ - 50x30x1.6	1.88	□ - 100x50x3.2	7.01	□ - 200x150x4.5	23.70
□ - 50x30x2.3	2.62	□ - 100x50x4.5	9.55	□ - 200x150x6	31.10
□ - 50x30x3.2	3.49	□ - 100x50x6	12.30	□ - 200x150x9	45.30
□ - 50x30x4.5	4.61	□ - 125x75x2.3	6.95	□ - 250x150x6	35.80
□ - 60x30x1.6	2.13	□ - 125x75x3.2	9.52	□ - 250x150x9	52.30
□ - 60x30x2.3	2.98	□ - 125x75x4.5	13.10	□ - 250x150x12	67.90
□ - 60x30x3.2	3.99	□ - 125x75x6	17.00	□ - 300x200x6	45.20
□ - 60x40x1.6	2.38	□ - 150x50x2.3	6.95	□ - 300x200x9	66.50
□ - 60x40x2.3	3.34	□ - 150x50x3.2	9.53	□ - 300x200x12	86.80
□ - 60x40x3.2	4.50	□ - 150x50x4.5	13.10	□ - 400x200x6	54.70
□ - 75x20x1.6	2.25	□ - 150x50x6	17.00	□ - 400x200x9	80.60
□ - 75x20x2.3	3.16	□ - 150x75x3.2	10.80	□ - 400x200x12	106.00
□ - 75x45x1.6	2.88	□ - 150x75x4.5	14.90		

# Table A.7 Downspout

Size	Diameter (m)	Thickness (m)
[]110x95	0.110 x 0.095	-
Ø90 x 2.9	0.090	0.0029
Ø114 x 3.8	0.114	0.0038
Ø168 x 4.3	0.168	0.0073
Ø220 x 6.6	0.220	0.0066
Ø250 x 7.3	0.250	0.0073
Ø280 x 8.2	0.280	0.0082
Ø315 x 9.2	0.315	0.0092
Ø400 x 11.7	0.400	0.0117

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